Handout 8
Modal Analysis of MDOF Systems with Proportional Damping

The governing equations of motion for a $n$-DOF linear mechanical system with viscous damping are:

$$
M \ddot{U} + D \dot{U} + K U(t) = F(t)
$$

(1)

where $U, \dot{U},$ and $\ddot{U}$ are the vectors of generalized displacement, velocity and acceleration, respectively; and $F(t)$ is the vector of generalized (external forces) acting on the system. $M, D, K$ represent the matrices of inertia, viscous damping and stiffness coefficients, respectively$^1$.

The solution of Eq. (1) is uniquely determined once initial conditions are specified. That is,

$$
at t = 0 \rightarrow U(0) = U_o, \quad \dot{U}(0) = \dot{U}_o
$$

(2)

Consider the case in which the damping matrix $D$ is of the form

$$
D = \alpha M + \beta K
$$

(3)

where $\alpha, \beta$ are constants$^2$, usually empirical. This type of damping is known as PROPORTIONAL, i.e proportional to either the mass $M$ of the system, or the stiffness $K$ of the system, or both.

---

1 The matrices are square with $n$-rows = $n$ columns, while the vectors are $n$-rows.
2 These constants have physical units, $\alpha$ is given in [1/sec] and $\beta$ in [sec]
Proportional damping is rather unique, since only one or two parameters, $\alpha$, $\beta$, appear to fully describe the complexity of damping, irrespective of the system number of DOFs, $n$. This is clearly not realistic. Hence, proportional damping is not a rule but rather the exception.

Nonetheless the approximation of proportional damping is useful since, most times damping is quite an elusive phenomenon, i.e. difficult to model (predict) and hard to measure but for a few DOFs.

Next, consider one already has found the natural frequencies and natural modes (eigenvectors) for the UNDAMPED case, i.e. given $M \ddot{U} + K U = 0$,

$$\{\omega_i, \phi_{(i)}\}_{i=1,2...n} \text{ satisfying } \begin{bmatrix} -M & \omega_i^2 + K \end{bmatrix} \phi_{(i)} = 0, \quad i=1,...,n. \quad (4)$$

with properties $\Phi^T M \Phi = [M]$; $\Phi^T K \Phi = [K]$ \quad (5)

As in the undamped modal analysis, consider the modal transformation $U_{(t)} = \Phi q_{(t)}$ \quad (6)

And with $\dot{U}_{(t)} = \Phi \dot{q}_{(t)}$; $\ddot{U}_{(t)} = \Phi \ddot{q}_{(t)}$, then EOM (1) becomes:

$$M \Phi \ddot{q} + D \Phi \dot{q} + K \Phi q = F_{(t)} \quad (7)$$

which offers no advantage in the analysis. However, premultiply the equation above by $\Phi^T$ to obtain

$$\left( \Phi^T M \Phi \right) \ddot{q} + \left( \Phi^T D \Phi \right) \dot{q} + \left( \Phi^T K \Phi \right) q = \Phi^T F_{(t)} \quad (8)$$

And using the modal properties, Eq. (5), and
\[ \Phi^T D \Phi = \Phi^T (\alpha M + \beta K) \Phi = \alpha \Phi^T M \Phi + \beta \Phi^T K \Phi \]

\[ \Phi^T D \Phi = \alpha [M] + \beta [K] \rightarrow [D] \]  

(9)

i.e. a diagonal matrix known as **proportional modal damping**.

Then Eq. (7) becomes

\[ [M] \ddot{q} + [D] \dot{q} + [K] q = Q = \Phi^T F(t) \]  

(10)

Thus, the equations of motion are **uncoupled in modal space**, since \([M]\), \([D]\), and \([K]\) are diagonal matrices. Eq. (10) is just a set of \(n\)-uncoupled ODEs. That is,

\[ M_1 \ddot{q}_1 + D_1 \dot{q}_1 + K_1 q_1 = Q_1 \]
\[ M_2 \ddot{q}_2 + D_2 \dot{q}_2 + K_2 q_2 = Q_2 \]

.....

\[ M_n \ddot{q}_n + D_n \dot{q}_n + K_n q_n = Q_n \]  

(11)

