

**Special Lecture
2022 Aircraft Engine
Technology Award**



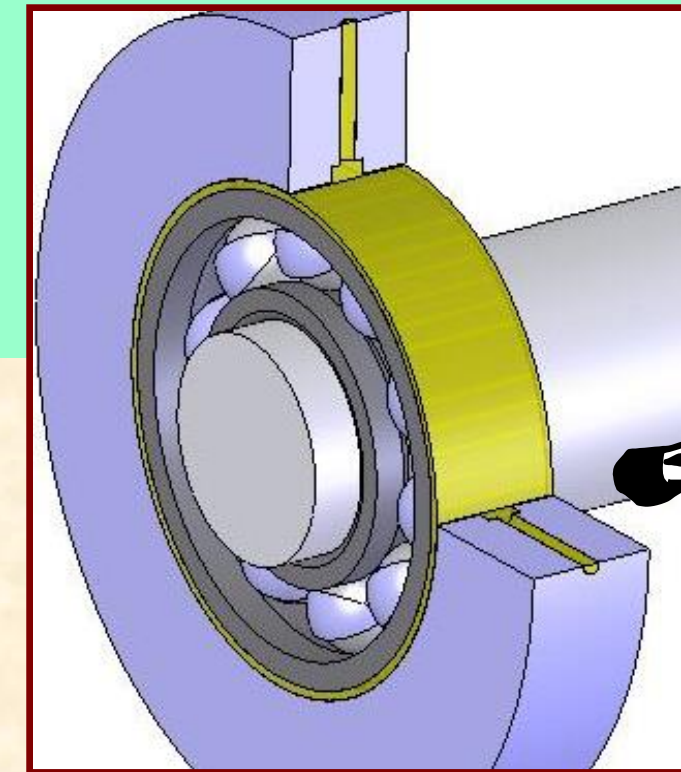
TURBO EXPO 2022

Turbomachinery Technical Conference & Exposition

JUNE 13–17, 2022

ROTTERDAM AHOY CONVENTION CENTRE
ROTTERDAM, NL

MEASUREMENTS AND MODELS OF SQUEEZE FILM DAMPERS' FORCED RESPONSE AND A BIRD'S VIEW TO AIR INGESTION & ENTRAPMENT



Luis San Andrés

Mast-Childs Chair Professor

Fellow ASME



TEXAS A&M UNIVERSITY

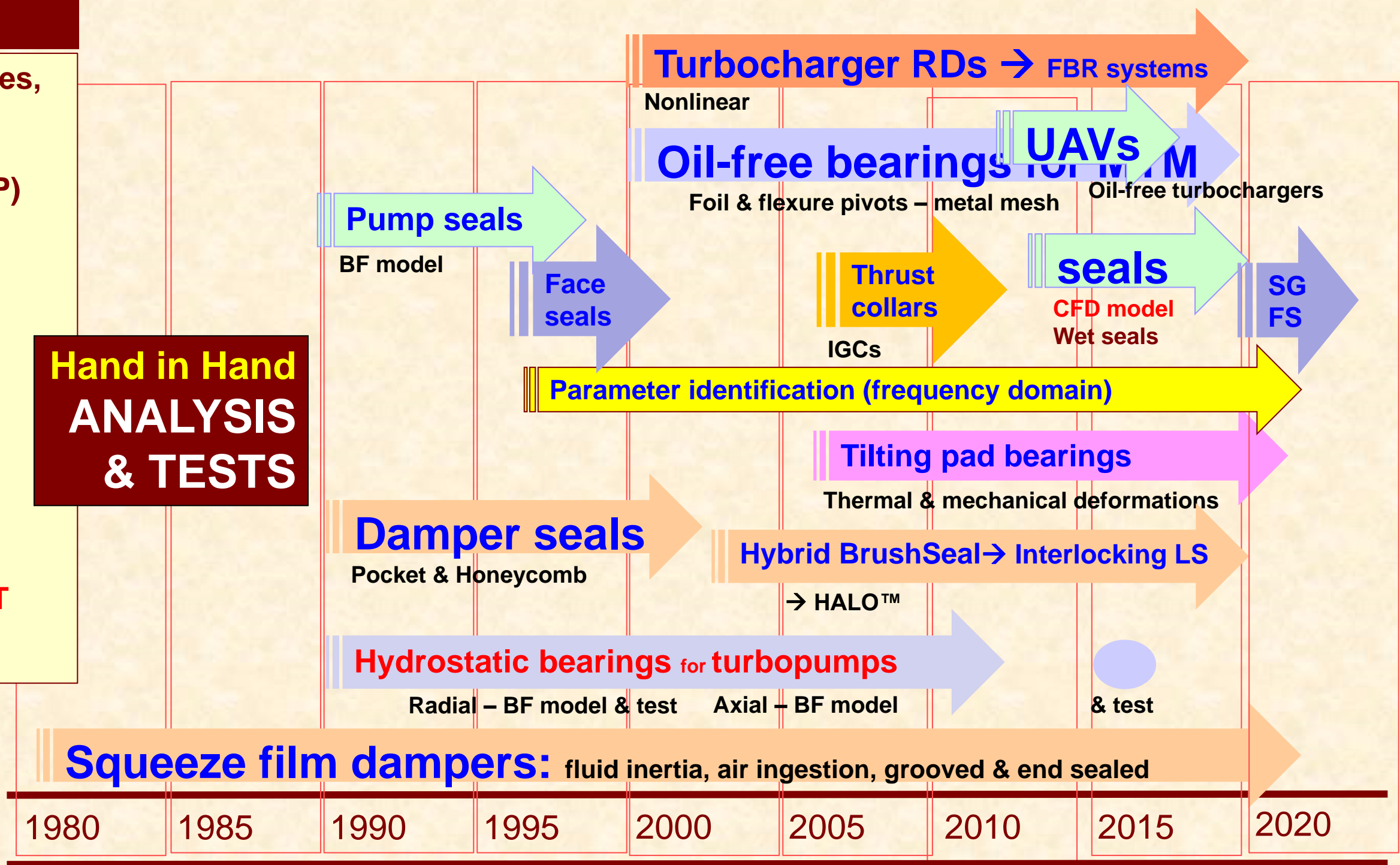
J. Mike Walker '66 Department of
Mechanical Engineering

Timeline since dinosaur age

Funding Sources

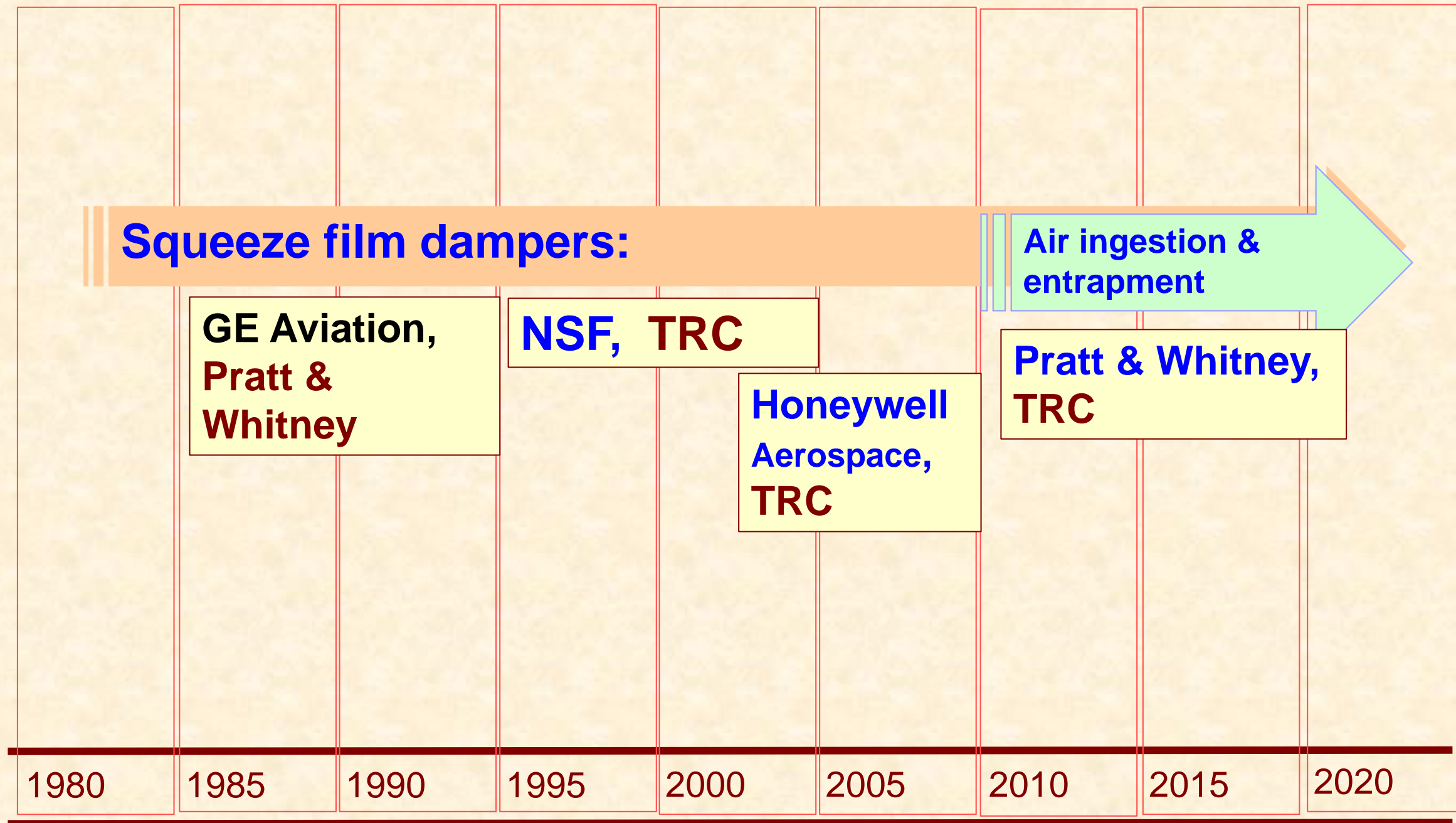
John Crane, Baker-Hughes, Trane, Elliott Co.
 Blue Origin,
 Army Research Lab (CUP)
 NSF, NASA GRC,
 Pratt & Whitney
 Northrop Grumman
 Rocketdyne
 Honeywell TT,
 Danfoss TurboCor
 Borg-Warner TC,
 Torishima Pumps
 MHI, Hitachi RL,
 Samsung, Key Yang,
 Hyundai HI, Capstone MT
 Siemens TRC

Hand in Hand ANALYSIS & TESTS



1980 1985 1990 1995 2000 2005 2010 2015 2020

Today a bird's view into SFD forced performance



TRC: Turbomachinery Research Consortium

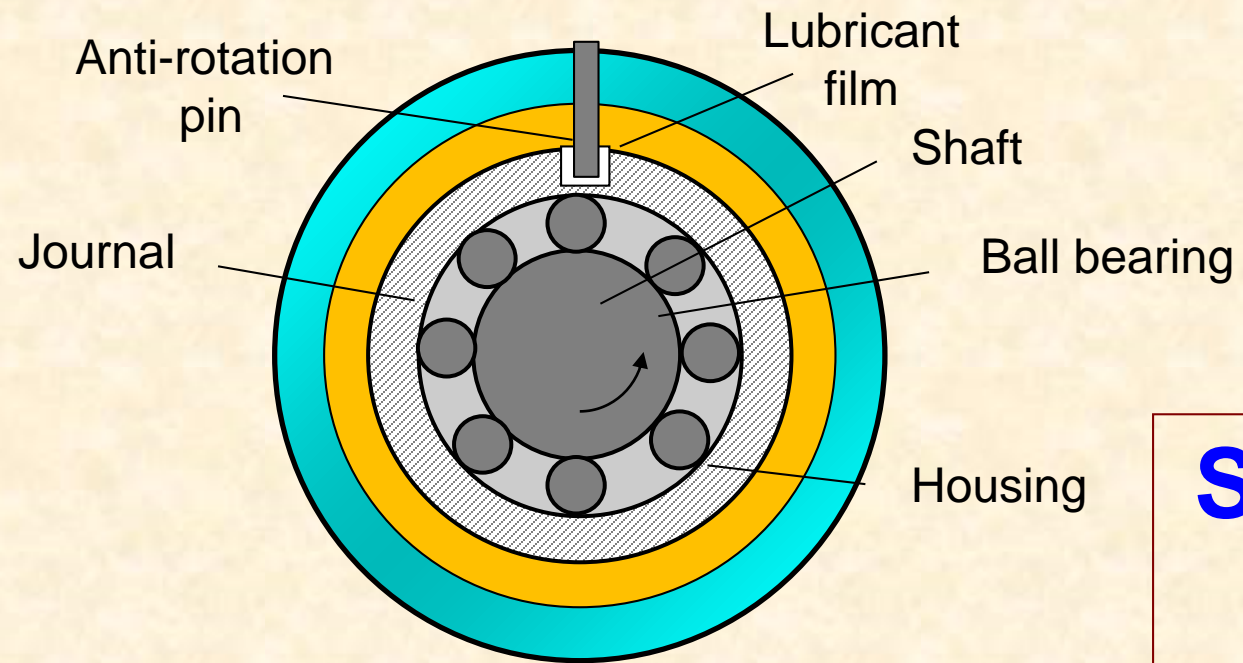
The early quest – last millennium

Whose woods these are I think I know.....

**The woods are lovely, dark and deep,
But I have promises to keep,
And miles to go before I sleep,
And miles to go before I sleep.**

Robert Frost, “Stopping by Woods on a Snowy Evening”

Squeeze Film Dampers (SFDs)



In a SFD, the **journal whirls but does not spin**. The lubricant film is squeezed due to rotor motions, and fluid film (damping) forces are generated

SFDs aid to attenuate rotor vibrations, suppress system instabilities, and provide mechanical isolation.

Too little damping may not be enough to reduce vibrations.

Too much damping may lock damper & will degrade system performance.

Most common problems in rotordynamics

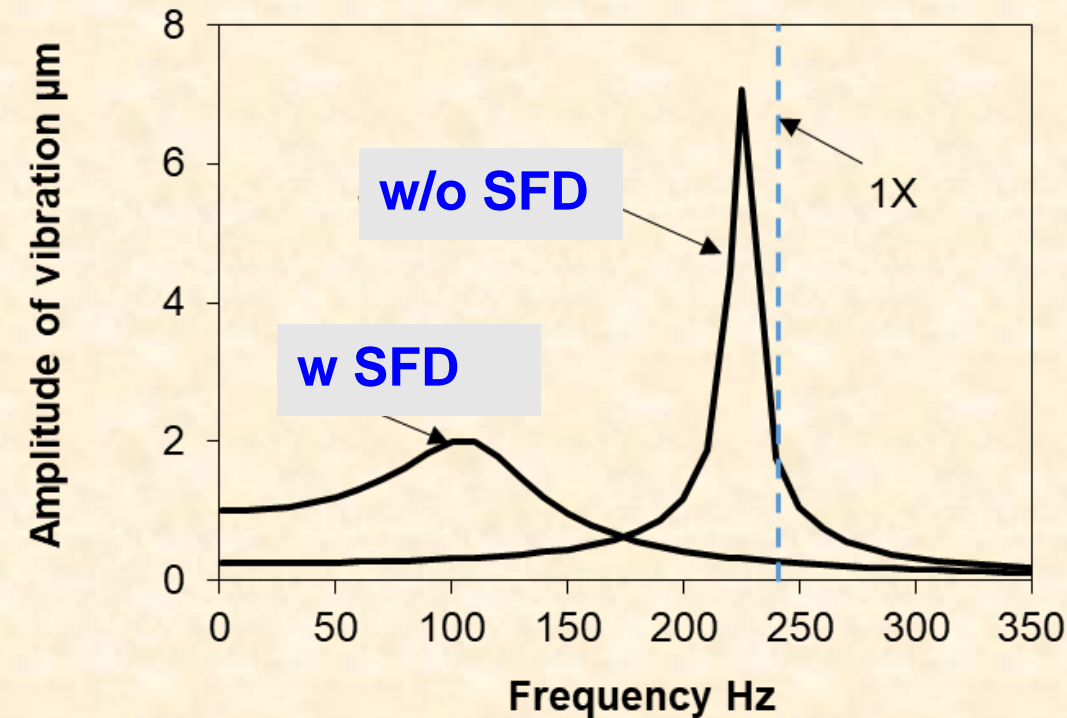
1. Excessive steady state synchronous vibration levels:

Improve balancing.

Modify rotor-bearing systems: tune system critical speeds out of RPM operating range.



Introduce damping to limit peak amplitudes at critical speeds that must be traversed.



2. Subharmonic rotor instabilities

Eliminate instability mechanism, i.e. change bearing design if oil whip is present.

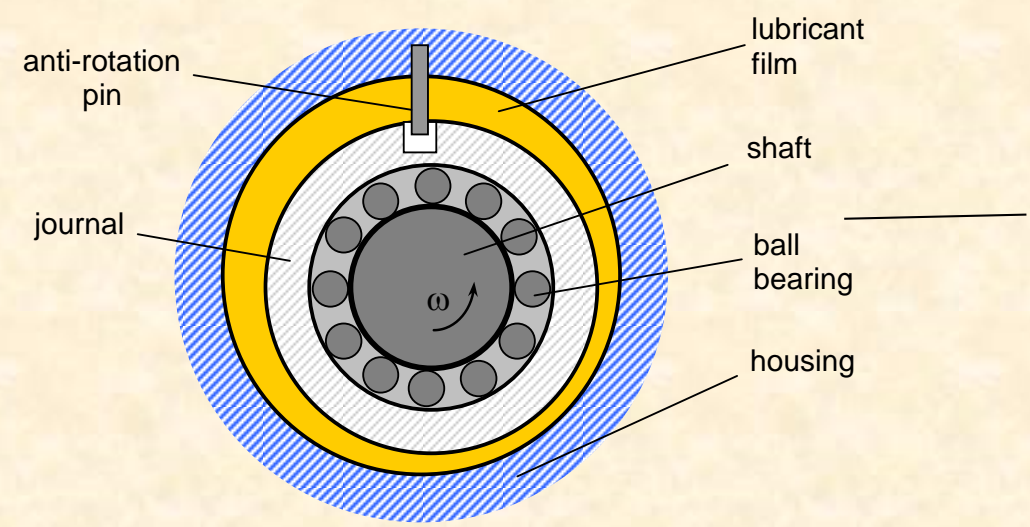
Rise natural frequency of rotor system as much as possible.



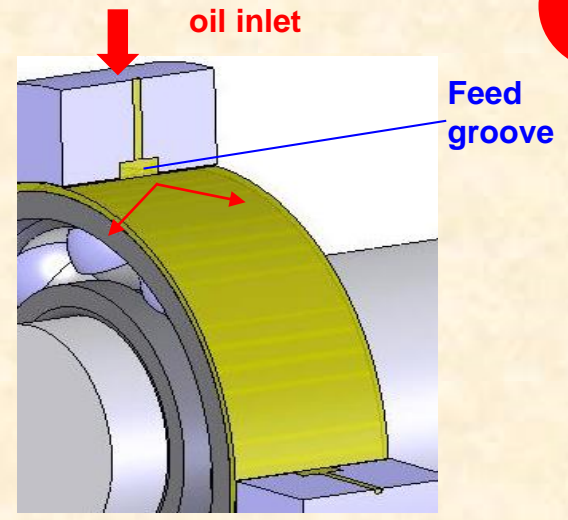
Introduce damping to increase onset rotor speed above the operating speed range.

SFD configurations

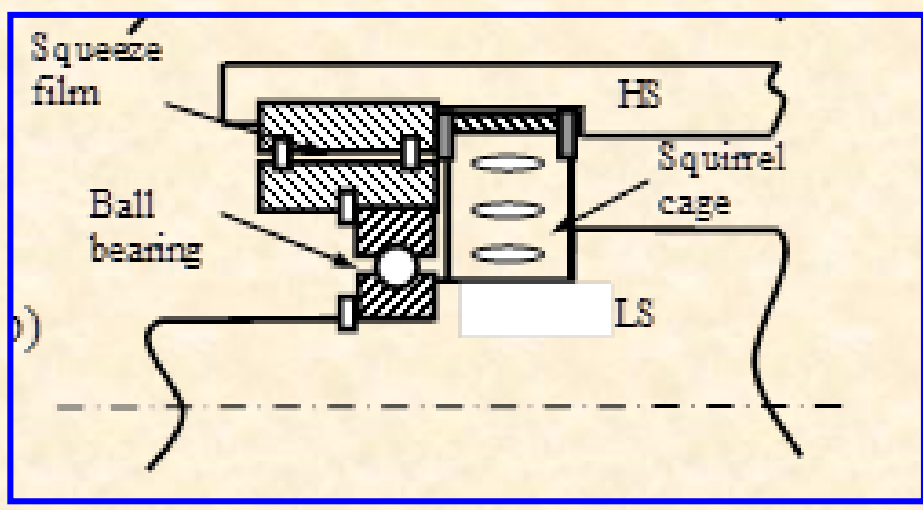
a SFD with anti-rotation pin



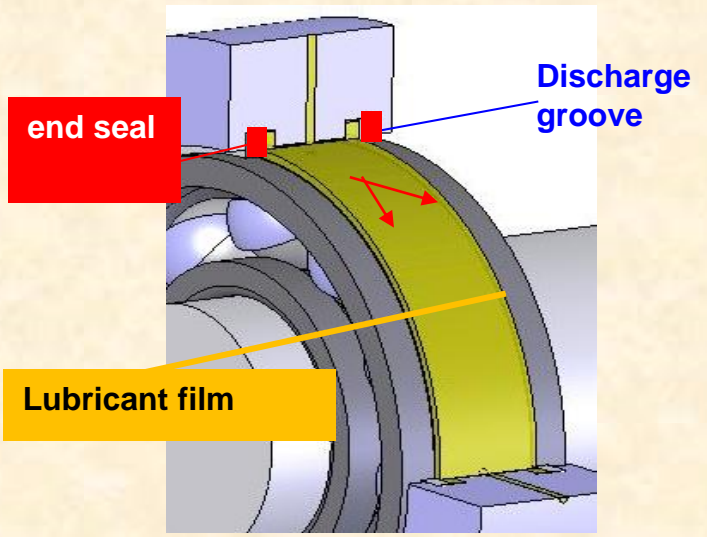
c



b SFD with elastic cage



d



(c) with a supply groove - open ends

(d) with a supply hole - end seals

Brief history of SFDs

Parsons (1889)

Discloses first use of a SFD in first modern-day steam turbine.

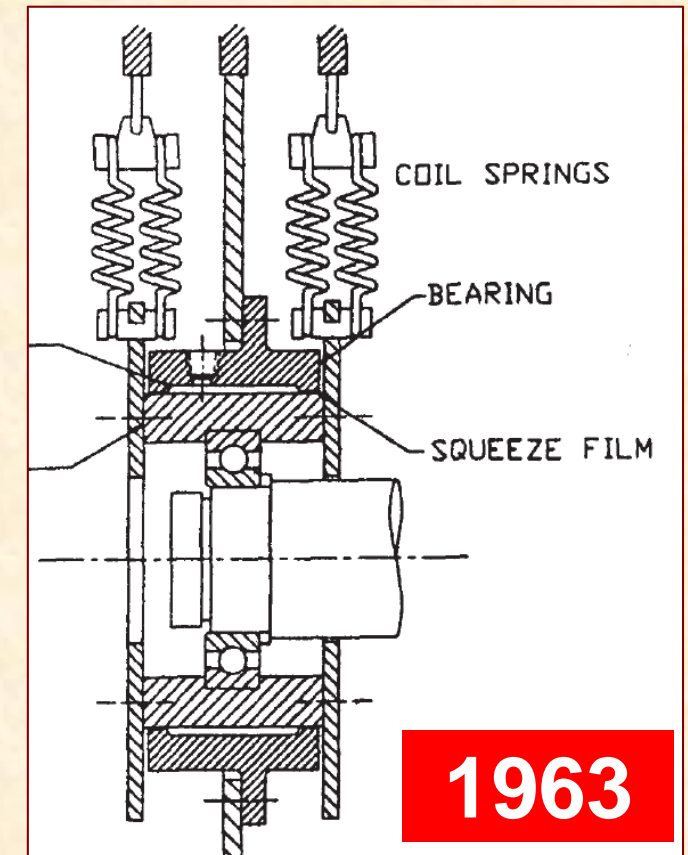
Cooper (1963)

Rolls Royce engineer investigates experimentally the performance of rotating machinery with a SFD.

Since the 1970s, SFDs are essential components in aircraft engines and high pressure centrifugal compressors.



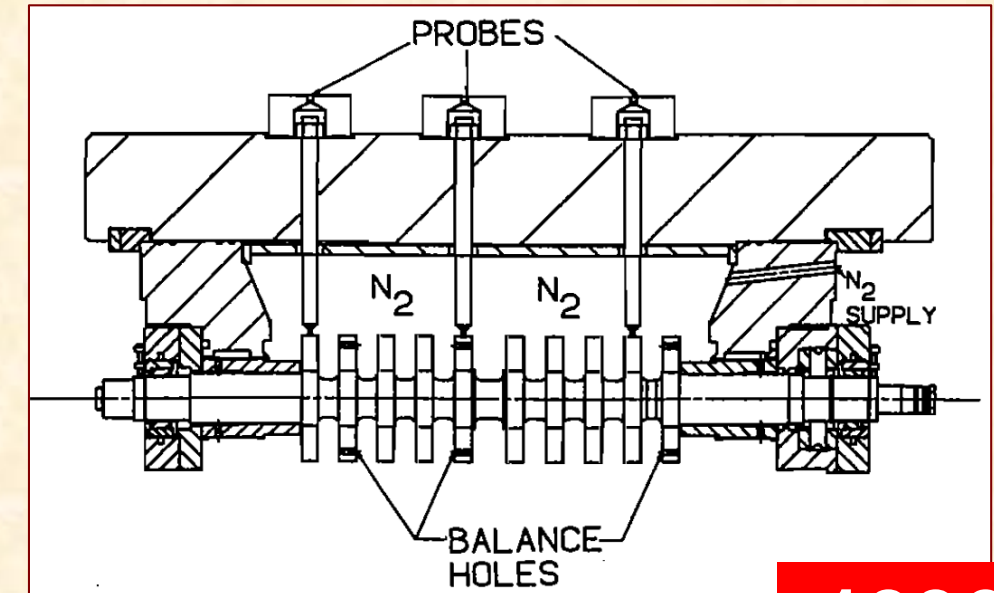
Marine Parsons turbine
(by Topori at Polish Wikipedia)



Brief history of SFD (Turbomachinery Symposium)

Zeidan et al. (1996)

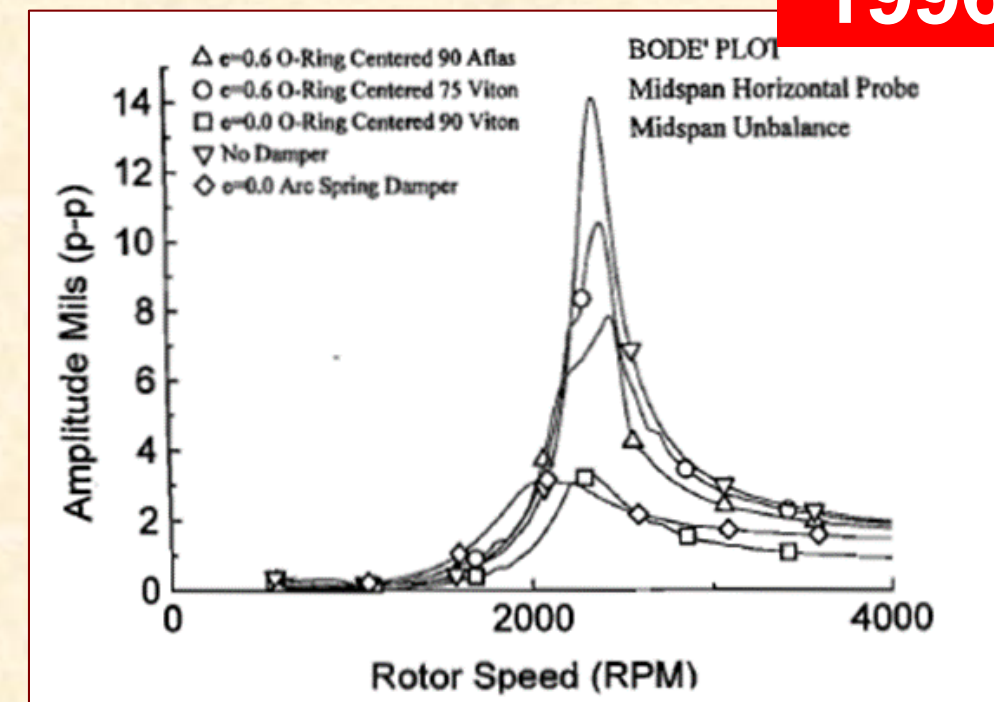
Discuss major technical issues for SFD integration into turbomachinery: oil cavitation vs. air ingestion and fluid inertia effects.



1996

Kuzdal and Hustak (1996)

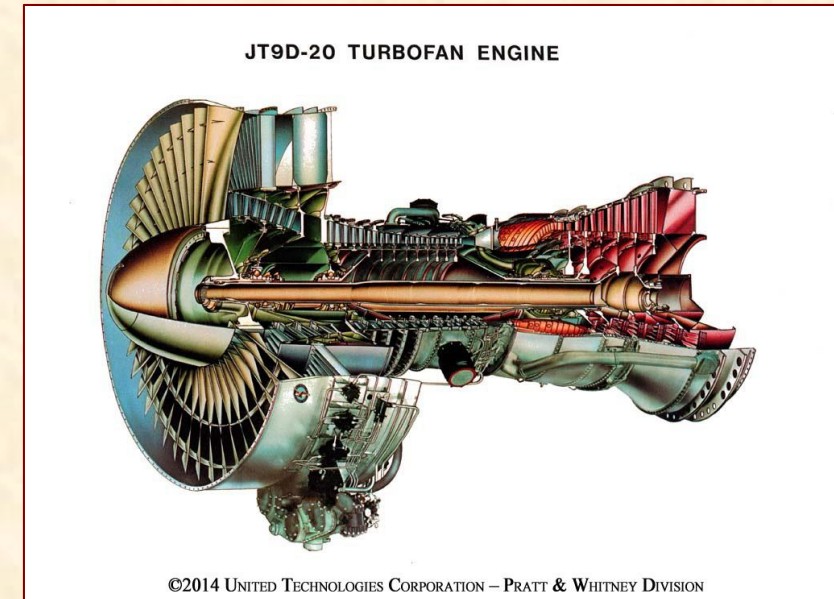
Test various damper configurations (open and sealed ends) → an optimized SFD reduces rotor synchronous motions and improves the stability threshold of rotor bearing systems.



SFD applications

Jet engines with rolling element bearings:

- a) To reduce synchronous peak amplitudes,
- b) To limit peak amplitudes at critical speeds,
- c) To isolate structural components (lower transmissibility), and
- d) To provide a margin of safety for blade loss.



Hydrocarbon compressors & steam turbines

- a) To stabilize unit by introducing damping and reducing cross-coupled effect of seals, hydrodynamic bearings, etc.
- b) To enhance limited damping available from tilting pad bearings.

Other benefits of SFDs on rotordynamic performance:

- Tolerance to larger rotor motions * Simpler alignment
- Reduced balancing requirements * Less mount fatigue

More recent literature on SFDs

Della Pietra and Adilleta (2002): Review of research conducted on SFDs over last 40 years (since 1960s')

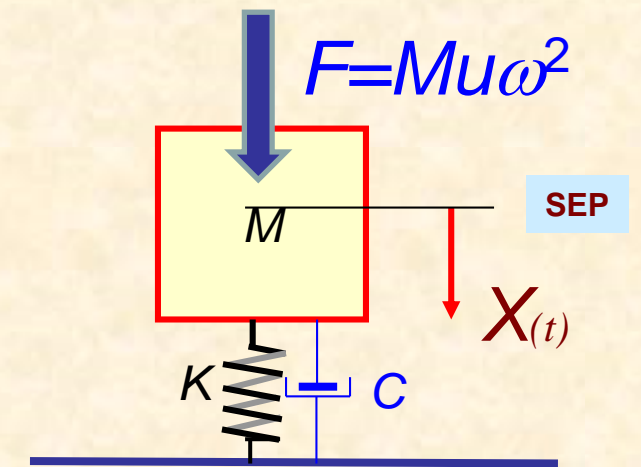
(2006-2010) San Andrés and Delgado (SFD & MECHANICAL SEAL, improved predictive models).

GT 2006-91238, GT 2007-24736, GT 2008-50528, GT 2009-50175

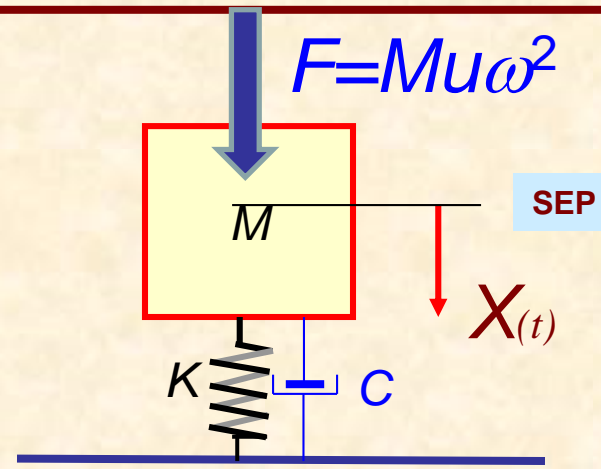
(2012-2020) San Andrés and students (SFDs for aircraft)

GT 2012-68212 , GT 2013-94273 , GT 2014-26413, GT 20015-43096,
GT 2016-43096, GT 2016-56695, 2016 A/TPS, GT2018-76224,
GT 2019-90330, GT2020-14182, GT2021-58627, GT2022-81990

How does viscous damping affect the response of a mechanical system?



