



Identification of SFD force coefficients

Large Clearance Open Ends SFD

TRC-SFD-01-2012

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Mast-Childs Professor

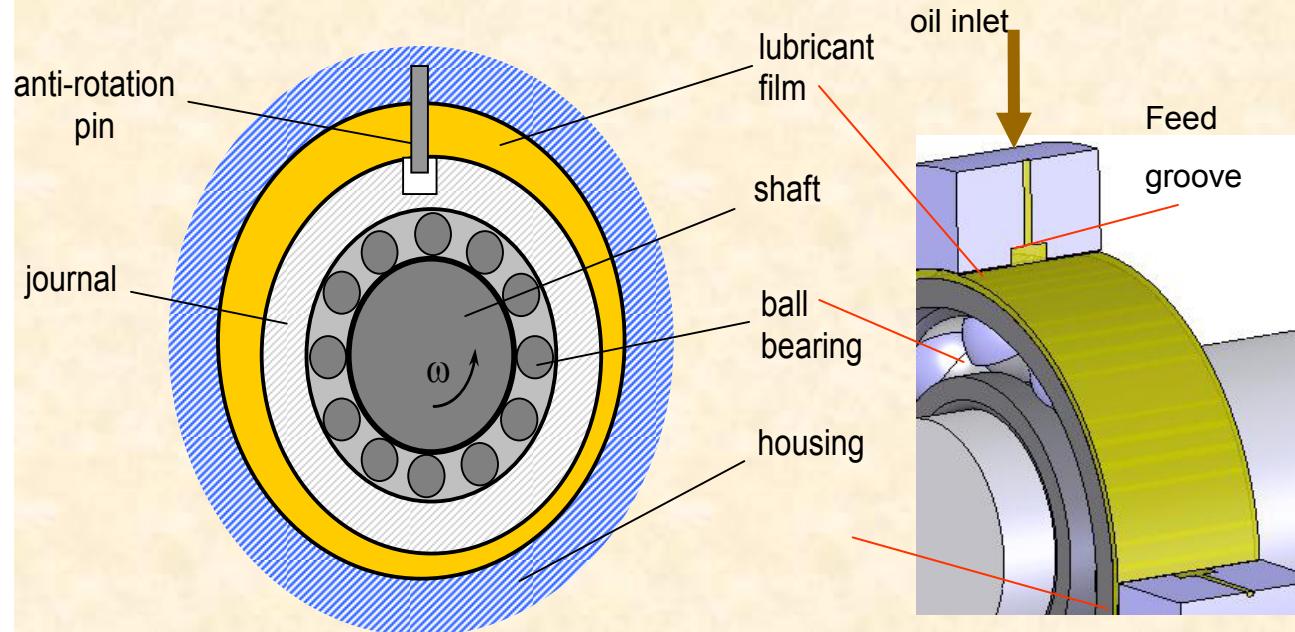
May 2012

TRC Project 32513/1519FB

Linear Nonlinear Force Coefficients for SFDs



SFD with a central groove



Typical squeeze film damper (SFD) with a central groove

Conventional knowledge regards a groove is indifferent to the kinematics of journal motion, thus effectively isolating the adjacent film lands.

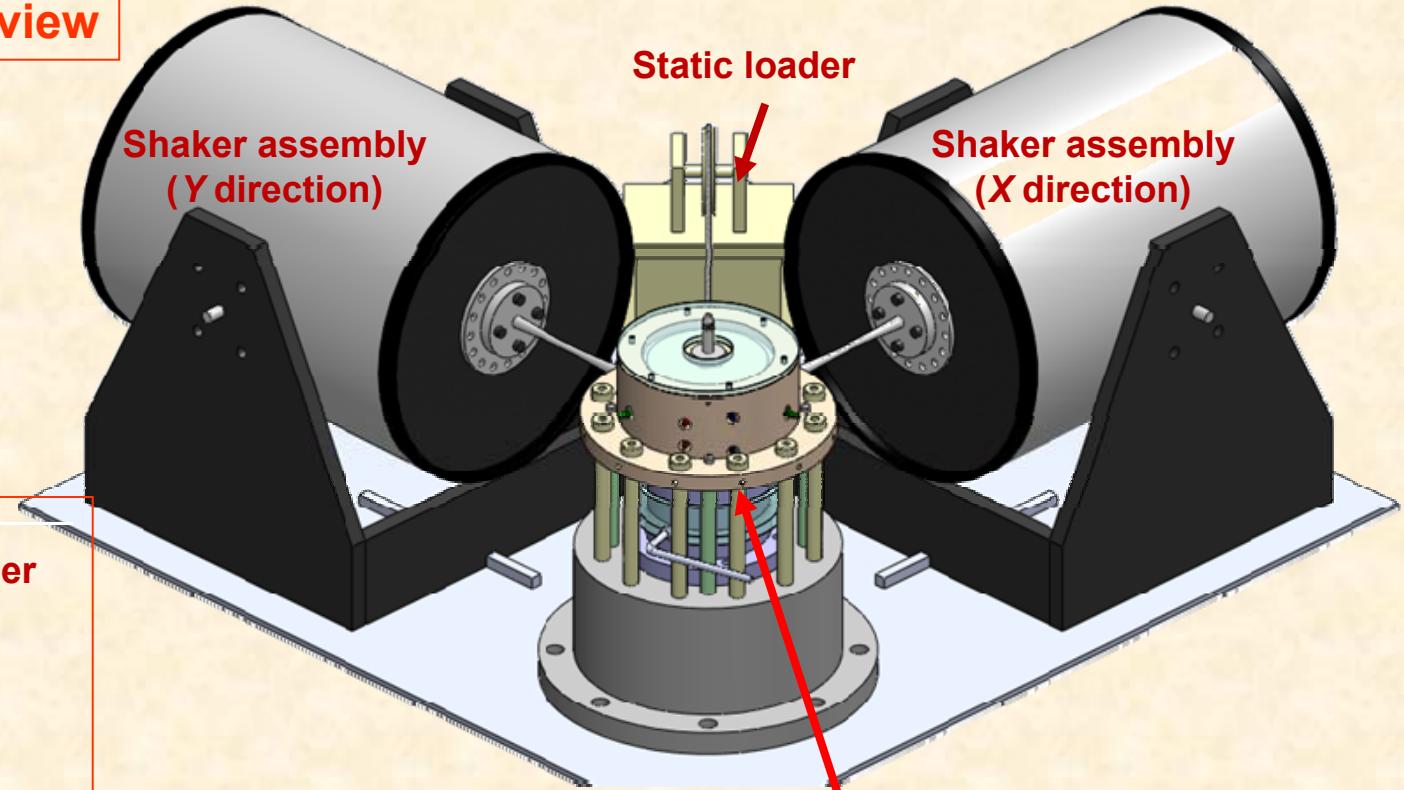
Pressurized lubricant flows through a **central groove** to fill the squeeze film lands.

Dynamic pressures in the film lands generate reaction forces aiding to damp excessive amplitudes of rotor whirl motion.

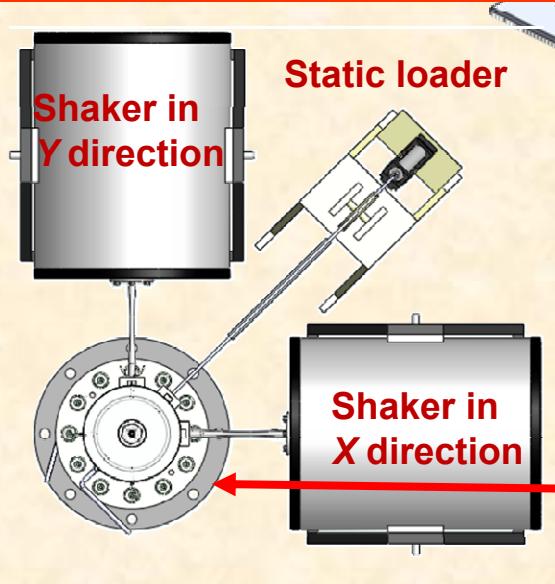
P&W SFD test rig



Isometric view



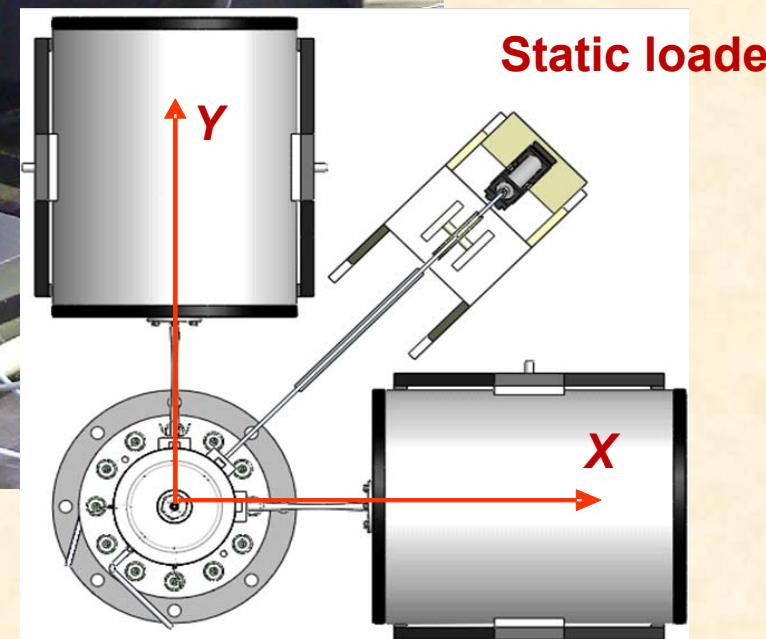
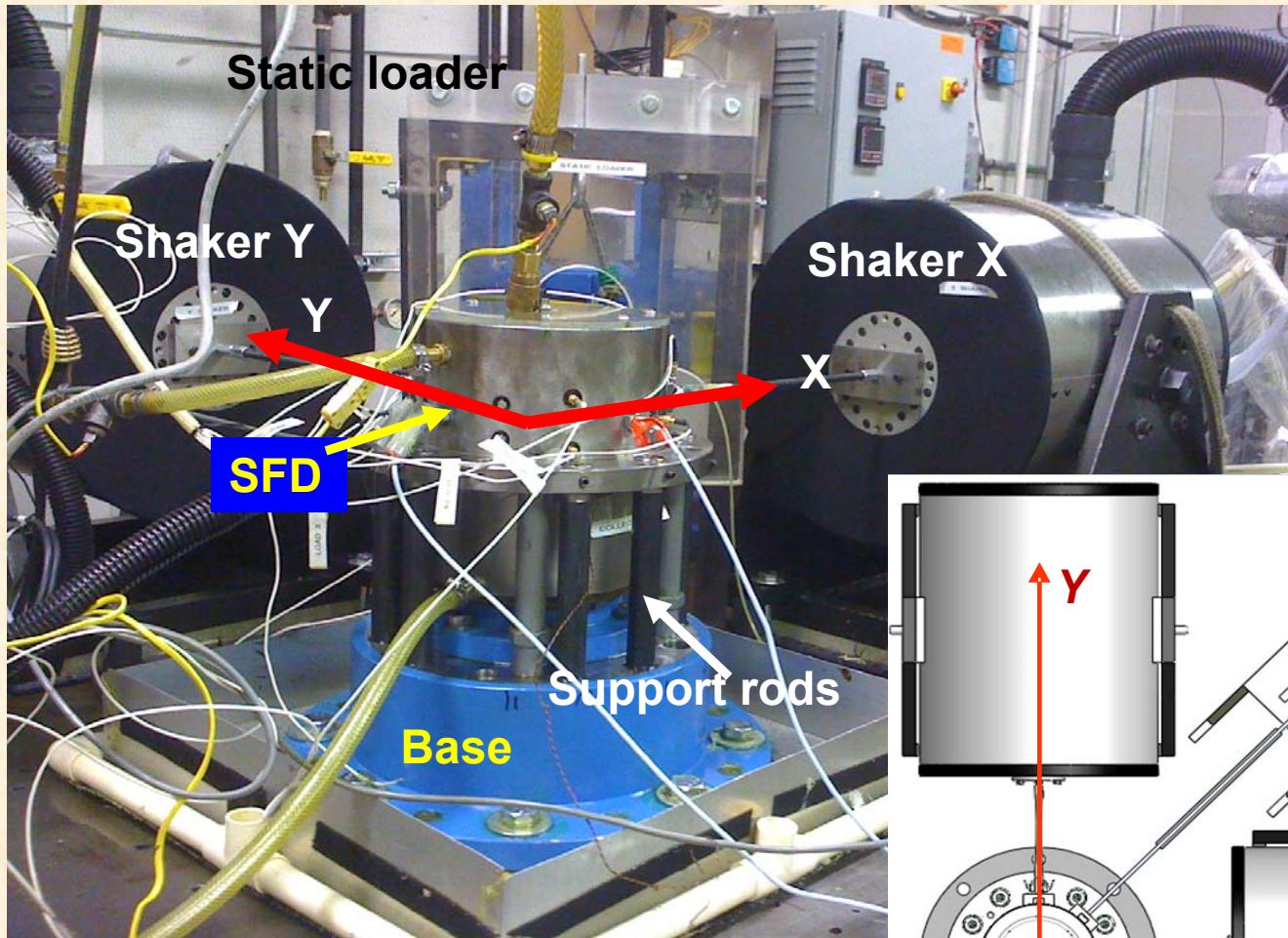
Top view