Or  
\[ M_j \ddot{q}_j + D_j \dot{q}_j + K_j q_j = Q_j, \quad j=1,2,...,n \]  

(12)

Where \( \omega_{n_j} = \sqrt{\frac{K_j}{M_j}} \) and \( D_j = \alpha M_j + \beta K_j \). Modal damping ratios are also easily defined as

\[ \zeta_j = \frac{D_j}{2 \sqrt{K_j M_j}} = \frac{\alpha M_j + \beta K_j}{2 \sqrt{K_j M_j}} ; j=1,2,\ldots,n \]  

(13)

For **damping proportional to mass only**, \( \beta = 0 \), and

\[ \zeta_j = \frac{\alpha M_j}{2 \sqrt{K_j M_j}} = \frac{\alpha}{2 \omega_{n_j}} \]  

(13a)
i.e., the $j$-modal damping ratio decreases as the natural frequency increases.

For **damping proportional to stiffness only**, $\alpha = 0$, **(structural damping)** and

$$\zeta_j = \frac{\beta K_j}{2 \sqrt{K_j M_j}} = \frac{\beta \omega_{nj}}{2} \quad (13b)$$

i.e., the $j$-modal damping ratio increases as the natural frequency increases. In other words, higher modes are more increasingly more damped than lower modes.

The response for each modal coordinate satisfying the modal Eqn. $M_j \ddot{q}_j + D_j \dot{q}_j + K_j q_j = Q_j, \quad j=1,2,...,n$ proceeds in the same way as for a single DOF system (See Handout 2).

First, find initial values in modal space $\{q_{o,j}, \dot{q}_{o,j}\}$. These follow from either

$$q_o = \Phi^{-1} U_o \quad ; \quad \dot{q}_o = \Phi^{-1} \dot{U}_o \quad (14)$$

or

$$q_o = [M]^{-1} \Phi^T M U_o ,$$

$$\dot{q}_o = [M]^{-1} \Phi^T M \dot{U}_o \quad (15a)$$

$$q_{o_k} = \frac{1}{M_k} \Phi^T_{(k)} \left( M U_o \right) , \quad \dot{q}_{o_k} = \frac{1}{M_k} \Phi^T_{(k)} \left( M \dot{U}_o \right) \quad (15b)$$

$k=1,\ldots,n$
**Free response in modal coordinates**

Without modal forces, $Q=0$, the modal EOM is

$$M_j \ddot{q}_j + D_j \dot{q}_j + K_j q_j = 0 = Q_j$$

with solution, for an elastic underdamped mode $\zeta_j < 1$

$$q_{H_j} = e^{-\zeta_j \omega_{d_j} t} \left( C_j \cos(\omega_{d_j} t) + S_j \sin(\omega_{d_j} t) \right)$$

if $\omega_{n_j} \neq 0$ \((17a)\)

where

$$\omega_{d_j} = \omega_{n_j} \sqrt{1 - \zeta_j^2}, \quad \omega_{n_j} = \sqrt{\frac{k_j}{M_j}}$$

and

$$C_j = q_{o_j} ; \quad S_j = \frac{\dot{q}_{o_j} + \zeta_j \omega_{n_j} q_{o_j}}{\omega_{d_j}}$$

\((17b)\)

See Handout (2a) for modal responses corresponding to overdamped and critically damped SDOF system.

**Forced response in modal coordinates**

For step-loads, $Q_{S_j}$, the modal equations are

$$M_j \ddot{q}_j + D_j \dot{q}_j + K_j q_j = Q_{S_j}$$

with solution, for an elastic underdamped mode $\zeta_j < 1$

$$q_j = e^{-\zeta_j \omega_{d_j} t} \left( C_j \cos(\omega_{d_j} t) + S_j \sin(\omega_{d_j} t) \right) + q_{S_j}$$

if $\omega_{n_j} \neq 0$ \((19a)\)
where \( \omega_d = \omega_n \sqrt{1 - \zeta^2} \), \( \omega_n = \sqrt{\frac{K}{M}} \) and

\[
q_{S_j} = \frac{Q_{S_j}}{K_j}; \quad C_j = \left(q_{o_j} - q_{S_j}\right); \quad S_j = \frac{\dot{q}_{o_j} + \zeta_j \omega_n C_j}{\omega_d_j}
\] (19b)

See Handout (2a) for physical responses corresponding to overdamped and critically damped SDOF system.