1DOF spring-damper-mass system



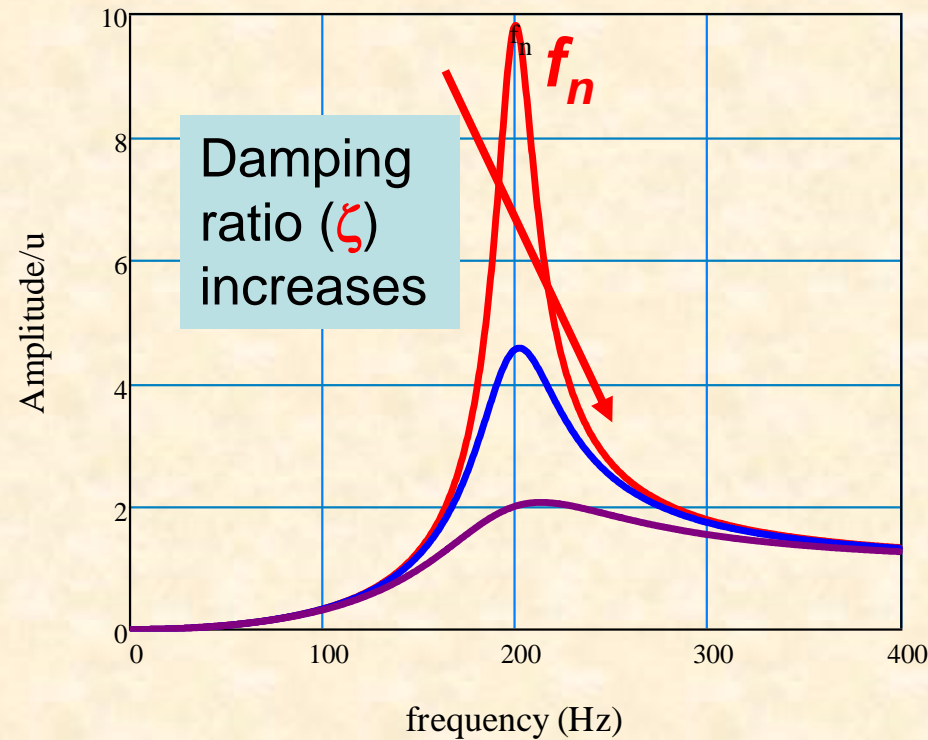
EOM: $M A x = F - F_{damper} - F_{spring}$

$$M \ddot{X} + C \dot{X} + K X = F_{(t)}$$

System response defined by natural frequency (f_n) & damping ratio (ζ)

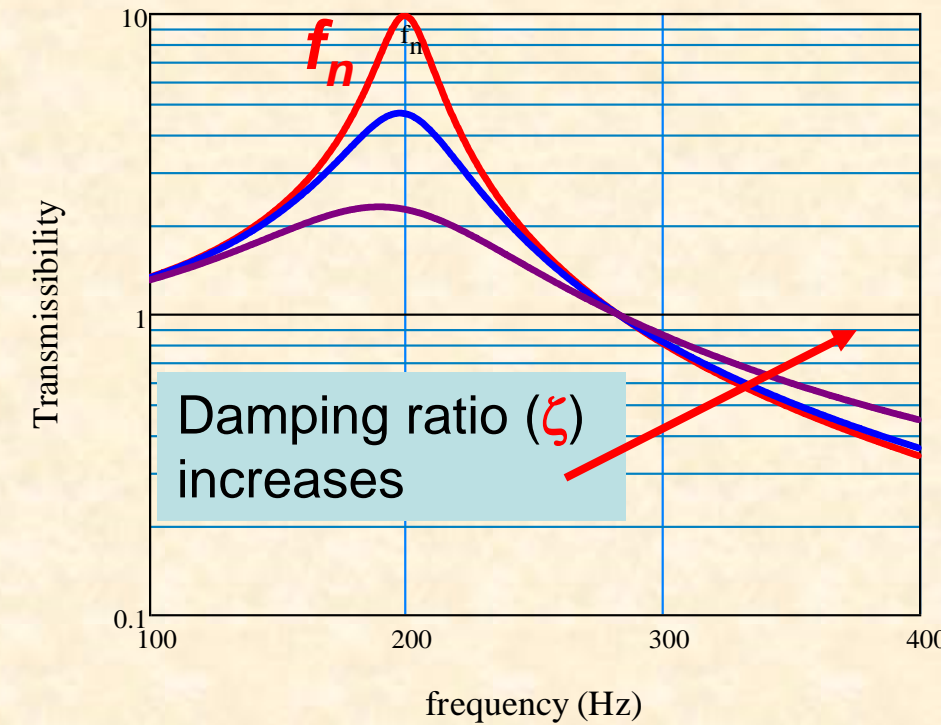
$$f_n = 2\pi \sqrt{K/M}; \quad \zeta = C / 2\sqrt{KM}$$

Response amplitude $|X/u|$



- Damping=0.05
- Damping=0.10
- Damping=0.25

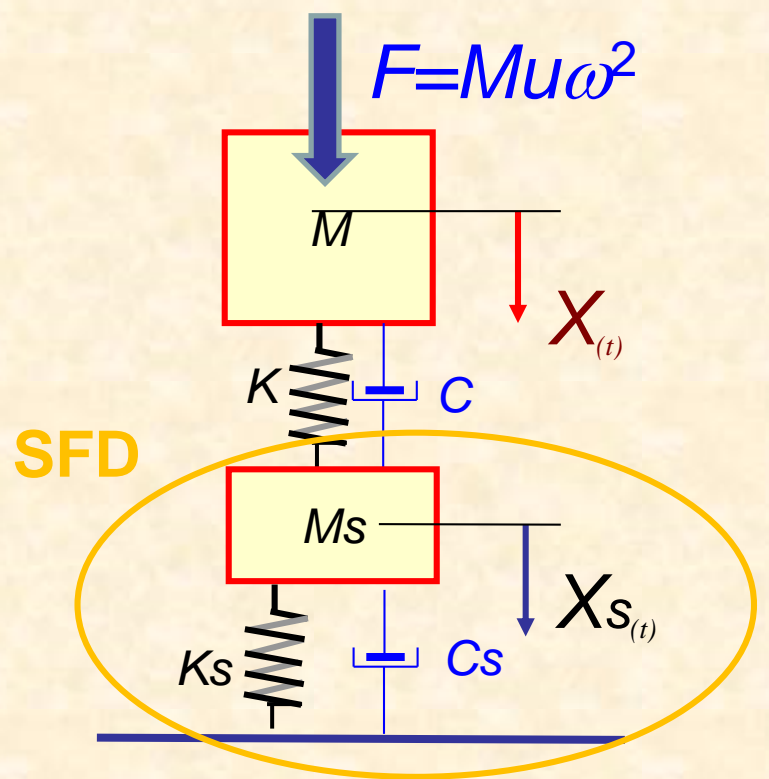
Transmissibility (to ground)



- Damping=0.05
- Damping=0.10
- Damping=0.25

Damping helps only when rotor traverses a critical speed (natural frequency= f_n) but increases force transmissibility for operation above $1.44 f_n$

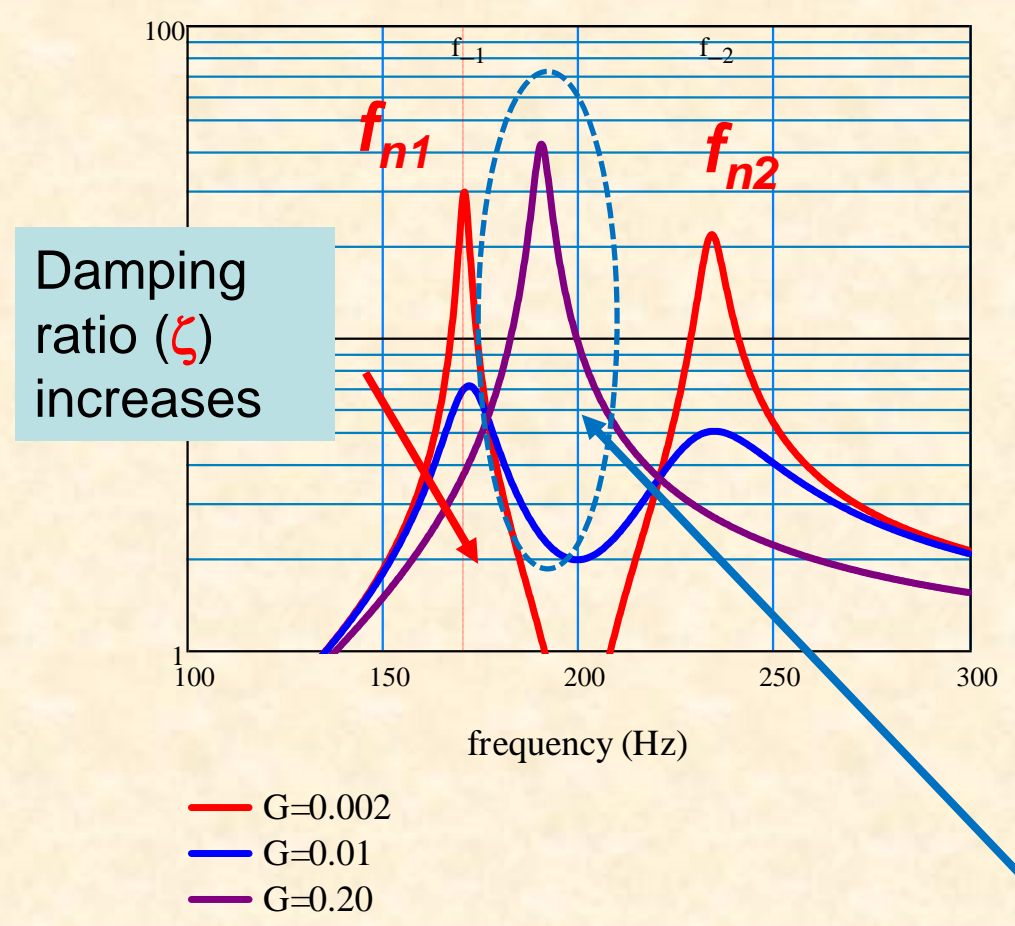
2 DOF K-C-M system : rotor on flexible supports



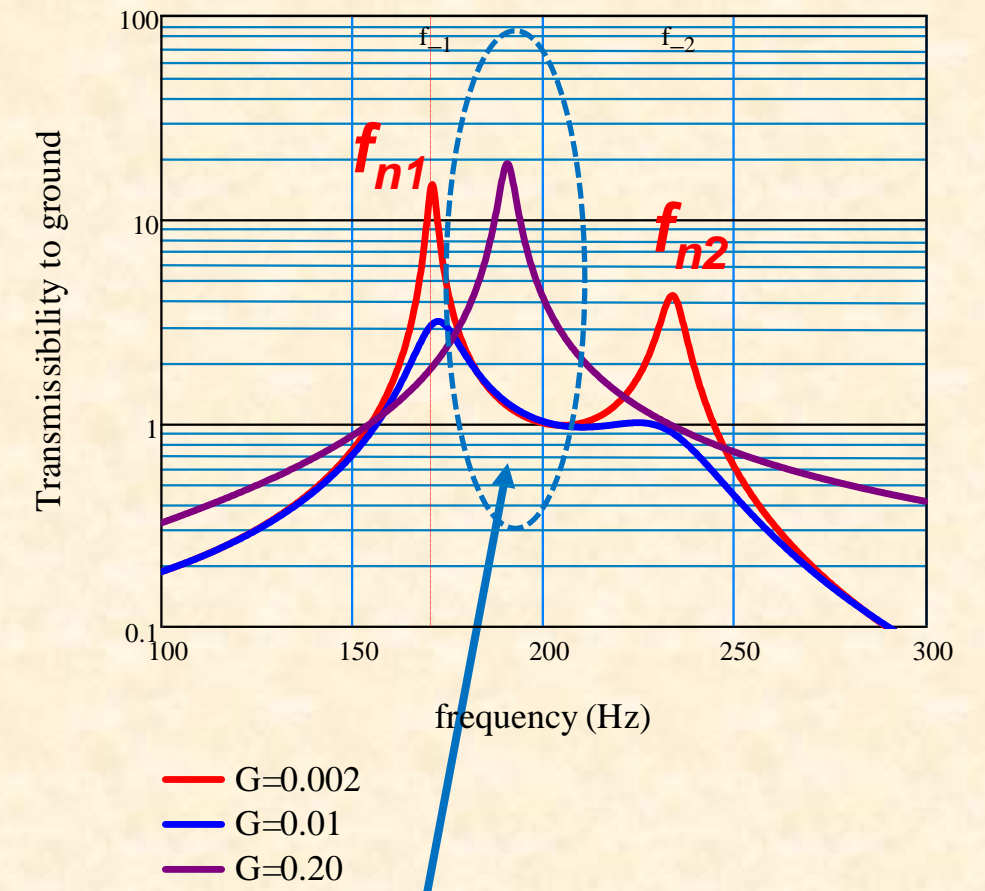
example

$$\frac{M_s}{M} = \frac{K_s}{K} = 0.1; C = 0; C_s \text{ varies}$$

Response amplitude $|X/u|$



Transmissibility (to ground)

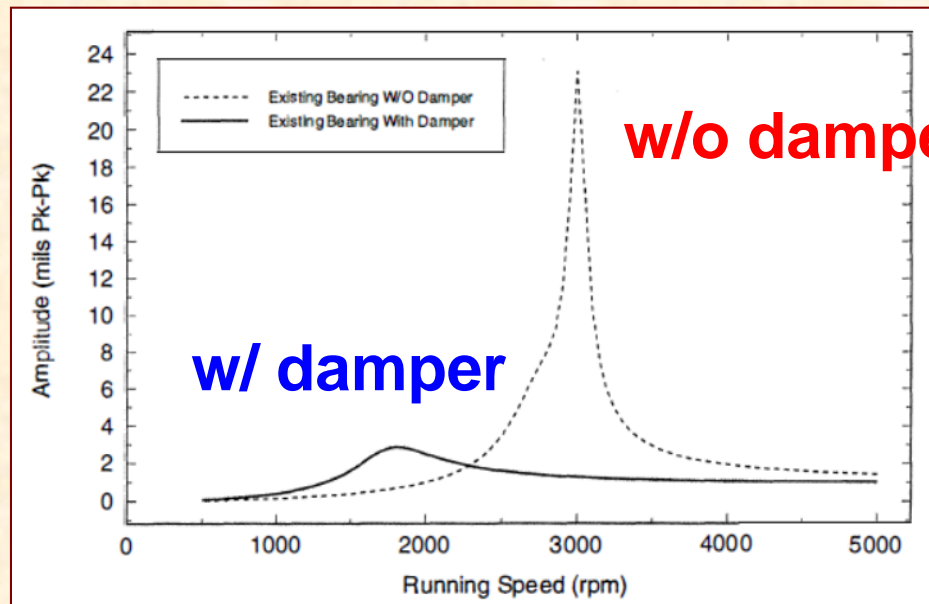


More complicated response. Damping helps only when rotor traverses a critical speed (natural frequency= f_{n1} and f_{n2}) but increases force transmissibility.

Excessive damping LOCKS supports and increases system response.

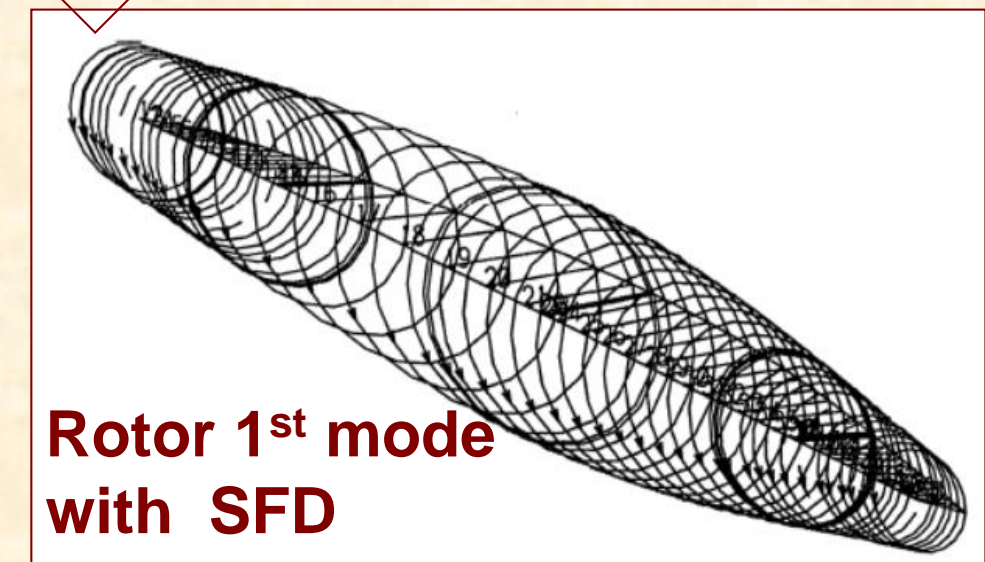
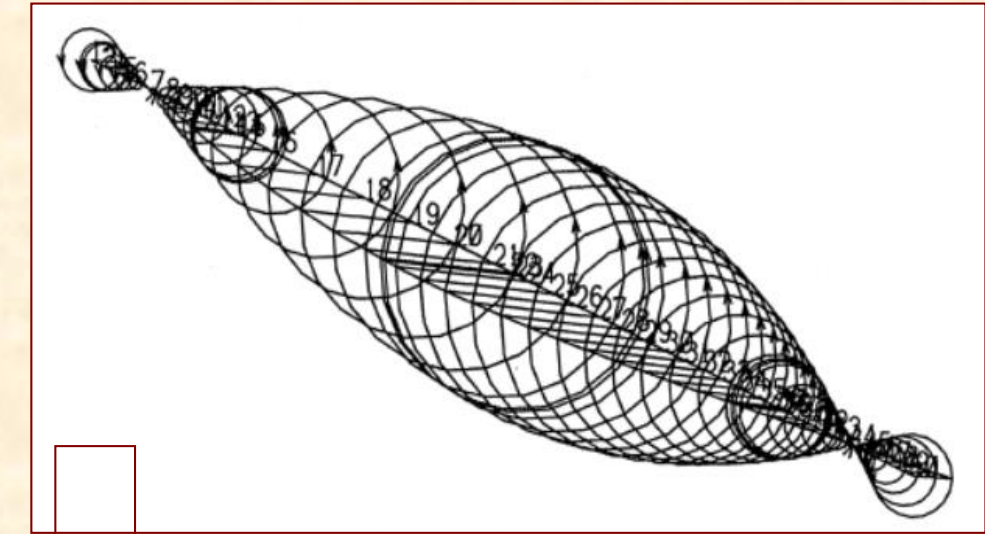
A typical SFD application

- A compressor vibrates ++ at its 1st forward mode. Bearings don't help since placed at mode-nodal locations.
- SFD support springs are soft → drop the system natural frequency and increase the effective damping ratio.
- Rotor motions greatly reduce while passing the (low) critical speed. SSV at the first forward mode eliminated.



Dampers often used to control placement of critical speed besides adding damping.

Rotor 1st forward mode w/o SFD



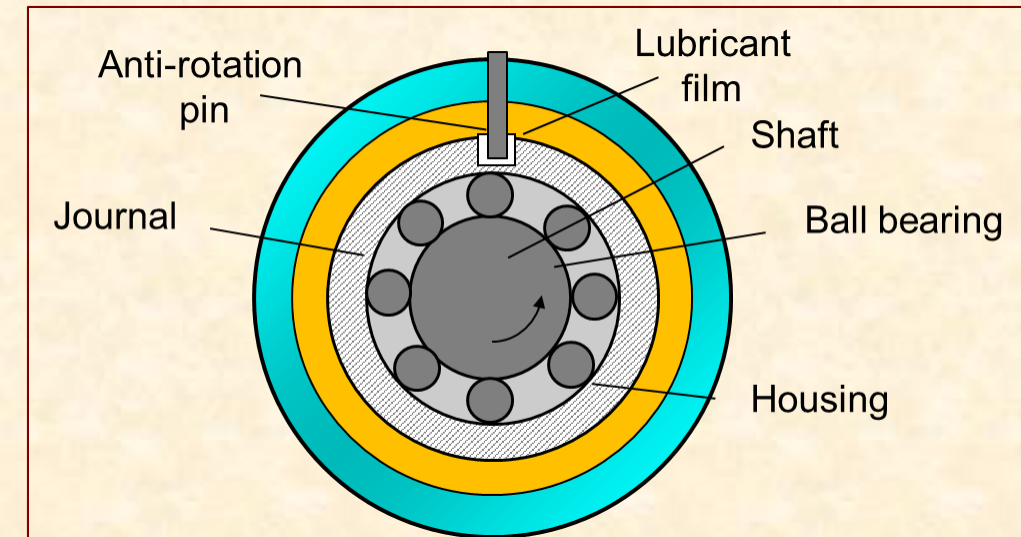
Rotor 1st mode with SFD

SFDs' fundamental design consideration

The amount of damping needed is critical.

If damping is too large the SFD acts as a rigid constraint to the rotor-bearing system with large forces transmitted to the supporting structure.

If damping is too low, the damper is ineffective and likely to permit large amplitude vibratory motion at synchronous and sub harmonic frequencies.



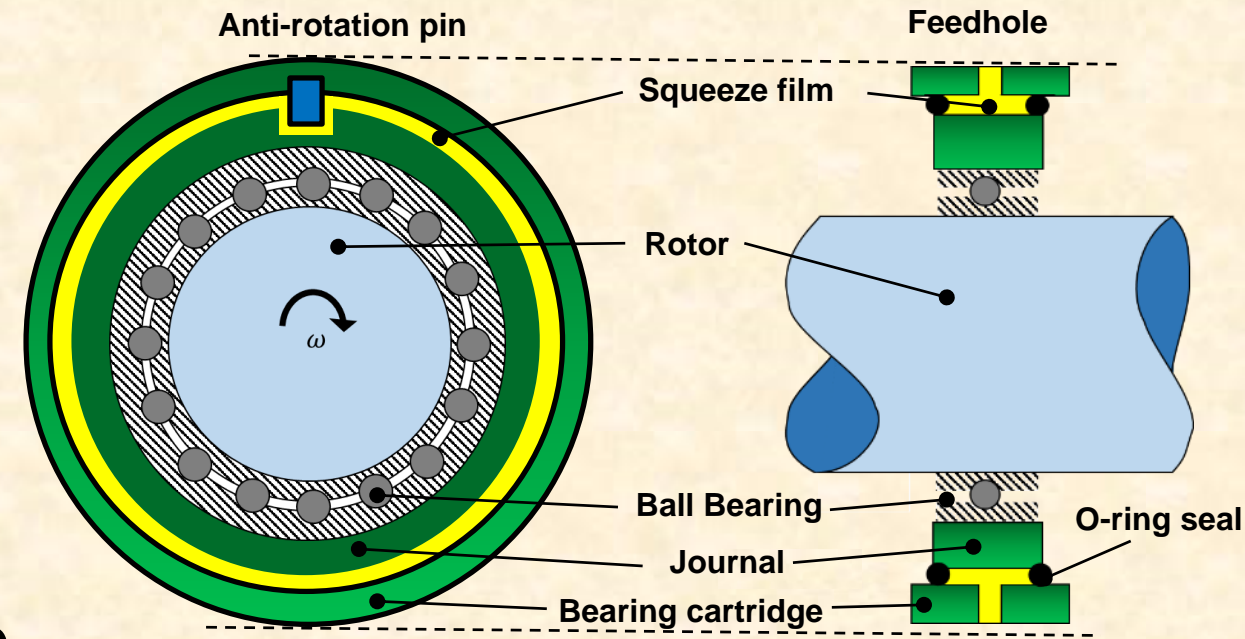
SFDs must be designed with consideration of the entire rotor-bearing system.

Physical damping is not as important as the system damping ratio!

$$\zeta = \frac{C}{C_{crit}} = \frac{C}{2\sqrt{KM}}$$

SFD performance is a function of

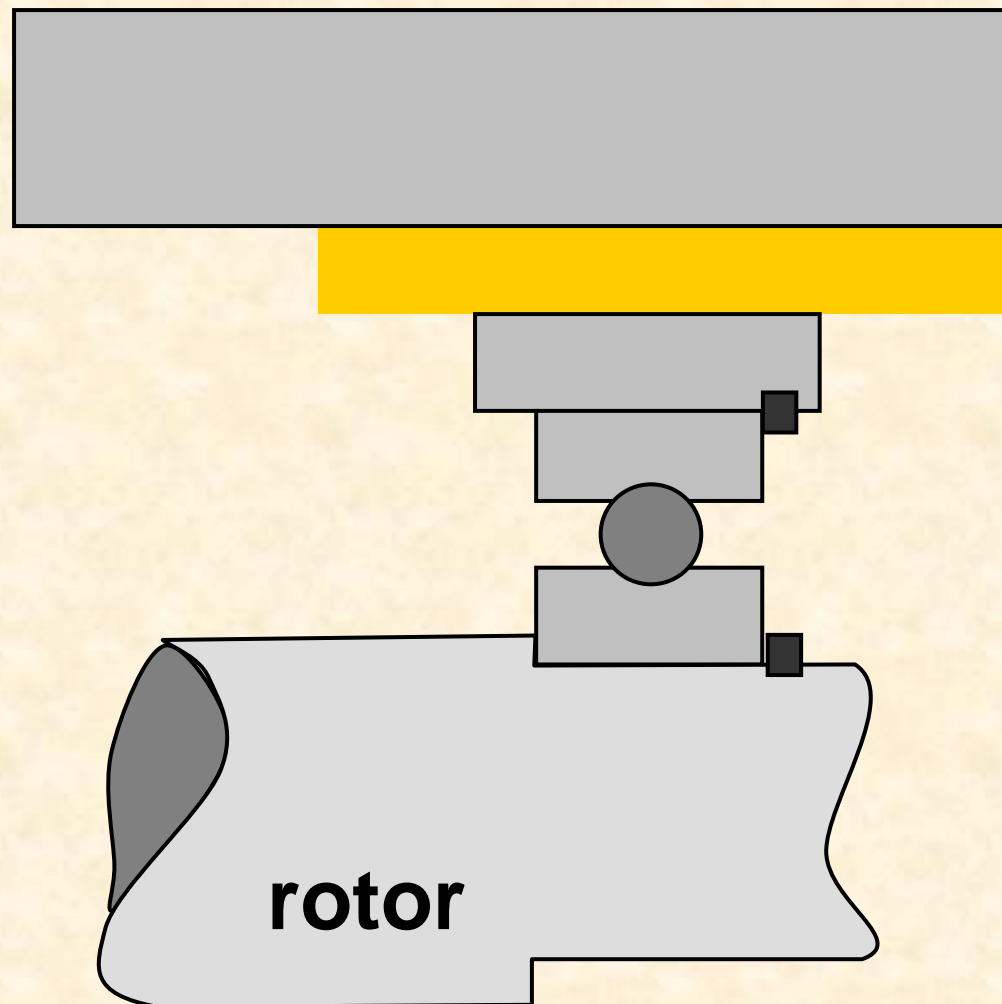
- Kinematics of rotor motion
- Geometry (land length, clearance, diameter)
- Lubricant (density, viscosity)
- Supply pressure and discharge
- Oil delivery and sealing mechanisms
- Operating speed (frequency)
- Flow Regime (lubricant cavitation: gaseous or vapor, air ingestion and entrapment)



Industry still relies on analyses that models SFDs as a **simple version** of a **hydrodynamic journal bearing**.

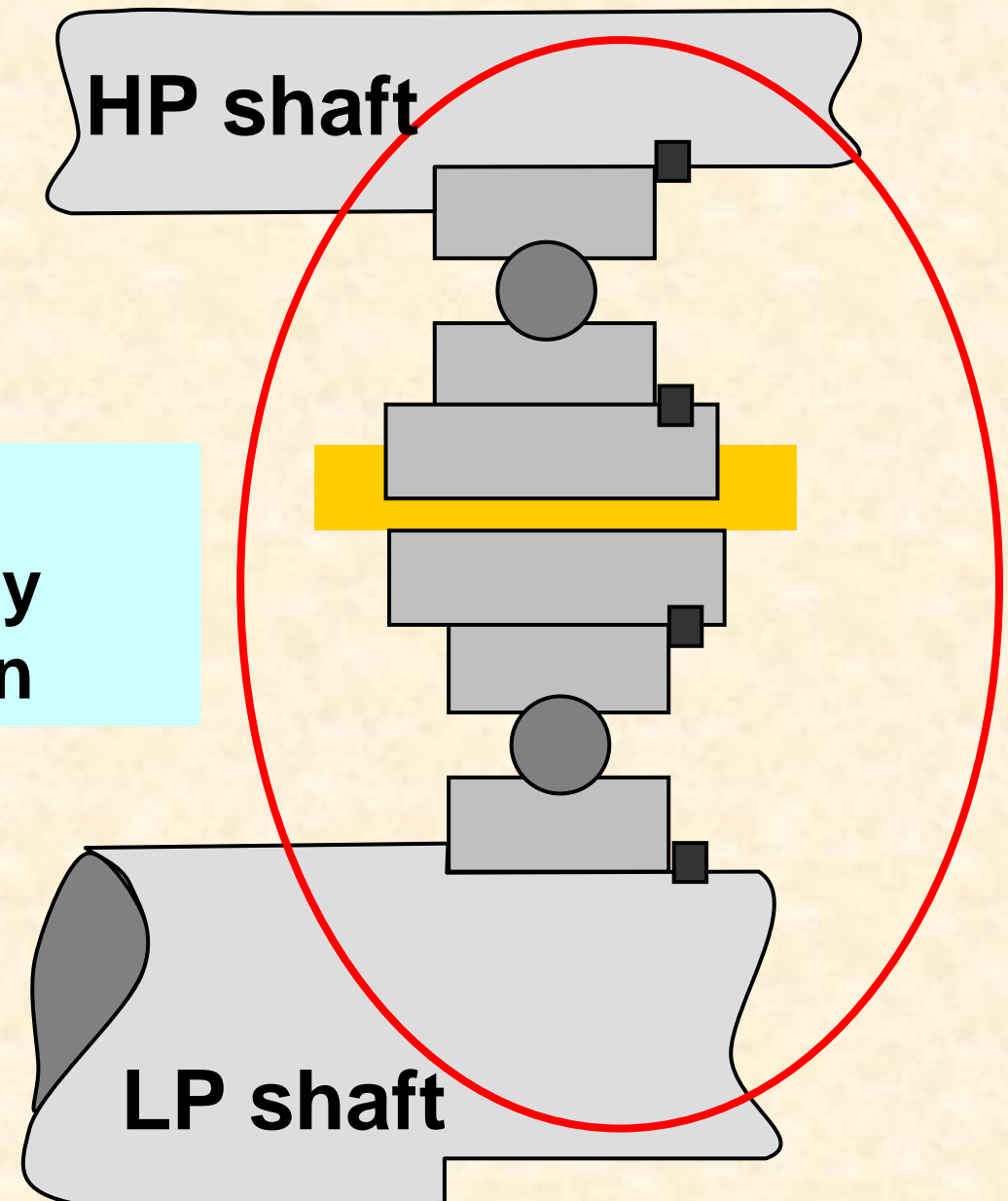
How does a SFD work? What is important?

Single frequency excitation

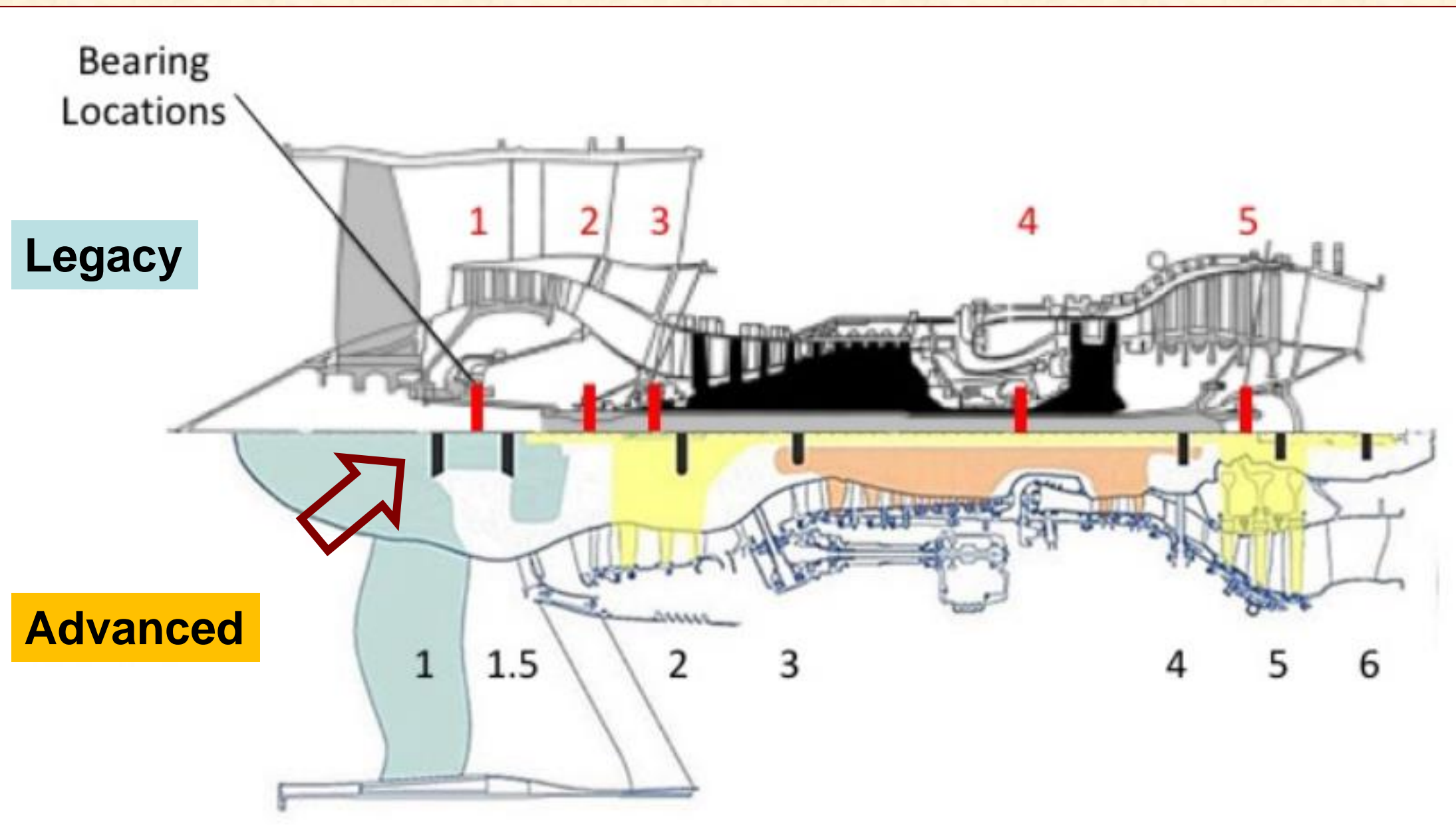


Multiple frequency excitation

Aircraft two spool rotor

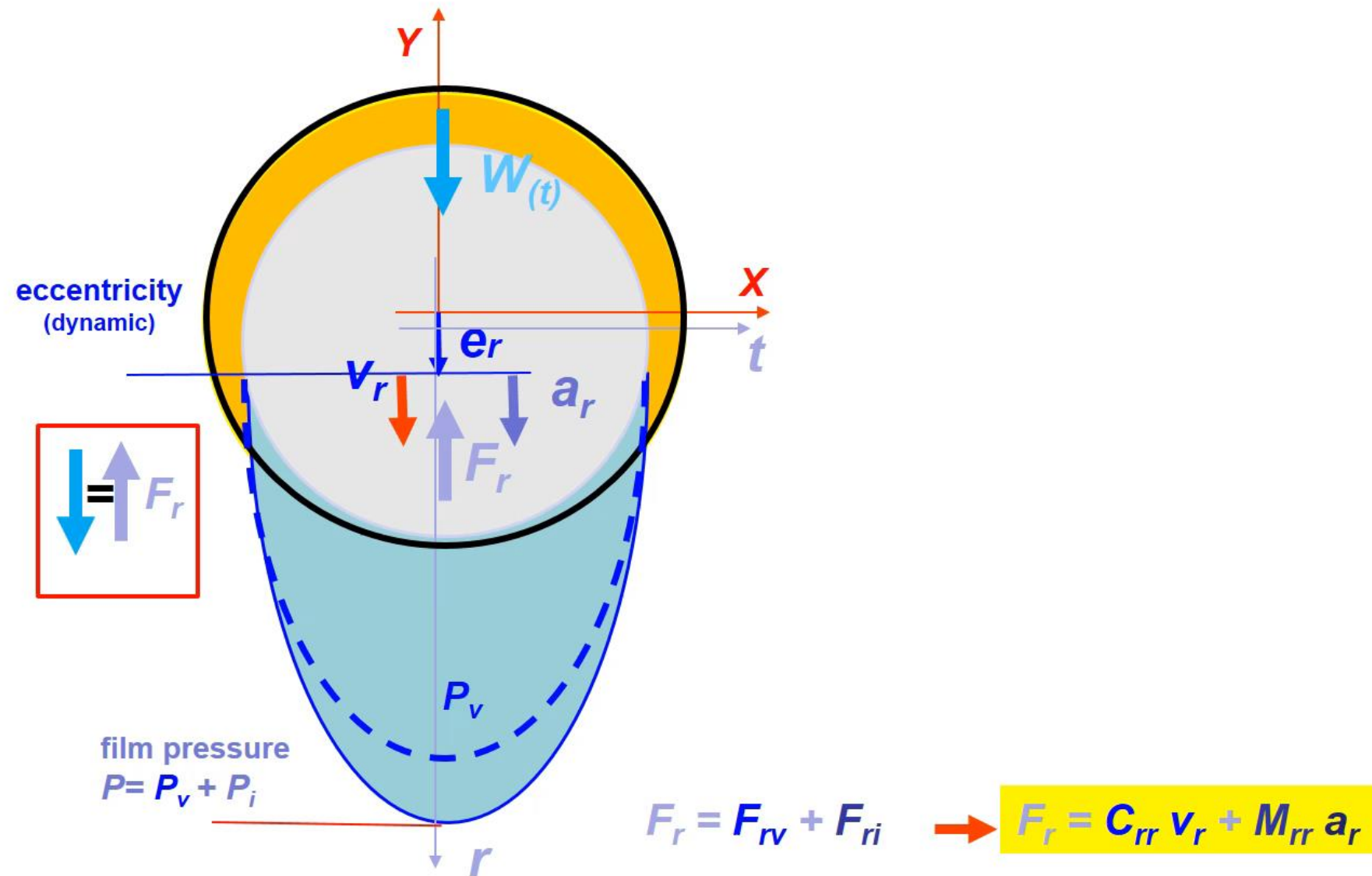


Bearings in modern aircraft



Slender & longer rotors demand more bearing supports. 4-6 SFDs are common.

SFD basics: pure squeeze (plunging motion)



Pressure field (P) changes with time.

$P \sim$ squeeze velocity (v_r) and acceleration (a_r).

Note pressure reversals.