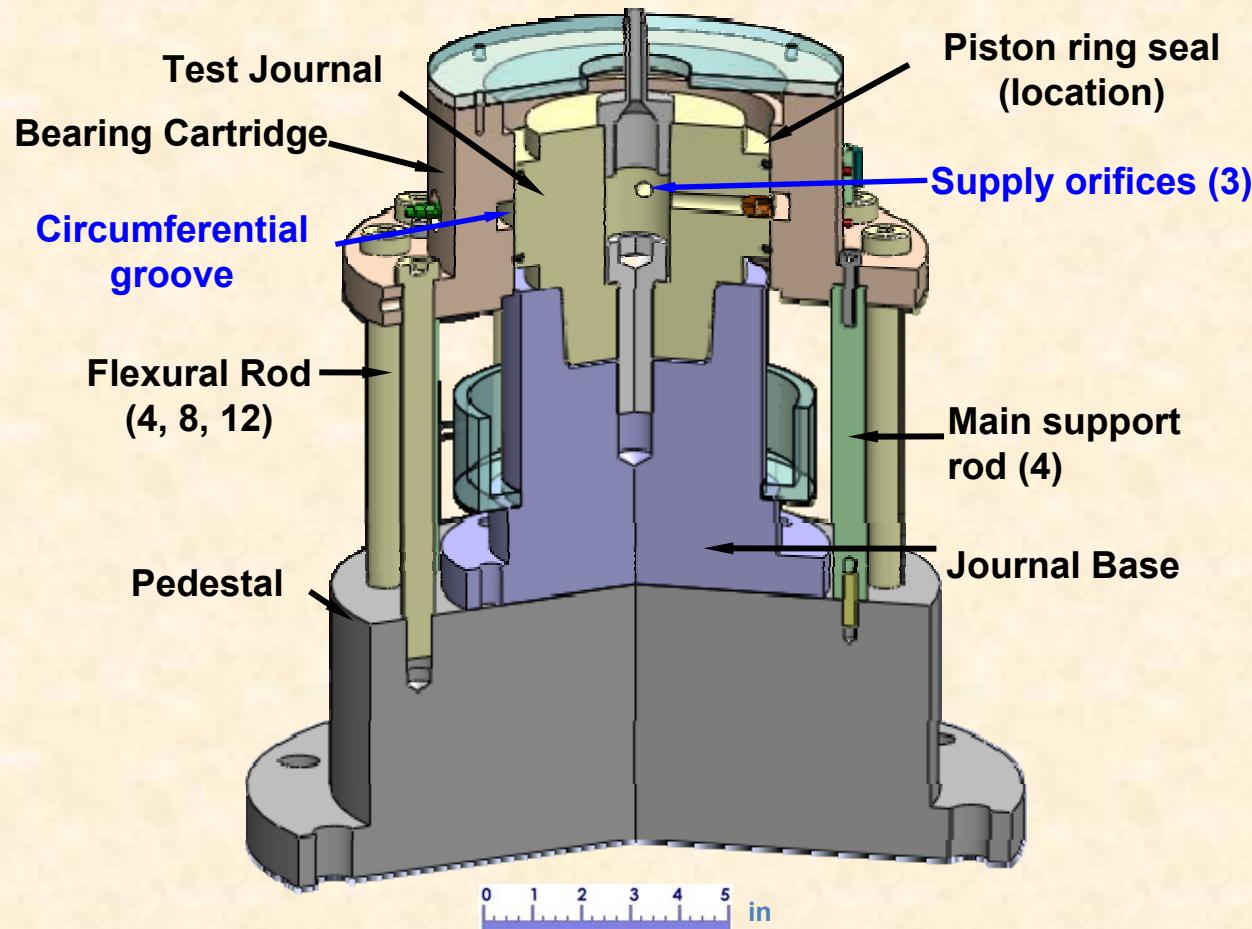
SFD test
bearing



Test rig description



SFD Test Rig – cut section



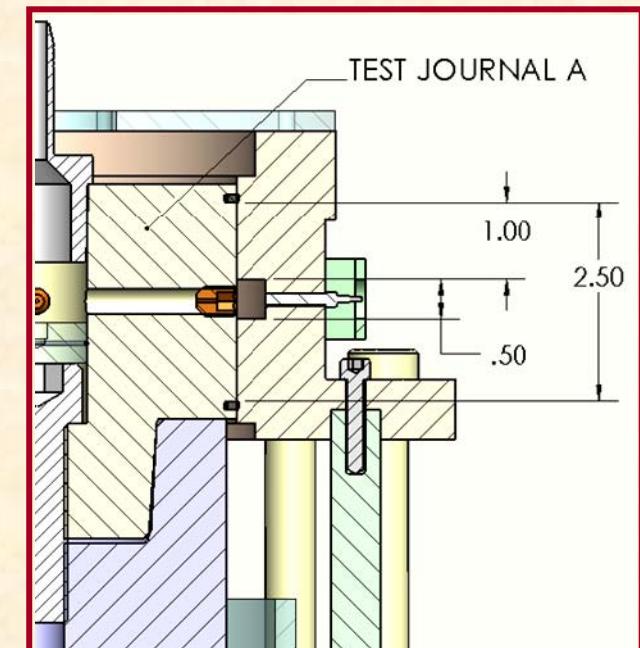
Test rig main features

Journal diameter: 5.0 inch

Film clearance: 9.9 mil

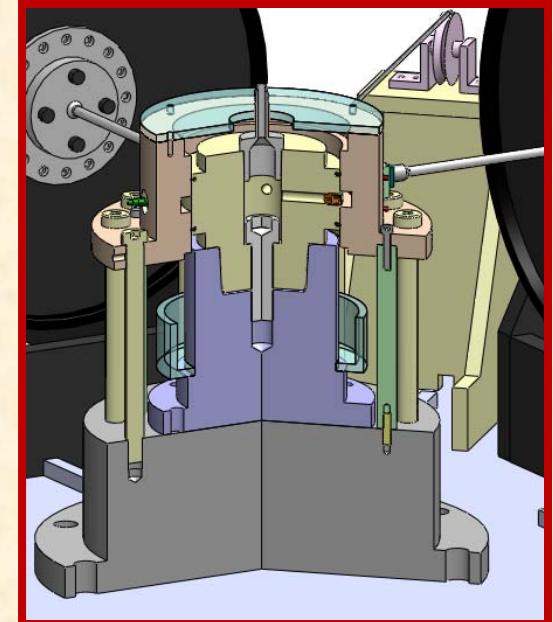
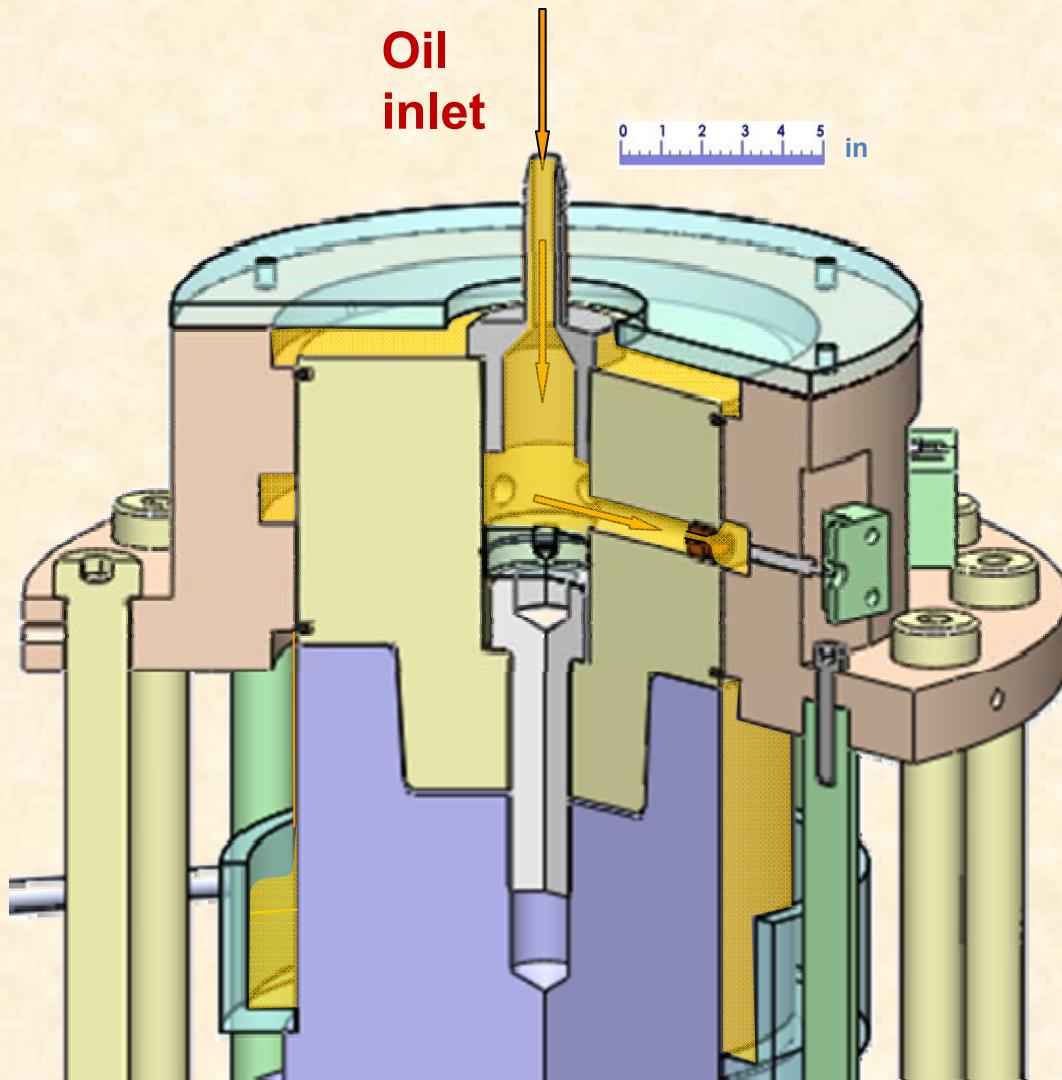
Film length: 2 x 1 inch

Support stiffness: 100 klf/in





Lubricant flow path



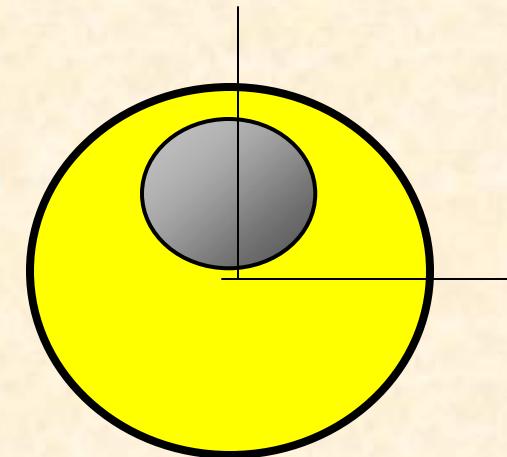
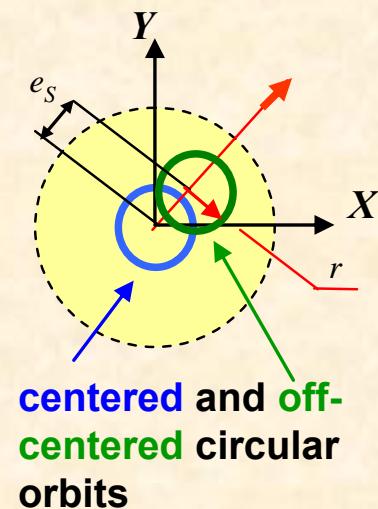
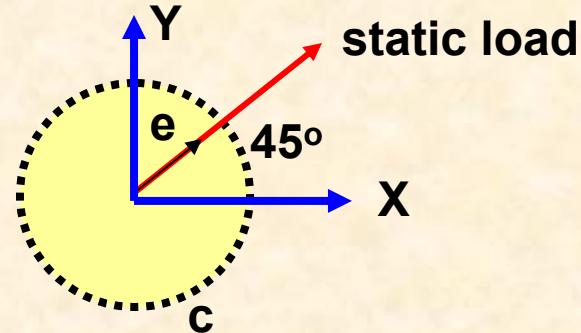
ISO VG 2 oil

Objective & tasks



Evaluate dynamic load performance of SFD with a central groove.