**For periodic-loads,**
Consider the case of force excitation with frequency \( \Omega \neq \omega_n \) and acting for very long times. The EOMs in physical space are

\[
M \ddot{U} + D \dot{U} + K U = F_p \cos(\Omega t)
\]

The modal equations are

\[
M_j \ddot{q}_j + D_j \dot{q}_j + K_j q_j = Q_p \cos(\Omega t)
\] (20)

with solutions

for **an elastic mode**, \( \omega_n \neq 0 \)

\[
q_j = q_{\text{transient}} + q_{ss(t)} = e^{-\zeta_j \omega_d_j t} \left(C_j \cos(\omega_d_j t) + S_j \sin(\omega_d_j t)\right) + C_{c_j} \cos(\Omega t) + C_{s_j} \sin(\Omega t)
\] (21)

The **steady state or periodic response** is of importance, since the transient response will disappear because of damping dissipative effects. Hence, the \( j \)-mode response is:
\[ q_{PS_j} = \left( \frac{Q_{P_j}}{K_j} \right) A_j \cos\left( \Omega t - \psi_j \right) \]  \hspace{1cm} (22)

Let \[ f_j = \frac{\Omega}{\omega_{n_j}} \] be a \textit{jth-mode} excitation frequency ratio. Then, define

\[
A_j = \frac{1}{\sqrt{(1 - f_j^2)^2 + (2\zeta_j f_j)^2}} \quad \text{and} \quad \tan(\psi_j) = \frac{2\zeta_j f_j}{(1 - f_j^2)} \]  \hspace{1cm} (23)

Recall that \( \varphi_j \) is a \textbf{phase angle} and \( A_j \) is an \textbf{amplitude ratio} for the \( j \text{th-mode} \).

Note that depending on the magnitude of the excitation frequency \( \Omega \), the frequency ratio for a particular mode, say \( k \), determines the regime of operation, i.e. below, above or around the natural frequency.

Using the \textbf{mode displacement method}, the response in physical coordinates is

\[
U \approx \sum_{j=1}^{m} \left( \varphi_j \frac{Q_{P_j}}{K_j} A_j \cos\left( \Omega t - \psi_j \right) \right) \]  \hspace{1cm} (24)

And recall that \( K_j = \omega_{n_j}^2 M_j = \phi_{(j)}^T K \phi_{(j)} \) and \( Q_{P_j} = \phi_{(j)}^T F_p \).

A \textbf{mode acceleration method} can also be easily developed to give

\[
U \approx U_{SP} \cos\left( \Omega t \right) - \sum_{j=1}^{m} \frac{2\zeta_j}{\omega_j} \varphi_j \ddot{q}_{PS_j} - \sum_{j=1}^{m} \frac{\varphi_j}{\omega_j^2} \dddot{q}_{PS_j} \]  \hspace{1cm} (25)

where \( U_{SP} = K^{-1} F_p \). (please demonstrate Eq. (25) above). Note that the mode acceleration method cannot be applied modes if there are any rigid body modes.
Frequency response functions for damped MDOF systems.

The steady state or periodic modal response for $j$-mode is:

$$q_{PS_j} = \left( \frac{Q_p}{K_j} \right) A_j \cos\left( \Omega t - \psi_j \right)$$  \hspace{1cm} (22)

Or, taking the real part of the following complex number expression

$$q_{PS_j} = \left( \frac{Q_p}{K_j} \right) H_j e^{i\Omega t}$$  \hspace{1cm} (26)

where

$$H_j = \frac{1}{\left(1 - f_j^2\right) + i\left(2\zeta_j f_j\right)}$$  \hspace{1cm} (27)

with $i = \sqrt{-1}$ is the imaginary unit, and where $f_j = \frac{\Omega}{\omega_{nj}}$ is the $j_{th}$-mode excitation frequency ratio. Then, recall from Eqs. (23)

$$A_j = |H_j| \frac{1}{\sqrt{\left(1 - f_j^2\right)^2 + \left(2\zeta_j f_j\right)^2}} \quad \text{and} \quad \psi_j = \arg(H_j)$$  \hspace{1cm} (28)