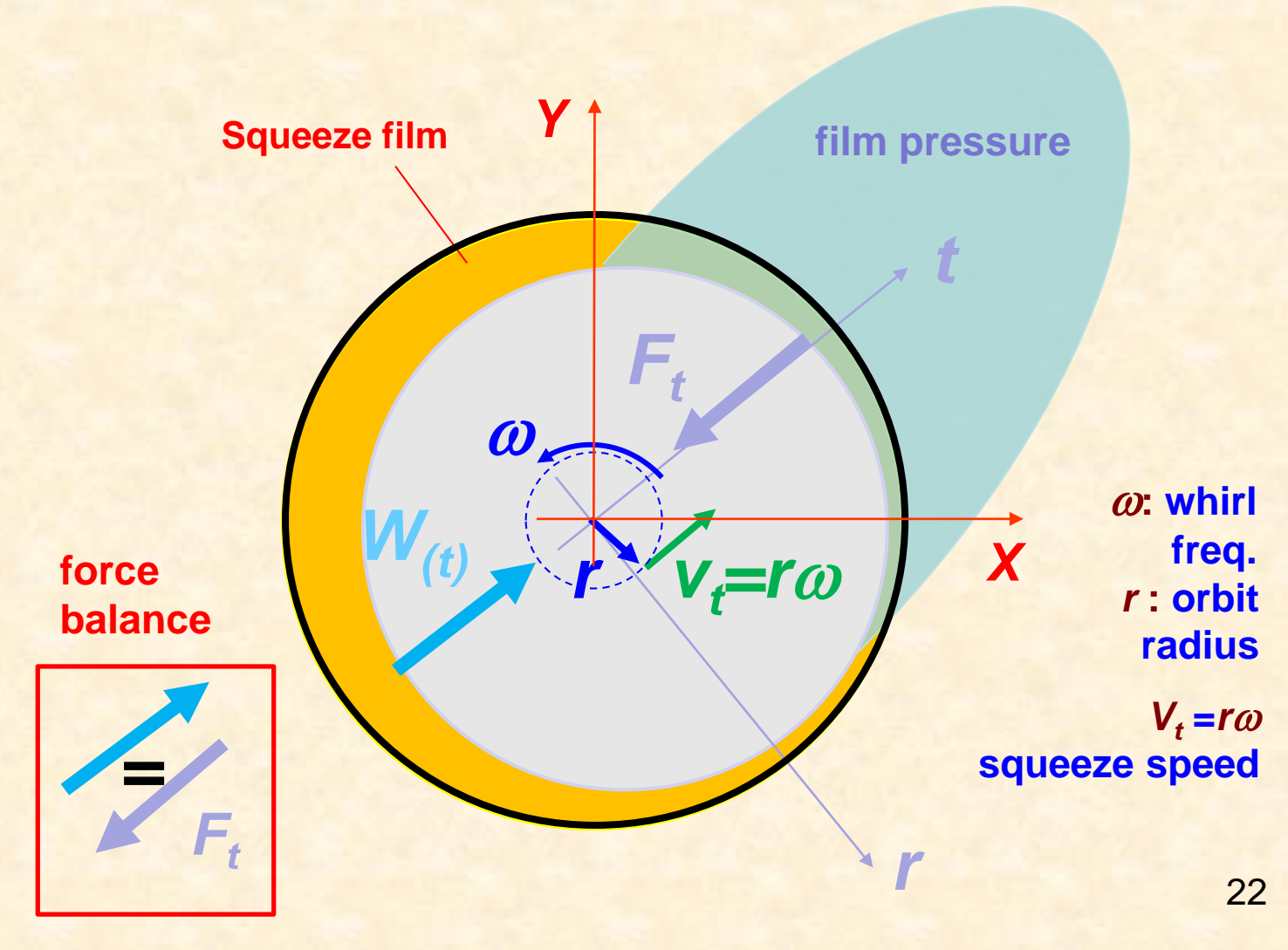
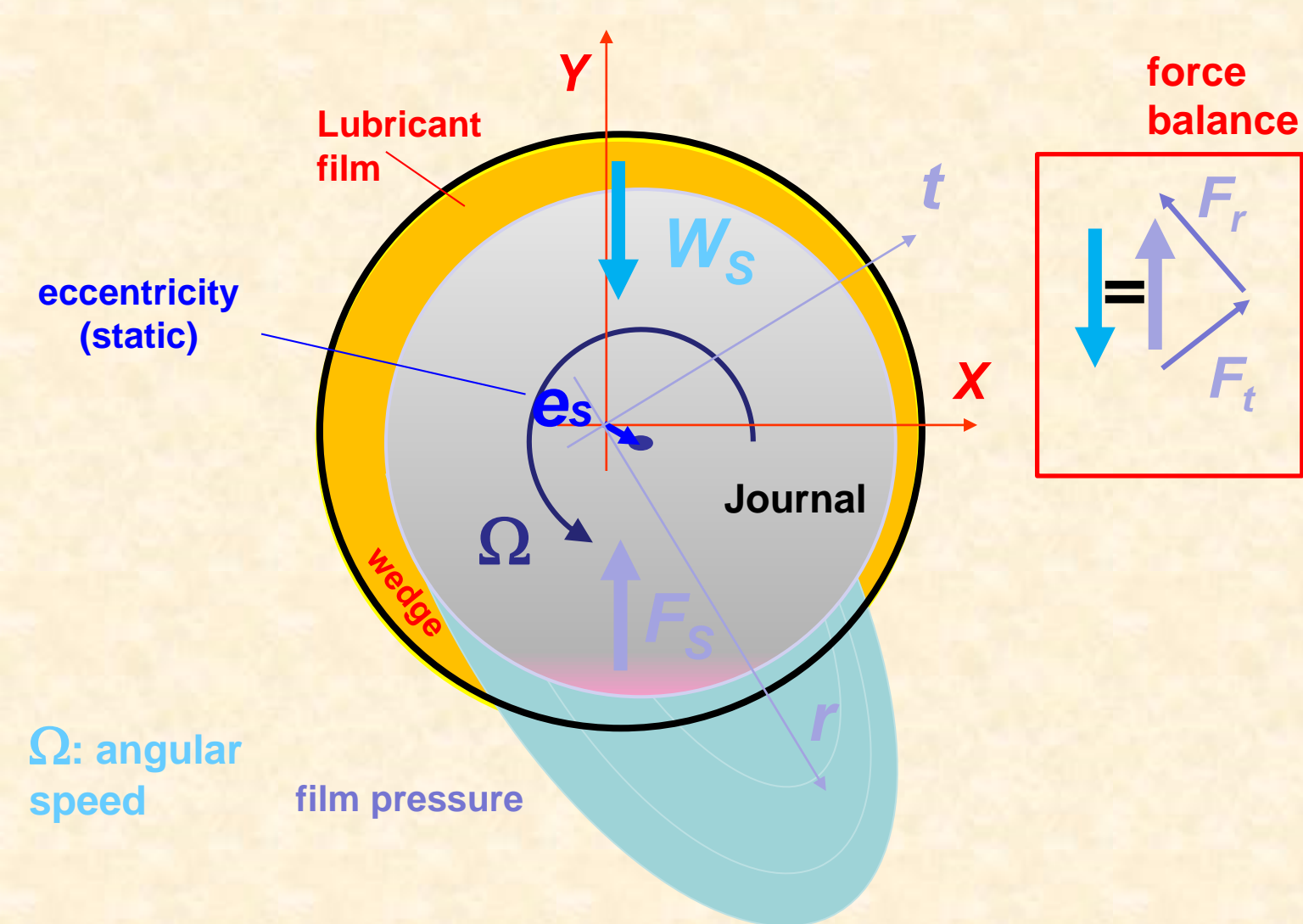
$$v_r = \dot{f}(t)$$
$$v_t = a_t = 0$$

Is a SFD a non-spinning journal bearing (JB)?

SFD & JB may have a similar configuration.

In a JB, the shaft **spins** with angular speed (Ω)

In a SFD, journal center **whirls** or **precesses**.



Fluid inertia effect in a SFD; when is it important?

SFD with circular whirl. **Fluid inertia** generates a pressure field (P_i)

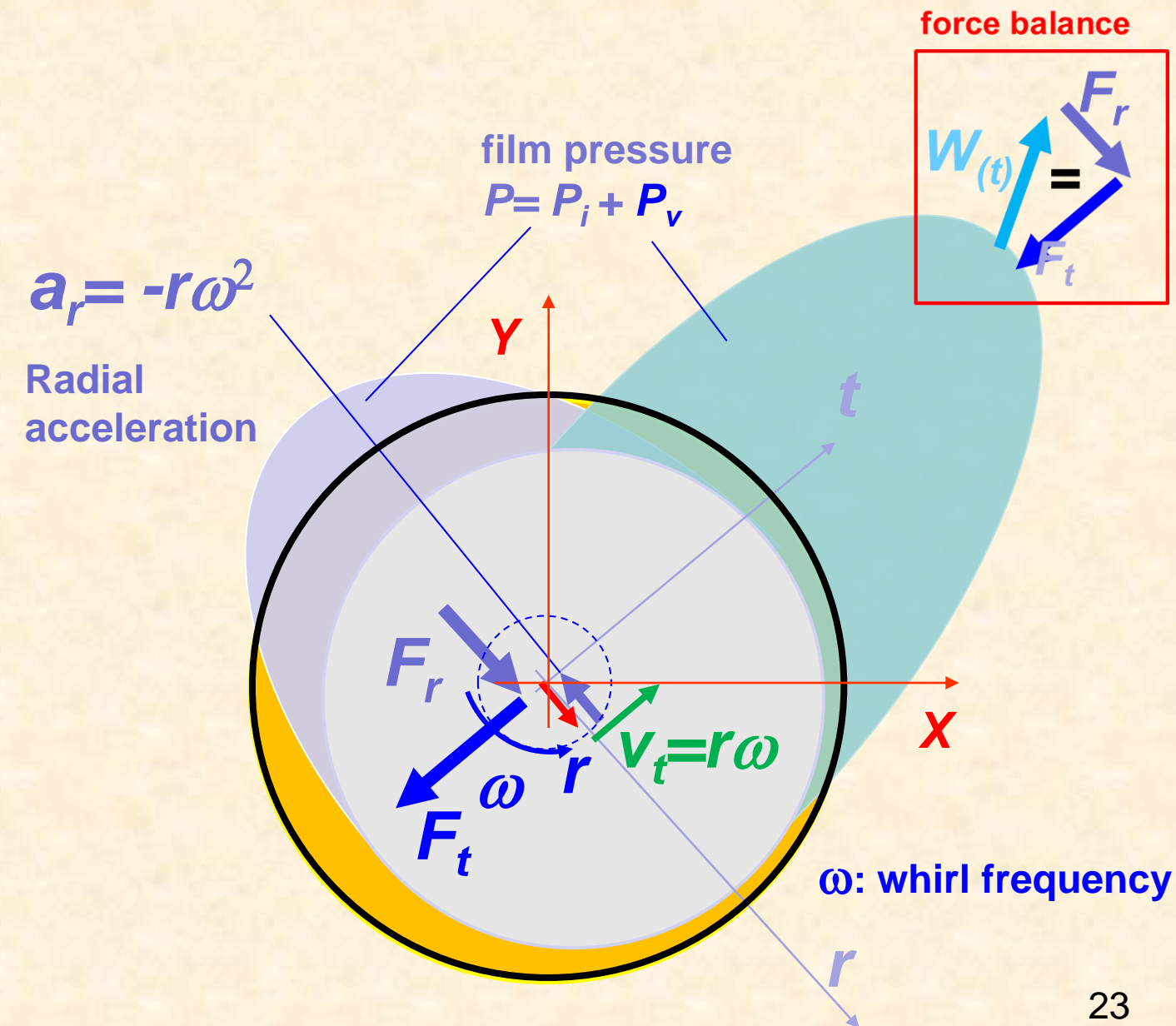
➔ If change in whirl speed is fast, $|a_r| \gg 0$, the damper reacts with a force larger than the viscous (damping) force.

$$F_r = F_{ri} \rightarrow = -M_{rr} a_r$$

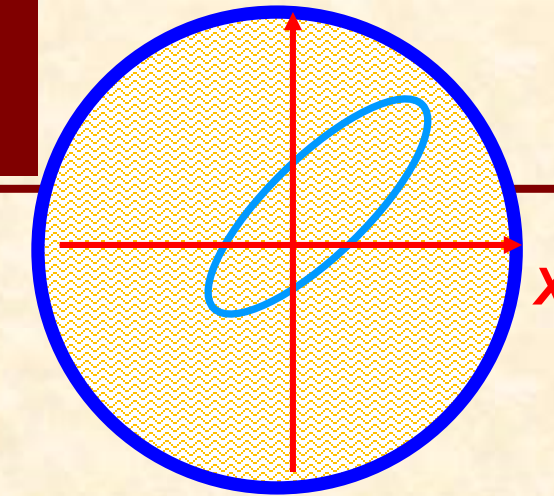
$$F_t = F_{tv} \rightarrow = -C_{tt} v_t$$

Reynolds number

$$Re_s = \frac{\rho \omega c^2}{\mu} > 12$$



SFD force coefficients



Rotordynamics models the SFD reaction force $F=\{F_x, F_y\}^T$ with constant force coefficients \rightarrow damping **C** & inertia **M**.

$$-\mathbf{F}_{(t)} = \mathbf{C}\dot{\mathbf{z}} + \mathbf{M}\ddot{\mathbf{z}}$$

$\mathbf{z}=\{x, y\}^T$: rotor (journal) displacement about static position (e).

$$-\begin{Bmatrix} F_x \\ F_y \end{Bmatrix} = \begin{bmatrix} C_{xx} & C_{xy} \\ C_{yx} & C_{yy} \end{bmatrix} \begin{Bmatrix} \dot{x} \\ \dot{y} \end{Bmatrix} + \begin{bmatrix} M_{xx} & M_{xy} \\ M_{xy} & M_{yy} \end{bmatrix} \begin{Bmatrix} \ddot{x} \\ \ddot{y} \end{Bmatrix}$$

SFDs cannot generate stiffness **K**

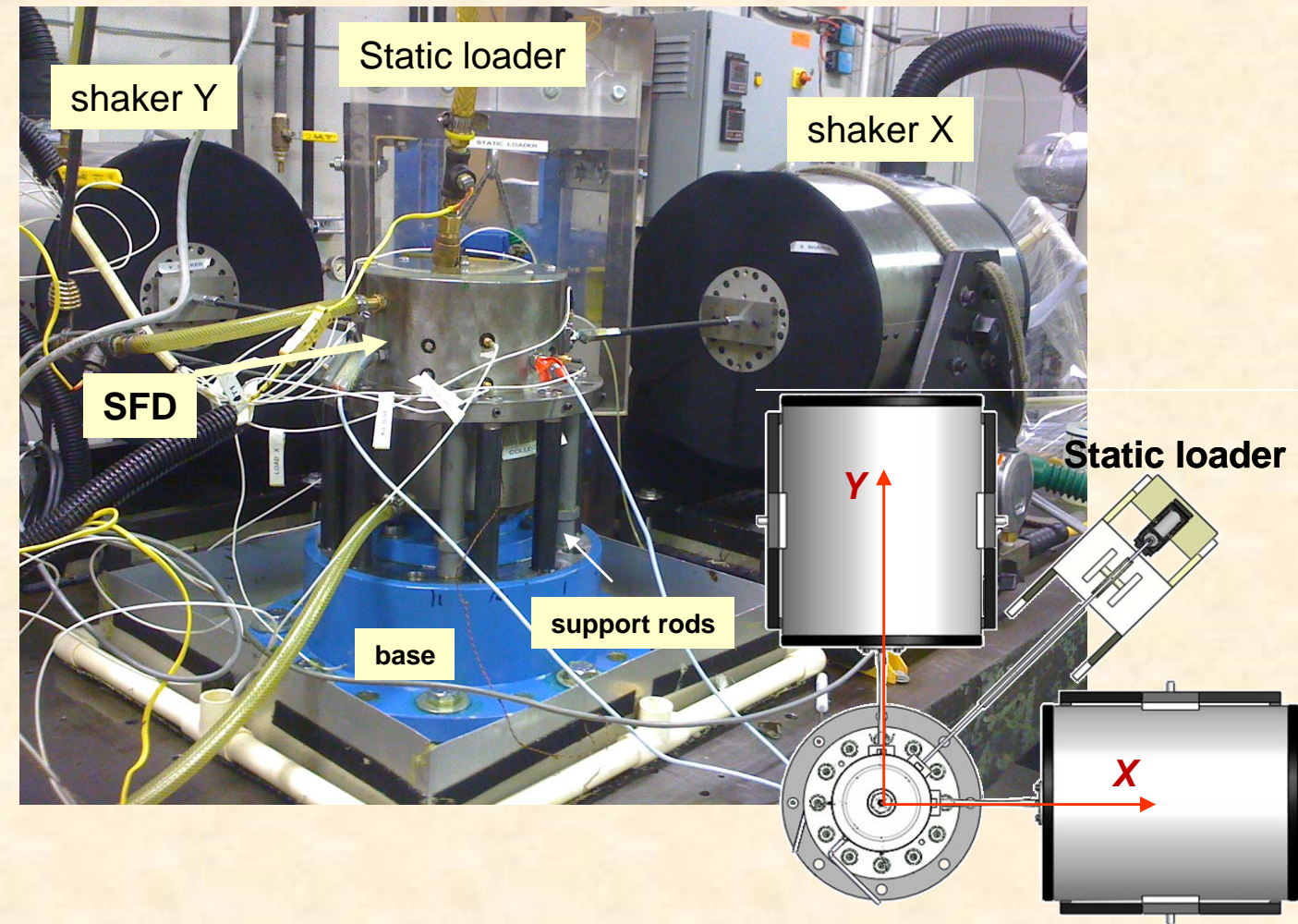
USED for prediction of synchronous rotordynamic response and rotor-bearing system stability.

SFD Test Program

Explore **novel SFD designs** & benchmark **SFD empirical data**.

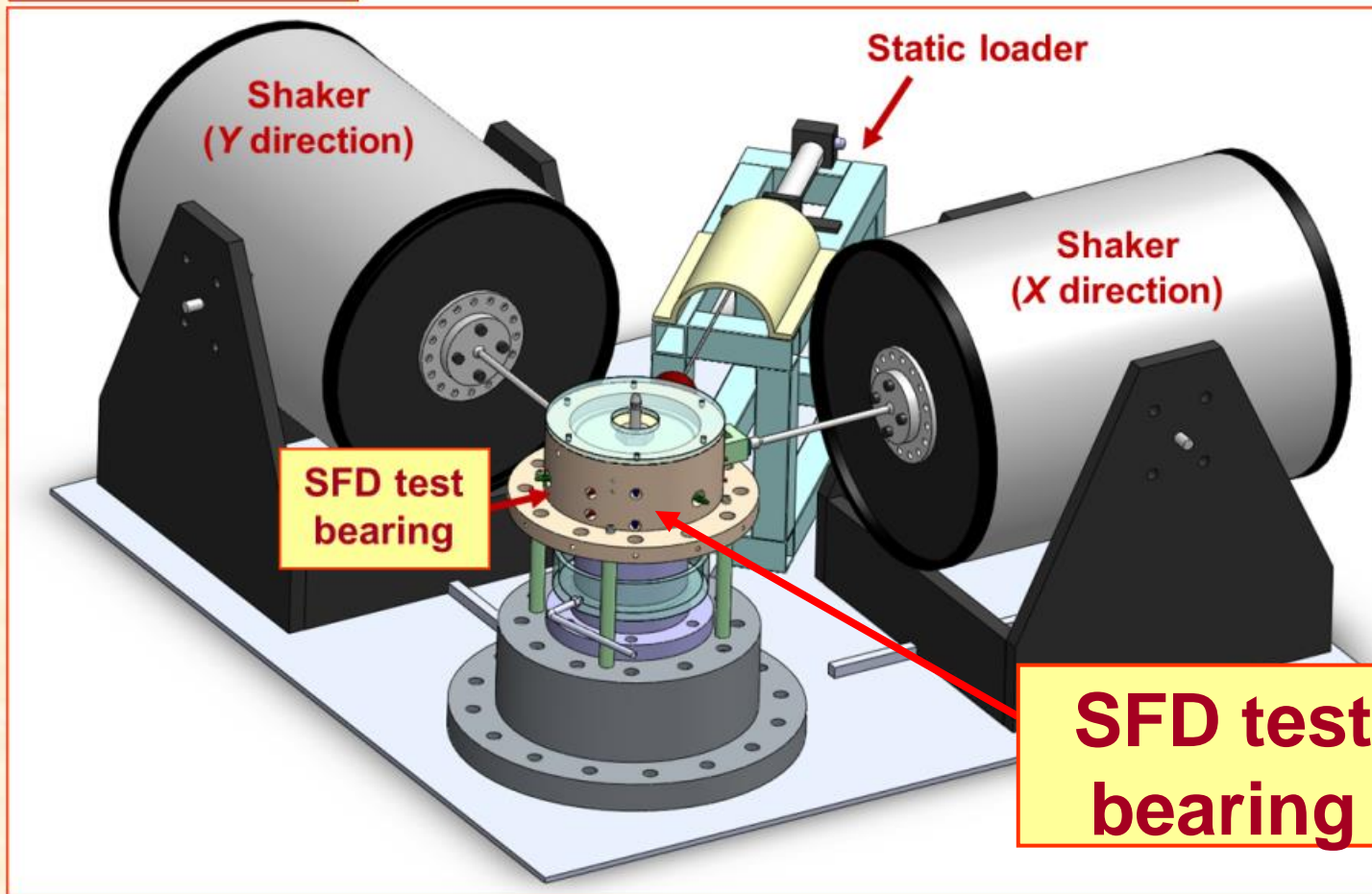
Develop & validate **SFD forced performance model** & improve its prediction.

Funded by Pratt & Whitney Engines (2008-2018)

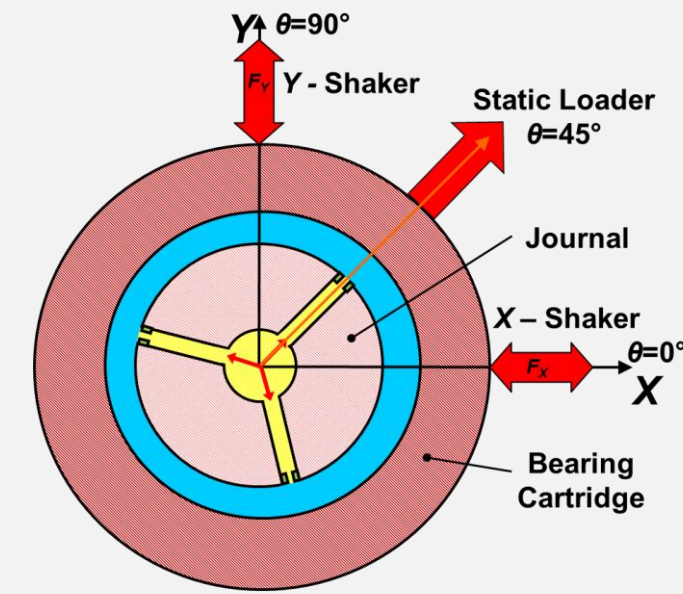
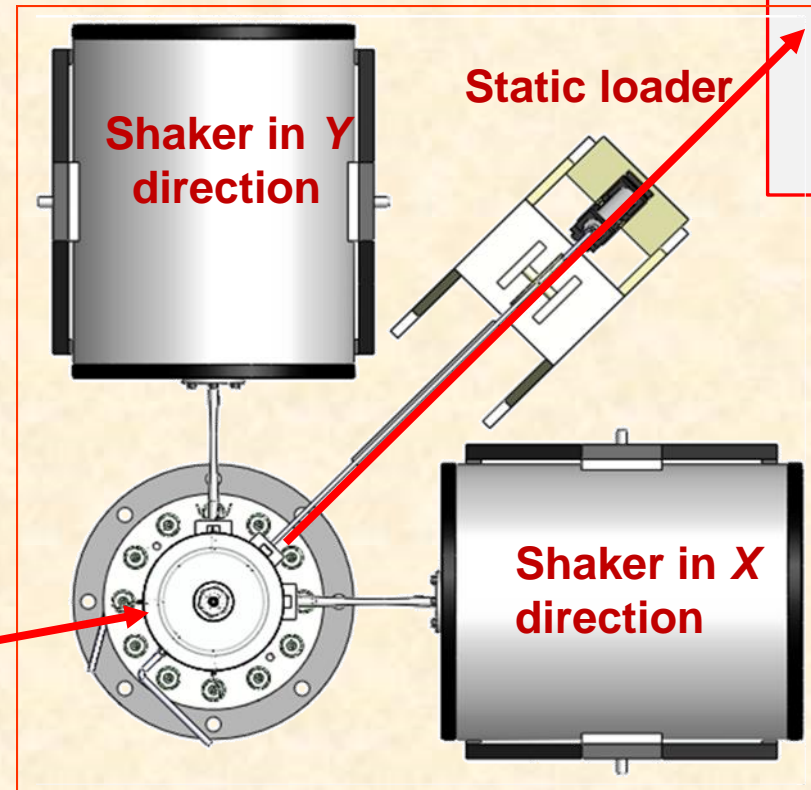


SFD test rig (2008-2018)

Isometric view



Top view

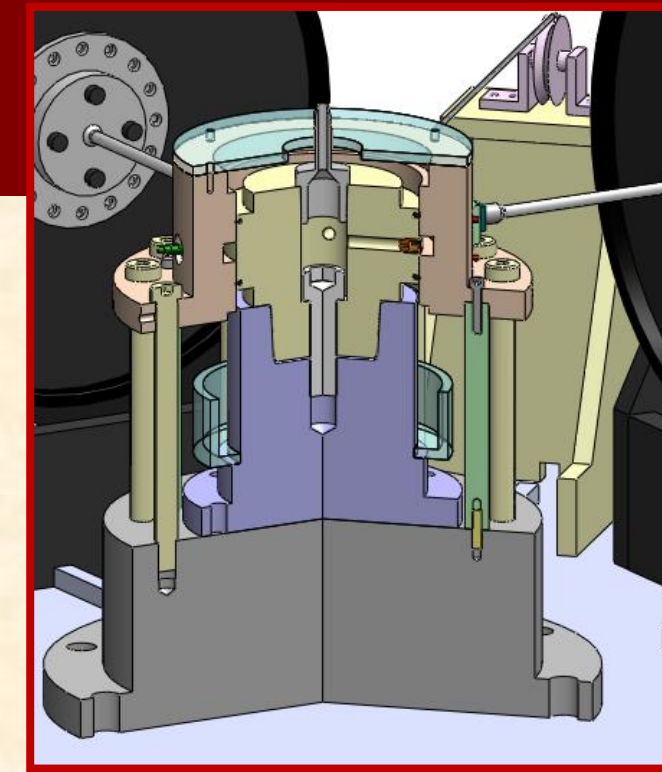
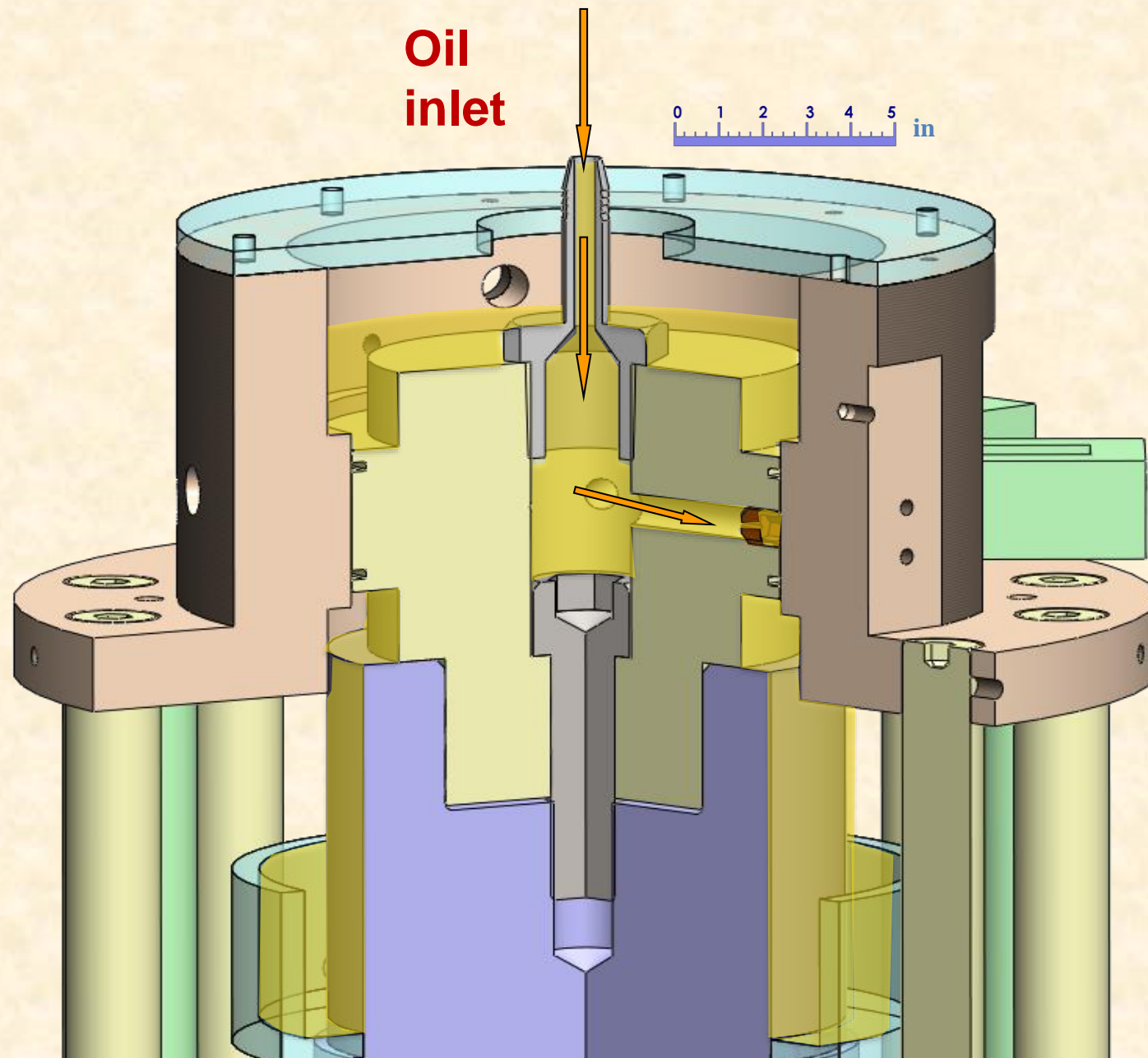


2 electro magnetic-shakers (2 kN ~ 550 lb_f)

Static loader (4 kN ~ 1 klbf) at 45°

Customizable SFD test bearing

Lubricant flow path



Oil physical properties similar to those in jet engines operating at high temperature.

ISO VG 2 oil

Oil inlet temperature, $T_s = 23 \text{ }^\circ\text{C}$

Density, $\rho = 800 \text{ kg/m}^3$

Viscosity μ at $T_s = 2.6 \text{ cPoise}$

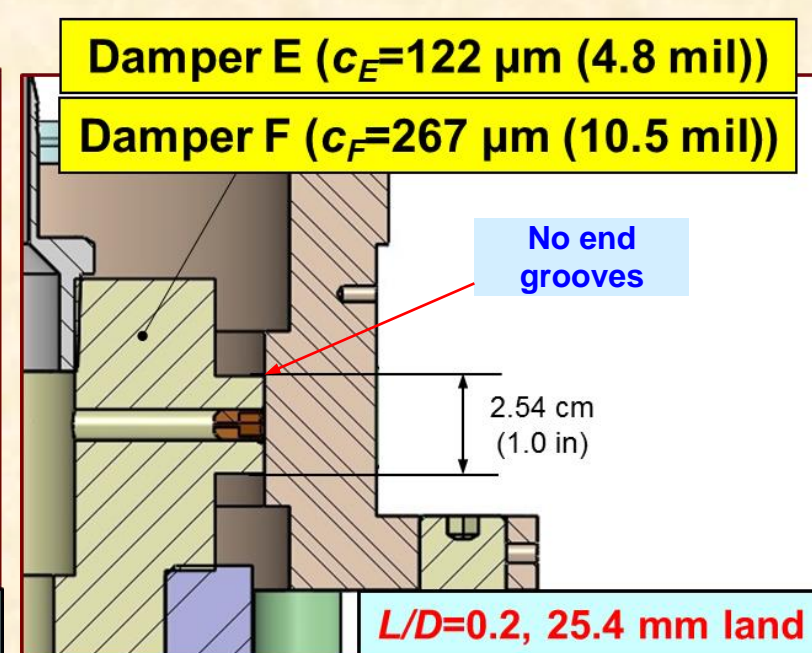
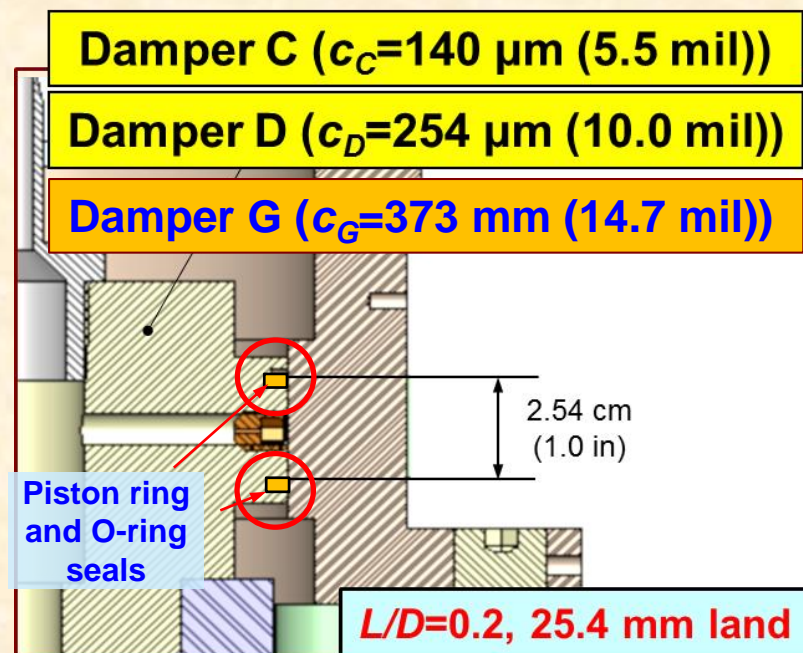
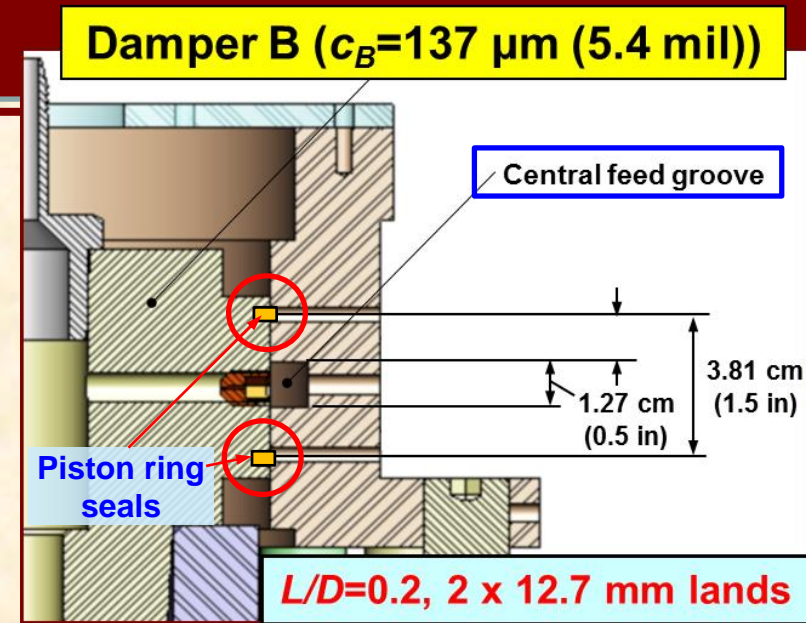
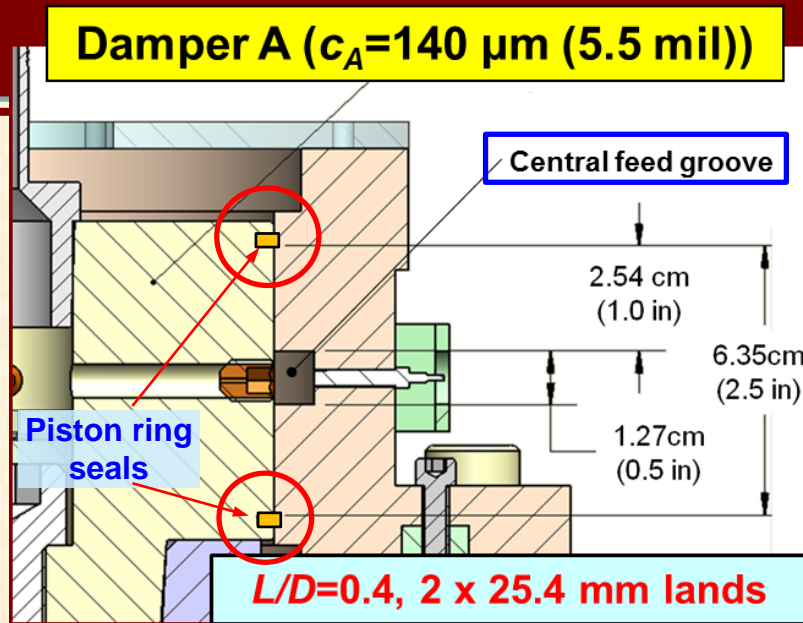
Flow rate, $Q_{in} = \text{varies}$

To explore

novel SFD designs
& benchmark SFD
empirical data & develop
& validate SFD forced
model.

Optimize SFD influence
on rotor dynamics.

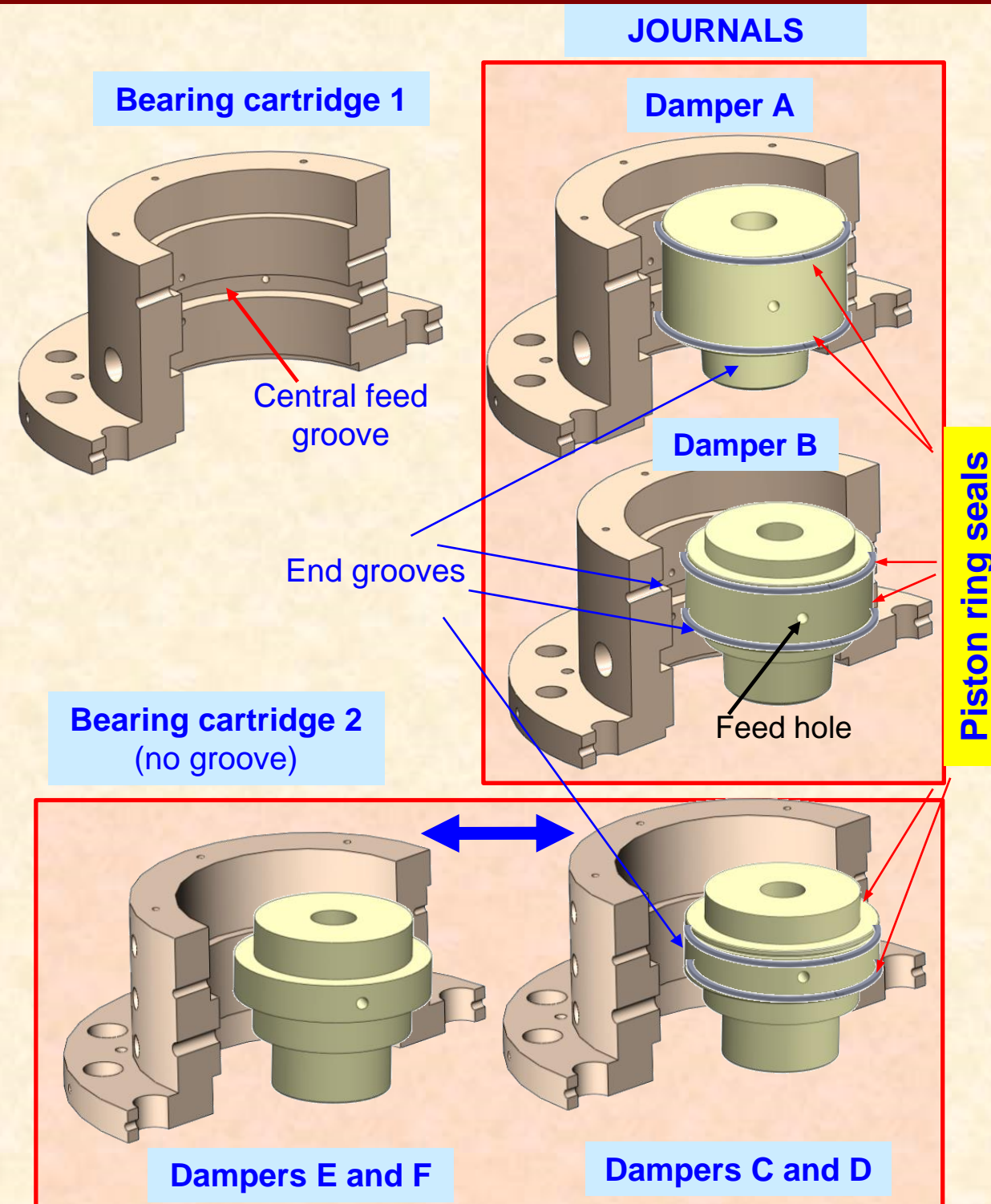
→ 25+ papers, computational
tool validated by test data, and
countless tech reports to
sponsor.



