



Imbalance Response of a Rotor Supported on Gas Foil Bearings



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Introduction

Gas Foil Bearings (GFBs) are a proven alternative to rolling element bearings to support rotating machinery operating at high speeds. Unlike rolling element bearings, GFBs eliminate the need for lubrication, have no DN limit, and are capable of withstanding high temperature operation. Drawbacks include limited load capacity, minimal damping, and until recently, absence of reliable prediction methods.

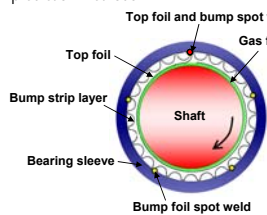


Fig 1. Schematic view of test gas foil bearings

The research objective is to measure and analyze the effect of imbalance masses on the rotordynamic response of a test rotor supported by GFBs. All data is collected during rotor coast down tests from 25 krpm. The bearings are supplied with pressurized air at 34.5 kPa (5 psi).

Test Apparatus

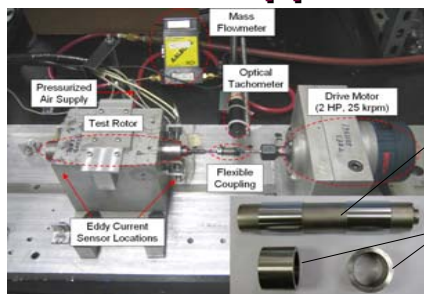


Fig 2. Photograph of GFB test rig. Inset shows rotor and test bearings

2.2 lb hollow test rotor made of AISI 4140 Steel. Chrome coating on journal surface to reduce wear.

Test foil bearings with spray-on Emralon® 333 coating to reduce friction during start up/shut down.

Imbalances: location and magnitudes

	Imbalance mass (m)		Imbalance displacement (u)	
	Drive End	Free End	Drive End	Free End
Test 1	60 mg (0°)	60 mg (0°)	1.4 μm	2.5 μm
Test 2	110 mg (0°)	110 mg (0°)	2.5 μm	4.6 μm
Test 3	160 mg (0°)	160 mg (0°)	3.7 μm	6.7 μm
Test 4	188 mg (0°)	188 mg (0°)	4.3 μm	7.9 μm
Test 5	240 mg (0°)	240 mg (0°)	5.5 μm	10.1 μm
Test 6	282 mg (0°)	282 mg (0°)	6.5 μm	11.8 μm
Test 7	334 mg (0°)	334 mg (0°)	7.7 μm	14.0 μm
Test 8	366 mg (0°)	366 mg (0°)	8.4 μm	15.3 μm
Test 9	400 mg (0°)	400 mg (0°)	9.2 μm	16.8 μm

M_{DE} & M_{FE} are fractions of rotor weight/g acting on each bearing, equal to 0.66 kg and 0.36 kg, respectively

All masses calibrated within ± 2 mg

Test Results

Cascade Analysis

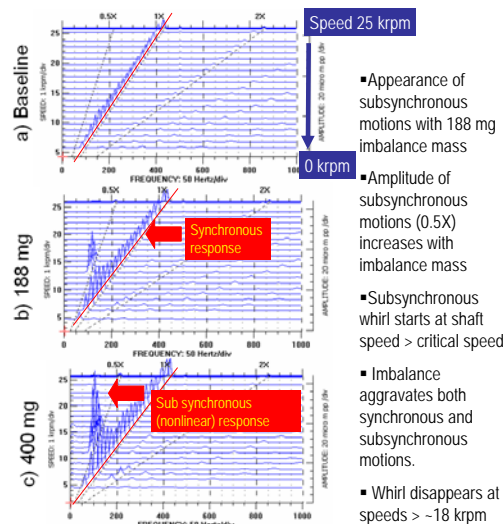


Fig 3. Cascades of rotor motions for (a) baseline condition, (b) 188 mg imbalance, and (c) 400 mg imbalance. Rotor free end, vertical plane. Horizontal axis: frequency, Vertical axis: rotor speed

Synchronous Response

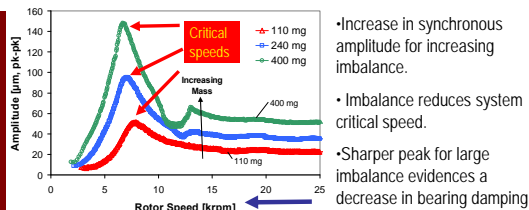


Fig 4. Amplitude (pk-pk) of synchronous (1X) rotor response for increasing imbalances. Rotor free end, vertical plane. Baseline subtracted

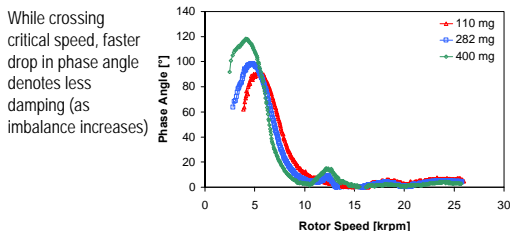


Fig 5. Phase angle of synchronous (1X) rotor response for increasing imbalances. Rotor free end, vertical plane. Baseline subtracted

Test Results Cont.

Subsynchronous Response

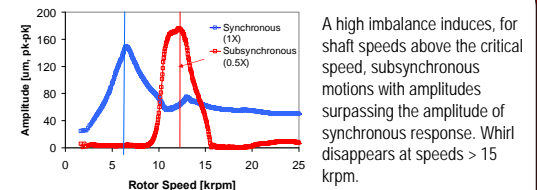


Fig 6. Amplitudes (pk-pk) of synchronous & subsynchronous whirl motions for 400 mg imbalance. Rotor free end, vertical plane. Baseline subtracted

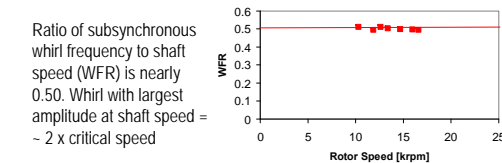


Fig 7. Subsynchronous frequency whirl ratio versus rotor speed. Response at rotor free end, vertical plane.

Coast Down Time

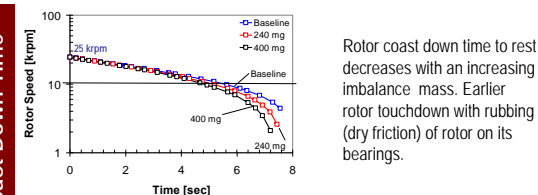


Fig 8. Rotor speed during coast down versus time for increasing imbalance masses

Future Work

- Identify fully nonlinearities in rotor responses at drive end and other motion plane (horizontal).
- Conduct tests to measure rotordynamic response due to out-of-phase imbalance conditions.
- Analyze motions in frequency domain to determine effect of imbalance on the onset, persistence and severity of subsynchronous whirl motions.
- Understand disappearance of whirl at highest shaft speeds is also of interest.

Acknowledgements

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