Using the modal transformation, the periodic response $U_P$ in physical coordinates is

$$U_P = \sum_{j=1}^{n} \left( \phi_j \frac{Q_p}{K_j} A_j \cos\left( \Omega t - \psi_j \right) \right)$$  \hspace{1cm} (24)
or take the real part of the equation below

\[
U_p = \Phi q = \sum_{j=1}^{n} (\phi_j q_j) = \sum_{j=1}^{n} \left( \phi_j \frac{\phi_j^T F_p}{K_j} H_j e^{i\Omega t} \right) = \left\{ \sum_{j=1}^{n} \left( \phi_j \phi_j^T \frac{H_j}{K_j} F_p \right) \right\} e^{i\Omega t}
\]

(29)

Now, the product \( \phi_j \phi_j^T = \text{matrix} (n \times n) \). That is, define the elements of the complex – frequency response matrix \( H \) as

\[
H_{p,q} = \left( \frac{\phi_{j_p} \phi_{j_q}^T}{K_j} \right) \left( \frac{1}{(1 - f_j^2) + i(2 \zeta_j f_j)} \right)
\]

(30)

\( p,q = 1,2,\ldots, n \). The response in physical coordinates thus becomes:

\[
U_p = HF_p e^{i\Omega t}
\]

(31)

Or in component form,

\[
U_{p_j} = \left( \sum_{r=1}^{n} H_{j,r} F_{p_r} \right) e^{i\Omega t}; \quad j=1,2,\ldots, n
\]

(32)

The components of the frequency response matrix \( H \) are determined numerically or experimentally. In any case, the components of \( H \) depend on the excitation frequency \( (\Omega) \). Determining the elements of \( H \) seems laborious and (perhaps) its physical meaning remains elusive.
Direct Method to Find Frequency Responses in MDOF Systems

Nowadays, with fast computing power at our fingertips, the young engineer prefers to pursue a more direct approach, one known as brute force or direct approach. Recall that the equation of motion is

\[ M \ddot{U} + D \dot{U} + K U = F_p \cos(\Omega t) \]

Or

\[ M \ddot{U} + D \dot{U} + K U = \text{Re} \left( F_p \ e^{i\Omega t} \right) \] (33)

Assume a periodic solution of the form

\[ U = V_p \ e^{i\Omega t} \] (34)

where \( V_p \) is a vector in the complex domain. Substitution of Eq. (34) into eq. (33) gives

\[
\begin{bmatrix}
K + i\Omega D - \Omega^2 M
\end{bmatrix} V_p = F_p
\] (35)

Define at each excitation frequency the complex impedance (dynamic stiffness) matrix as:

\[ K_{D(\Omega)} = \begin{bmatrix}
K + i\Omega D - \Omega^2 M
\end{bmatrix} \] (36)

And find the vector of physical responses (amplitude and phase) as

\[ V_p = \left[ K_{D(\Omega)} \right]^{-1} F_p \] (37)

Since \( V_p = V_{p_{\text{real}}} + iV_{p_{\text{imaginary}}} \), the physical response for each DOF follows as:
\[ U_r = V_{p_r} \cos (\Omega t - \gamma_r) ; \quad r=1,2,...,n \]
\[ V_{p_r} = \sqrt{V_{p_r\text{--real}}^2 + V_{p_r\text{--imaginary}}^2} ; \quad \tan (\gamma_r) = - \left( \frac{V_{p_r\text{--imaginary}}}{V_{p_r\text{--real}}} \right) \]  

(38)

The direct method requires calculating the inverse of the dynamic stiffness matrix at each excitation frequency. The computational effort to perform this task could be excessive but for systems with a few DOFs \( n \) small.
The equations of motion are:
\[ \mathbf{M} \ddot{\mathbf{X}} + \mathbf{C} \dot{\mathbf{X}} + \mathbf{K} \mathbf{X} = \mathbf{F}(t) \quad (1) \]

where \( \mathbf{M}, \mathbf{K}, \mathbf{C} \) are \( nxn \) matrices of inertia, stiffness, and damping force coefficients, and \( \mathbf{X}, \mathbf{V} = \frac{d\mathbf{X}}{dt}, \frac{d\mathbf{X}}{dt^2} \), and are the \( nx1 \) vectors of displacements, velocity and accelerations, respectively. \( \mathbf{F}(t) \) is a vector of nodal forces. At \( t=0, \mathbf{X}_0, \mathbf{V}_0 = \frac{d\mathbf{X}}{dt} \) are known.