SFD test configurations

Ultra short length desired

Configuration	Land Length, Central Groove	End Grooves	Radial Clearance
A*	2 X L=25.4 mm film lands - Central groove	Yes	$c_{A-1}=141 \mu\text{m}$ $c_{A-2}=251 \mu\text{m}$
B*	2 X L=12.7 mm film lands - Central groove		$c_B=138 \mu\text{m}$
C*	L=25.4 mm film land - No feed groove	Yes	$c_C=130 \mu\text{m}$
D*			$c_D=254 \mu\text{m}$
E*	L=25.4 mm film land - No feed groove	No	$c_E=122 \mu\text{m}$
F*			$c_F=267 \mu\text{m}$

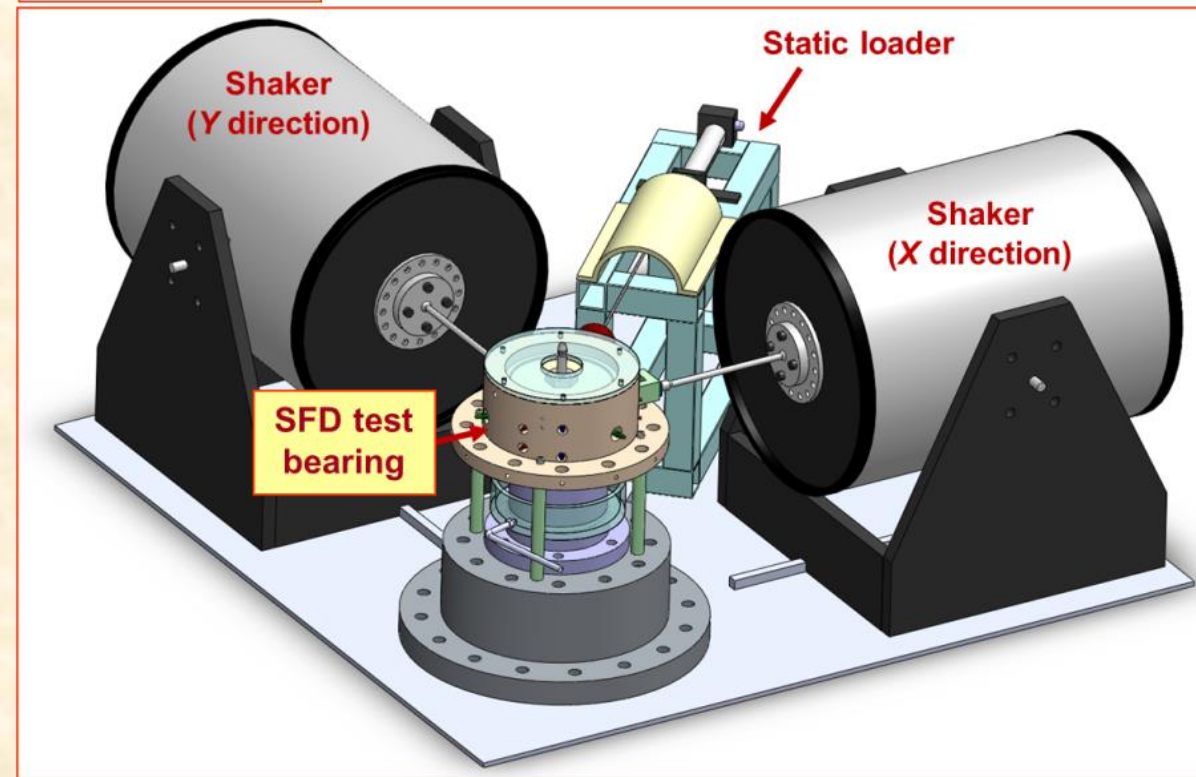


* SFD: open ends and sealed ends: piston rings or Orings

Experimental estimation of force coefficients

Designed and built by students at TAMU

Isometric view

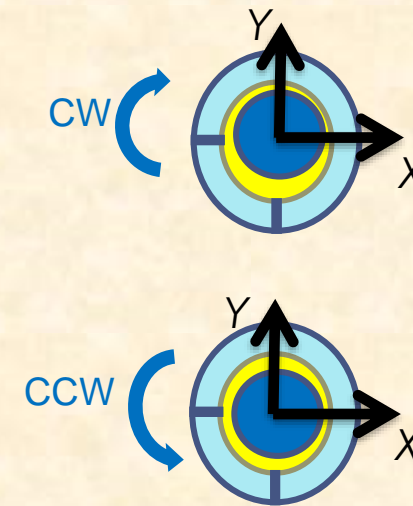
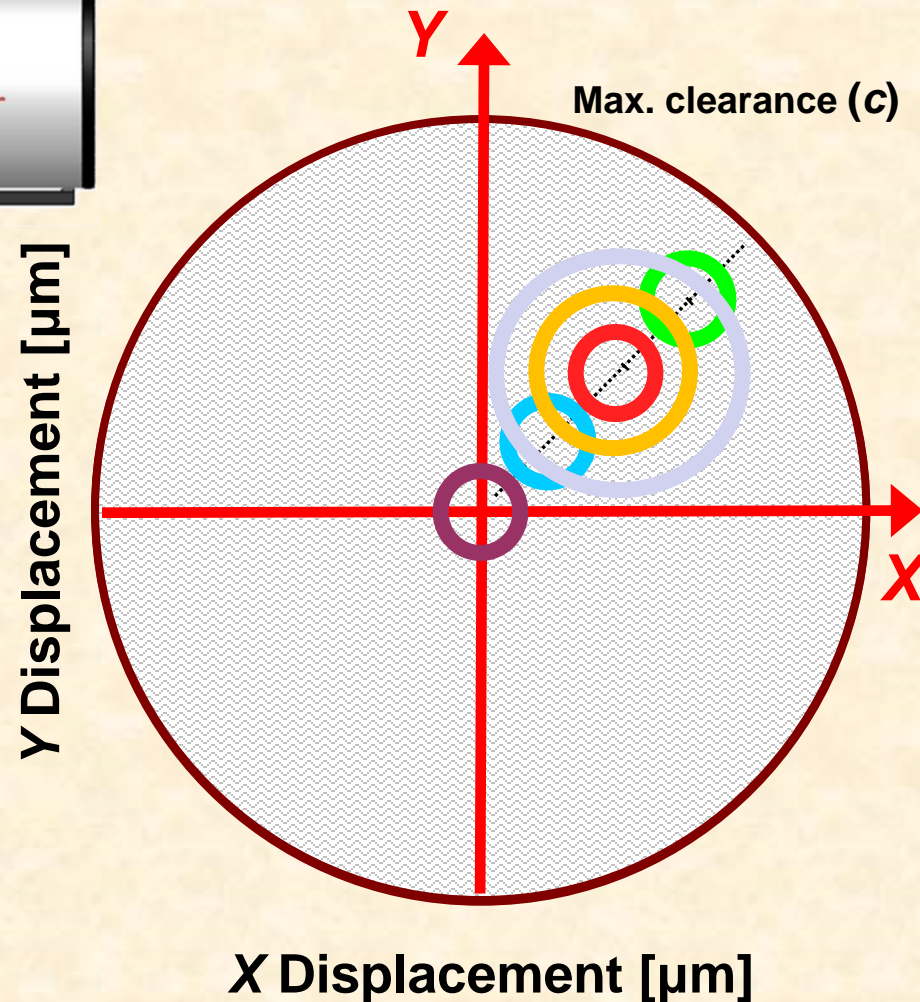
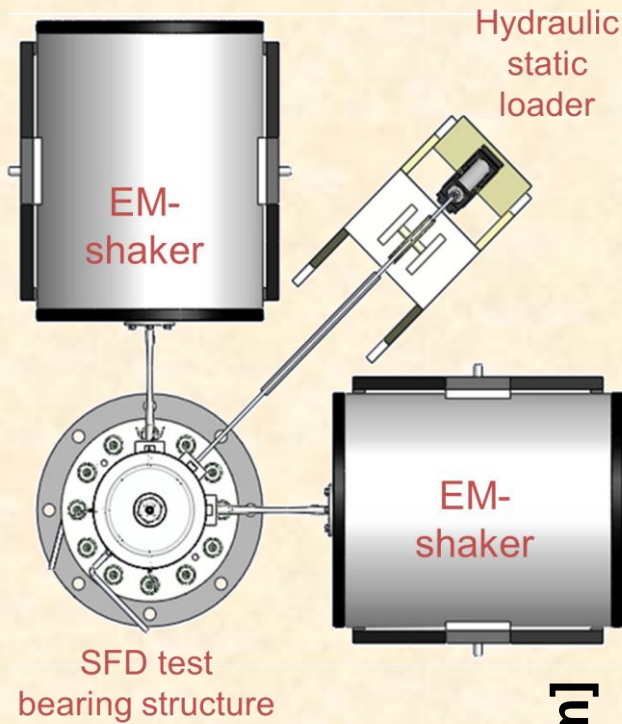


Bearing mass, M_{BC}	15 kg
Structure stiffness, K_s	10 MN/m
damping, C_s	0.9 kN-s/m

$$f_n = 131 \text{ Hz}, \zeta = 0.03$$

Imposed whirl motions

Evaluate SFD force coefficients from
whirl orbits: amplitude (r) grows.
with offset or **static eccentricity**
(e_s) – 45° away.



Measurement procedure and identification

Step 1 : Apply loads and measure BC motions

Shakers apply forces

$$\mathbf{F}^1 = \text{Re} \left(\begin{bmatrix} F_X^1 \\ iF_Y^1 \end{bmatrix} e^{i\omega t} \right)$$

$$\mathbf{F}^2 = \text{Re} \left(\begin{bmatrix} F_X^2 \\ -iF_Y^2 \end{bmatrix} e^{i\omega t} \right)$$

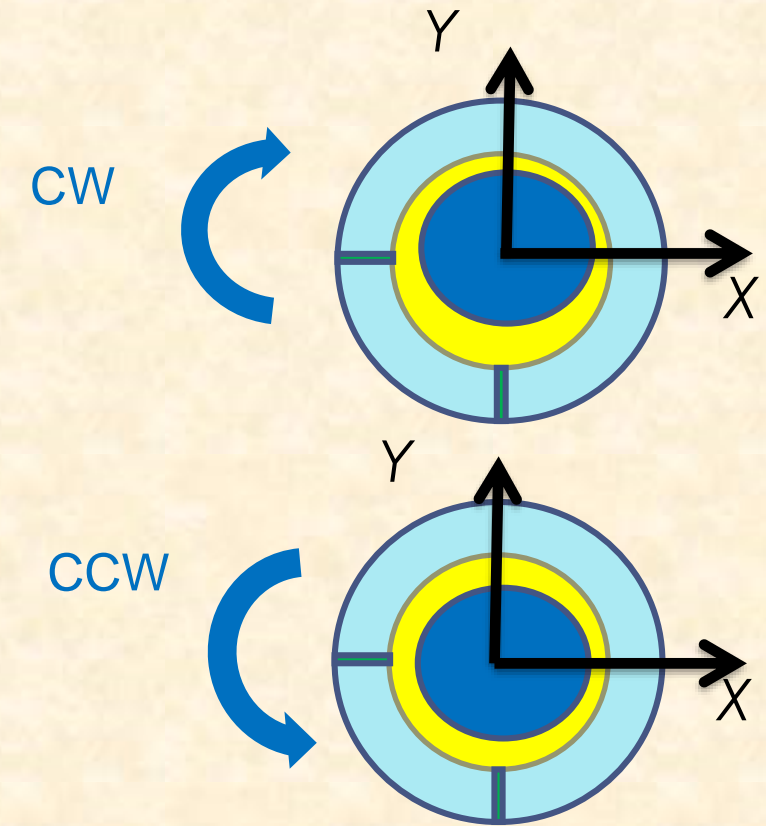
Record BC displacements and accelerations

$$\mathbf{z}^1 = \begin{bmatrix} x_{(t)}^1 \\ y_{(t)}^1 \end{bmatrix} = \begin{bmatrix} X^1 \\ Y^1 \end{bmatrix} e^{i\omega t}$$

$$\mathbf{z}^2 = \begin{bmatrix} x_{(t)}^2 \\ y_{(t)}^2 \end{bmatrix} = \begin{bmatrix} X^2 \\ Y^2 \end{bmatrix} e^{i\omega t}$$

\mathbf{a}^1

\mathbf{a}^2



Load $\mathbf{F}_{(t)}$, displacement $\mathbf{z}_{(t)}$ and acceleration $\mathbf{a}_{(t)}$ recorded at each frequency

EOM: Frequency Domain

$$[\mathbf{K}_L + i\omega\mathbf{C}_L - \omega^2\mathbf{M}_L]\bar{\mathbf{z}} = \bar{\mathbf{F}} - M_{BC}\bar{\mathbf{a}} \rightarrow \mathbf{H}_L \mathbf{z}$$

Unknown Parameters:

$$\mathbf{K}_L, \mathbf{C}_L, \mathbf{M}_L$$

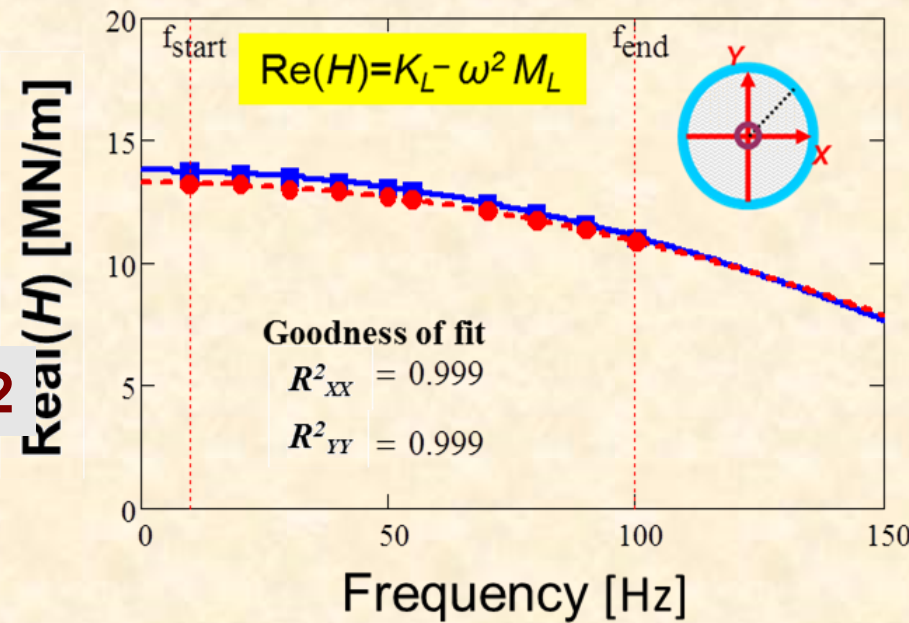
Identification of parameters

Step 2 : Transform to frequency domain and curve fit H_L 's

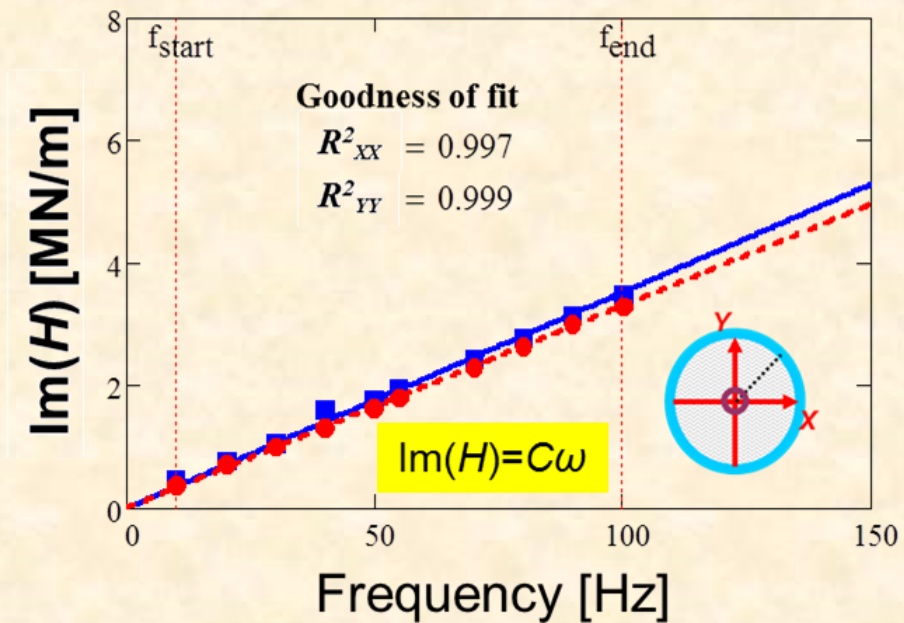
Complex dynamic stiffness

$r/c_A = 0.2$

$$\text{Re}\left([\bar{\mathbf{F}} - M_{BC}\bar{\mathbf{a}}]\bar{\mathbf{z}}^{-1}\right) \rightarrow \mathbf{K}_L - \omega^2 \mathbf{M}_L$$



$$\text{Im}\left([\bar{\mathbf{F}} - M_{BC}\bar{\mathbf{a}}]\bar{\mathbf{z}}^{-1}\right) \rightarrow \mathbf{C}_L \omega$$



■ Test data ● Test data
 — Model - - - Model

TYPICALLY physical model $\text{Re}(H_{xx}) \rightarrow K - \omega^2 M$ and $\text{Im}(H_{xx}) \rightarrow C\omega$ agree with experimental data. Damping C is constant over frequency range

SFD coefficients

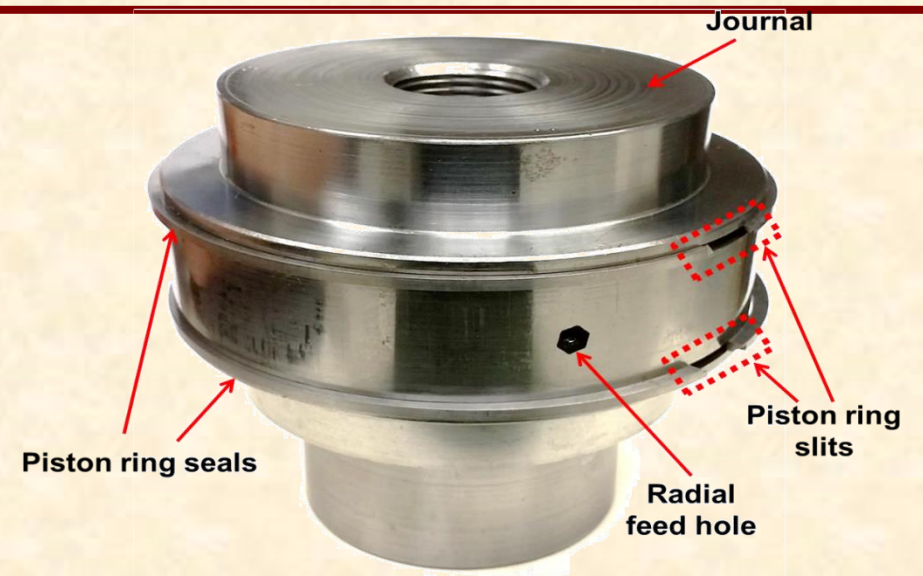
$$(\mathbf{K}, \mathbf{C}, \mathbf{M})_{\text{SFD}} = (\mathbf{K}, \mathbf{C}, \mathbf{M})_L - (\mathbf{K}, \mathbf{C}, \mathbf{M})_S$$

↑
SFD

↑
Test system
(lubricated)

↑
Dry structure

Comprehensive flow model for prediction of SFD forced performance



**Bubbly mixture (2001, 2019) and orbit-model (2016)
..... with a major departure!**

San Andrés, L., and Koo, B., **2019**, "Model and Experimental Verification of the Dynamic Forced Performance of a Tightly Sealed Squeeze Film Damper Supplied with a Bubbly Mixture," ASME 2019-90330

San Andres, L., and Jeung, S-H, **2016**, "Orbit-Model Force Coefficients for Fluid Film Bearings: A Step Beyond Linearization," ASME J. Eng. Gas Turb. Pwr., 138(2)

Diaz and San Andres, **2001**, "A Model for Squeeze Film Dampers Operating with Air Entrainment and Validation with Experiments," ASME J. Tribol., 123.

Squeeze film pressure and boundary conditions

Extended Reynolds Eq.

$$\frac{\partial}{R\partial\Theta} \left(\frac{\rho h^3}{12\mu} \frac{\partial P}{R\partial\Theta} \right) + \frac{\partial}{\partial z} \left(\frac{\rho h^3}{12\mu} \frac{\partial P}{\partial z} \right) = \frac{\partial}{\partial t} (\rho h) + \frac{\rho h^2}{12\mu} \frac{\partial^2}{\partial t^2} (\rho h)$$

density

viscosity

$$\rho_m = \beta_{(P)} \rho_g + (1 - \beta_{(P)}) \rho_{oil}$$

$$\mu_m = \beta_{(P)} \mu_g + (1 - \beta_{(P)}) \mu_{oil}$$

Gas volume fraction $GVF = \beta$

$$\beta_{(P)} = \frac{1}{1 + \frac{P_{(\Theta,z,t)} - P_v}{P_s - P_v} \left(\frac{1 - \beta_s}{\beta_s} \right)}$$

Temporal fluid inertia

See also ASME GT2022-81990

feedhole

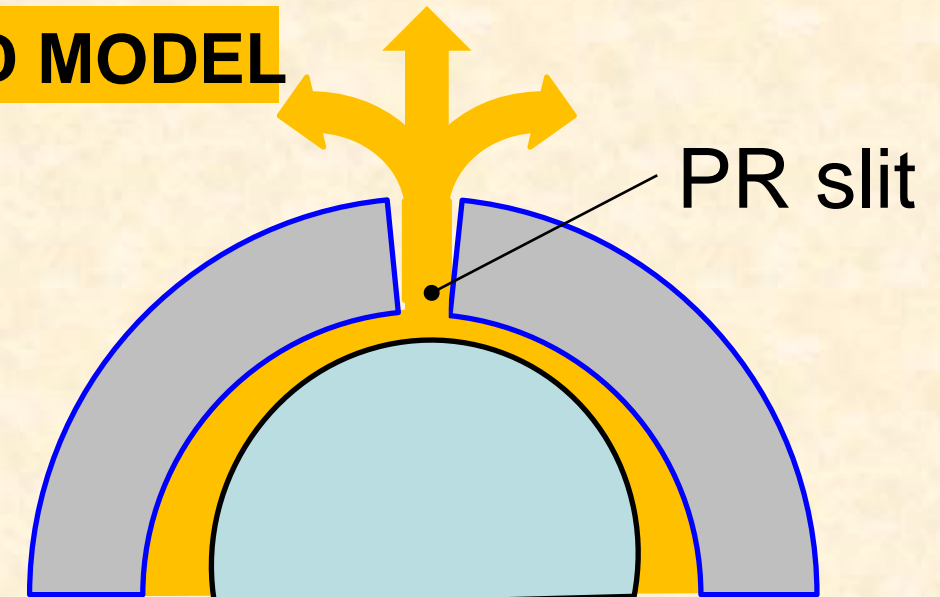
$$M_{in} = C_{in} A_{in} \sqrt{2\rho |P_s - P_{(\theta_{in},0,t)}|}$$

PR-slit

$$M_{exit} = C_{slit} A_{slit} \sqrt{2\rho |P_{(\theta_{slit}, \frac{L}{2}, t)} - P_a|}$$

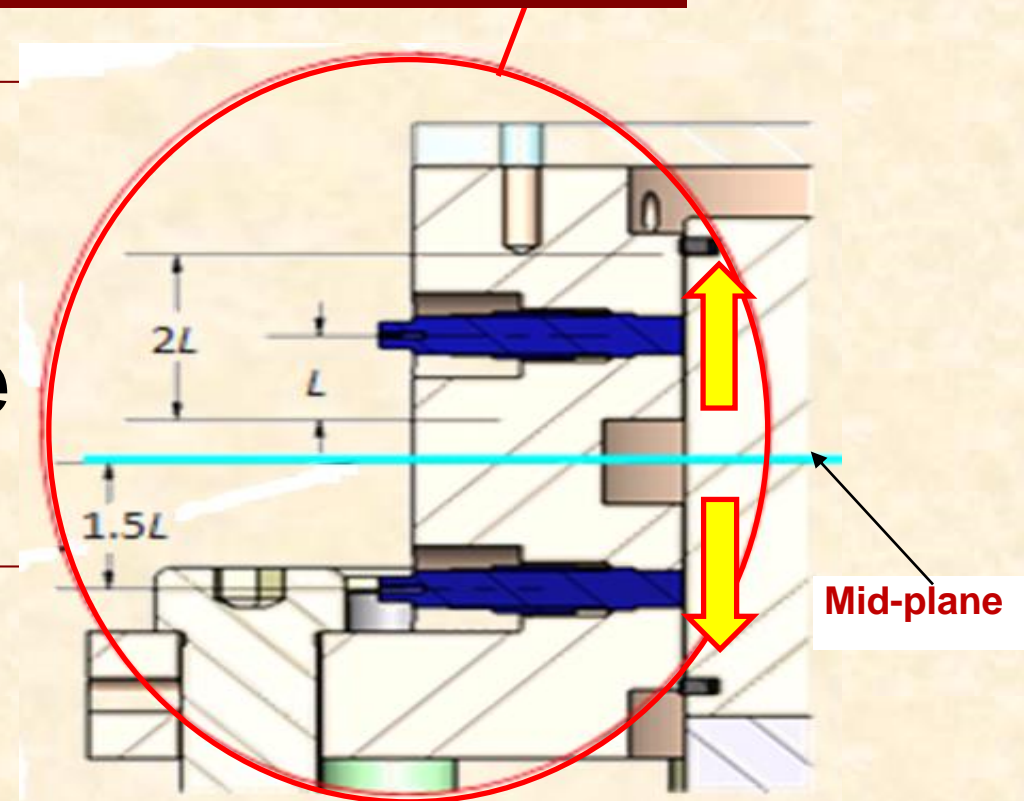
PR flow: a major departure from simple practice: $Q \sim C_{seal} \Delta P$

OLD MODEL

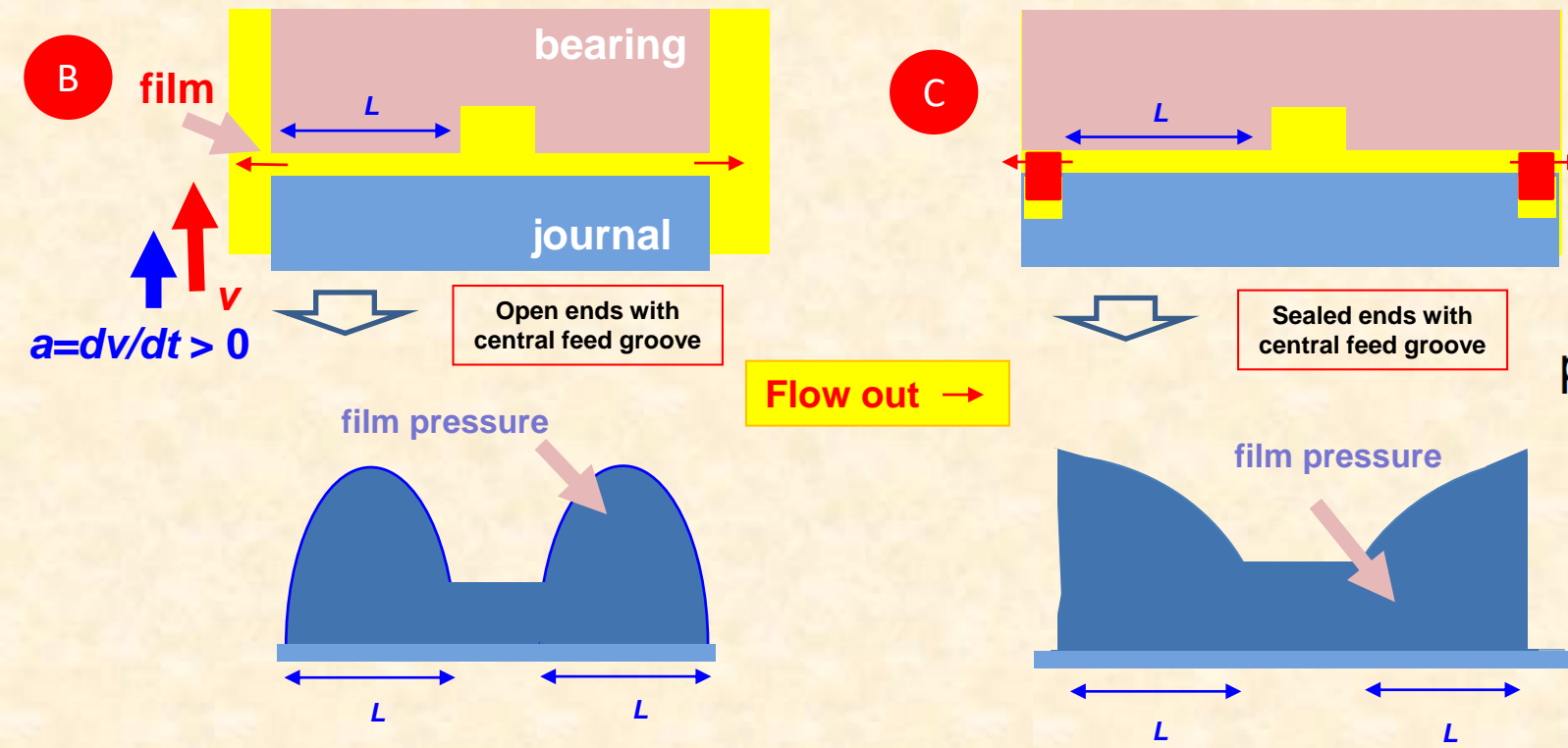


Do dynamic pressures appear in a deep feed groove?

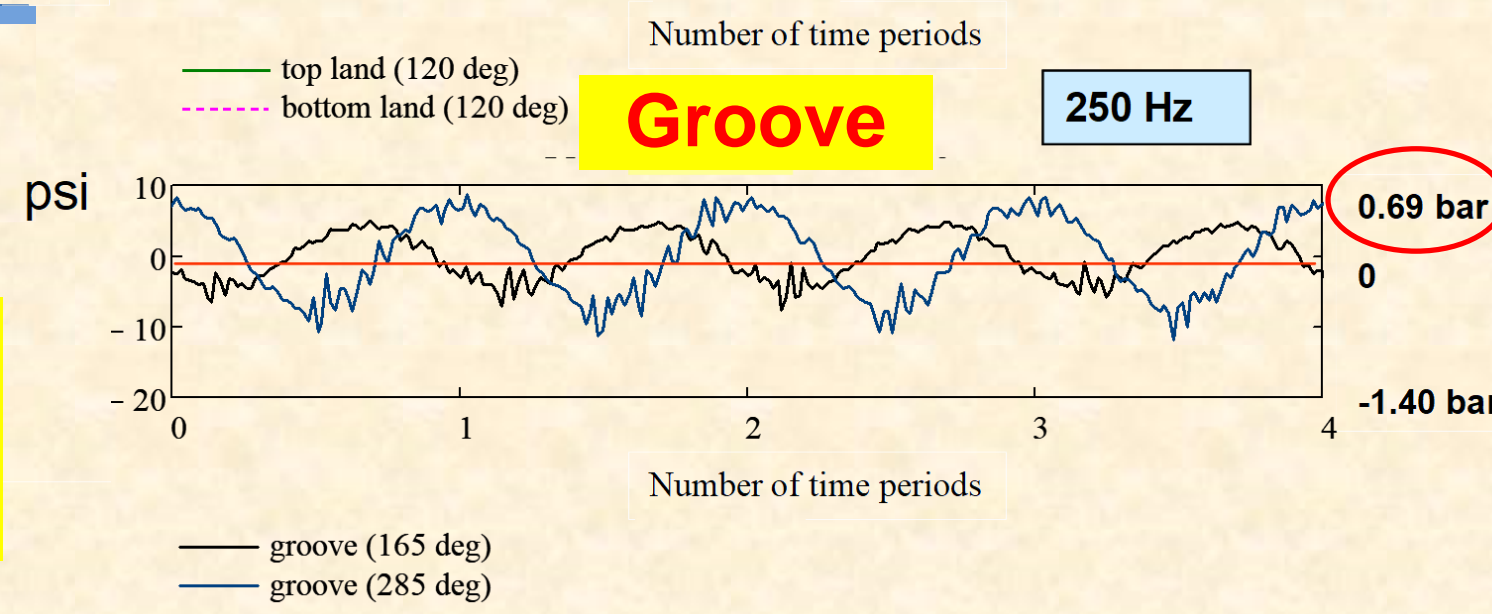
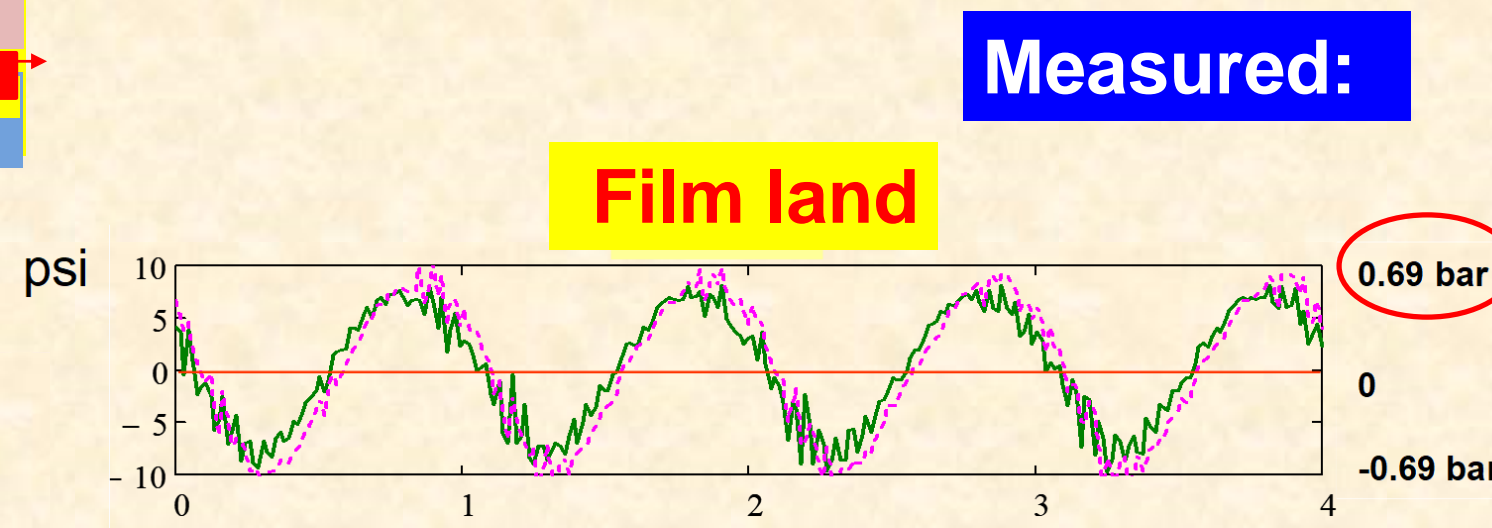
Conventional knowledge asserts that deep grooves keep a constant pressure that pushes lubricant into the adjacent film lands.