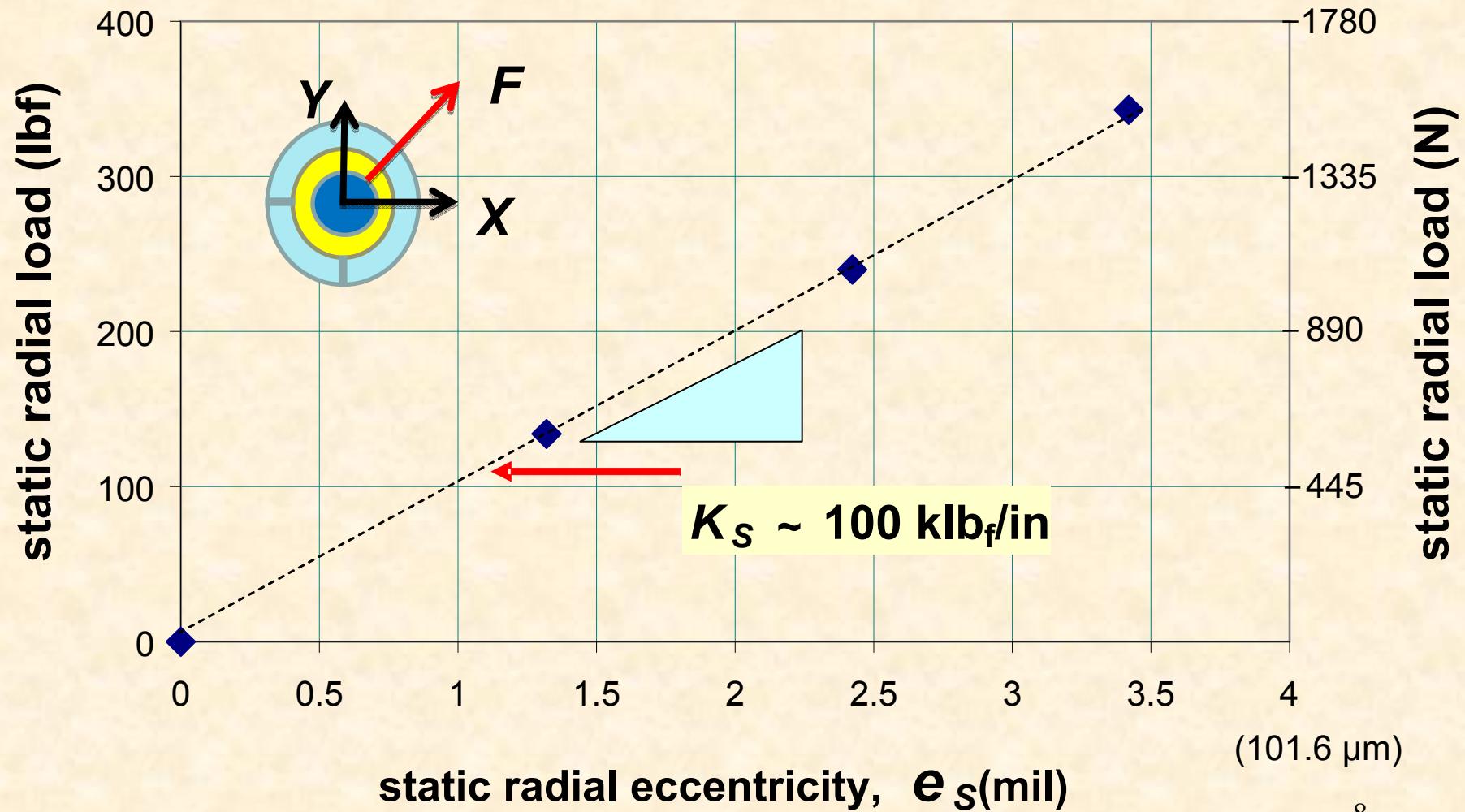
Dynamic load measurements: circular orbits (centered and off centered) and identification of test system and SFD force coefficients



Structure static stiffness



- Pull test using static loader to determine static structure stiffness





Structural parameters

- Dry test system
- Circular Centered Orbit
- Frequency **50-210 Hz**

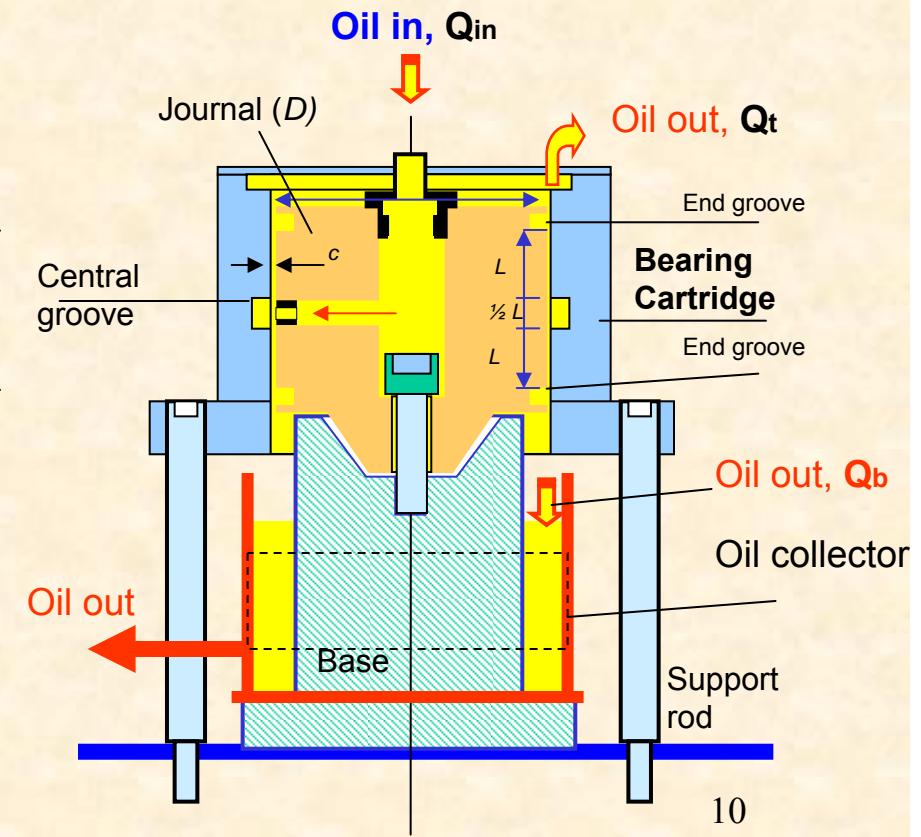
		Direct			
		XX		YY	
		US	SI	US	SI
Stiffness	K_s	107 klf/in	19 MN/m	120 klf/in	21 MN/m
Damping	C_s	8 lbf-s/in	1.4 kN-s/m	9 lbf-s/in	1.6 kN-s/m
Mass	M	-4 lb	-2 kg	-3 lb	-1 kg
System Mass	M_{BC}	48 lb	22 kg	48 lb	22 kg
Natural frequency	f_{ns}	148Hz		156Hz	
Damping ratio	ξ_s	4%		4%	

SFD dimensions & operating conds.



- Maximum static load 324 lbf
- Centered and off-centered, $e_s = 1, 2, \text{ and } 3 \text{ mil}$
- Frequency range: 50-210 Hz, Orbit amplitude $r = 0.5 \text{ mil}$

ISO VG 2 Oil	
Viscosity at 73 °F [cPoise]	3.10
Density [kg/m ³]	785
Inlet pressure [psig]	1.6
Outlet pressure [psig]	0
Radial Clearance [mil]	9.9
Journal Diameter [inch]	5.0
Central groove length [inch] & depth	0.500 0.375
Land length, L [inch]	1.0 x 2
Total Length [inch]	2.5



SFD force coefficients



IVFM parameter identification method

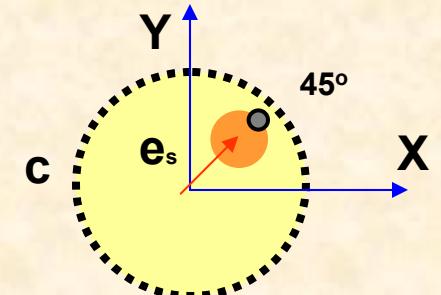
SFD

Difference between lubricated system and dry system (baseline) coefficients

$$C_{SFD} = C_{lubricated} - C_s$$

$$M_{SFD} = M_{lubricated} - M_{BC}$$

$$K_{SFD} = K_{lubricated} - K_s$$



DRY system parameters

$$K_s = 100 \text{ klf/in}$$

$$M_{BC} = 48 \text{ lb}$$

$$C_s = 8-9 \text{ lbf-s/in}$$

$$\text{Nat freq} = 148-156 \text{ Hz}$$

$$\text{Damping ratio} = 4\%$$

SFD force coefficients - theory



Centered journal ($e_s=0$), no lubricant cavitation

Two film lands separated by a plenum: central groove has no effect on squeeze film forces.