For proportional damping, \( \mathbf{C} = a \mathbf{M} + b \mathbf{K} \), so the undamped mode shapes are still valid. \( a \) & \( b \) are physical constants usually determined from measurements.

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1. Define elements of inertia, damping & stiffness matrices:

\[
\begin{align*}
\mathbf{M} &:= 10^2 \begin{pmatrix} 1 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{pmatrix} \\
\mathbf{K} &:= 10^7 \begin{pmatrix} 2 & -1 & 0 \\ -1 & 2 & -1 \\ 0 & -1 & 1 \end{pmatrix} \\
\end{align*}
\]

\( n := 3 \) # of DOF

\( a := 0 \)
\( b := .001 \)
\( \mathbf{C} := a \cdot \mathbf{M} + b \cdot \mathbf{K} \)

2. Calculate the undamped natural frequencies and natural mode shapes from the fundamental relationship:

\[
\begin{pmatrix} -\omega^2 \mathbf{M} + \mathbf{K} \end{pmatrix} \phi = 0 \quad (2)
\]

Let:
\[
\lambda = \omega^2 \quad (3)
\]

If \( \mathbf{M} \) is invertable, then define
\[
\mathbf{A} := \mathbf{M}^{-1} \cdot \mathbf{K} \quad (4)
\]

and write Eq. (2) as:
\[
\lambda \cdot \phi = \mathbf{A} \cdot \phi \quad (5)
\]

\( \lambda := \text{sort(eigenvals}(\mathbf{A})) \)

\(<--- \text{find eigenvalues}\)

\( j := 1 \ldots n \)

\[
\begin{align*}
\omega_n^j &:= \sqrt{\lambda^j} \\
\mathbf{f}_n^j &:= \frac{\omega_n^j}{2 \pi} \\
\end{align*}
\]

The undamped natural frequencies are:

\[
\omega_n^T = (120.57 \quad 374.57 \quad 495.14) \quad \text{rad/s}
\]

\[
\mathbf{f}_n^T = (19.19 \quad 59.61 \quad 78.8) \quad \text{Hz}
\]

and undamped natural modes:

\[
\phi_j := \text{eigenvec}(\mathbf{A}, \lambda^j)
\]

\( \phi^j := \phi_j \)
3. Plot the natural mode shapes:

![Graph showing mode shapes](image)

This is the matrix of undamped modal vectors:

\[
\Phi = \begin{pmatrix}
0.33 & 0.53 & -0.88 \\
0.61 & 0.32 & 0.4 \\
0.72 & -0.79 & -0.27
\end{pmatrix}
\]

4. Modal transformation of physical equations to (natural) modal coordinates

Using the transformation \( X = \Phi \cdot q \), the equations of motion (1) become uncoupled in the modal space (principal coordinates):

\[
M_m \frac{d^2q}{dt^2} + C_m \frac{dq}{dt} + K_m \cdot q = Q
\]

where:

\[
M_m := \Phi^T \cdot M \cdot \Phi
\]

\[
K_m := \Phi^T \cdot K \cdot \Phi
\]

\[
C_m := \Phi^T \cdot C \cdot \Phi
\]

and modal forces:

\[
Q = \Phi^T \cdot F
\]

and with the initial conditions:

\[
q_0 = M_m^{-1} \left( \Phi^T \cdot M \cdot X_0 \right) \quad \frac{dq_0}{dt} = M_m^{-1} \left( \Phi^T \cdot M \cdot V_0 \right)
\]
Define the damping ratios and damped natural freqs. in modal space:

\[ \zeta_k := \frac{C_{m,k,k}}{2M_{m,k,k}\omega_{n_k}} \]  

(10a)  

\[ \zeta^T = (0.06 \quad 0.19 \quad 0.25) \]  

note all damping ratios < 1

\[ \omega_{d_k} := \omega_{n_k}\left[1 - \left(\zeta_k\right)^2\right]^{\frac{5}{4}} \]  

(10b)  

UNDERDAMPED CASE

The **modal responses** for arbitrary excitations are easily obtained for each natural mode (based on response of simple 1DOF system).