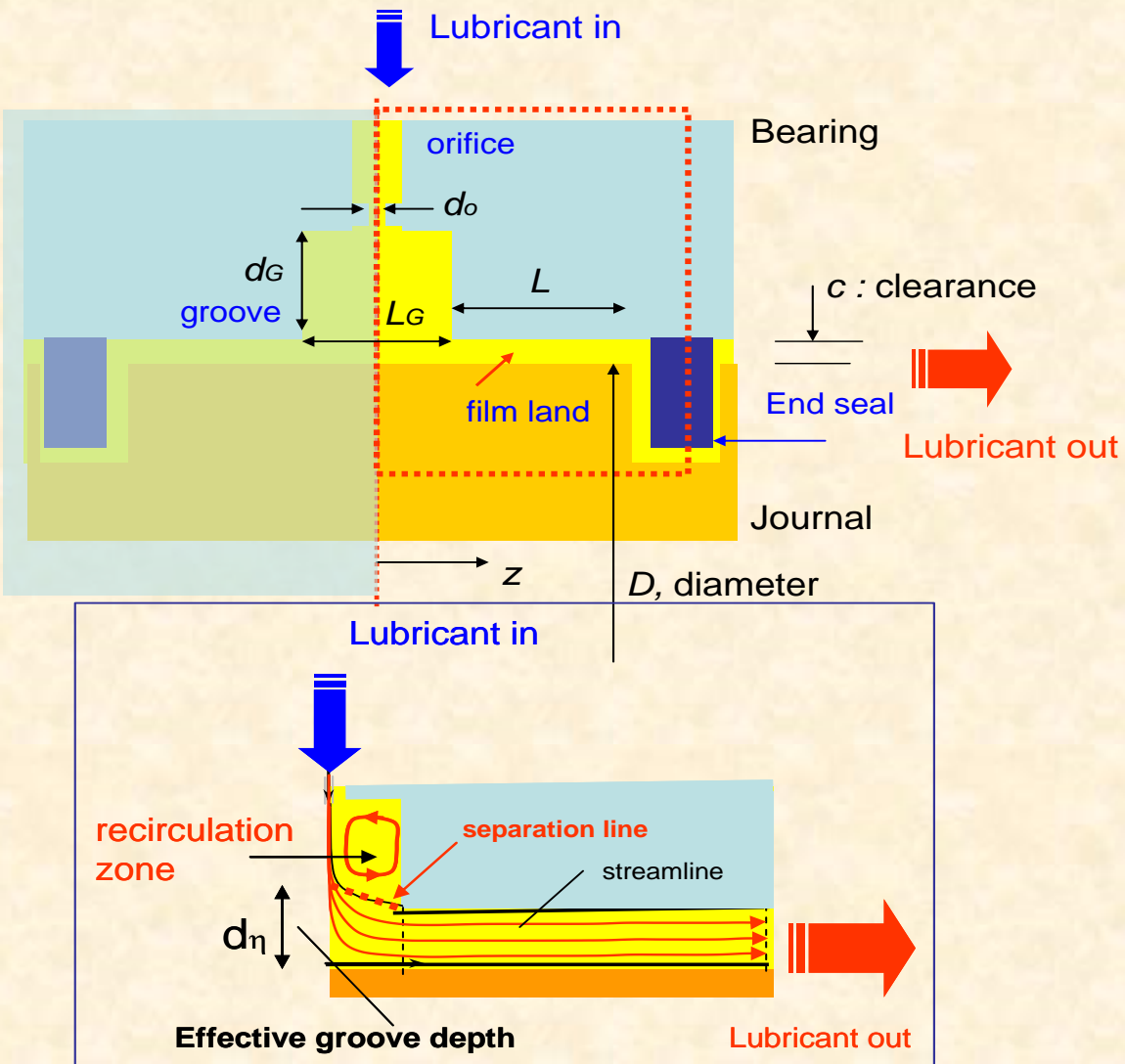
Do deep grooves isolate film lands?



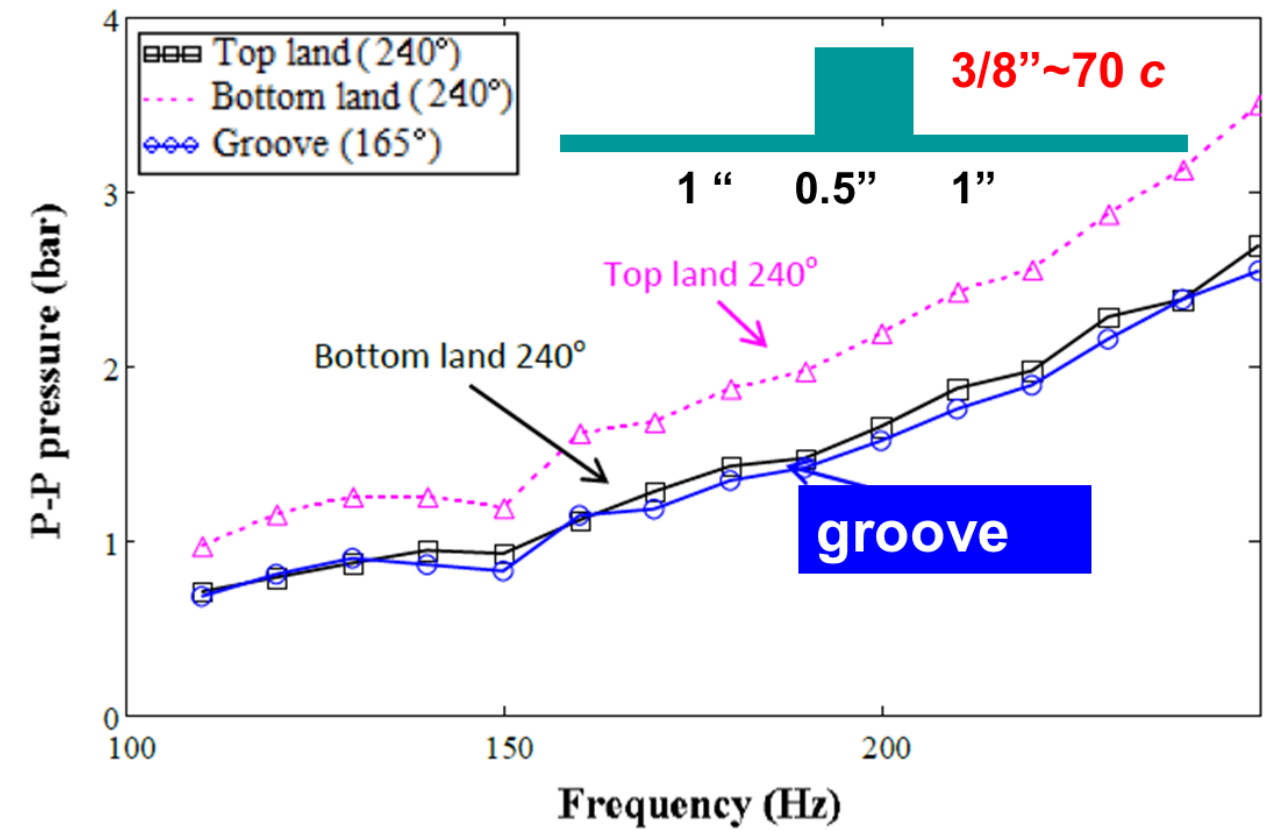
No! grooves lead to a significant squeeze film (pressure) action



Flow model SFD with grooves



SFD with PR seals

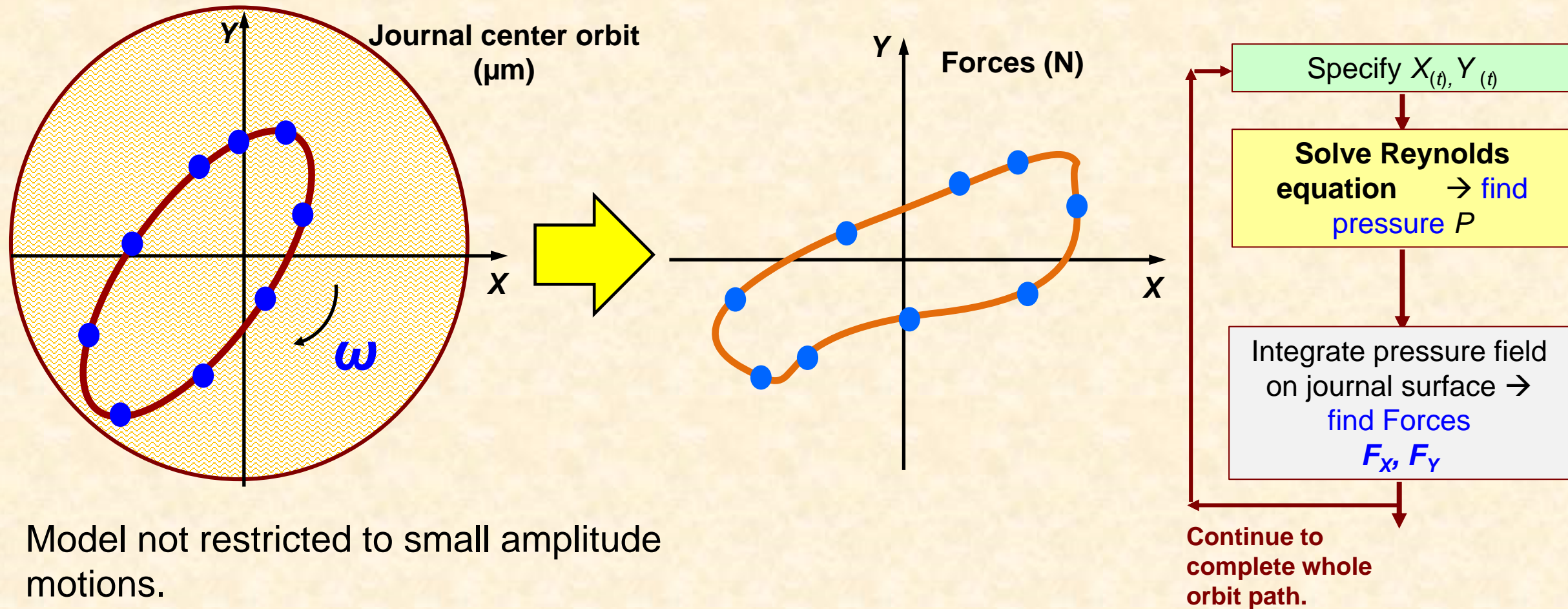


Pressures in groove are **as large as** those in the film lands.

Accounts for dynamic pressure generation in deep grooves as measurements show!

Journal center kinematics and forces

Journal motion (X vs Y) \rightarrow bearing reaction forces (F_x vs F_y).



Model not restricted to small amplitude motions.

Procedure reproduces experimental one and *estimates* (numerically) force coefficients over a wide frequency range.

See also ASME
GT2022-81990

Practical questions answered by R&D:

- How do the film length and clearance affect SFD force coefficients?
- Does the amplitude / shape of whirl motion affect the force coefficients?
- Is a damper configuration with feed holes as effective as one containing a feed groove?
- What if one feed hole plugs, is a damper still effective?
- What if the damper operates largely off-centered; does its performance become nonlinear?

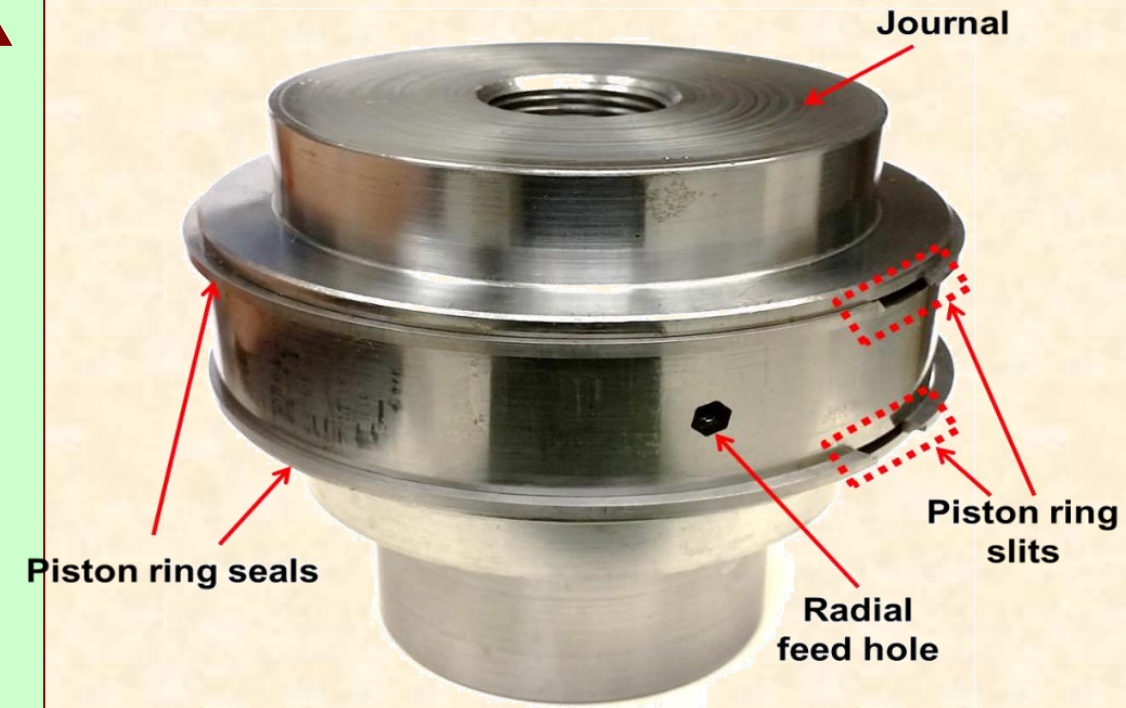
today

- Do end seals work? how much more damping do they enable?

Learn
more

San Andrés, L., Jeung, S.-H, Den, S., and Savela, G., 2016
“**Squeeze Film Dampers: A Further Experimental Appraisal of their Dynamic Performance,**” *Proceedings of the 45th Turbomachinery Symposium, Houston, TX*

A SQUEEZE FILM DAMPER SEALED WITH PISTON RINGS OR WITH ORINGS: WHICH ONE TO SELECT?



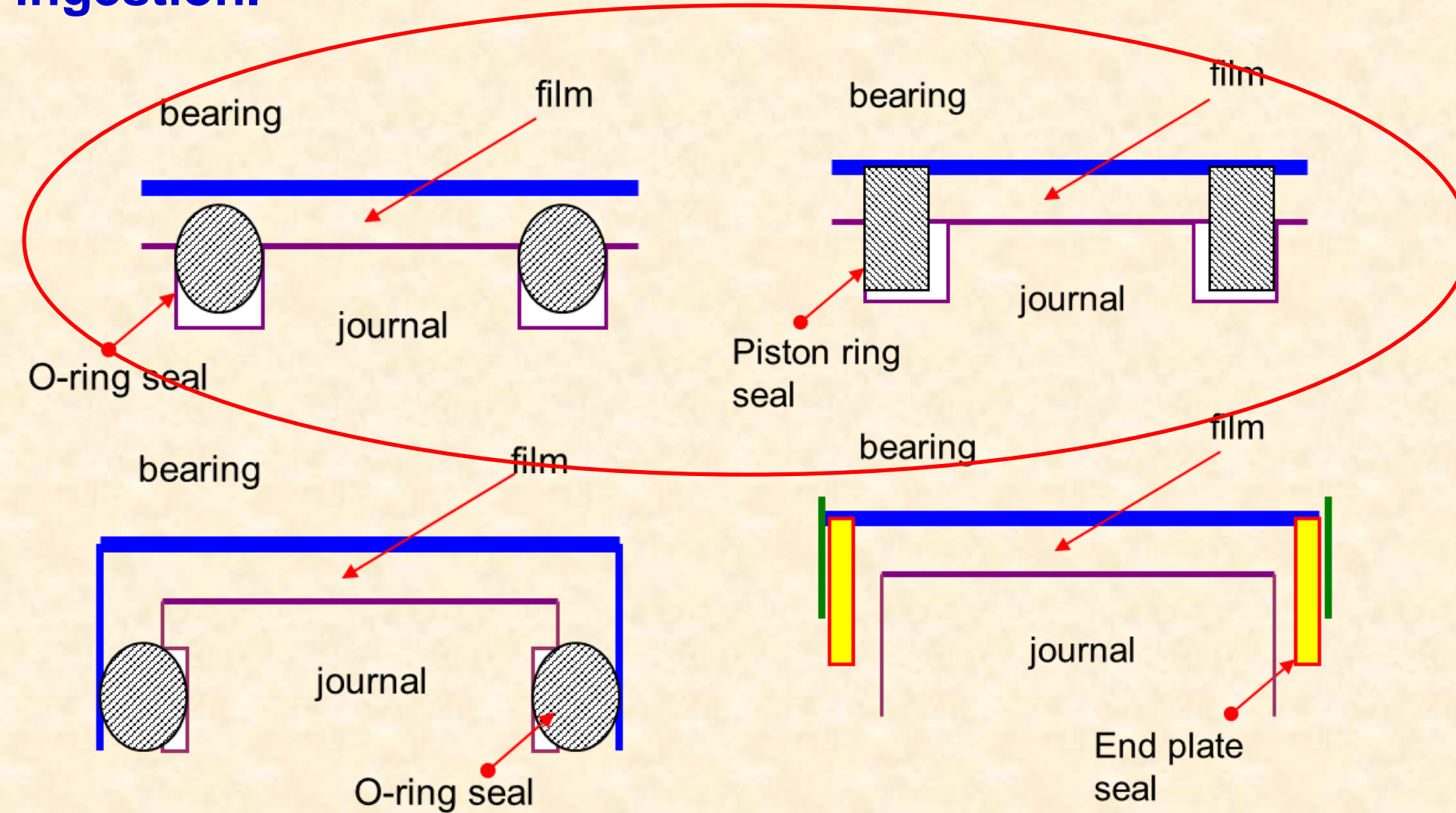
San Andrés, L., Jeung, S-H, Koo, B., **2018**, ASME GT2018-76224

San Andrés, Koo, B., **2019**, ASME GT 2019-90330

End seals for SFDs

Compressors use O-rings, and commercial jet engines implement piston rings.

Reduce thru flow and increase damping but **cannot always prevent air ingestion.**

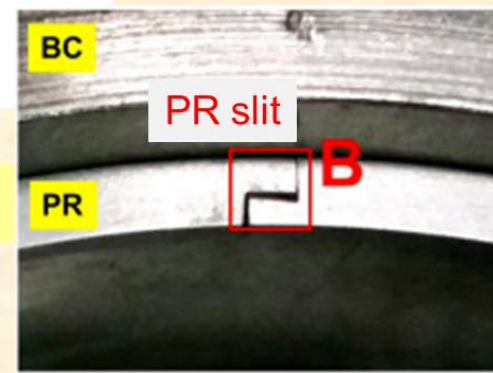
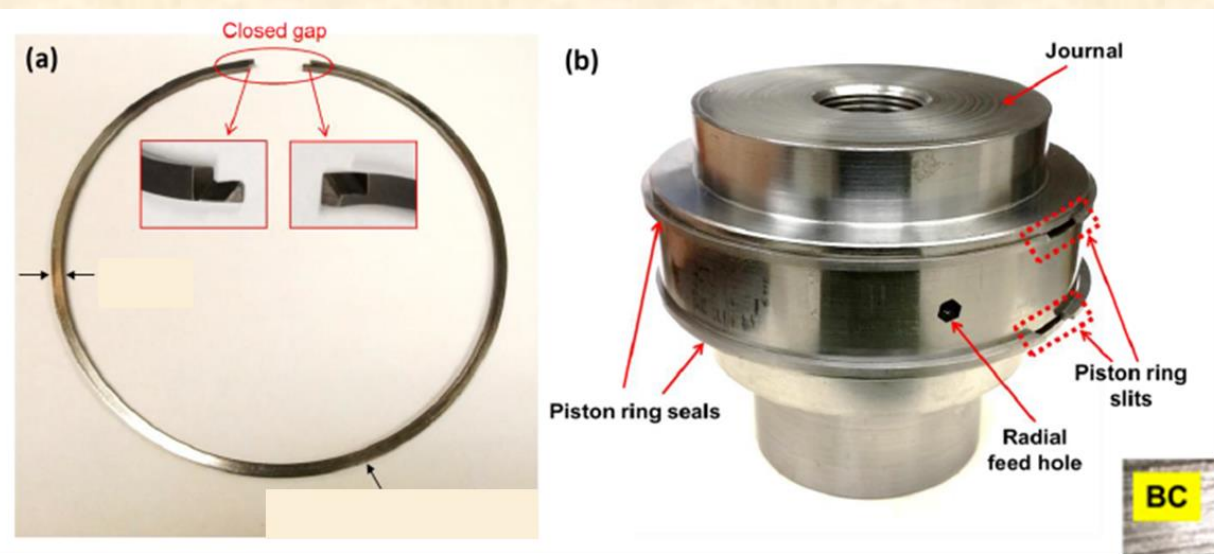
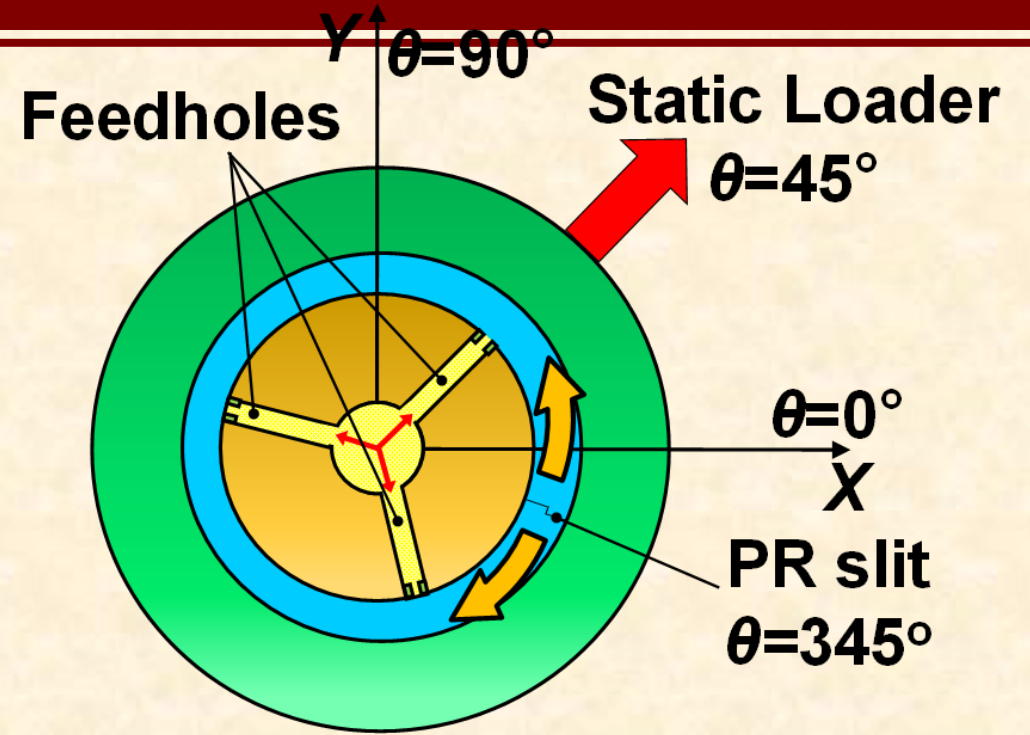
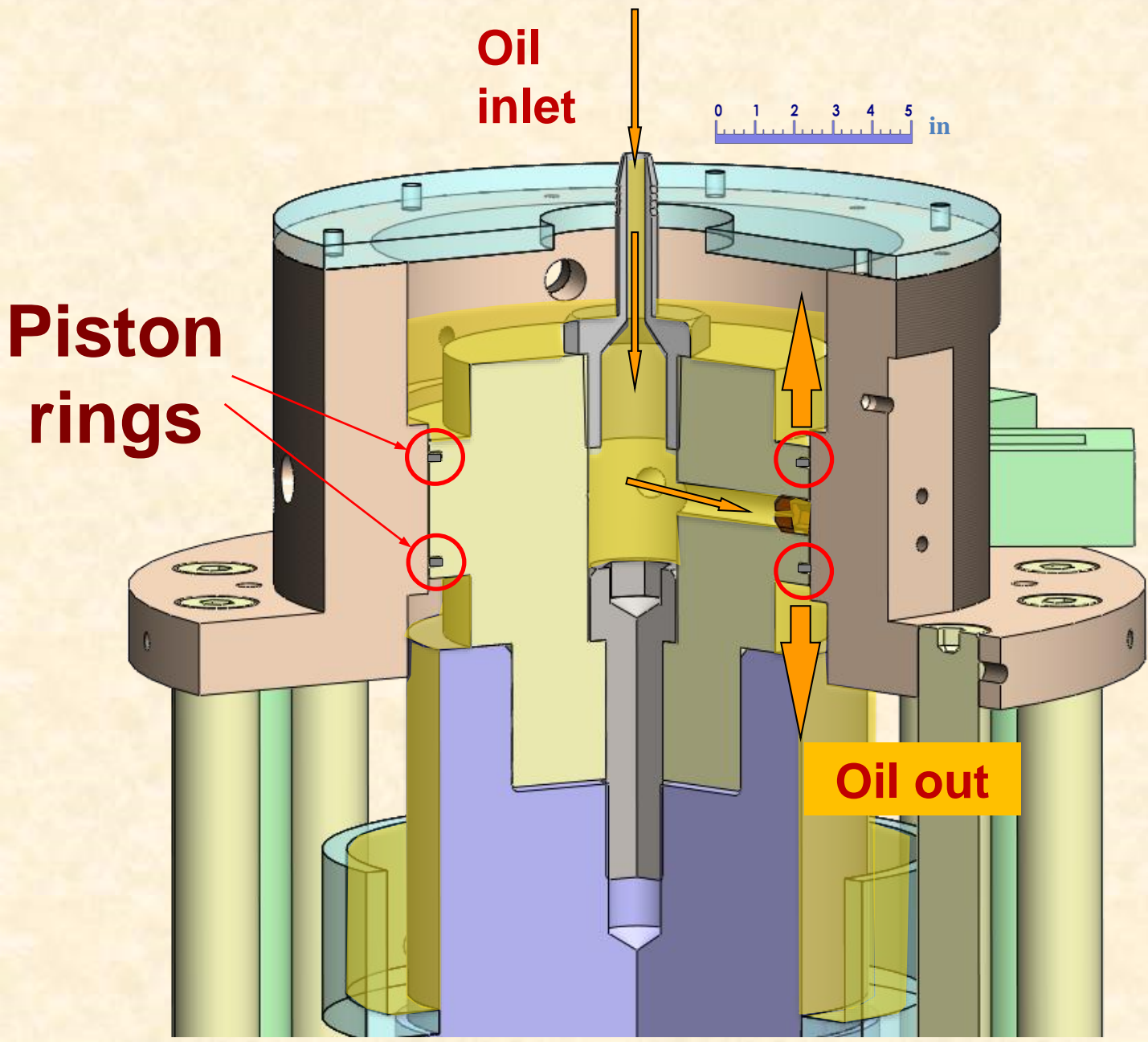


O-ring issues:
Special groove machining,
Material compatibility,
Add viscoelastic effect.

Piston ring issues:
Cocking and locking
Slits – leak too much

Design is highly empirical, except for end plate seals.

PR-SFD lubricant flow path

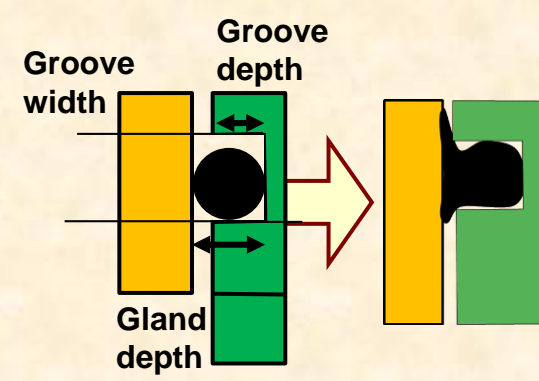
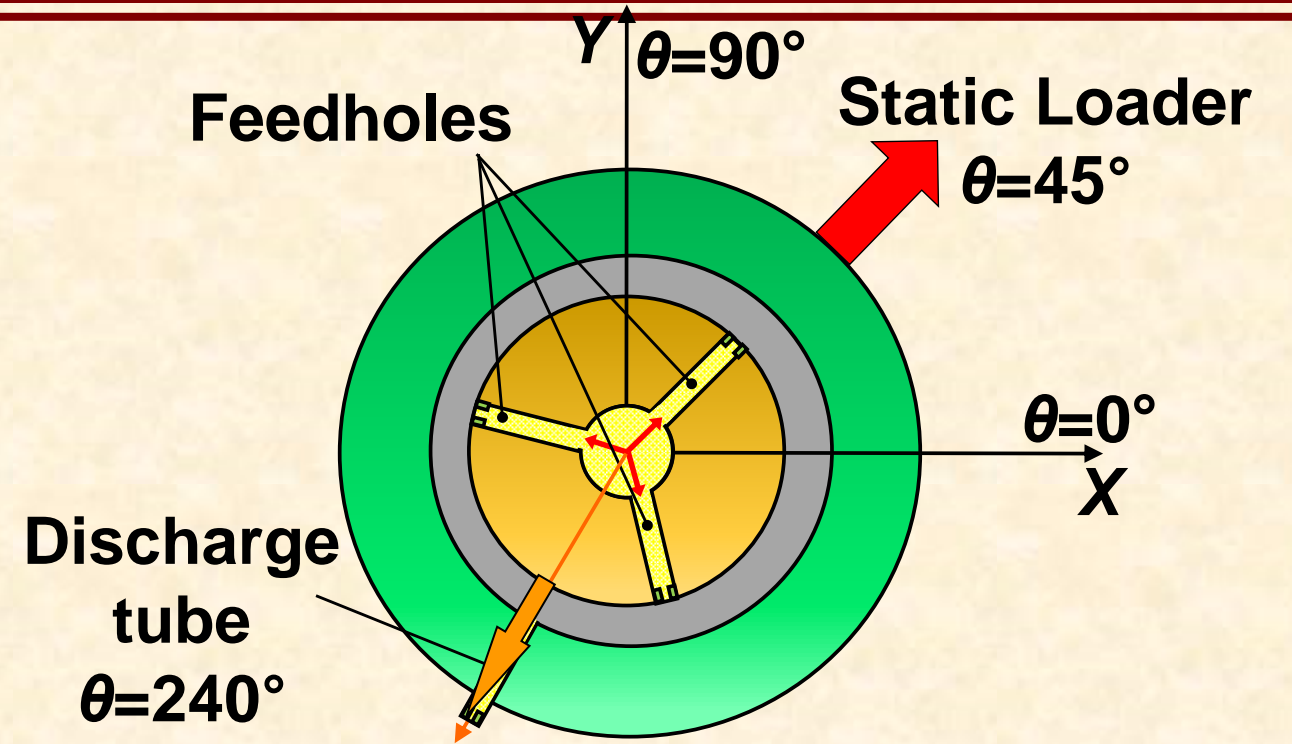
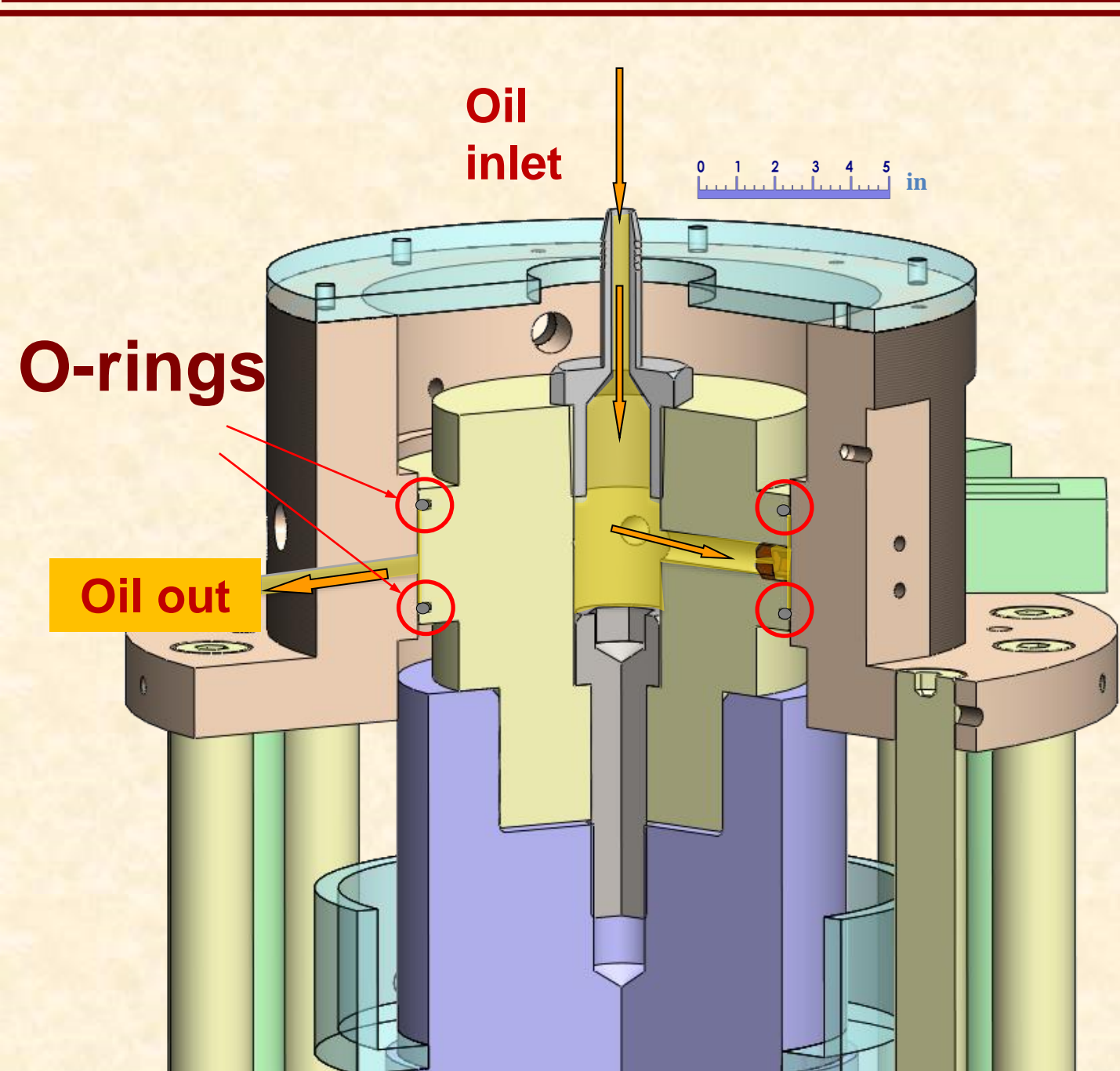


Proper installation of PR with slit on correct side is important.

PR slits allow oil leakage and air ingestion



OR-SFD lubricant flow path

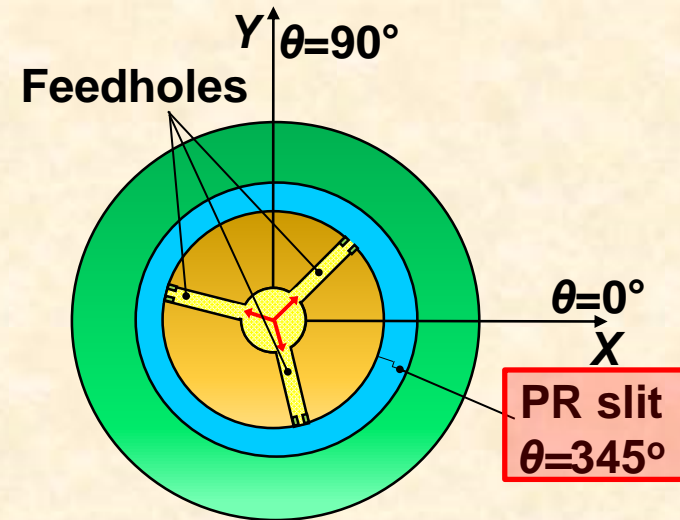


Adequate squeeze and volume fill ensure proper sealing. If too low, the OR does not seal; if too tight OR permanently deforms and becomes overly stiff.

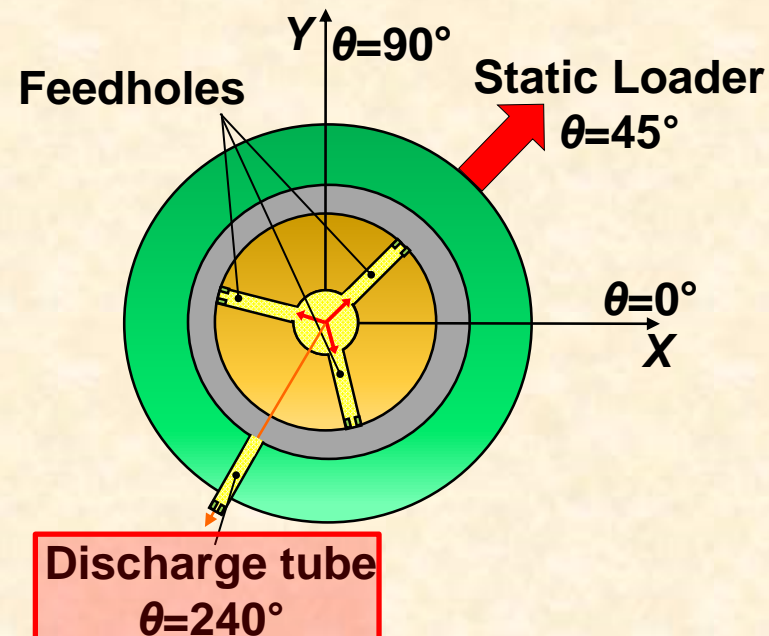
ORs fully seal film land; hence oil evacuation through hole needed.