Damping \rightarrow

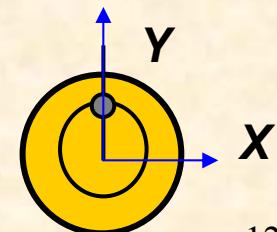
$$C^* = C_{XX}^* = C_{YY}^* = 2 \times 12 \pi \mu L \left(\frac{R}{c} \right)^3 \left[1 - \frac{\tanh[L/D]}{L/D} \right]$$

Inertia \rightarrow

$$M^* = M_{XX}^* = M_{YY}^* = 2 \times \frac{\pi \rho L R^3}{c} \left[1 - \frac{\tanh[L/D]}{L/D} \right]$$

Stiffness \rightarrow

$$K_{XX} = K_{YY} = K_{XY} = K_{YX} = 0$$



SFD force coefficients - theory



Damping

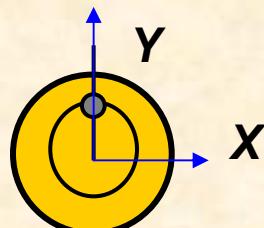
$$C^* = C_{xx}^* = C_{yy}^* = 2 \times 12\pi\mu L \left(\frac{R}{c}\right)^3 \left[1 - \frac{\tanh\left[\frac{L}{D}\right]}{\frac{L}{D}} \right]$$

$c=5.5 \text{ mil}$ $C_* = 7,121 \text{ N.s/m (40.7 lb}_f\text{.s/in)}$

$c=9.9 \text{ mil}$ $C_* = 1,255 \text{ N.s/m (7.16 lb}_f\text{.s/in)}$

Inertia

$$M^* = M_{xx}^* = M_{yy}^* = 2 \times \frac{\pi\rho LR^3}{c} \left[1 - \frac{\tanh\left[\frac{L}{D}\right]}{\frac{L}{D}} \right]$$



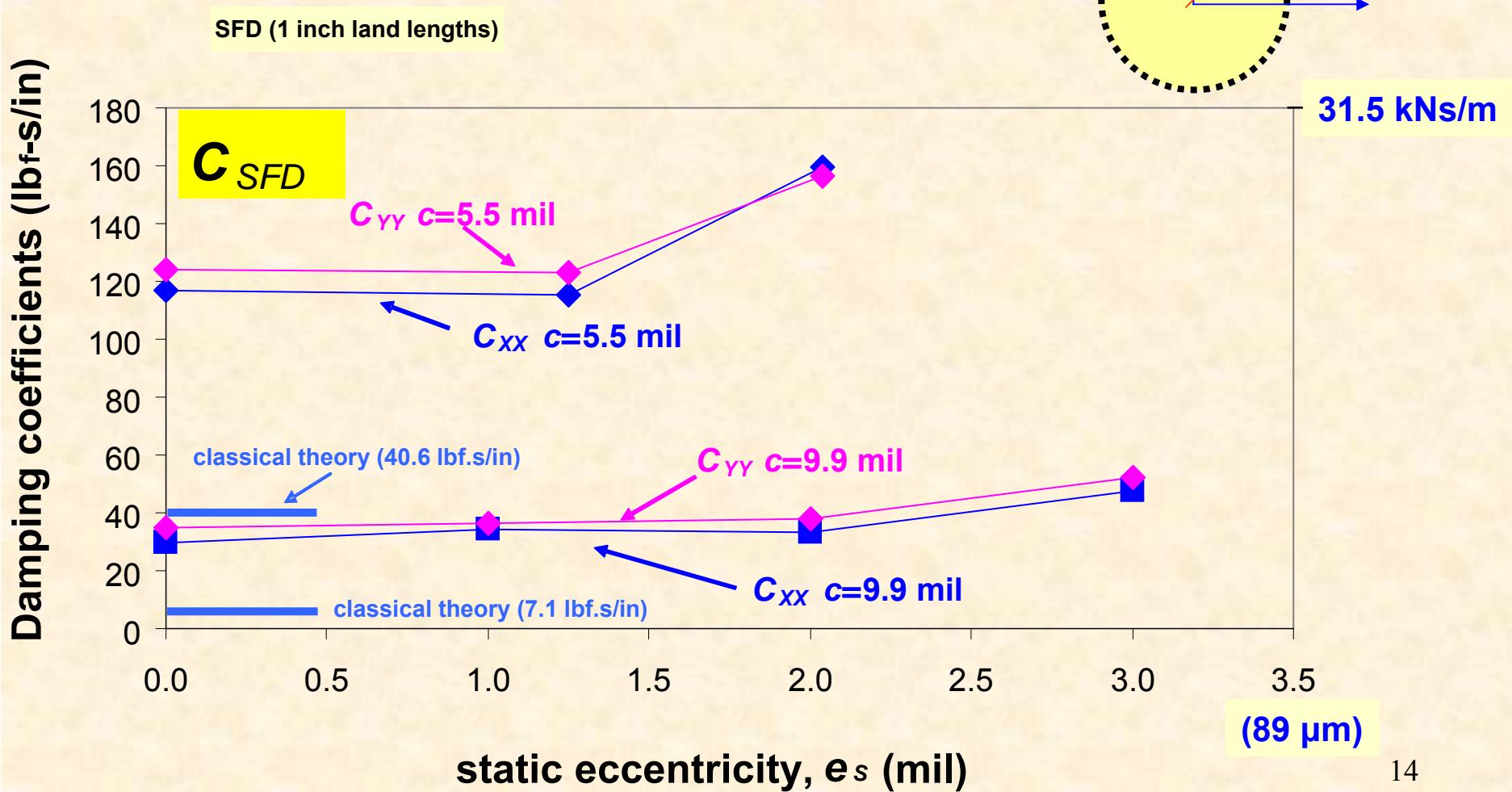
$c=5.5 \text{ mil}$ $M_* = 2.98 \text{ kg (6.58 lb}_m\text{)}$

$c=9.9 \text{ mil}$ $M_* = 1.67 \text{ kg (3.69 lb}_m\text{)}$

Experimental SFD damping coeffs.



- Open ends SFD
- Circular orbits ($r = 0.5$ mil)

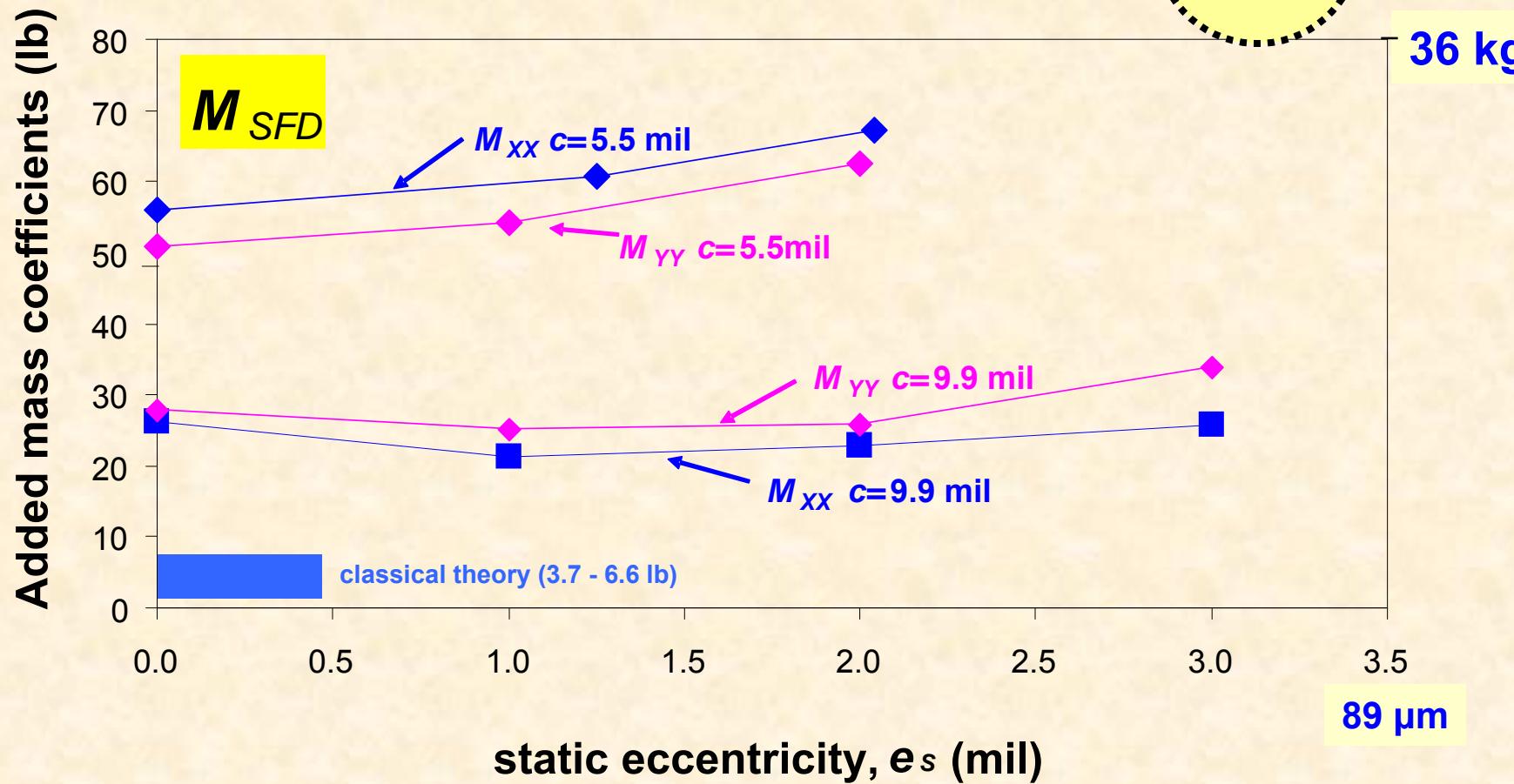


Experimental SFD inertia coeffis.



- Open ends SFDs
- Circular orbits ($r = 0.5$ mil)

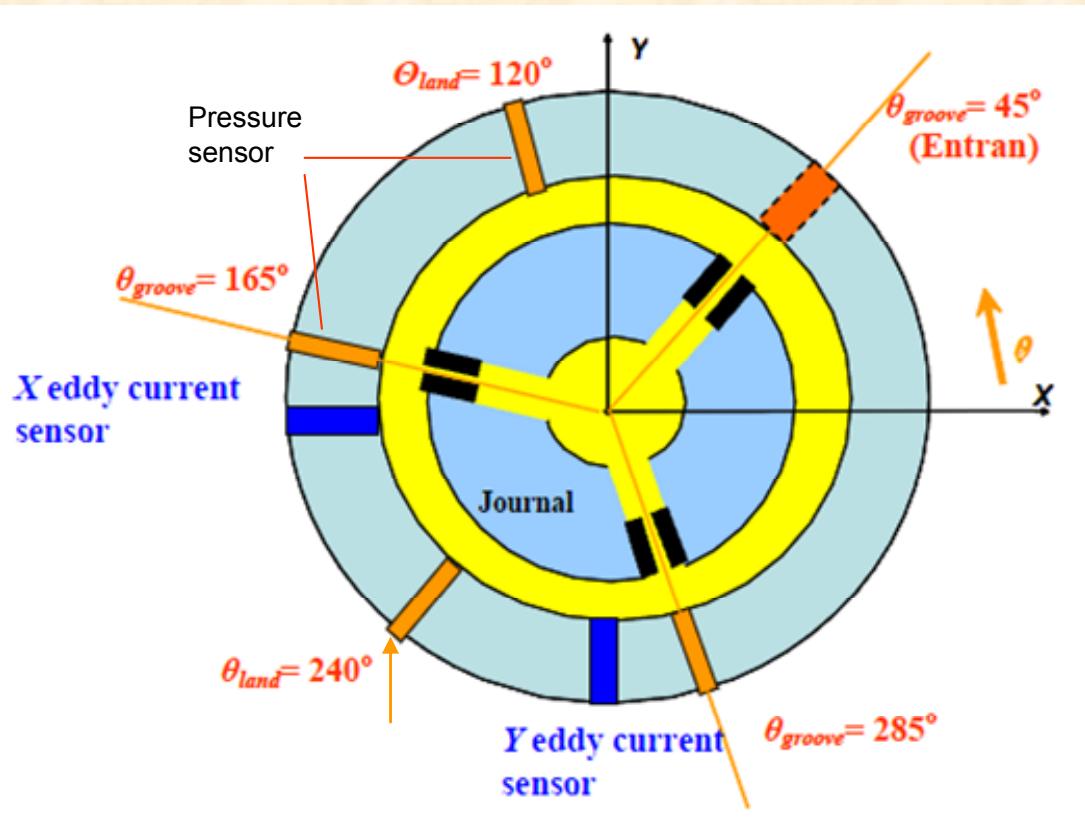
SFD (1 inch land lengths)