And, the response in the physical coordinates is given by the superposition of the modal responses, i.e.

\[ X(t) = \Phi \cdot q(t) \]  

(11)

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### Response to a STEP Load

**Set STEP load vector:**

\[ F := \begin{pmatrix} 2000 \\ -3000 \\ 1000 \end{pmatrix} \]

modal force is:

\[ Q := \Phi^T \cdot F \]

and set **vectors of initial conditions**

\[ X_0 := \begin{pmatrix} 0 \\ 0 \\ 0 \end{pmatrix} \quad V_0 := \begin{pmatrix} 0 \\ 0 \\ 0 \end{pmatrix} \]

**Calculate response**

calculate modal initial displs. & vels.

\[ q_0 := M_m^{-1}(\Phi^T \cdot M \cdot X_0) \quad q_{\text{dot0}} := M_m^{-1}(\Phi^T \cdot M \cdot V_0) \]

**Evaluate response in modal coordinates:**

\[ j := 1..n \]

\[ q_s_j := \frac{Q_j}{K_{m,j,j}} \quad \text{: static displacement in modal space} \]

\[ A_c_j := (q_0_j - q_s_j) \quad A_s_j := \frac{(q_{\text{dot0}}_j - \zeta_j \cdot \omega_{n_j} \cdot A_c_j)}{\omega_{d_j}} \]

\[ p := 1..N \text{points} \]

\[ t_p := (p - 1) \cdot \Delta t \]

\[ q_j,p := e^{-\zeta_j \cdot \omega_{n_j} \cdot t_p} \cdot (A_c_j \cdot \cos(\omega_{d_j} \cdot t_p) + A_s_j \cdot \sin(\omega_{d_j} \cdot t_p)) + q_{s_j} \]  

see your cheat-sheet  
(1 DOF response)
The response in physical coordinates is:  

\[ X(t) = \Phi \cdot q(t) \]

Let: \( m := 3 \)  

Recall I.C's

\[ X_0 = \begin{pmatrix} 0 \\ 0 \\ 0 \end{pmatrix} \quad V_o = \begin{pmatrix} 0 \\ 0 \\ 0 \end{pmatrix} \]

Steady state response (long times)

\[ X_s = \begin{pmatrix} 0 \\ -2 \times 10^{-4} \\ -1 \times 10^{-4} \end{pmatrix} \]

Check steady-state (t infinite) response:

\[ X_s, \text{Npoints} = \begin{pmatrix} 7.86 \times 10^{-6} \\ -1.85 \times 10^{-4} \\ -8.29 \times 10^{-5} \end{pmatrix} \quad X_s = \begin{pmatrix} 0 \\ -2 \times 10^{-4} \\ -1 \times 10^{-4} \end{pmatrix} \]
PERIODIC RESPONSE of MDOF system with proportional damping

The equations of motion are:

\[ M \ddot{X} + C \dot{X} + K X = F_0 \cos(\Omega t) \]  

(1)

where \( M, K, C \) are \( n \times n \) matrices of inertia, stiffness, and damping force coefficients, and \( X, V = \dot{X}/dt, \; \ddot{X}/dt^2 \), and are the \( n \times 1 \) vectors of displacements, velocity, and accelerations, respectively.

\( F(t) \) is a vector of nodal forces - periodic. At \( t=0 \), \( X_0, V_0 = \dot{X}/dt \) are known.

For proportional damping, \( C = aM + bK \), so the undamped mode shapes are still valid. \( a \) & \( b \) are physical constants usually determined from measurements.