Short length damper ($L/D=0.2$, $D/c=340$)

PR-SFD



OR-SFD



Journal diameter, D	127 mm
Axial film land length, L	25.4 mm
Radial clearance, c	0.373 mm

Feedhole diameter, ϕ_{in}	2.5 mm
angular location, θ_{in}	45°, 165°, 285°

OR-SFD Discharge hole diameter, ϕ_{exit}	1.0 mm
hole location, θ_{exit}	240°

PR-SFD slit location, θ_{slit}	345°
--	------

Max squeeze velocity $v_s = r\omega \sim 60$ mm/s
 $\rightarrow Re_s = (\rho / \mu) \omega c^2 \sim 65$

ISO VG 2 oil

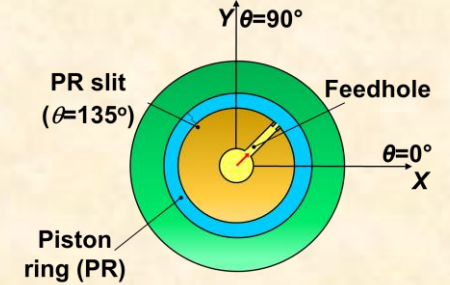
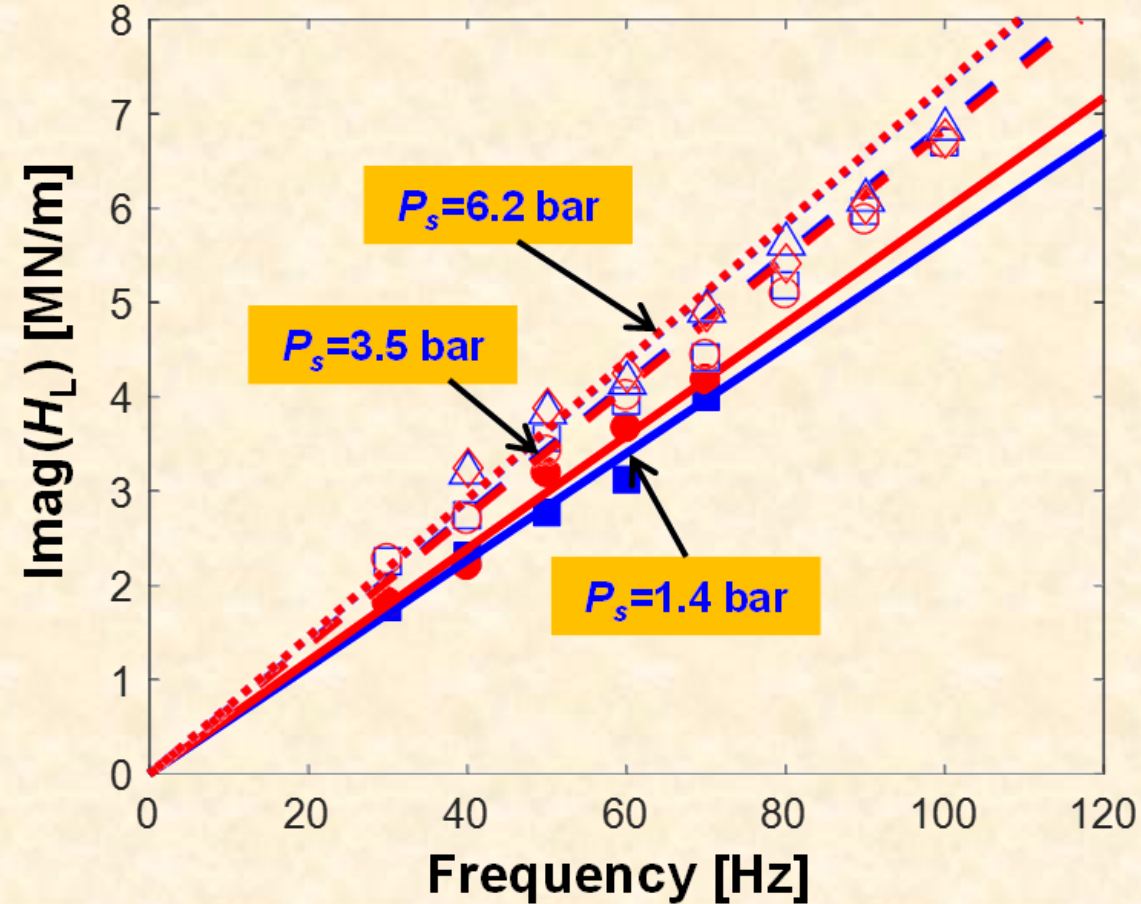
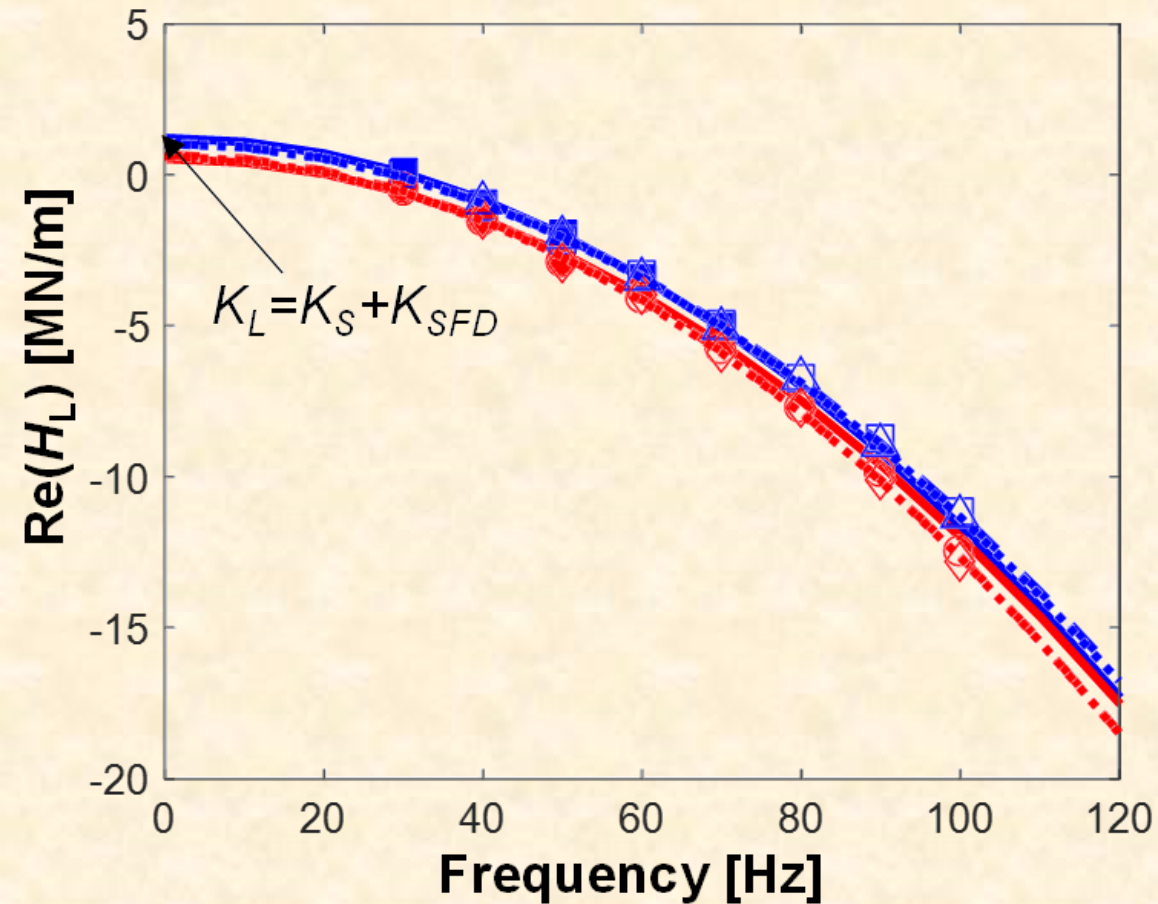
Oil inlet temperature $T_s = 23$ °C
 Density $\rho = 799$ kg/m³
 Viscosity μ at $T_s = 2.7$ cPoise

Typical PR-SFD complex stiffness

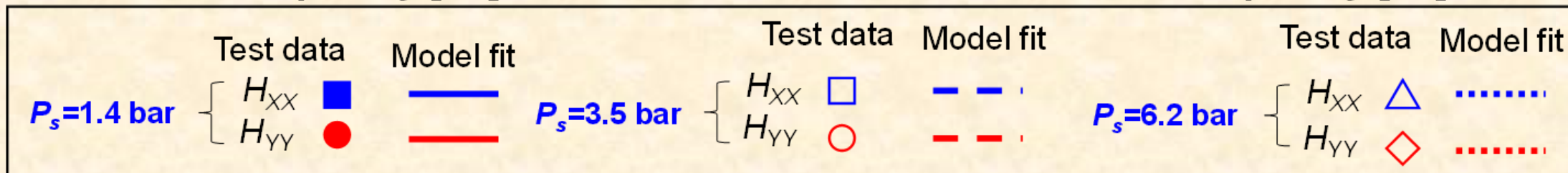
$c=0.37$ mm, orbit size $r/c=0.3$

$$\text{Re}([\bar{\mathbf{F}} - M_{BC}\bar{\mathbf{a}}]\bar{\mathbf{z}}^{-1}) \rightarrow \mathbf{K}_L - \omega^2\mathbf{M}_L$$

$$\text{Im}([\bar{\mathbf{F}} - M_{BC}\bar{\mathbf{a}}]\bar{\mathbf{z}}^{-1}) \rightarrow \mathbf{C}_L\omega$$



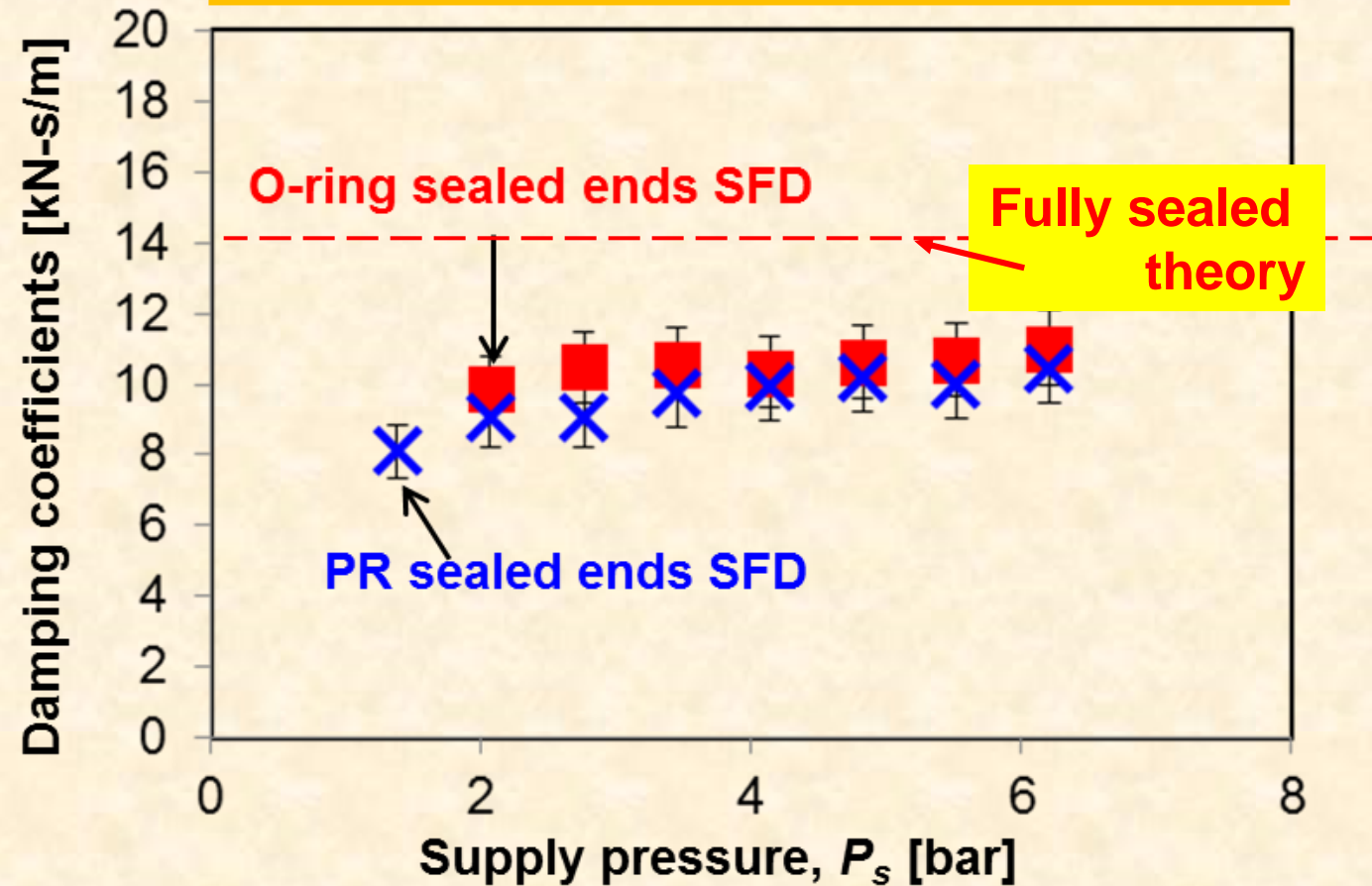
$K_{S+SFD} \sim$
 $K_S = 1.6$ MN/m.
 Damping $C \uparrow$
 as $P_s \uparrow$.
 Added mass
 $M \sim$ same.



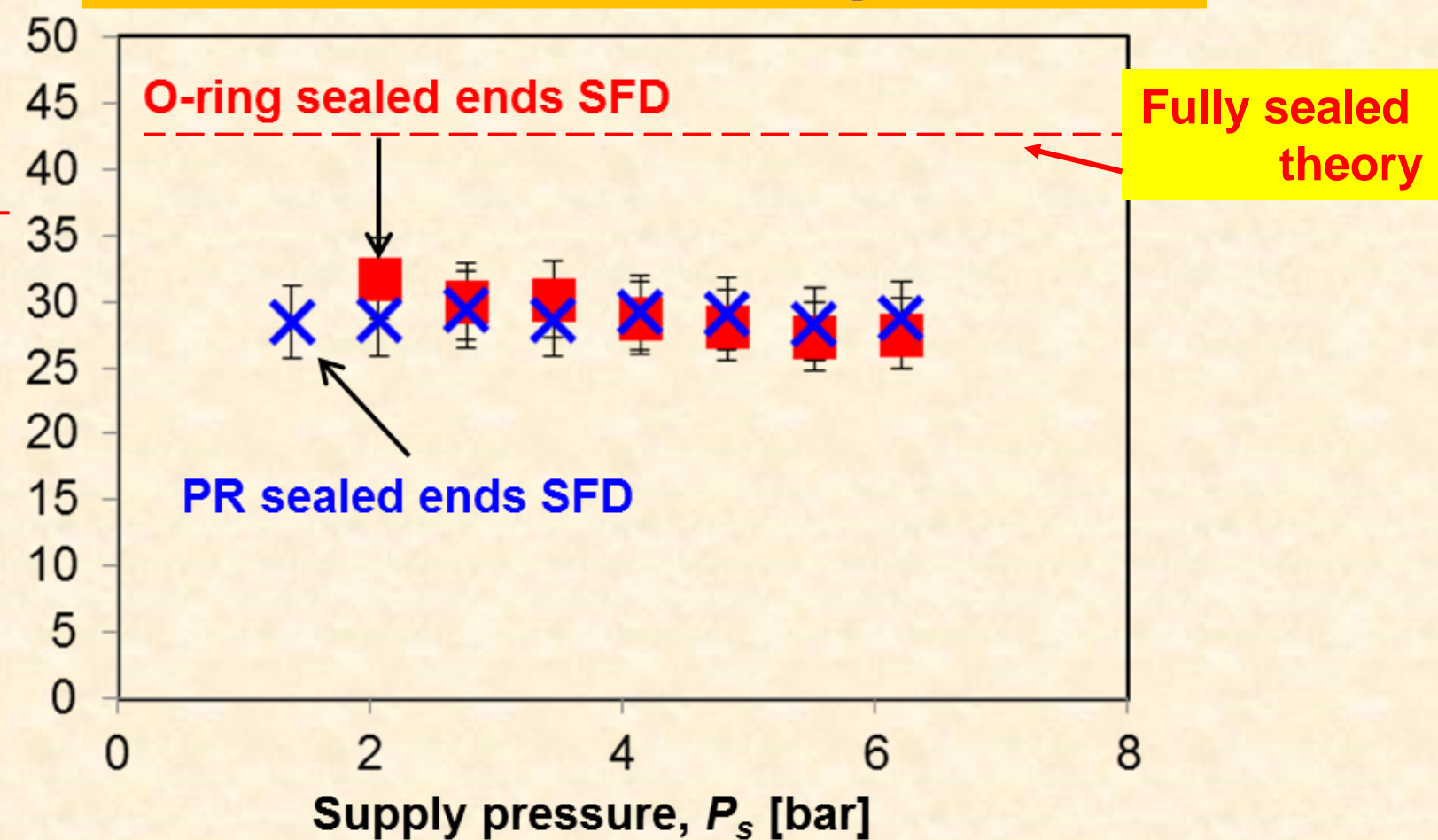
→ Extract SFD force coefficients from curve fits to complex stiffnesses

OR-SFD vs PR-SFD C and M vs. supply pressure

(a) Damping coefficients, $C_{avg} = \frac{1}{2}(C_{XX} + C_{YY})$



(b) Added inertia coefficients, $M_{avg} = \frac{1}{2}(M_{XX} + M_{YY})$

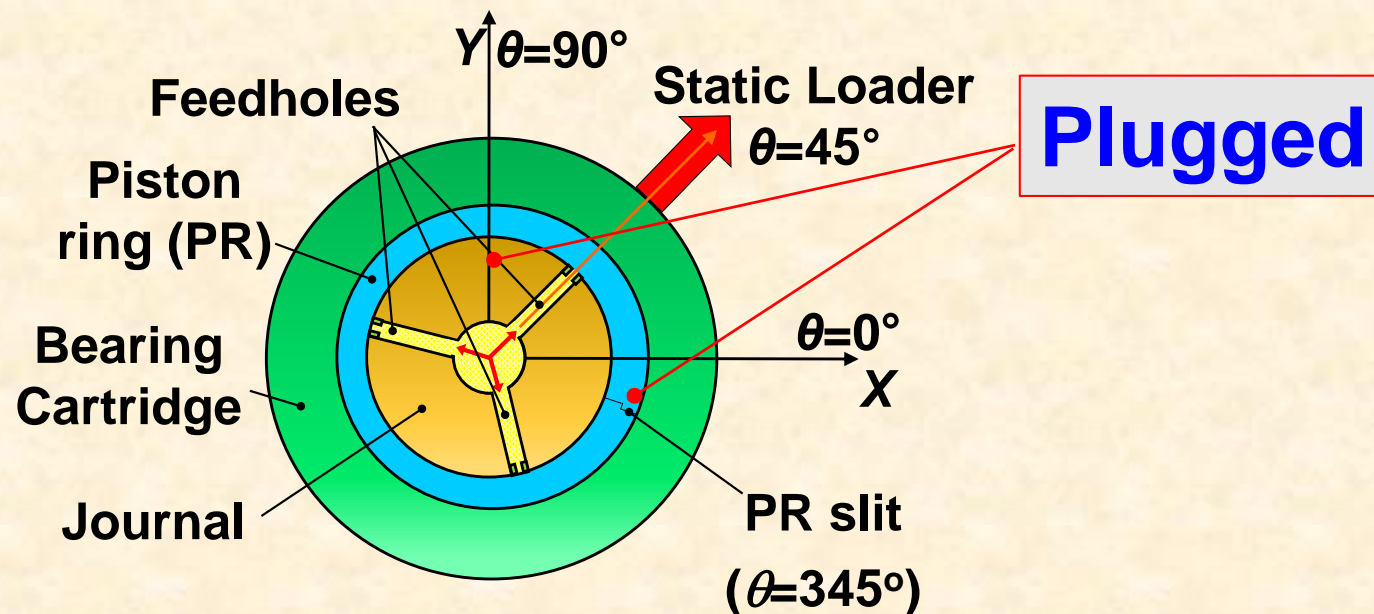


- Damping coefficients $C \downarrow$ as lubricant supply pressure \downarrow
- Damping C for OR-SFD is 11% larger than C for PR-SFD.
- Added mass $M \sim 30$ kg as supply pressure decreases.

$c=373 \mu\text{m}$, $r/c=0.3$, $\omega=10-100$ Hz \rightarrow Max. $v_s=r\omega=70$ mm/s, Max. $Re_s=26$

Effect of number of **open** feed holes on SFD forced performance

OR-SFD vs. PR-SFD



Dimensionless force coefficients

$$\underline{C} = C/C^*$$

$$\underline{M} = M/M^*$$

Long bearing model

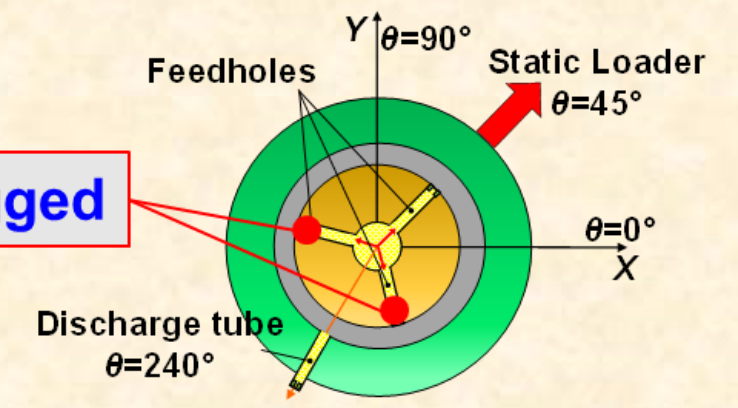
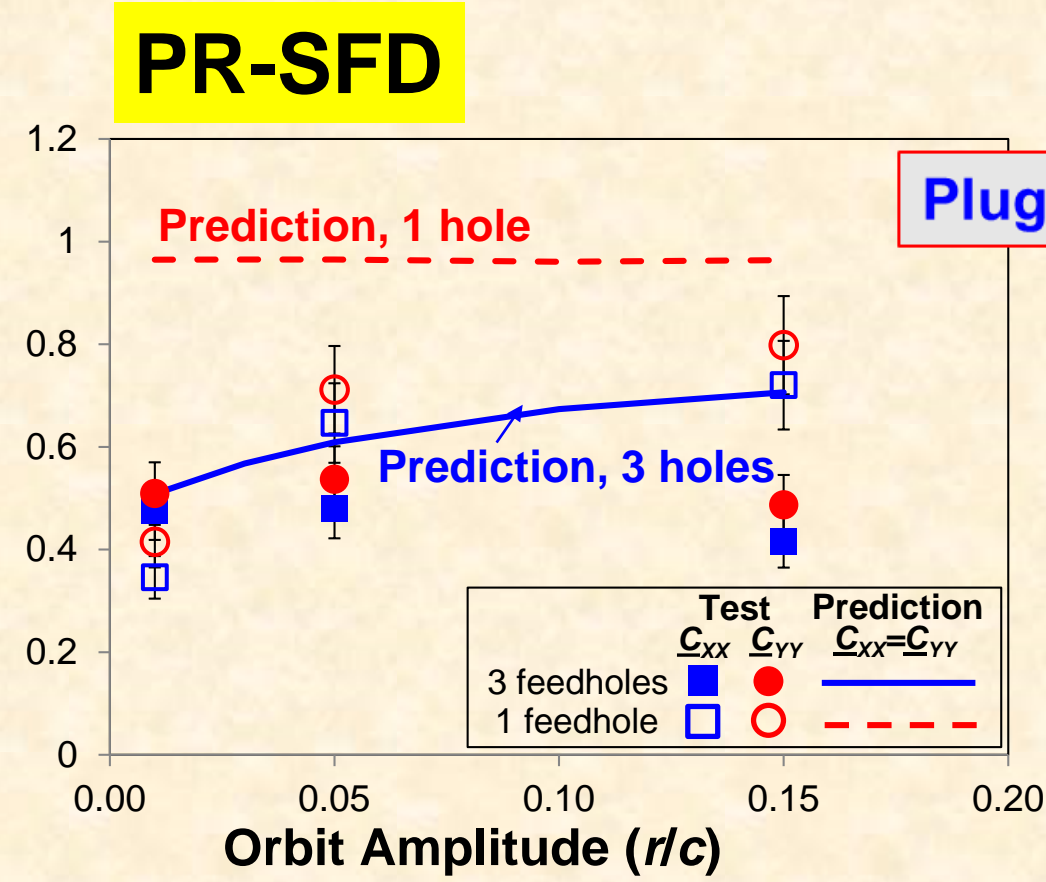
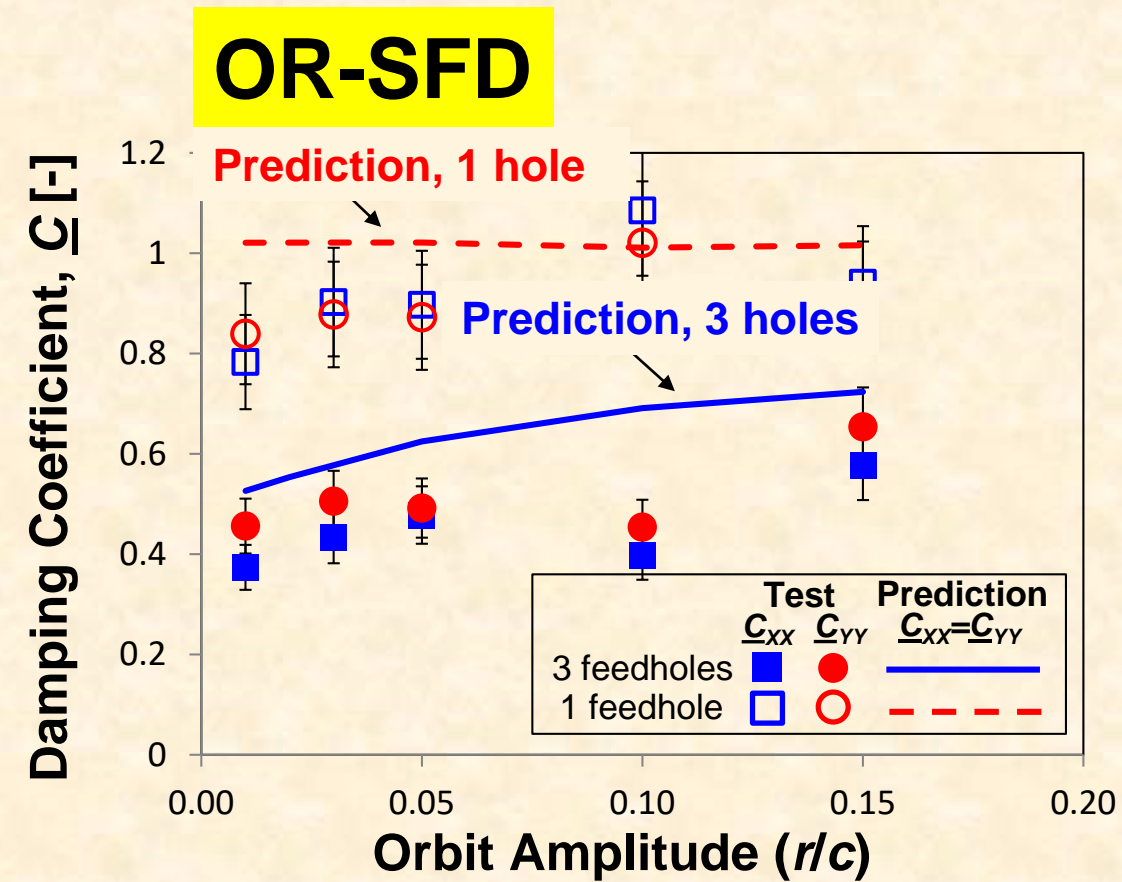
$$C^* = 12\pi\mu L \left(\frac{D}{2c}\right)^3 \quad M^* = \frac{\rho\pi L}{c} \left(\frac{D}{2}\right)^3$$

$$= 12.85 \text{ kN s/m}$$

$$= 44 \text{ kg}$$

of feedholes → damping C_{SFD}

$P_s = 69 \text{ kPa(g)}$



Damper with one open feedhole produces 60% +damping than SFD with three holes.

For damper with one feedhole, $C_{PR-SFD} < C_{OR-SFD}$ due to air ingestion through PR slits.

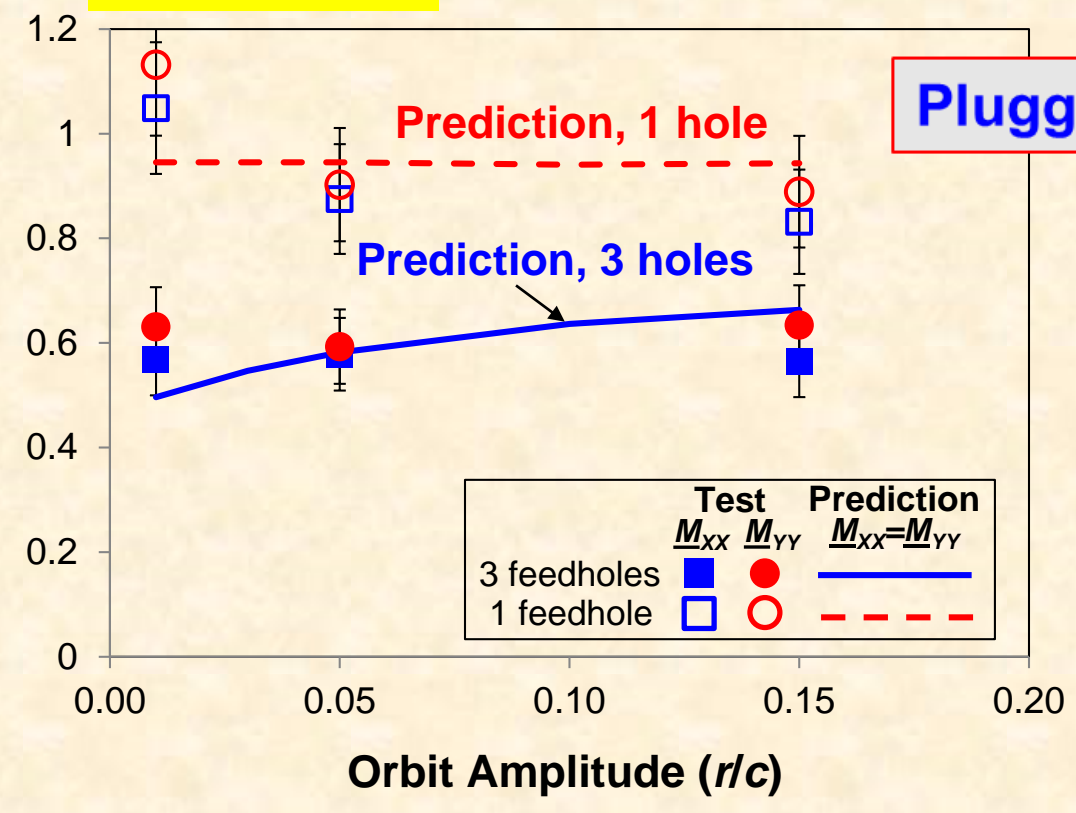
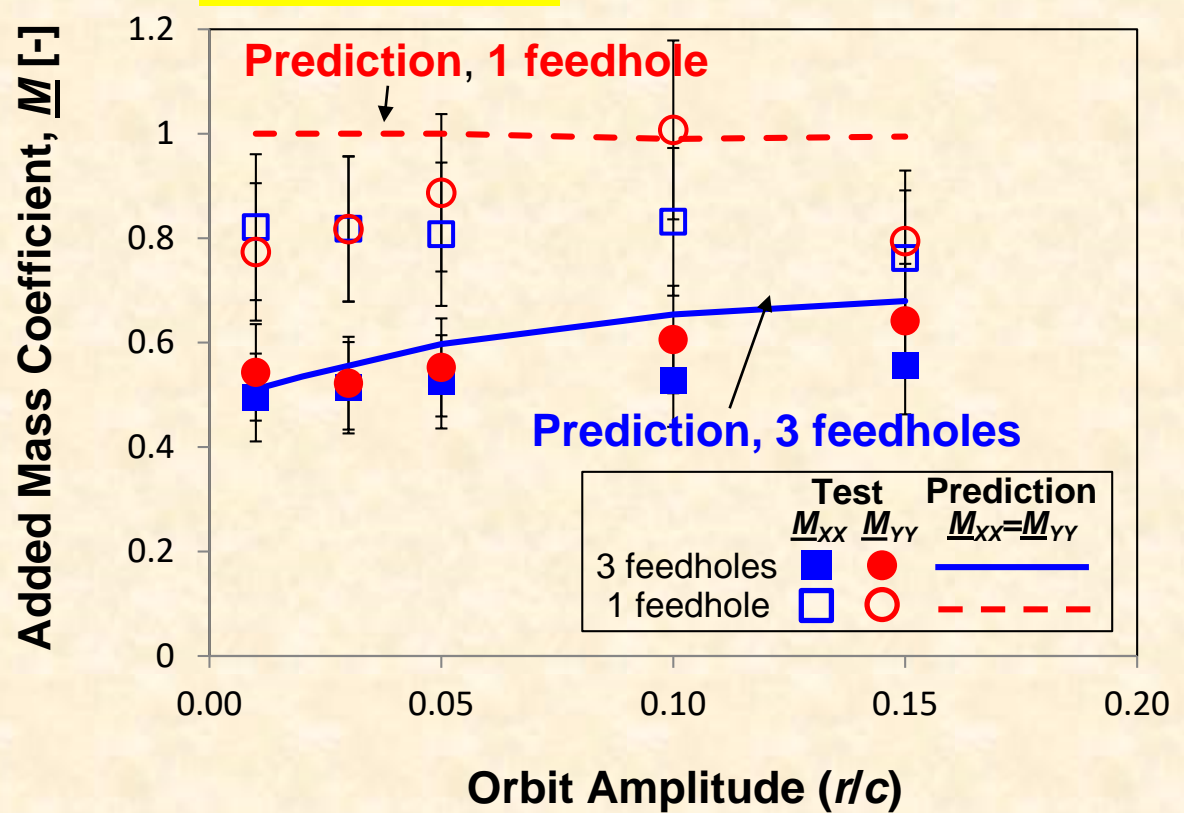
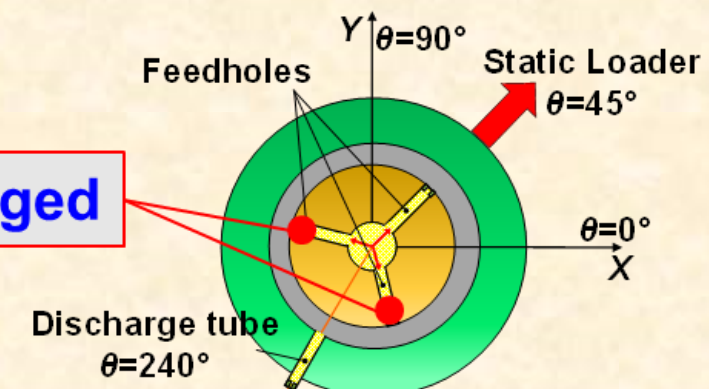
of feedholes → inertia M_{SFD}

$P_s=69$ kPa(g)

OR-SFD

PR-SFD

Plugged



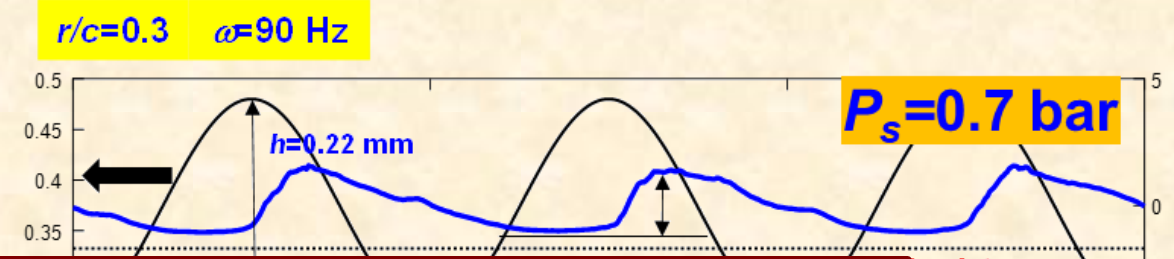
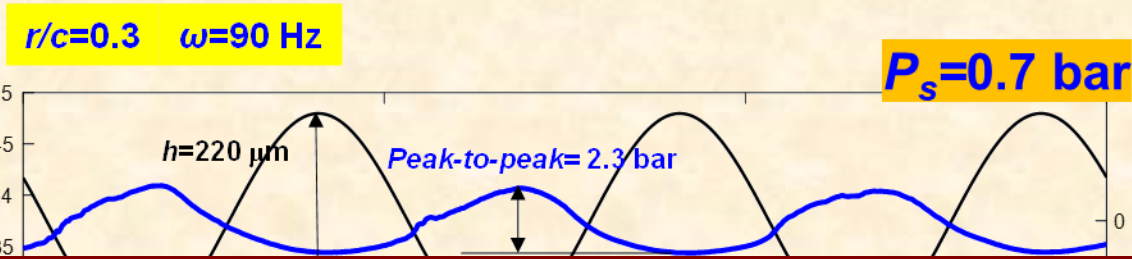
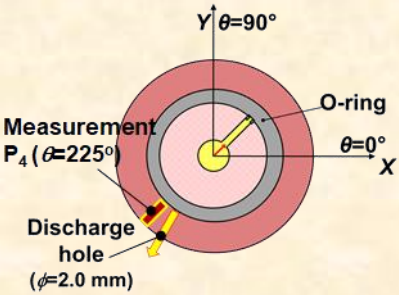
Damper with one feedhole produces more inertia (~80%) than damper with three feedholes. OR-SFD and PR-SFD produce same M .

Although ends are sealed, test coefficients are 50% of long bearing model due to pressure distortions.