Pressure sensors in bearing

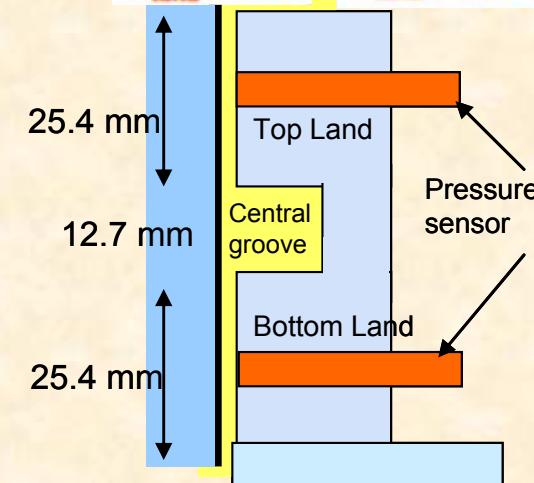


Top view: Sensors around bearing circumference

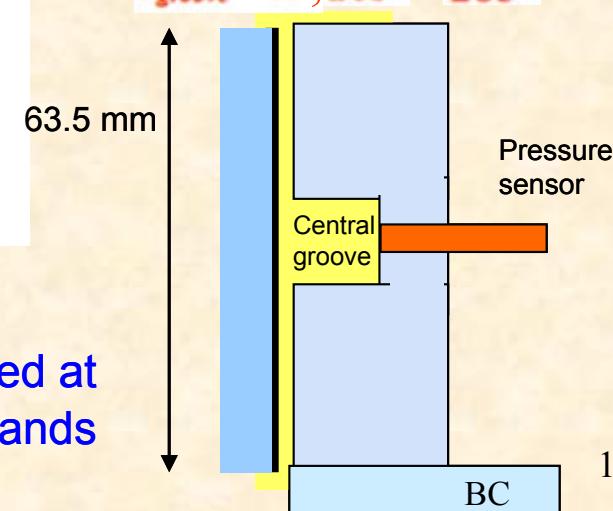


Pressure sensor locations

$\theta_{land} = 120^\circ$ and $\theta_{land} = 240^\circ$



$\theta_{groove} = 45^\circ, 165^\circ$ and 285°



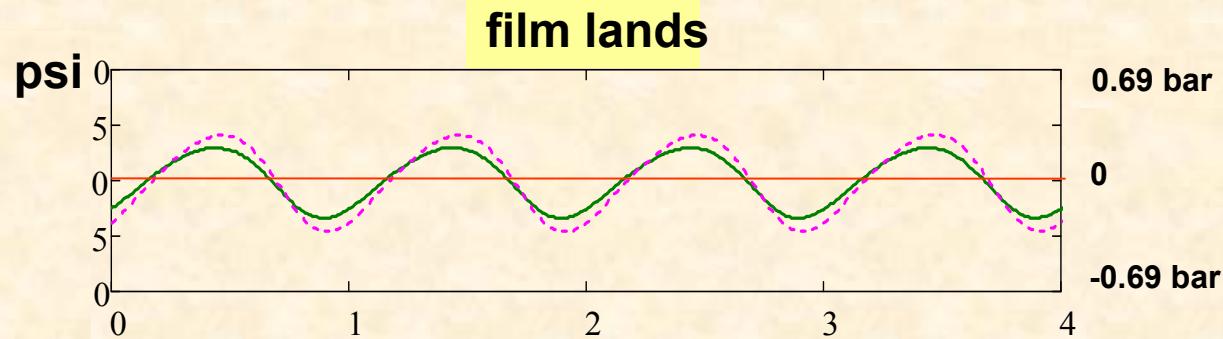
Side view: Sensors located at middle plane of film lands

Dynamic pressures: films & groove

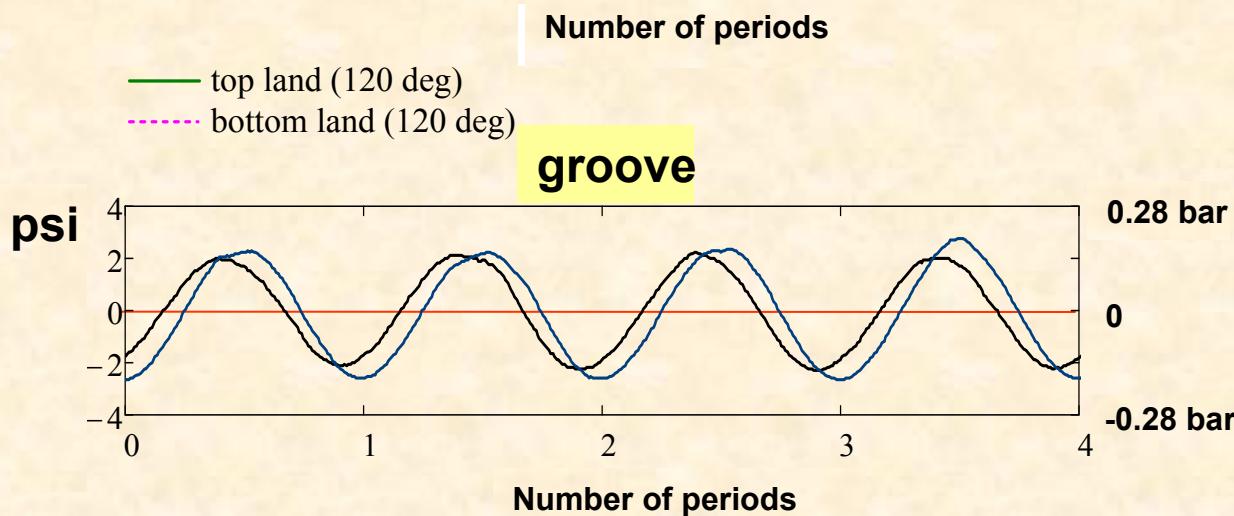


Whirl frequency 130 Hz

ASME GT2012-68212



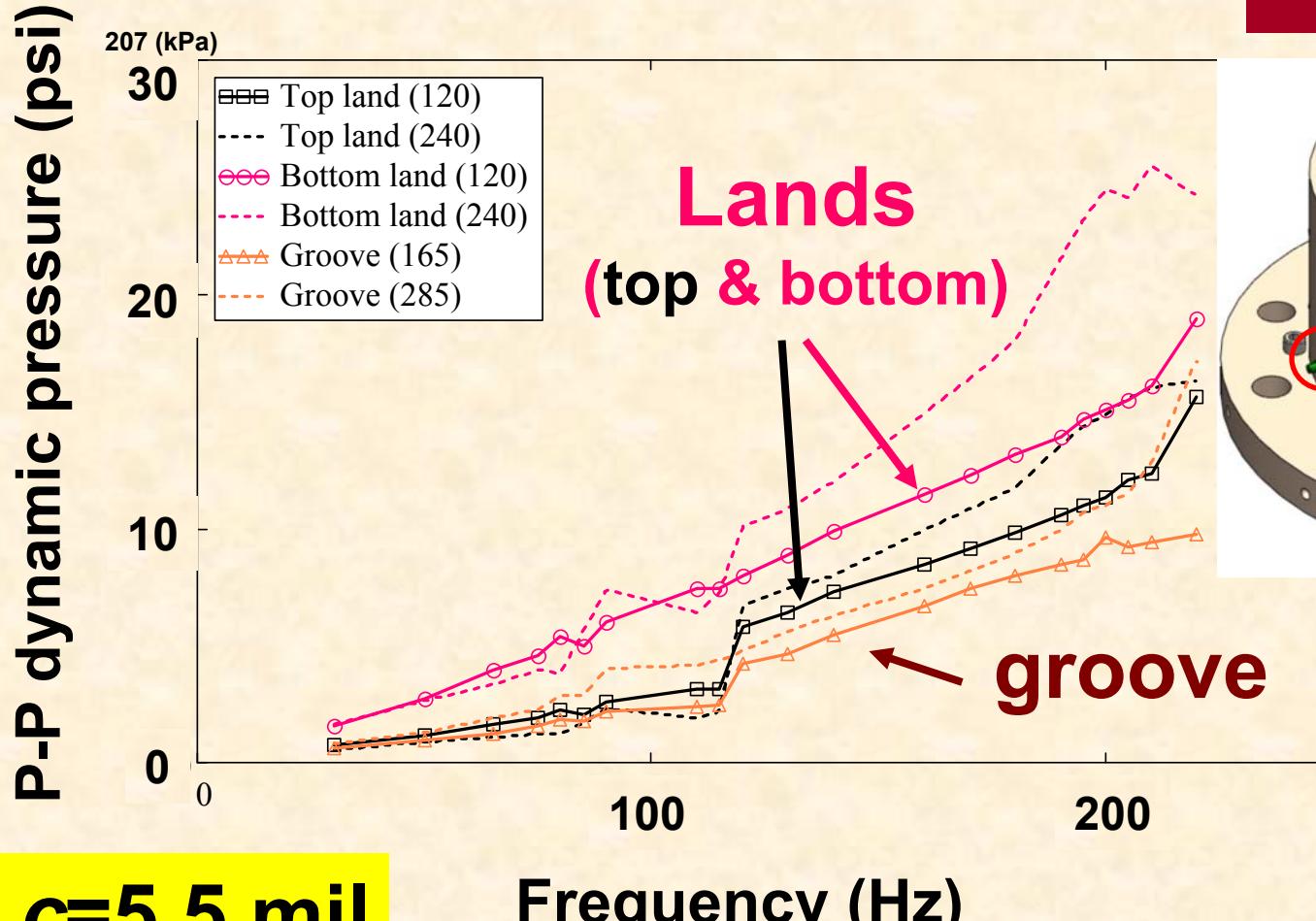
Top and bottom film lands show similar pressures.



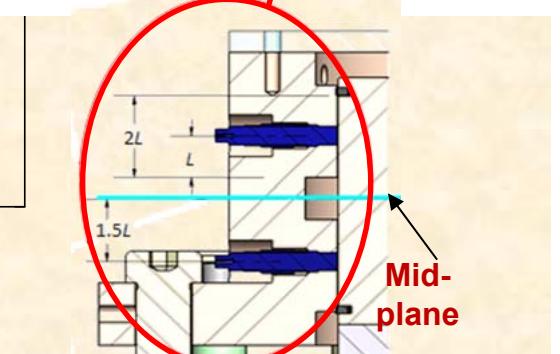
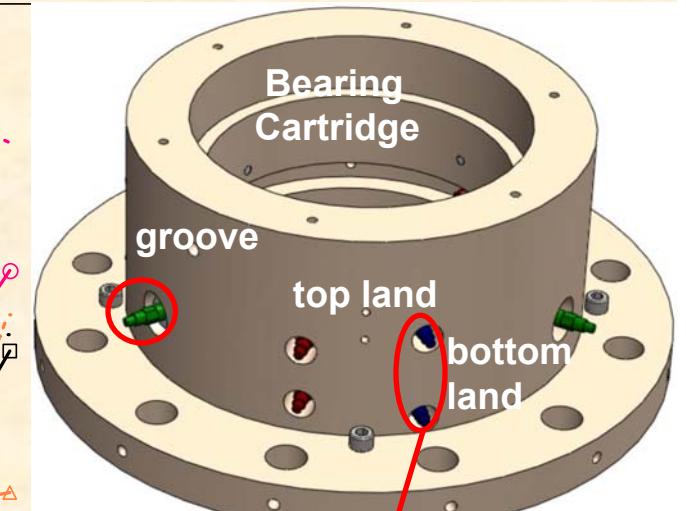
Dynamic pressure in the groove is not zero!