1. Define elements of inertia, damping & stiffness matrices:

\[
M := 10^2 \begin{pmatrix} 1 & 0 & 0 \\ 0 & 2 & 0 \\ 0 & 0 & 1 \end{pmatrix} \quad K := 10^7 \begin{pmatrix} 2 & -1 & 0 \\ -1 & 2 & -1 \\ 0 & -1 & 1 \end{pmatrix}
\]

\[
\Phi \quad \text{mode 1} \quad \Phi \quad \text{mode 2} \quad \Phi \quad \text{mode 3}
\]

\[
\Phi = \begin{pmatrix} 0.33 & 0.53 & -0.88 \\ 0.61 & 0.32 & 0.4 \\ 0.72 & -0.79 & -0.27 \end{pmatrix}
\]

\[
f_n^T = (19.19 \; 59.61 \; 78.8) \quad \text{Hz}
\]

\[
\zeta^T = (0.02 \; 0.05 \; 0.06)
\]
Response to a PERIODIC Load

Set amplitude load vector:

\[
F_o := \begin{pmatrix} 2000 \\ -4000 \\ 6000 \end{pmatrix}
\]

modal force is:

\[
Q_o := \Phi^T \cdot F_o
\]

\[
Q_o = \begin{pmatrix} 2.51 \times 10^3 \\ -4.92 \times 10^3 \\ -4.97 \times 10^3 \end{pmatrix}
\]

Assume effect of initial conditions has vanished since periodic load acts for very long time.

Calculate response

Vary the excitation frequency \( \Omega \) to determine the amplitude and phase of the FRF for each mode:

\[
\Omega_{\text{max}} := 4 \cdot \omega_n
\]

(larger than highest natural freq)

\[
\omega_n = 495.14 \quad \text{rad/s}
\]

\[
\Omega_{\text{max}} = 1.98 \times 10^3
\]

Evaluate response in modal coordinates:

Frequency ratio

\[
r_{i,k} := \frac{\Omega_k}{\omega_n}
\]

\[
q_{s_i} := \frac{Q_{o,i}}{K_{m_{i,i}}}
\]

\[
q_{s}^T = \begin{pmatrix} 1.26 \times 10^{-3} \\ -3.19 \times 10^{-4} \\ -1.75 \times 10^{-4} \end{pmatrix}
\]

The MODAL complex amplitudes are:

\[
q_{i,k} := q_{s_i} \begin{pmatrix} \left[ 1 - \left( \frac{2 \cdot \zeta_i r_{i,k}}{2\pi} \right)^2 \right] + j \left( \frac{2 \cdot \zeta_i r_{i,k}}{2\pi} \right) \end{pmatrix}
\]

and MODAL Phase Angles:

\[
\alpha_{i,k} := \begin{cases} 
\arctan \left( \frac{2 \cdot \zeta_i r_{i,k}}{1 - \left( \frac{2 \cdot \zeta_i r_{i,k}}{2\pi} \right)^2} \right) \cdot \frac{180}{\pi} & \text{if } (\omega_n) - (\Omega_k) \geq 0 \\
180 + \arctan \left( \frac{2 \cdot \zeta_i r_{i,k}}{1 - \left( \frac{2 \cdot \zeta_i r_{i,k}}{2\pi} \right)^2} \right) \cdot \frac{180}{\pi} & \text{otherwise}
\end{cases}
\]

\[
\alpha := -\alpha
\]

See your cheat-sheet

(1 DOF response)
Natural Mode Response Amplitude

Magnitude of Modal Amplitude (in m)

Forcing Frequency (Hz)

Natural Mode Response Phase angle

Phase Angle (in Degrees)

Forcing Frequency (Hz)

\[ f_d^T = (19.19 \ 59.55 \ 78.65) \] Hz

\[ \zeta^T = (0.02 \ 0.05 \ 0.06) \]
The response in physical coordinates is: \( X(t) = \Phi \cdot q(t) \)

Let: \( m := 3 \) ← select number of modes for physical response

\[ \zeta^T = (0.02 \ 0.05 \ 0.06) \]

\[ f_d^T = (19.19 \ 59.55 \ 78.65) \text{ Hz} \]
DIRECT METHOD

\[ K_{D_k} := K - M \cdot (\Omega_k)^2 + i \cdot \Omega_k \cdot C \]

for each \( \Omega_k \)

and solve

\[ Z_k := (K_{D_k})^{-1} \cdot F_0 \]

Amplitude Physical Response - Direct

![Graph showing amplitude physical response for different forcing frequencies and modal amplitudes.]

- **Forcing Frequency (Hz)**: \( f_d^T = (19.19, 59.55, 78.65) \) Hz
- **Damping Ratios**: \( \zeta^T = (0.02, 0.05, 0.06) \)

- X1
- X2
- X3