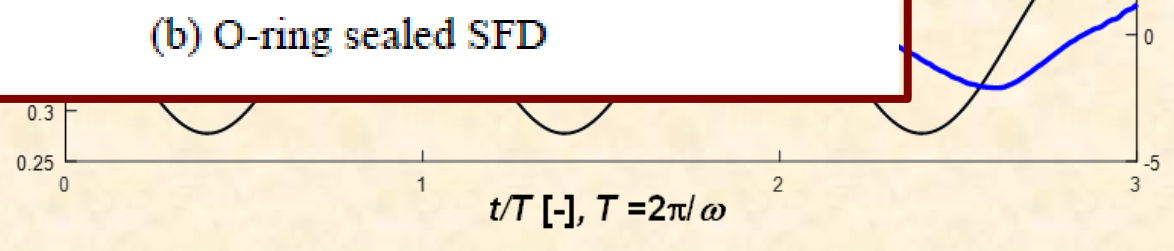
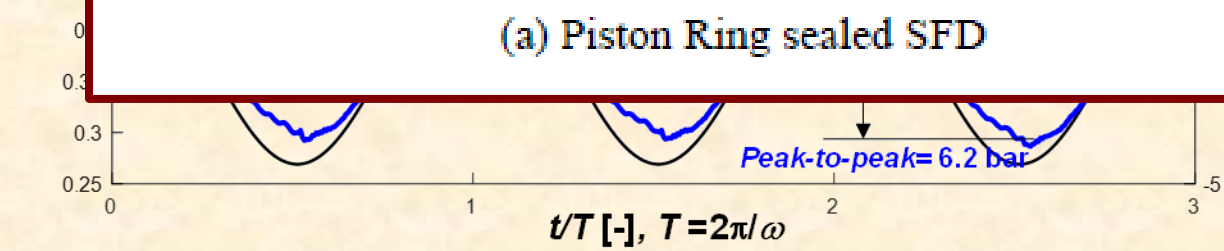
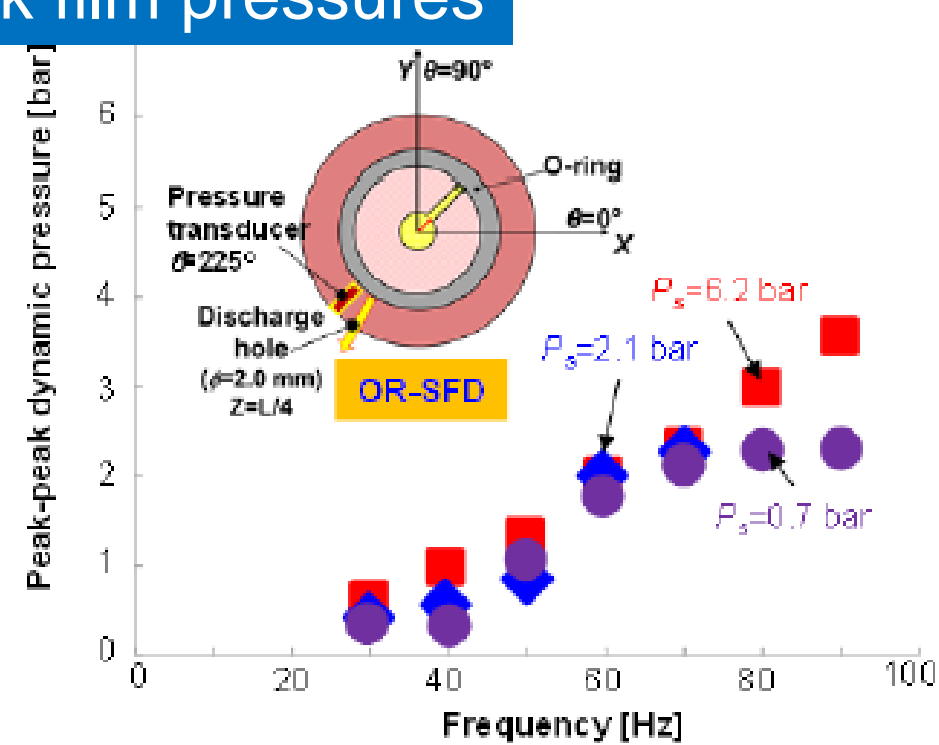
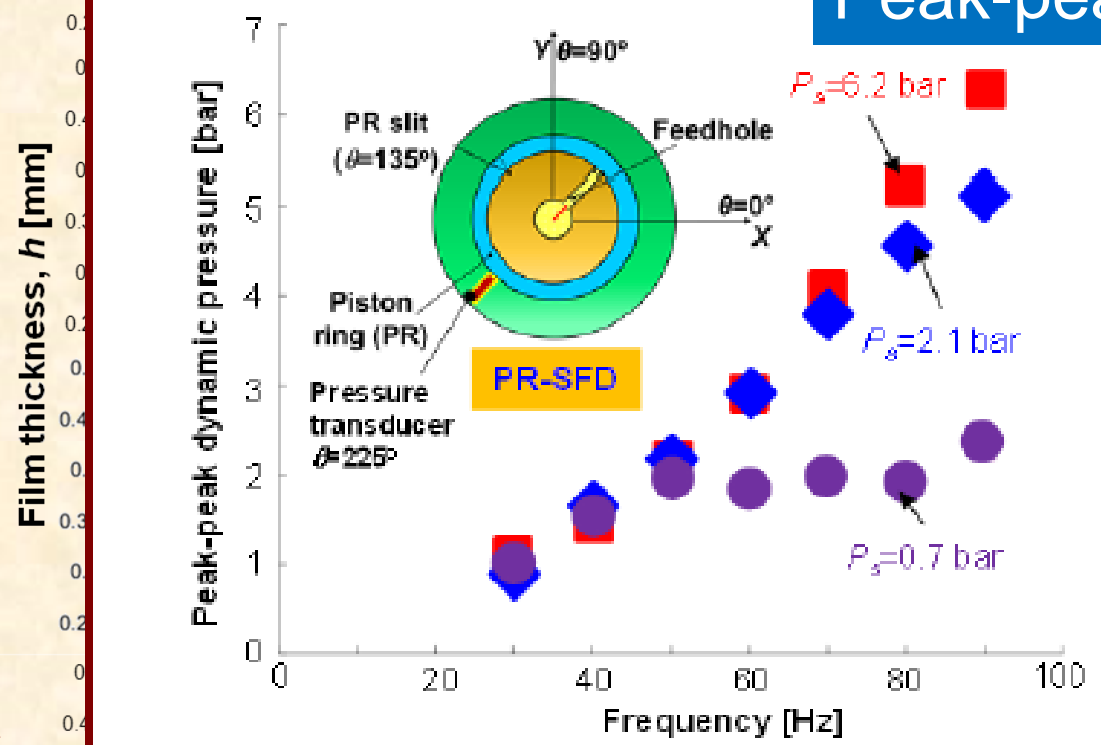
PR-SFD & OR-SFD pressure profiles

$v_s = 63.2 \text{ m/s} = (r=0.3 \text{ c} \times \omega = 90 \text{ Hz})$

PR-SFD shows air ingestion at low feed pressures.



Peak-peak film pressures



PR-SFD

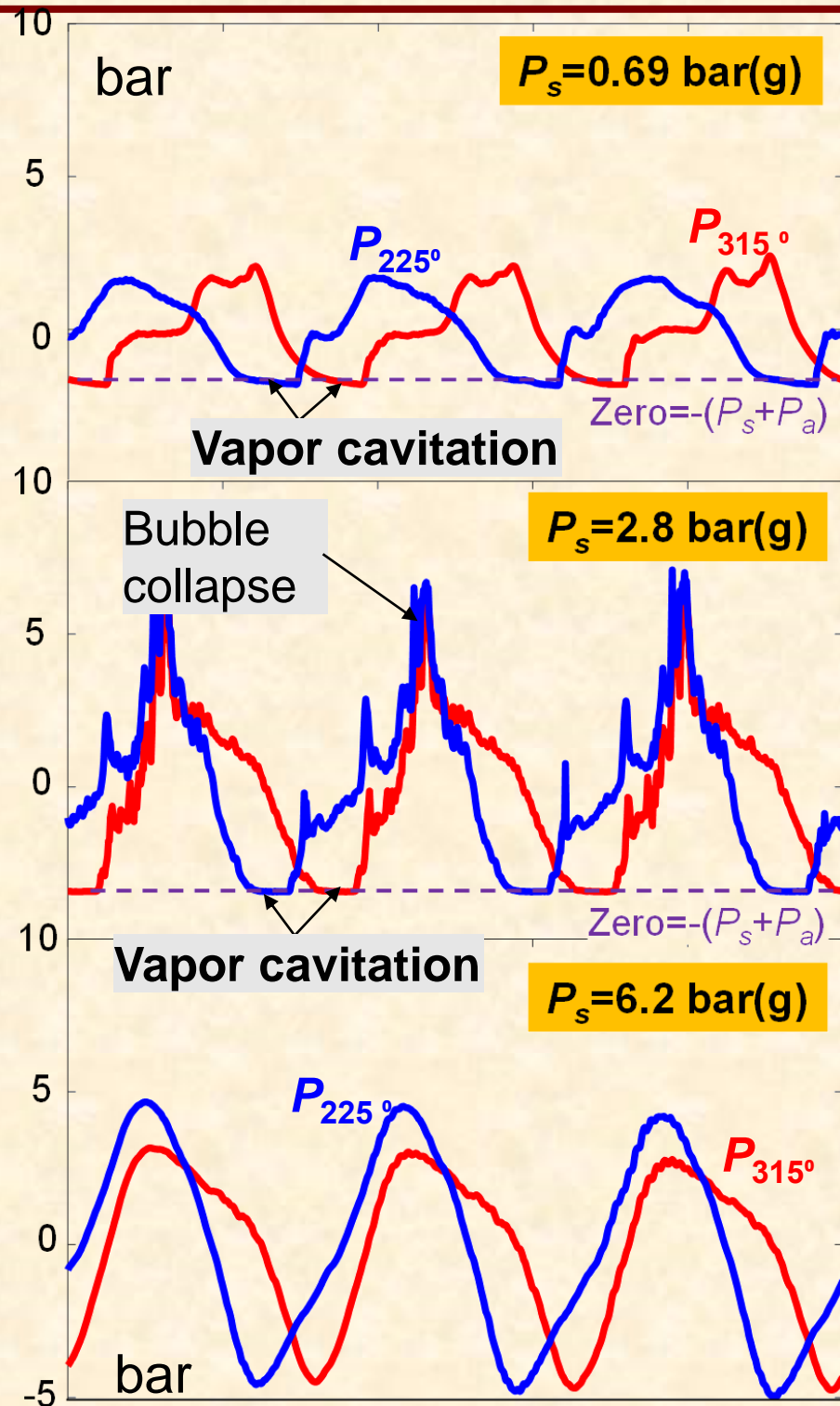
OR-SFD

Closure OR-SFD vs PR-SFD

- (a) O-ring damper provides more damping as it avoids air ingestion. O-rings add stiffness and viscoelastic damping to test system.**
- (b) For both PR-SFD and OR-SFD, the more # feedholes, the lower the damping coefficient.**
- (c) SFD coefficients from dynamic film pressure are largely in error. Pressure field does not simply rotate!**
- (d) Film pressures show oil vapor cavitation and persistent air ingestion for operation at a low supply pressure and/or with a large squeeze velocity (v_s).**

PR-SFD more film pressures

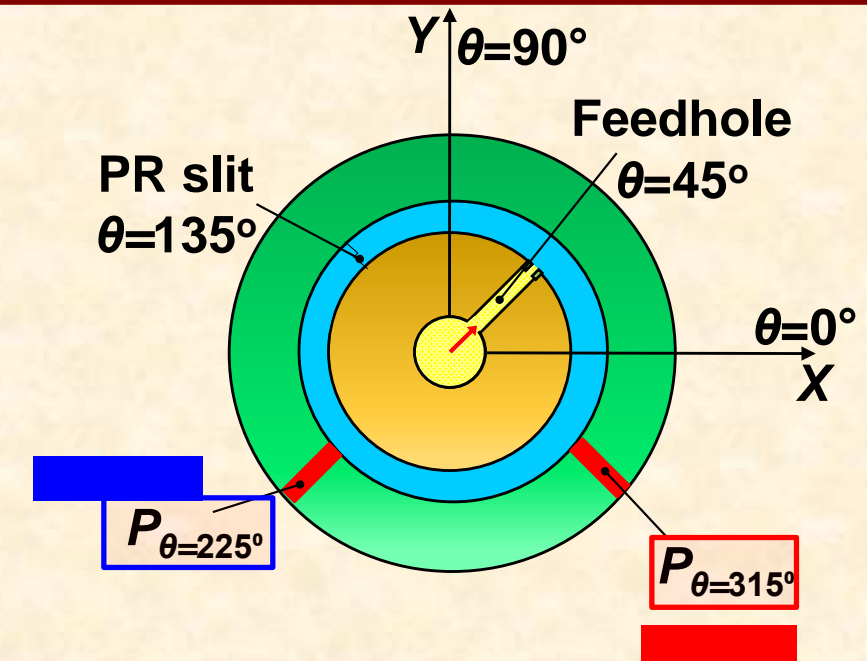
60 Hz, $r=0.65c$: $v_s \sim 90$ mm/s $\rightarrow Re_s \sim 65$



Supply pressure increases

Air ingestion + Vapor cavitation

No air ingestion or oil vapor



Pressures show both oil vapor cavitation & air ingestion with large amplitude/high frequency spikes from bubbles collapsing.

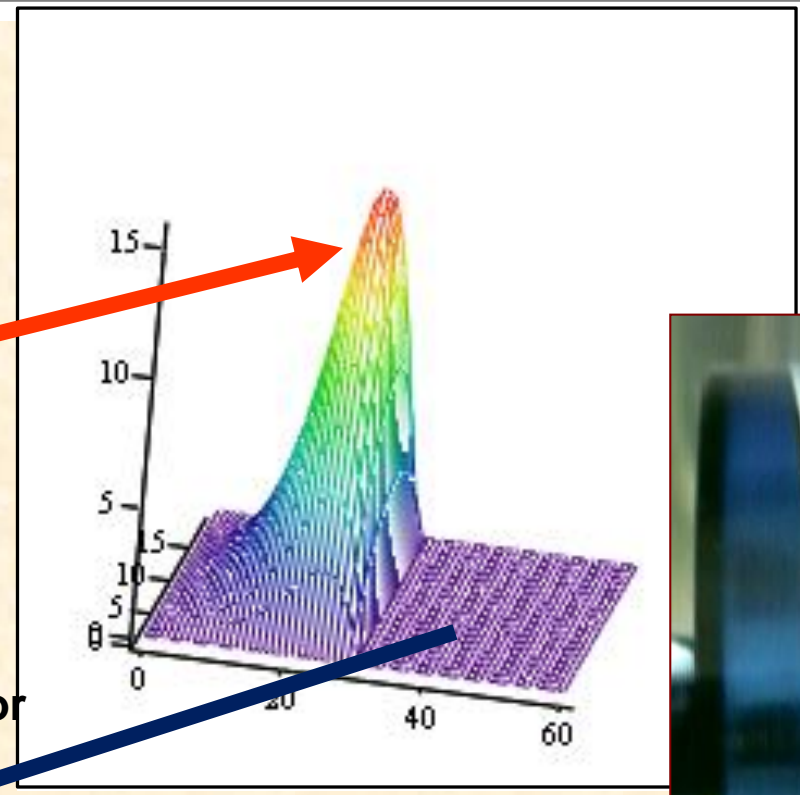
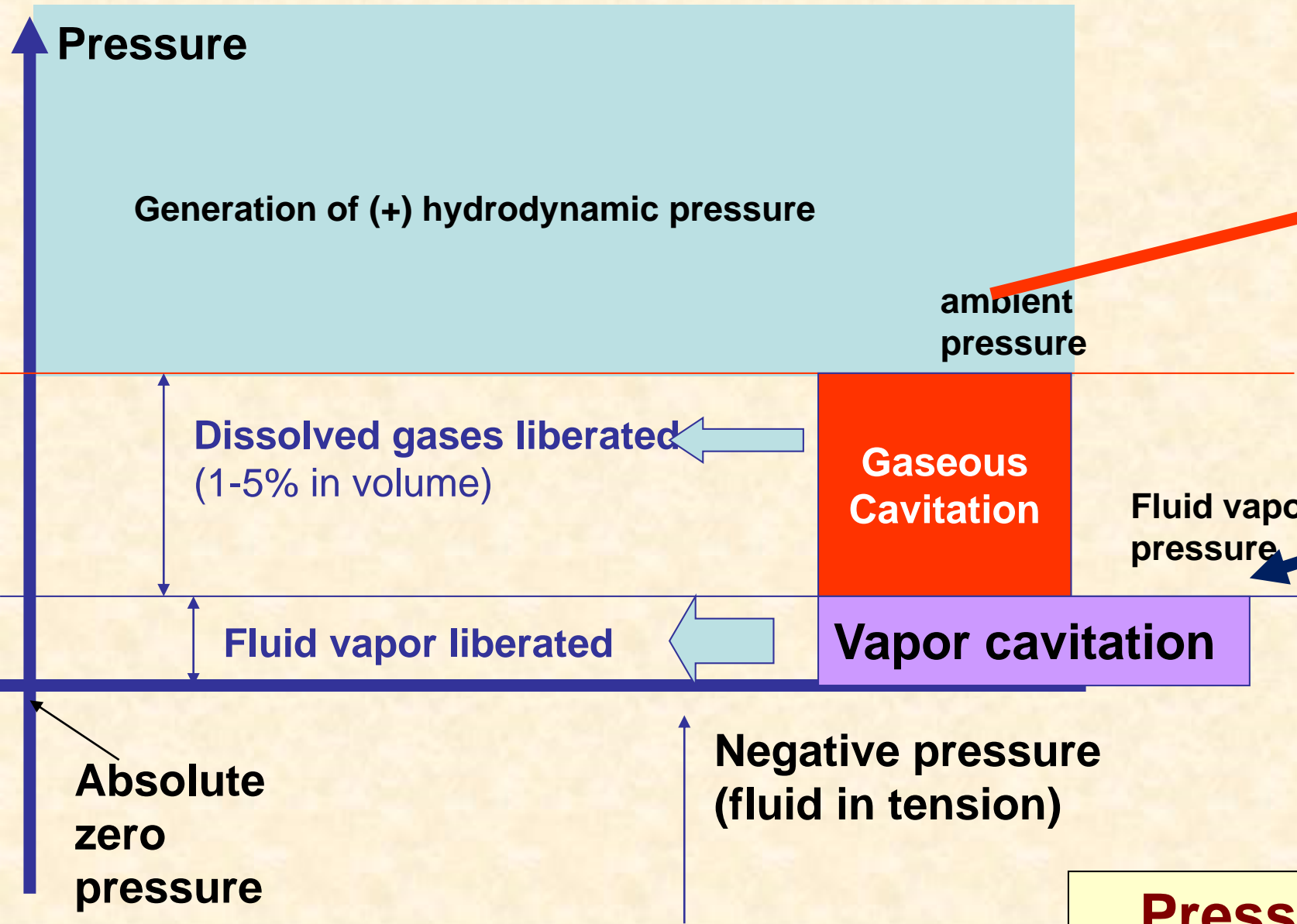
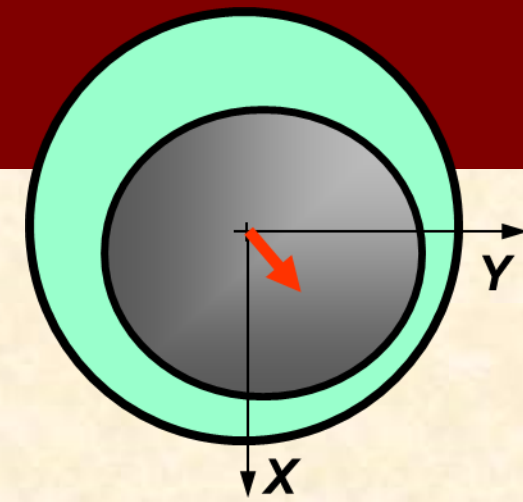
Pressures are distorted & do NOT displace with whirl speed!

Oil cavitation vs. air ingestion and entrapment?

Major issue for reliable SFD forced performance!

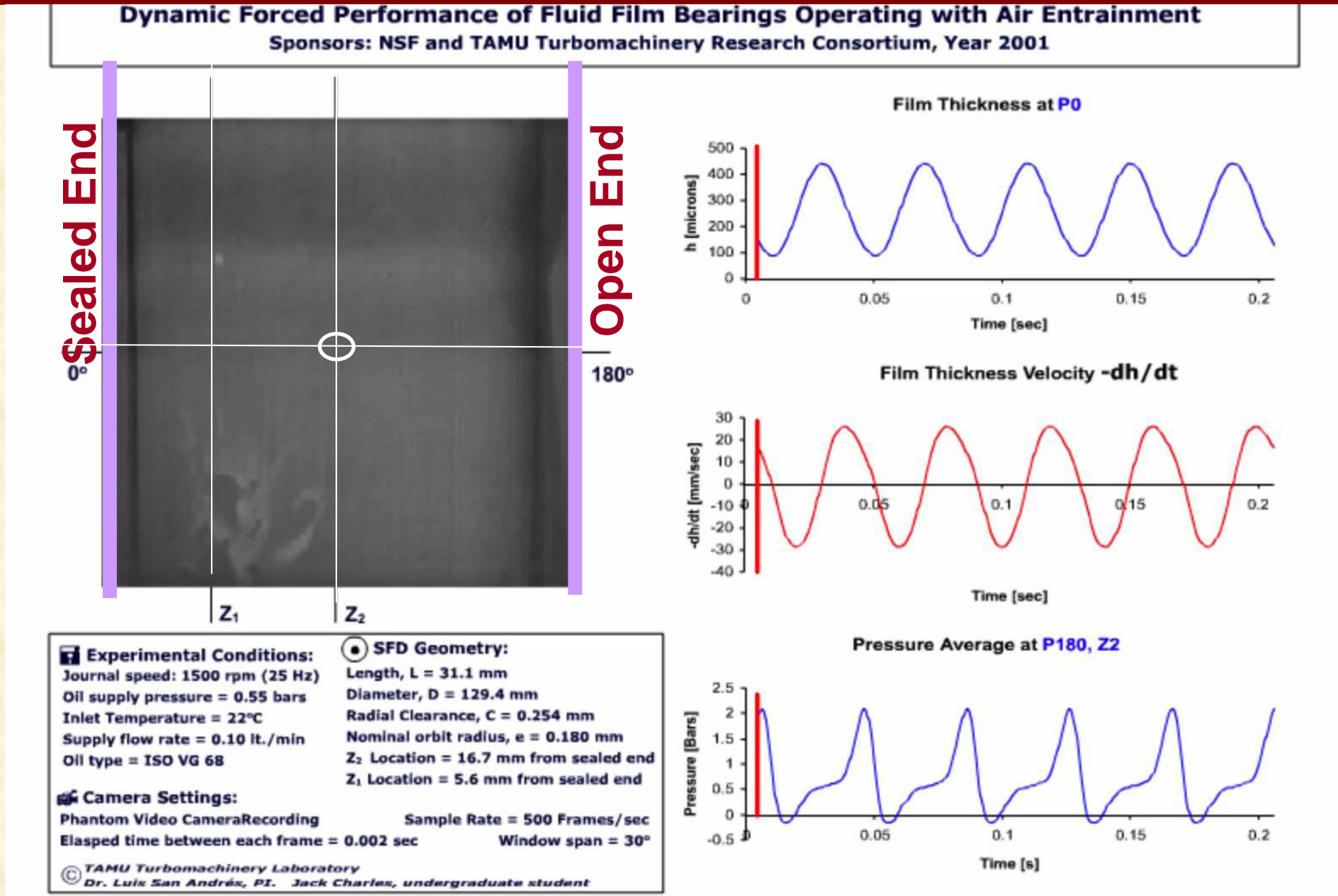


Cavitation in oil lubricated bearings



Pressure is uniform (constant) inside cavitation "bubble" – Flow reformation at trailing edge of bubble.

SFD flow visualization of air ingestion (2001 NSF)

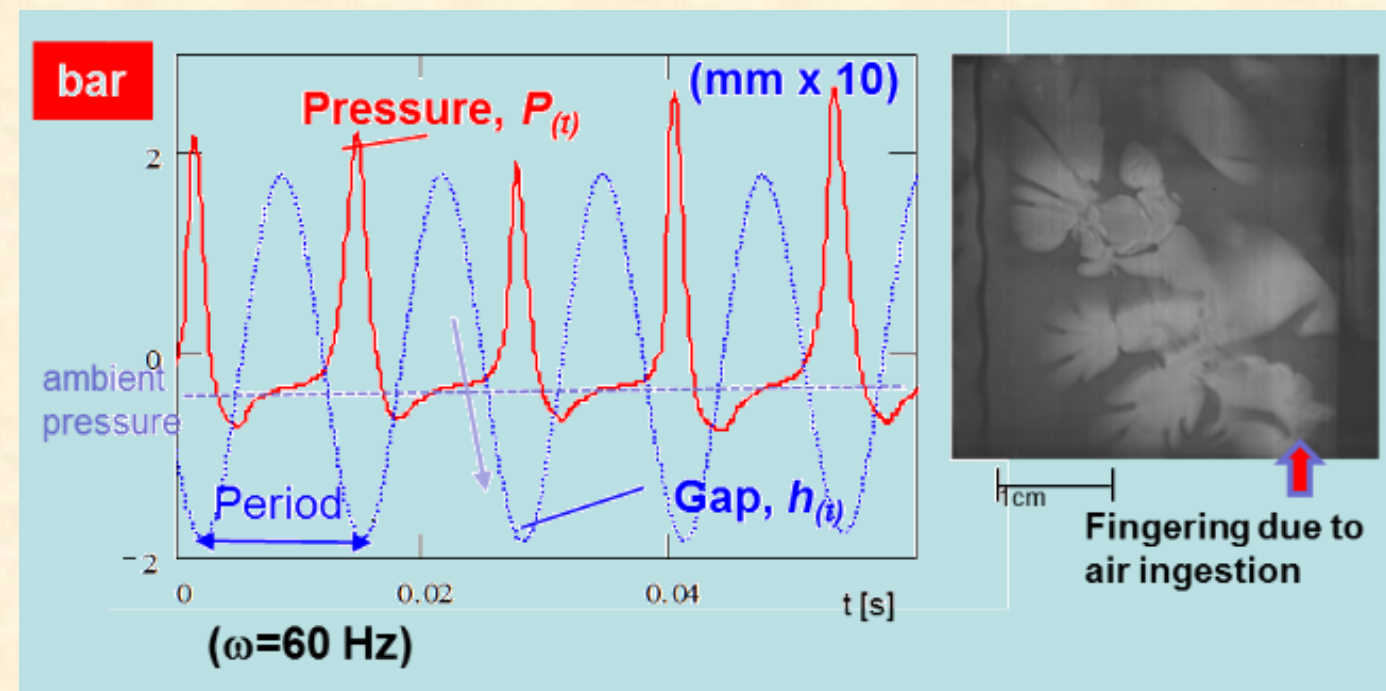
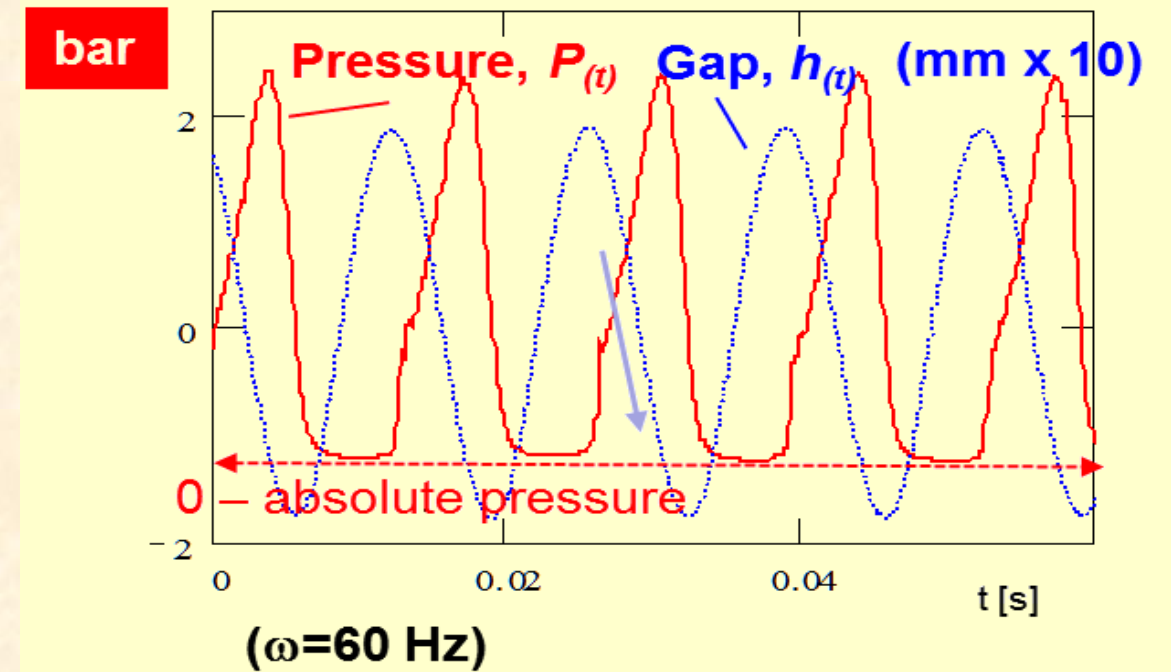


Lubricant cavitation vs. air ingestion in SFJs

Lubricant vapor cavitation: A **constant pressure zone** at nearly zero absolute pressure.

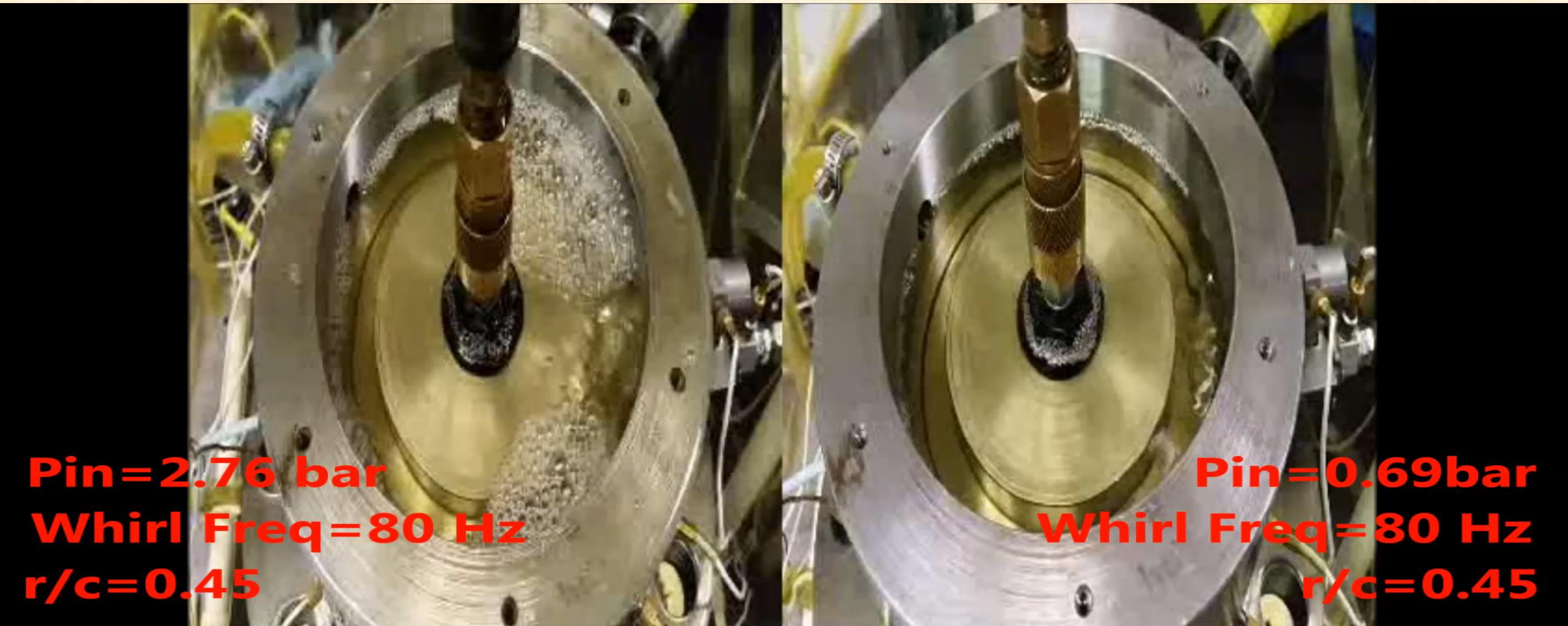
Gas cavitation: Cavitation zone, contains released **dissolved gas** in lubricant and appears steady in a rotating frame.

Air ingestion and entrapment: as local gap opens, **air is drawn** to fill in empty volume.



Onset of air ingestion

→ High and low oil feed pressures



Oil foamy mixture evolves through piston ring slit.

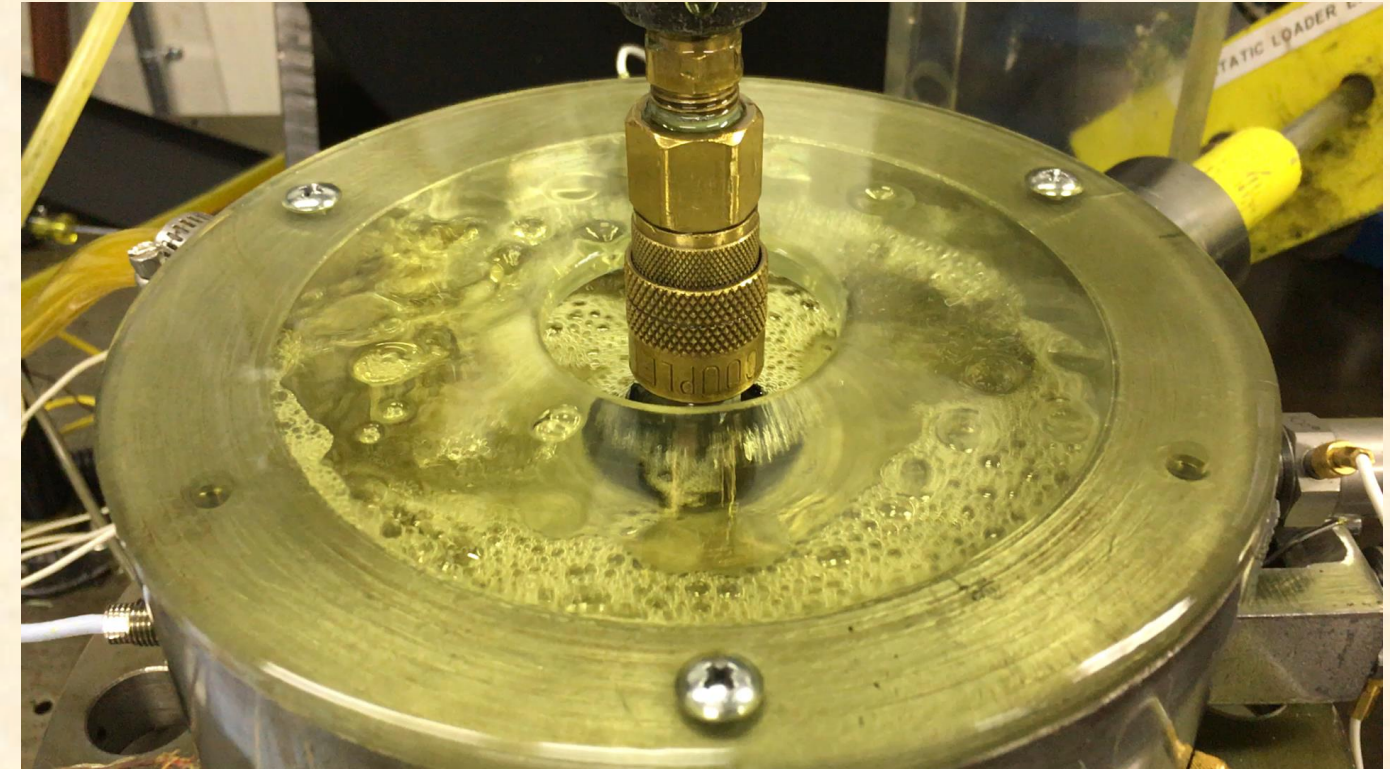
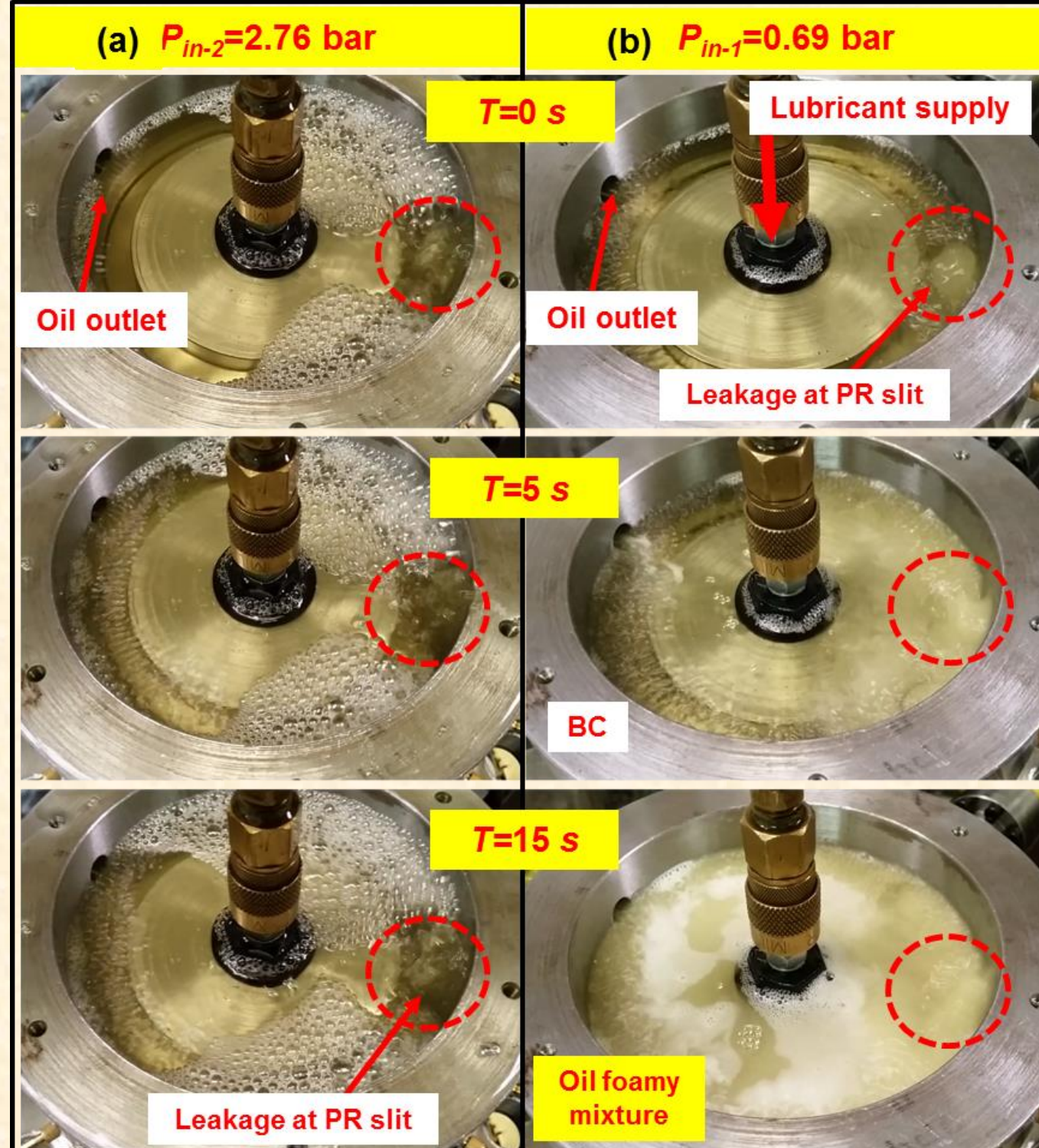
PR-SFD

Time evolution of exit flow

$r/c=0.45, \omega=80\text{Hz}$

high feed pressure

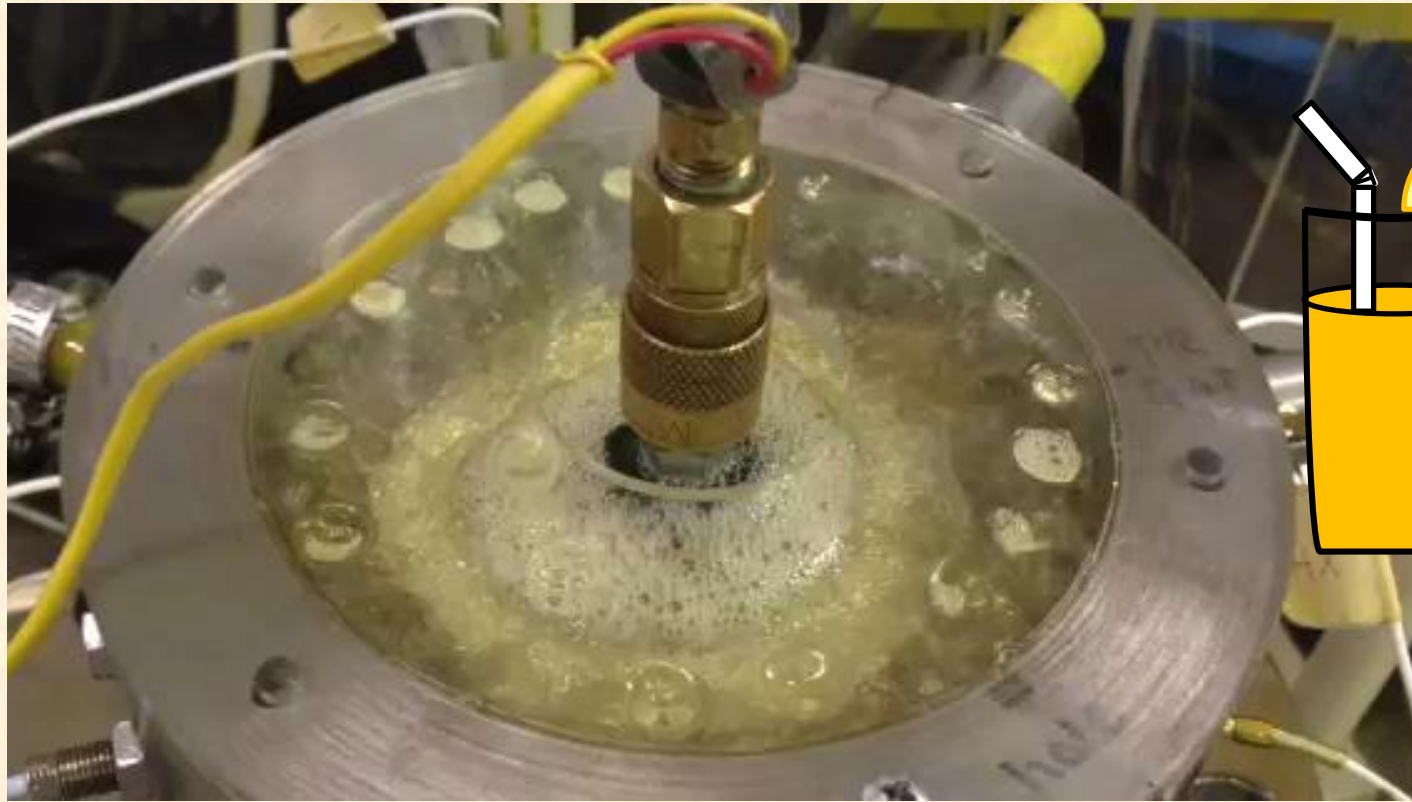
low feed pressure



→ Bubbly mixture makes a foam.