$e_s=0$, circular orbit $r=0.5$ mil. Groove pressure $P_G = 0.72$ bar

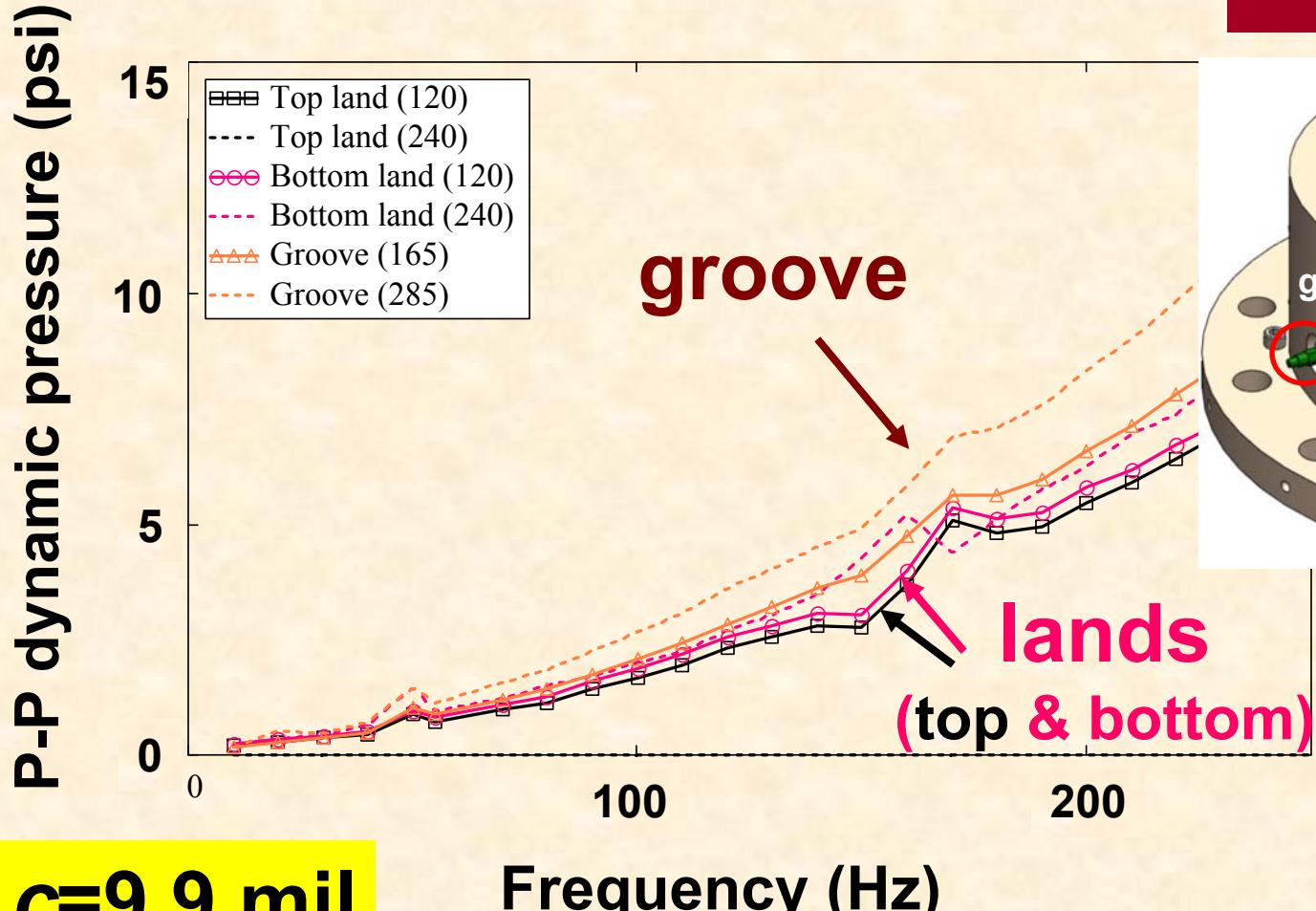
Peak-peak lubricant pressures



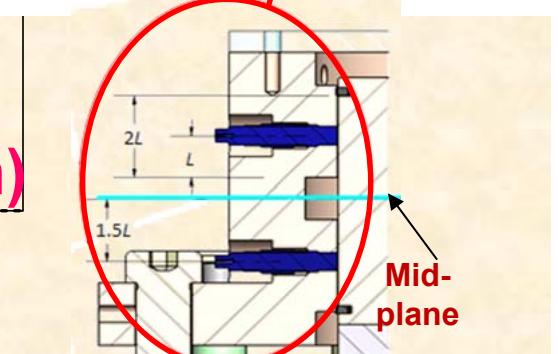
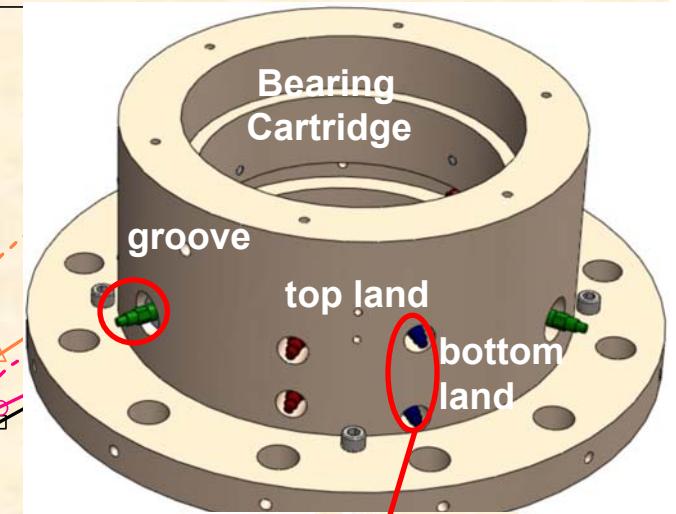
Piezoelectric pressure sensors
(PCB) location



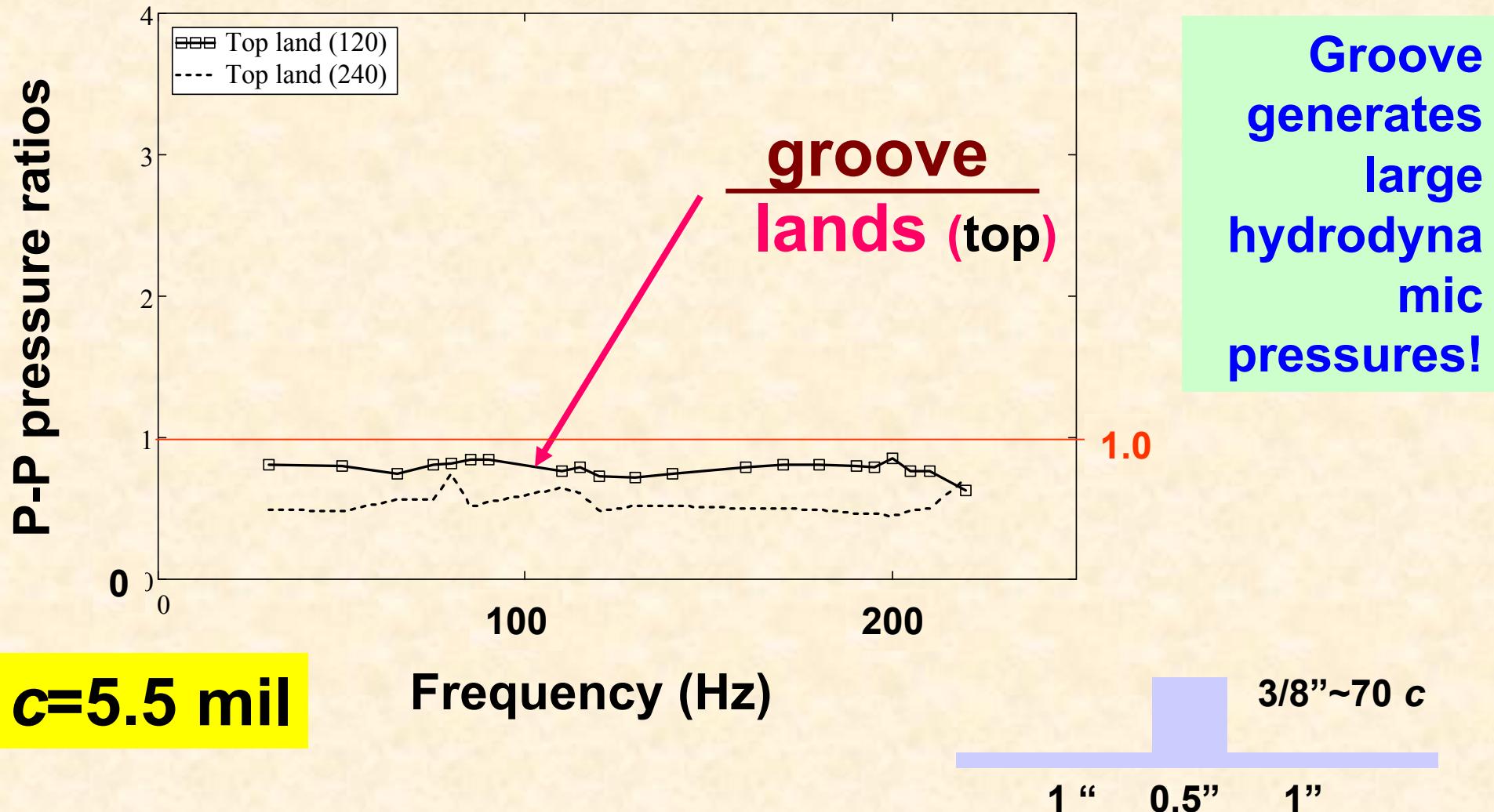
Peak-peak lubricant pressures



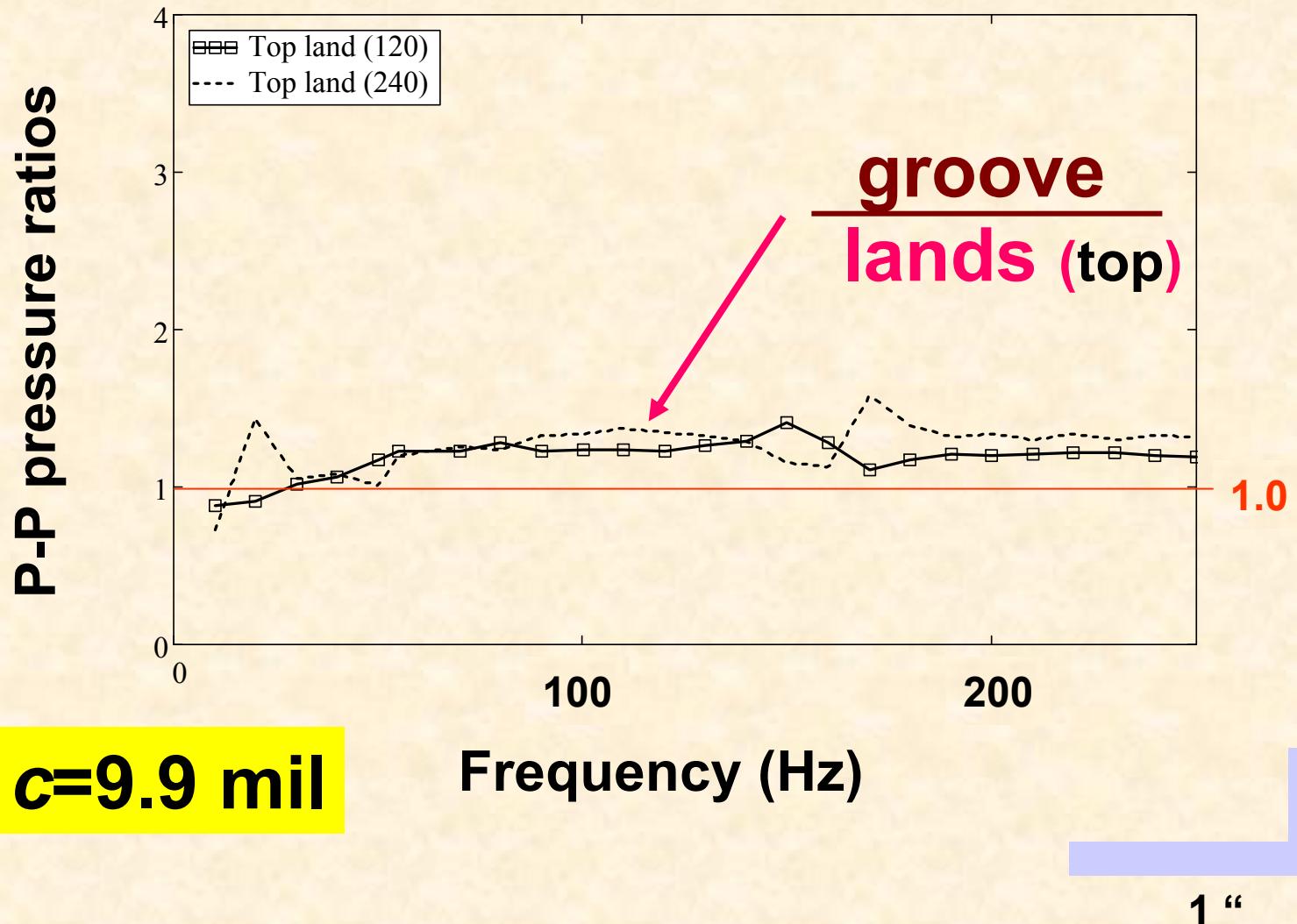
Piezoelectric pressure sensors
(PCB) location



Ratio of groove/film land pressures



Ratio of groove/film land pressures

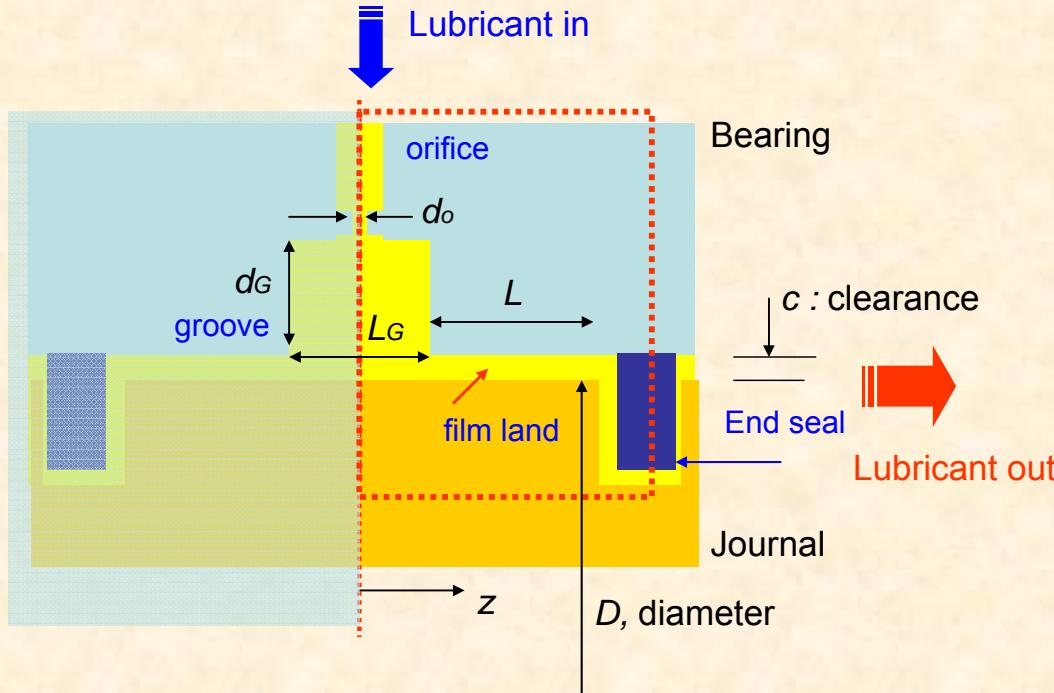


Groove generates larger hydrodynamic pressures!!
Larger than in the film!

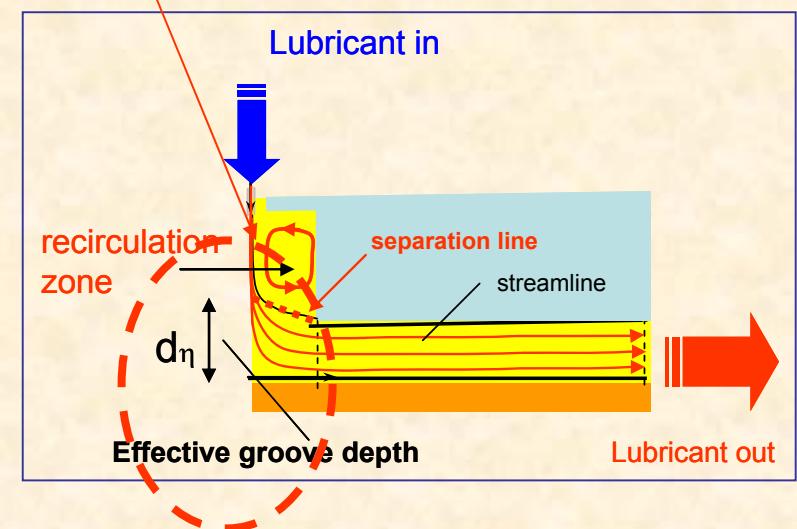
Model SFD with a central groove



SFD geometry and nomenclature



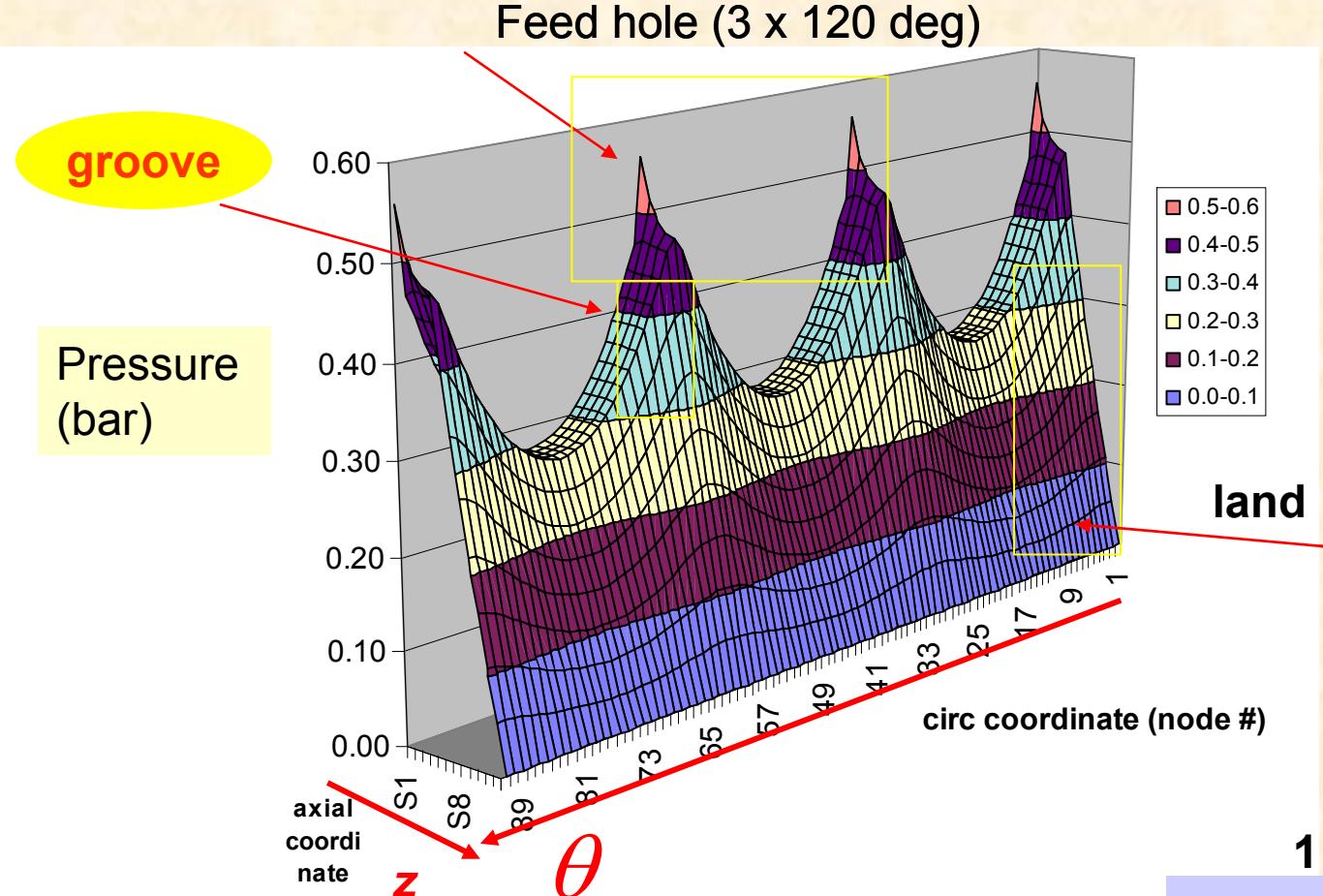
Use effective depth
 $d_\eta = 1.6c$



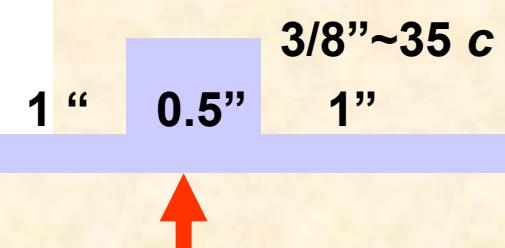
Solve modified Reynolds equation (with fluid inertia)

$$\frac{\partial}{R \partial \Theta} \left(h^3 \frac{\partial P}{R \partial \Theta} \right) + \frac{\partial}{\partial z} \left(h^3 \frac{\partial P}{\partial z} \right) = 12 \mu \frac{\partial h}{\partial t} + \rho h^2 \frac{\partial^2 h}{\partial t^2}$$

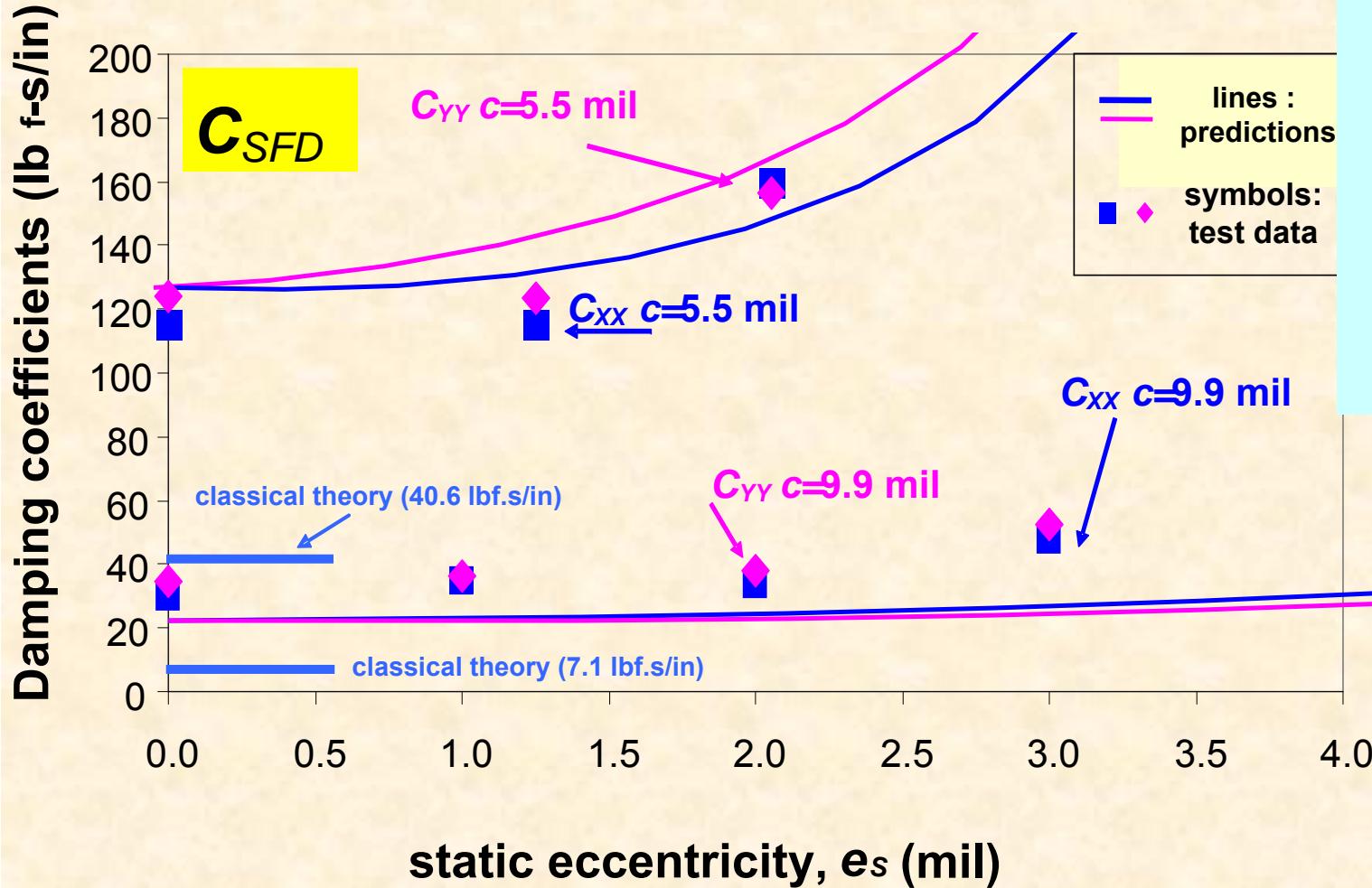
Example predicted pressure field



Static pressure at groove shows circumferential variation due to feed holes spacing