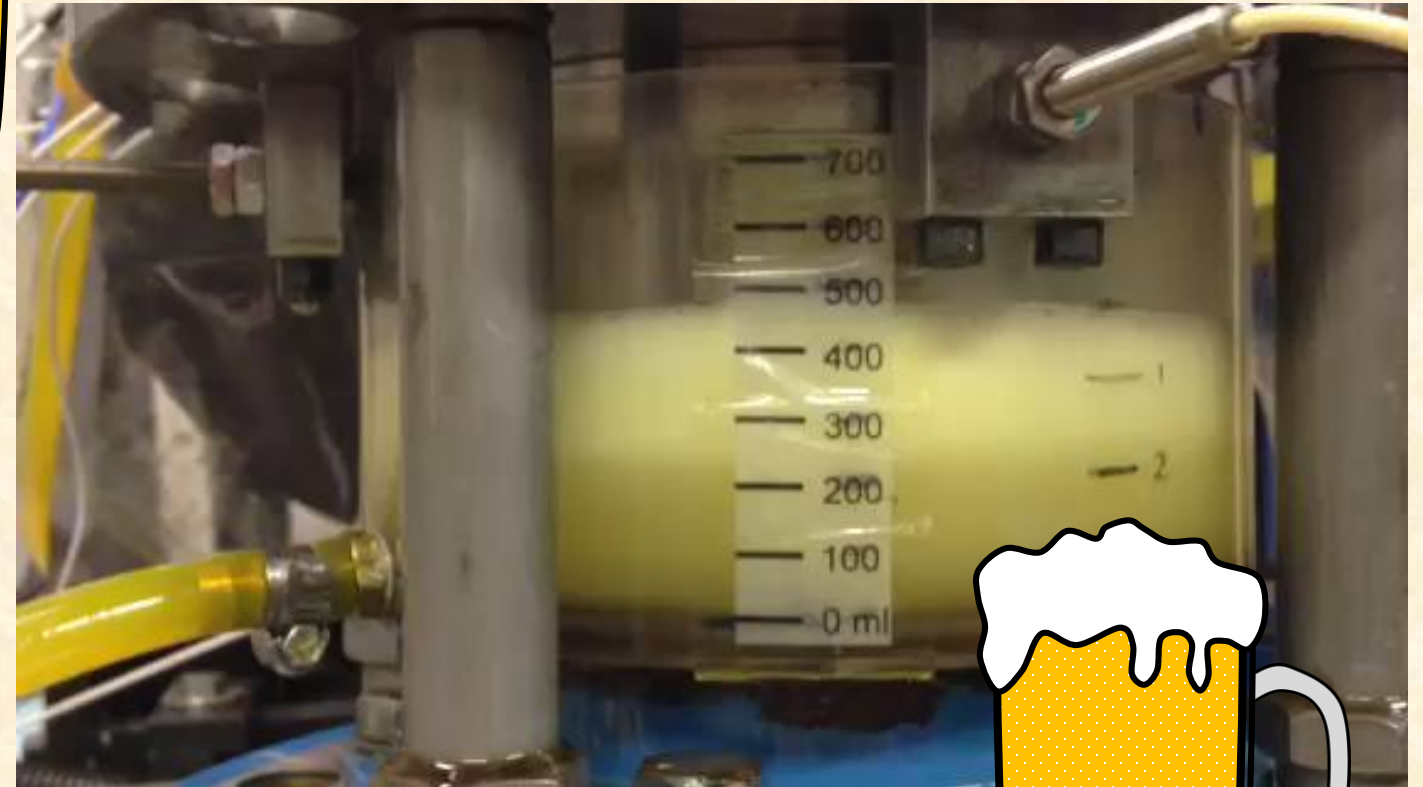
SFD operational issue

Air ingestion and entrapment



Bubbly lubricant leaves SFD through top and bottom ends.

PR-SFD



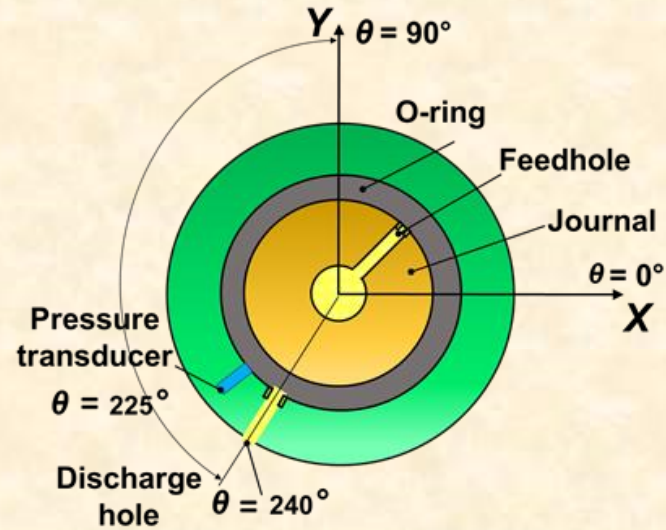
How much gas is ingested?



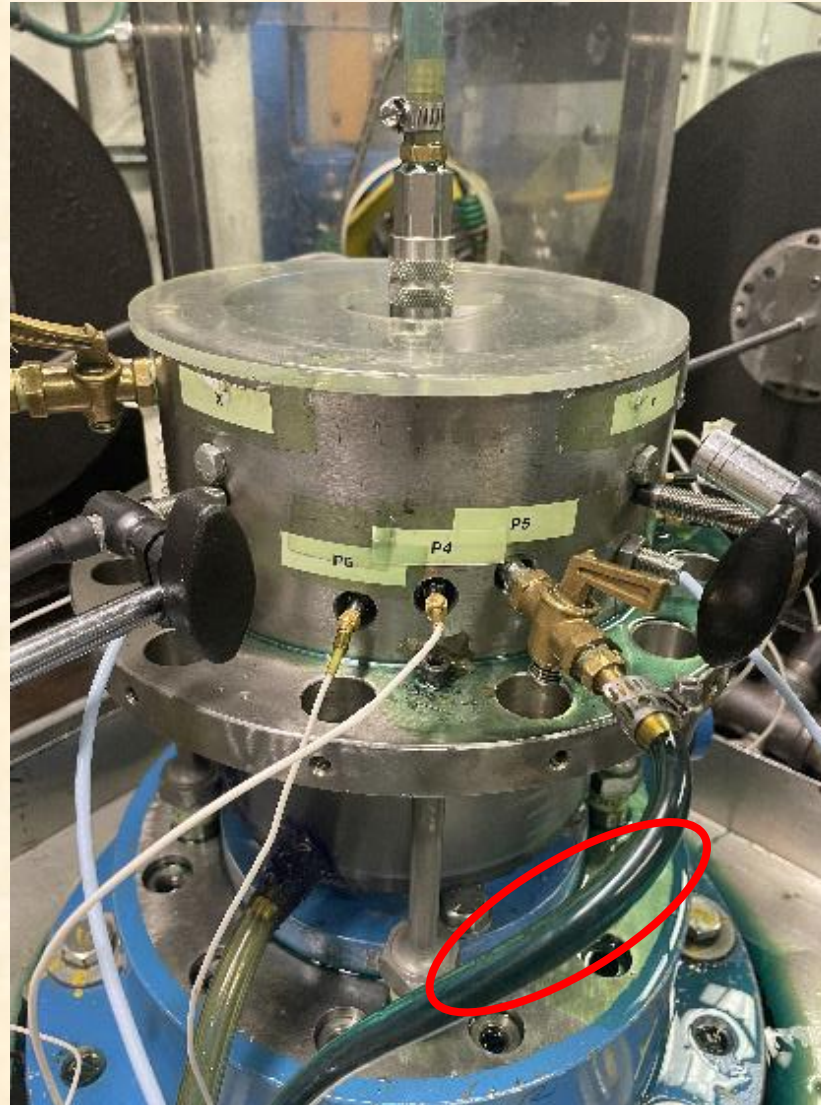
Quantifying air ingestion could aid selecting adequate operating conditions and to produce accurate predictions.

Estimation of gas content in film

OR-SFD



Rodriguez, L., San Andrés, L., 2012, TRC-SFD-01-2022.



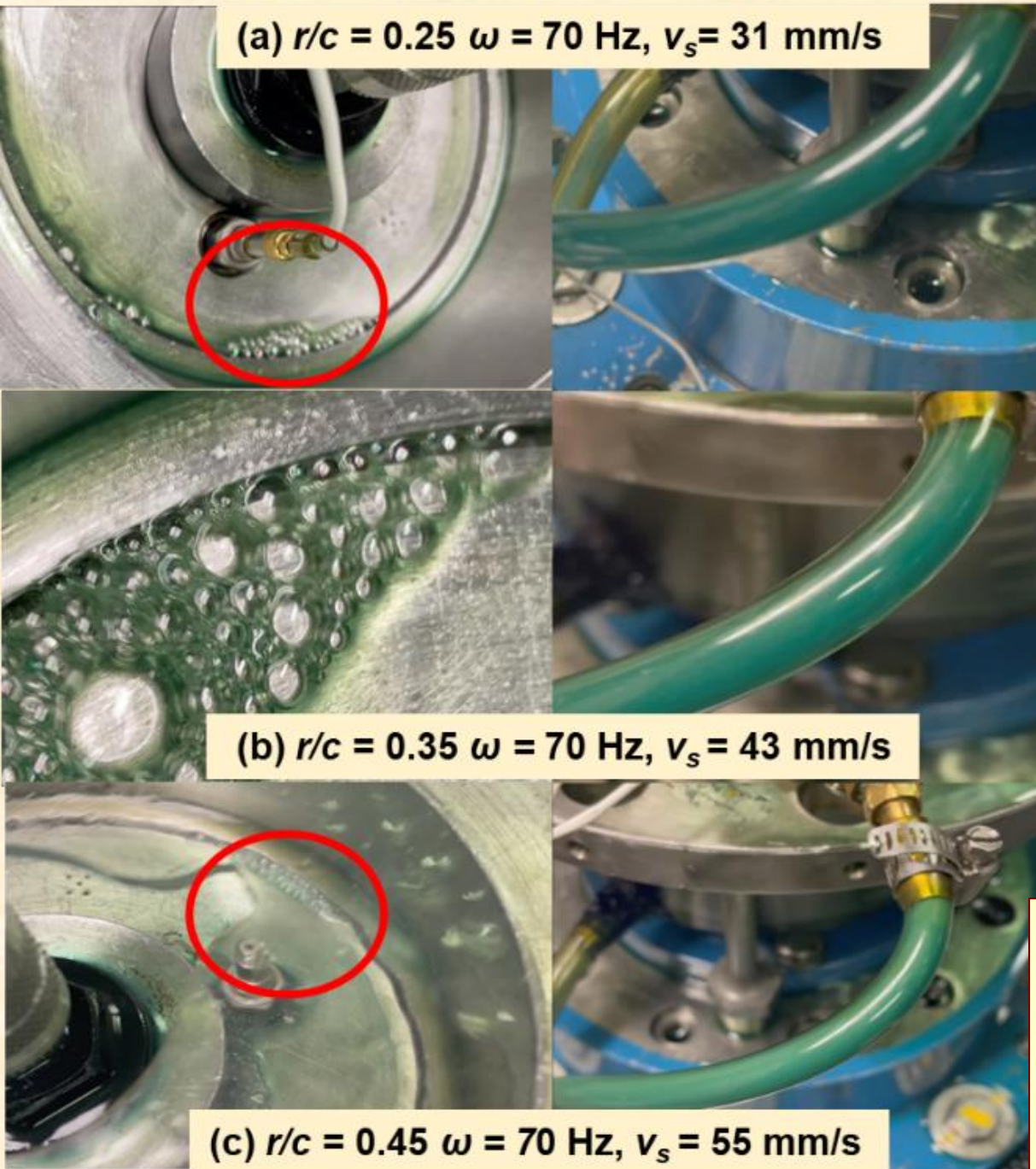
Blue oil (no gas)



Bubbly oil (foam)

Visual records showing **air ingestion**

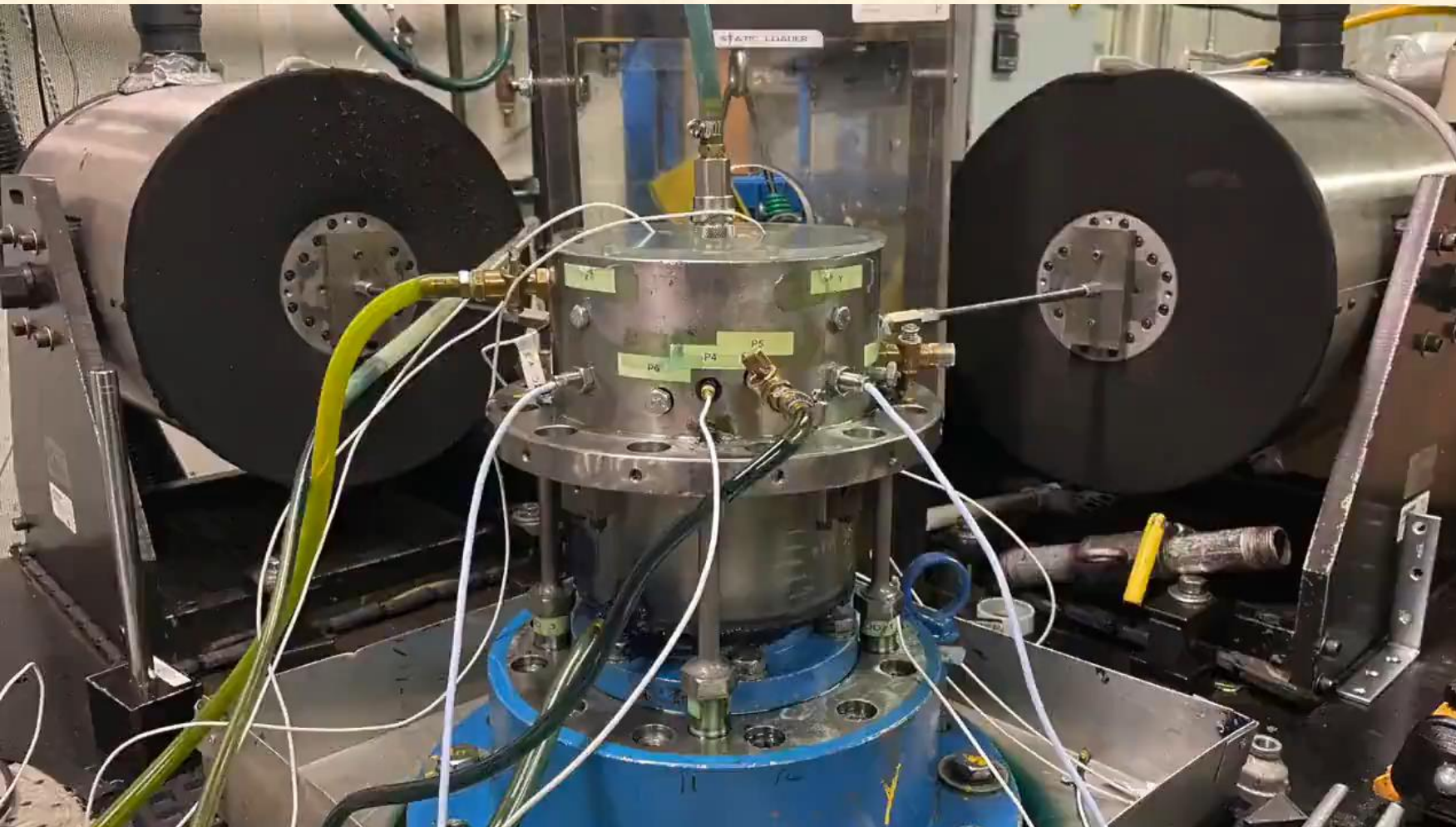
Squeeze Velocity [mm/s]	Orbit radius (r/c) [-]
31	0.25
43	0.35
55	0.45



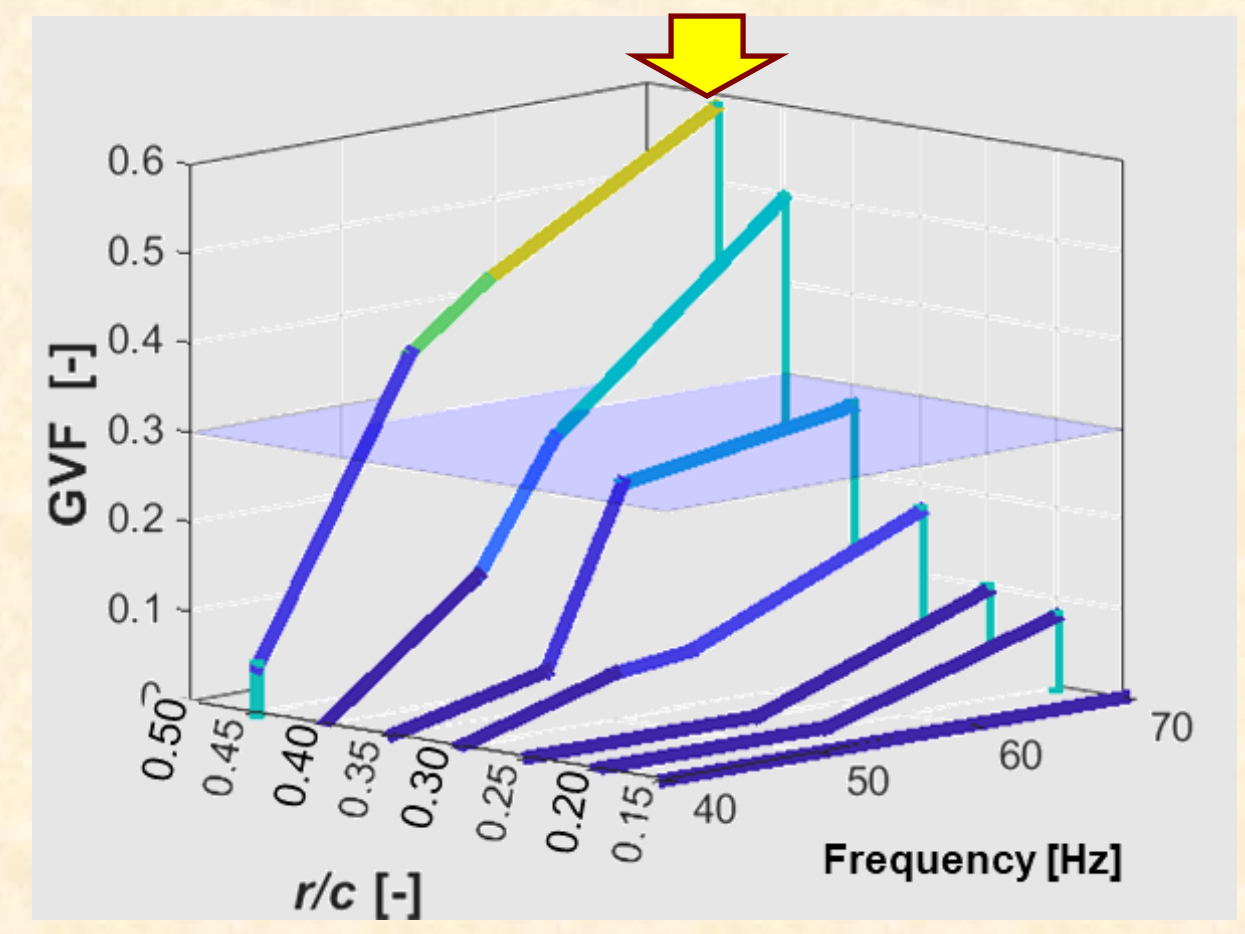
ORs perfectly seal land for $v_s < 25$ mm/s. For larger v_s , lubricant leaks through damper top end. Exit lubricant line shows gas content.

Estimation of gas volume fraction (GVF)

Squeeze Velocity [mm/s]	Orbit radius (r/c) [-]
24	0.20
31	0.25
37	0.30
55	0.45



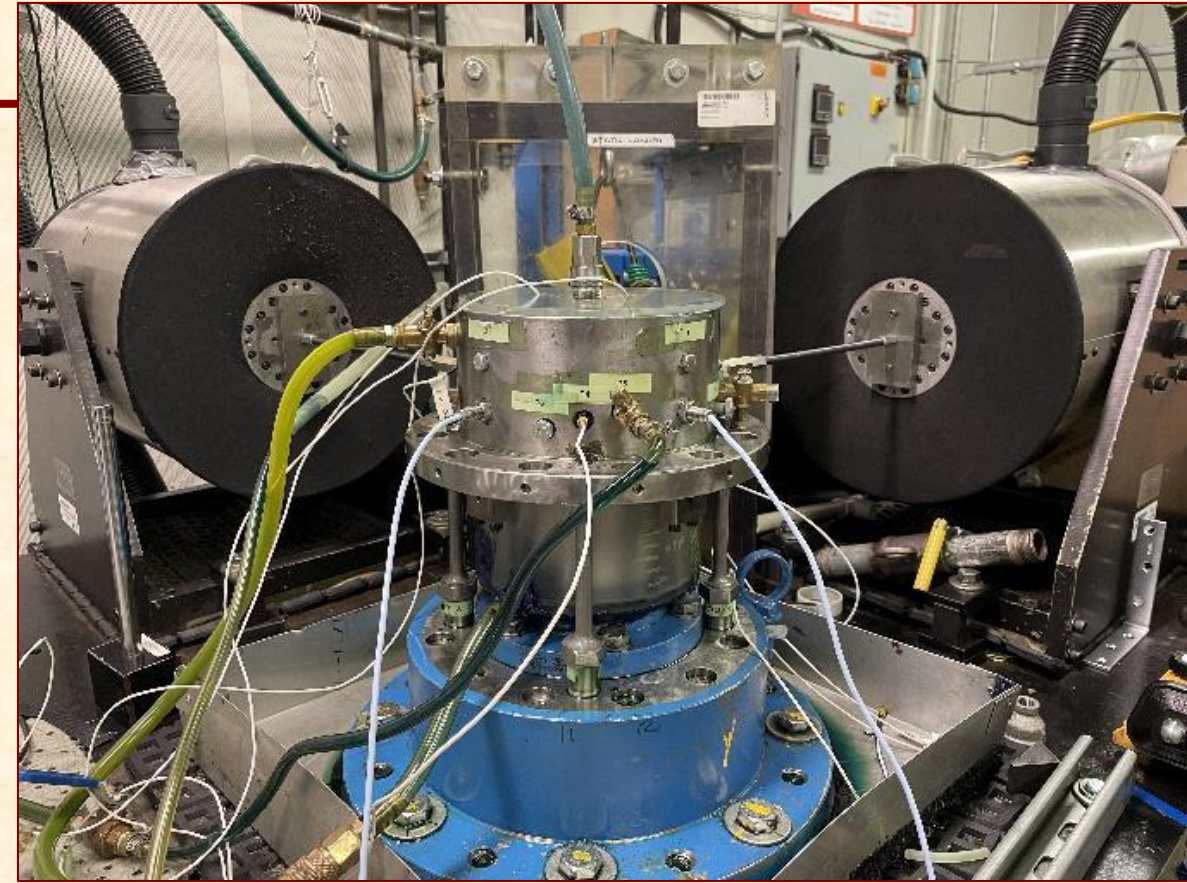
$r=0.45 \text{ c} \times \omega=70\text{Hz} = v_s=55 \text{ mm/s} \rightarrow \text{GVF} \sim 58\%$



Method catches air ingestion, but (clearly) not oil vapor cavitation

Closure

**MEASUREMENTS AND MODELS OF
SQUEEZE FILM DAMPERS' FORCED
RESPONSE AND A BIRD'S EYE VIEW
TO AIR INGESTION & ENTRAPMENT**



After 25+ years of work,

what is the learning?

Closure (1):

(a) Damping (C) and inertia (M) coefficients are \sim isotropic, i.e., $C_{xx} \sim C_{yy}$ and $M_{xx} \sim M_{yy}$. Cross-coupled coefficients are negligible for most whirl type motions.

(b) Classical lubrication theory does a poor job in producing physically accurate results for test SFDs with feed groove.

➔ Advanced accurate model includes fluid inertia, two-phase flow transport and correct boundary conditions.

(c) SFDs generate large added mass coefficients, in particular for configurations with feed and discharge grooves.

➔ Grooves generate dynamic pressures \rightarrow affect force coefficients

Closure (2):

- (d) A damper with one feed hole is more effective than other with three feed holes.
- (e) A sealed SFD produces significantly (**4X**) more damping and (++) more added mass than an open ends SFD.
- (f) The amplitude and shape of whirl motion have a small effect on the SFD force coefficients.
- (g) **Air ingestion** impairs the growth of film pressures for increasing squeeze velocities = (orbit amplitude x whirl frequency) → **damping coefficients decrease**.

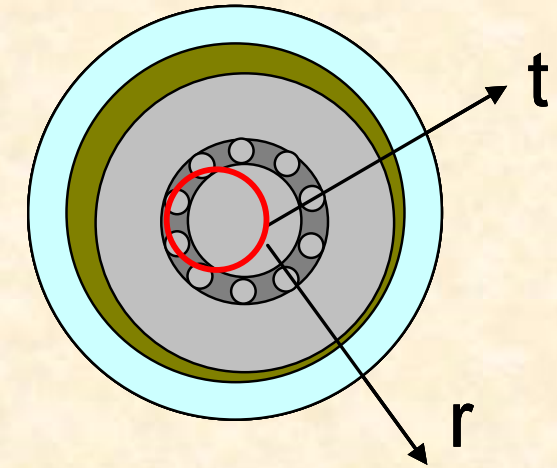


The experimental record shows SFDs perform as a **linear** mechanical element.

Fundamental learning

The amount of damping (needed) is a critical design consideration

If damping is too large the SFD acts as a rigid constraint to the rotor-bearing system with large forces transmitted to the supporting structure.



If damping is too low, the damper is ineffective and likely to permit large amplitude vibratory motion at synchronous and sub harmonic frequencies.

SFDs must be designed with consideration of the entire rotor-bearing system.

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The end of the quest

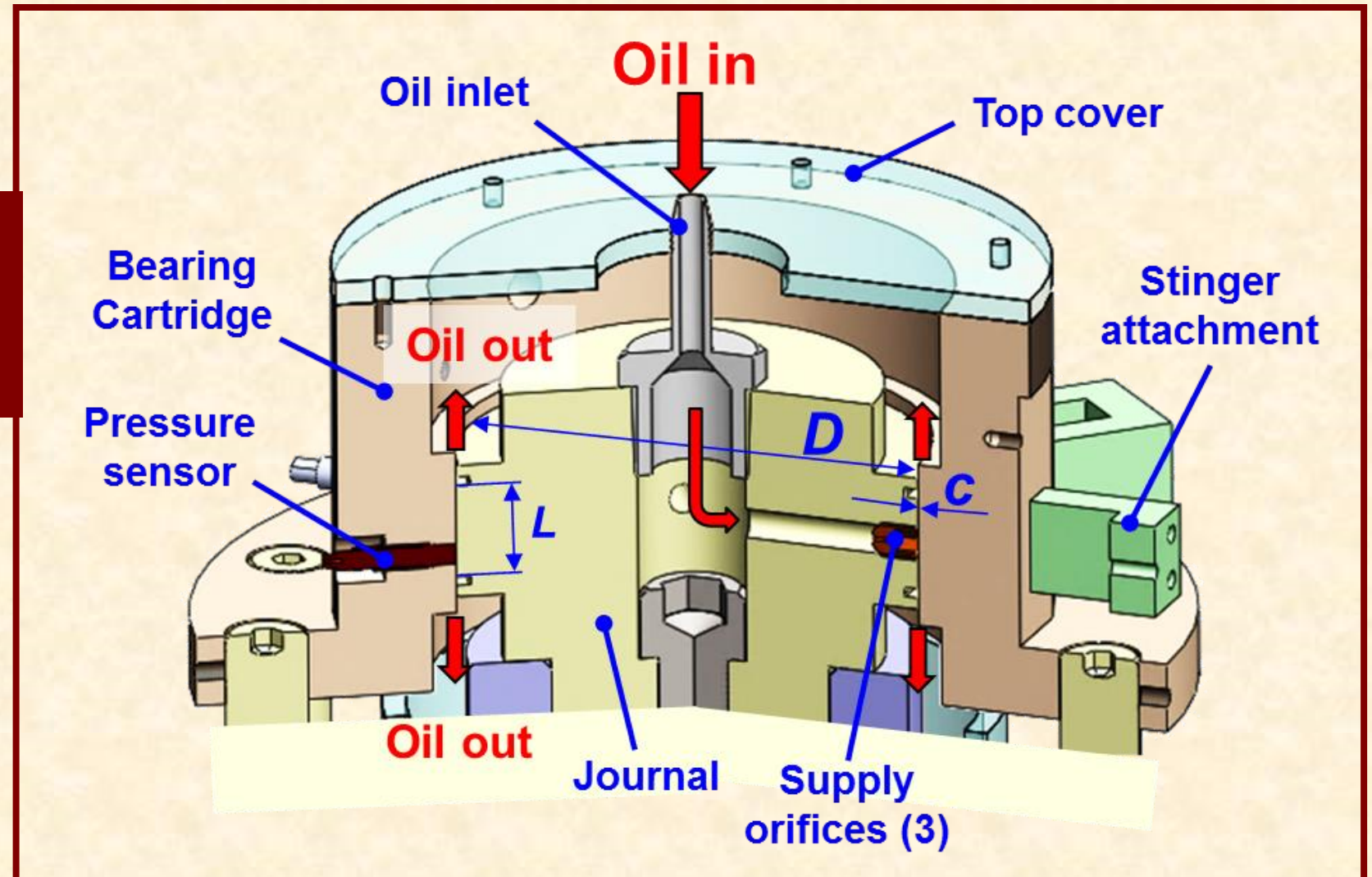
**When the music's over,
When the music's over, ...
turn out the lights,
turn out the lights...**

Jim Morrison (The Doors)



Questions (?)

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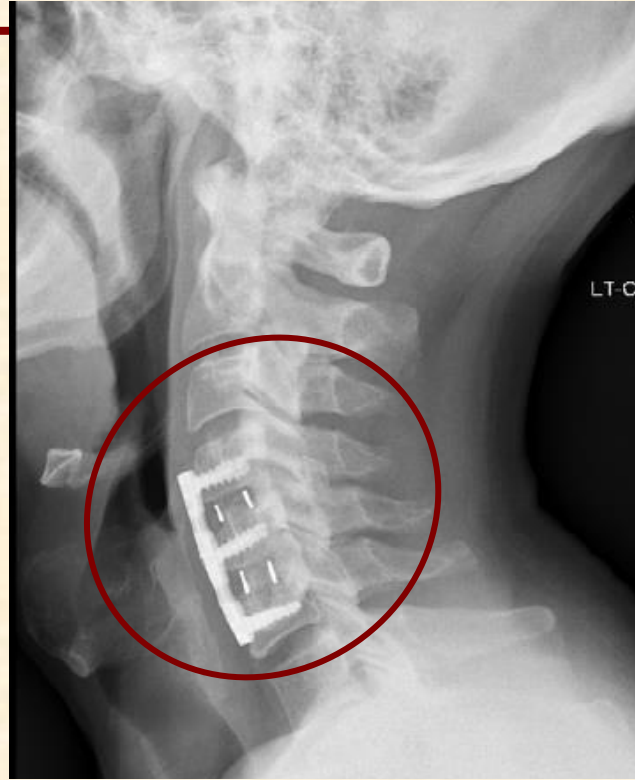


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Being a pain in the neck



... to being screwed



... to being totally screwed





1991-2022

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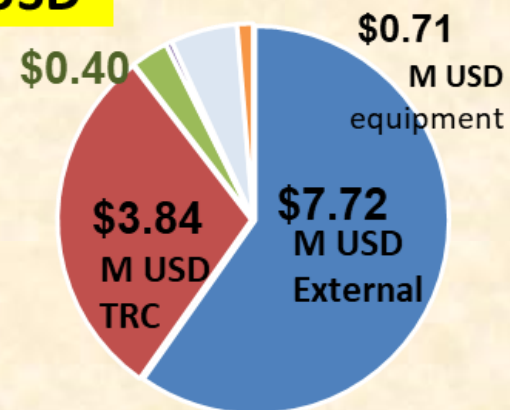
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52	138	Conference (peer reviewed)
7	42	Conference (NOT peer reviewed)
		Accepted/awaiting publication
104	376	

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- 2008 IGTI S&D - Rotordynamics
- 2005 IGTI S&D - Rotordynamics
- 2003 IGTI S&D - Rotordynamics

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