Damping coefficients: test & predictions

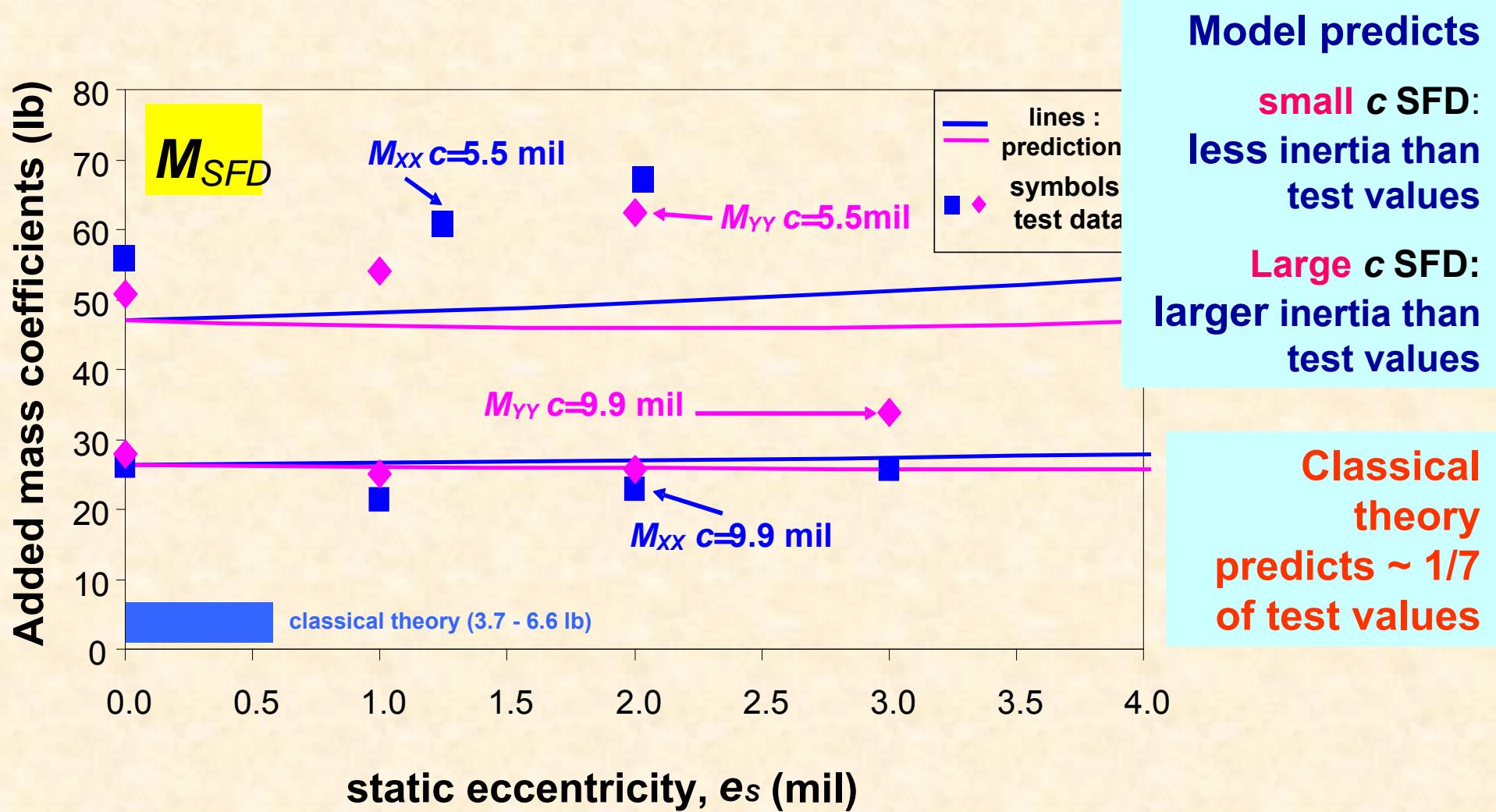


Model predicts

small c SFD:
larger damping
coefficient than
test values

large c SFD:
less damping
than test values

Inertia coefficients: test & predictions





Conclusions

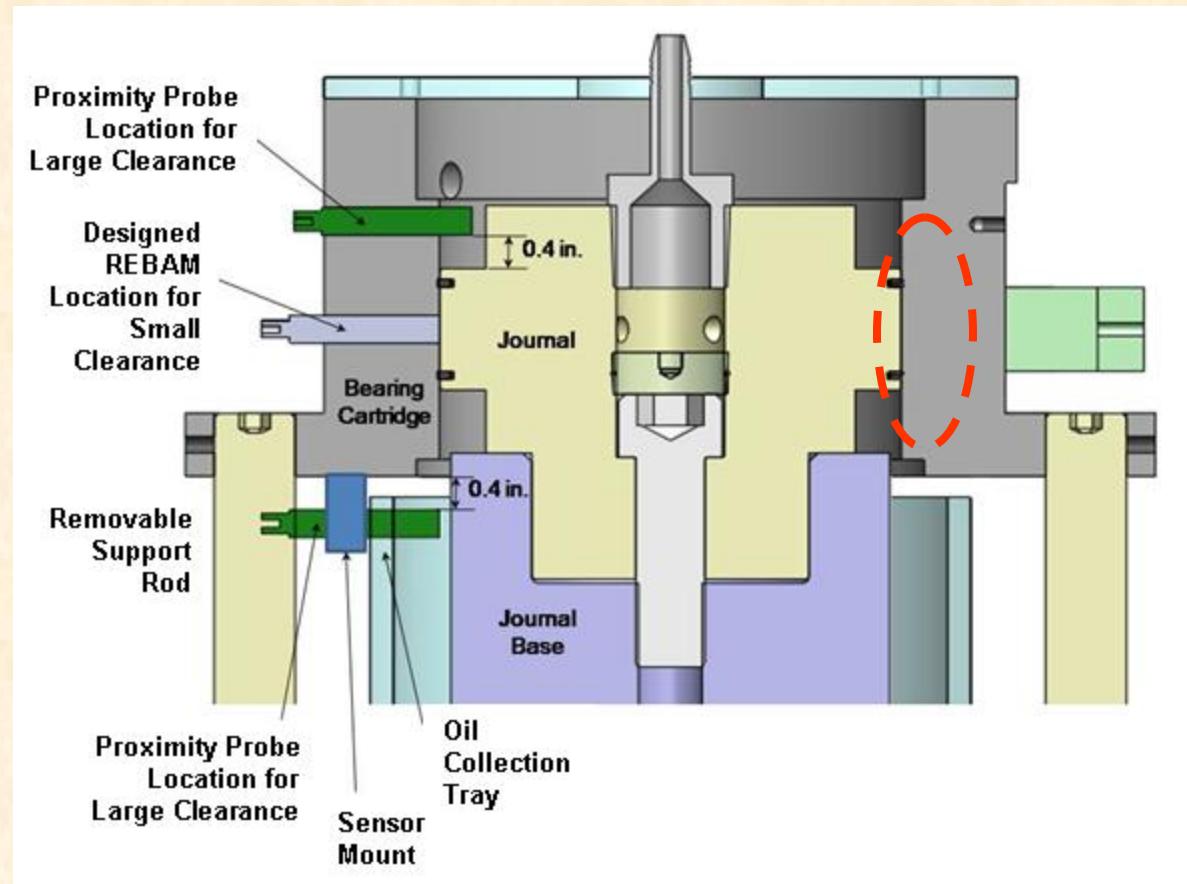
Conducted measurements of dynamic load response in large clearance ($c=9.9$ mil) open ends SFD with circular orbits, centered and off-centered.

- Central groove is NOT a zone of constant pressure: dynamic pressures as large as in film lands.
- Classical theory predicts too low SFD added masses:
1/7 of test values
- Using an effective shallow groove depth, new model predictions agree well with test results.

P&W funded project (2012)



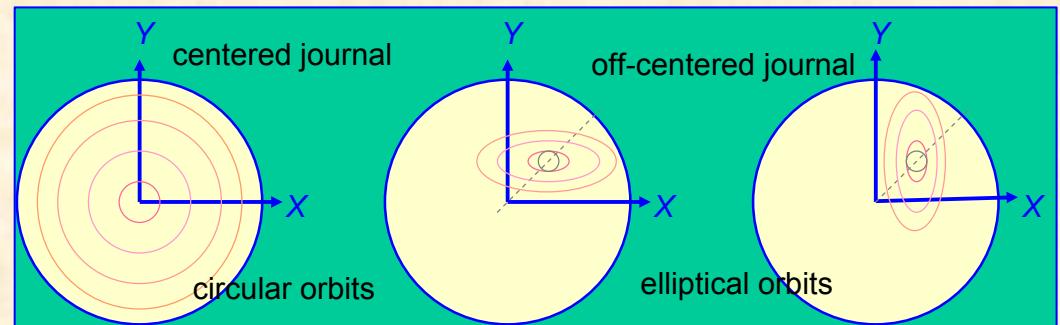
Modify test rig and construct SFD w/o a central groove, conduct measurements of film pressures and identify force coefficients.



Proposed tasks TRC (2012-13)



1. Test damper w/o groove with dynamic loads (20-300 Hz) inducing off-centered elliptical orbital motions to reach 0.8c.
2. Identify SFD force coefficients from test impedances, and correlate coefficients with linear force coefficients and experimental coefficients for smallest whirl amplitude (0.05c).
3. **Perform numerical experiments**, similar to the physical tests, to extract linearized SFD force coefficients from the nonlinear forces. Quantify goodness of linear-nonlinear representation from an equivalence in mechanical energy dissipation.



TRC Budget (2012-13)

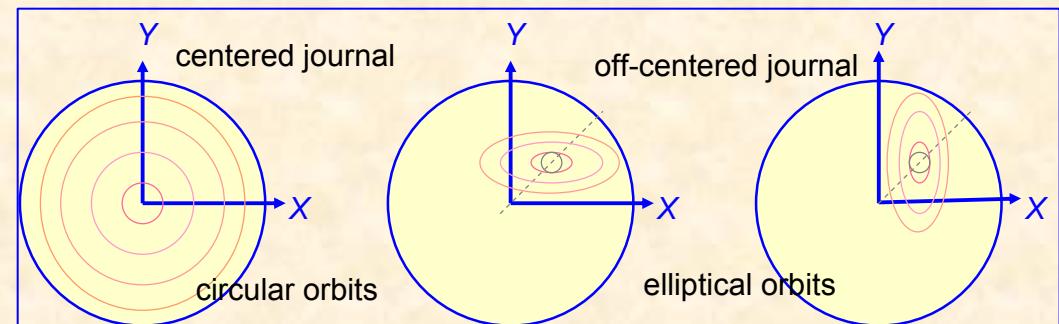


eight months

Year II

Support for graduate student (20 h/week) x \$ 2,200 x 8 months	\$ 17,600
Fringe benefits (0.6%) and medical insurance (\$197/month)	\$ 1,682
Travel to (US) technical conference	\$ 1,200
Tuition three semesters (\$227 credit hour x 15 ch x 1.7 fees multiplicative factor)	\$ 5,789
Supplies for test rig	<u>\$ 2,200</u>
	Total Cost: \$ 28,470

Year I started
on Jan 2012





Acknowledgments

Thanks to

- Pratt & Whitney Engines
- Turbomachinery Research Consortium
- Sung-Hwa Jeng, RA for making the presentation

Learn more

<http://rotorlab.tamu.edu>

Questions (?)