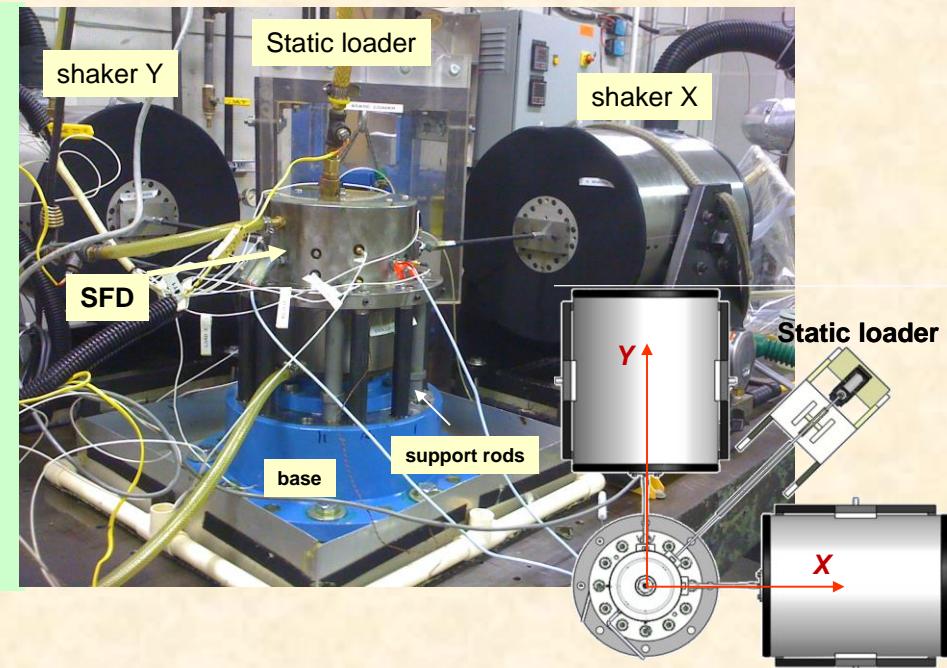

An example of parameter identification

Identification of force coefficients in a test damper



Luis San Andrés

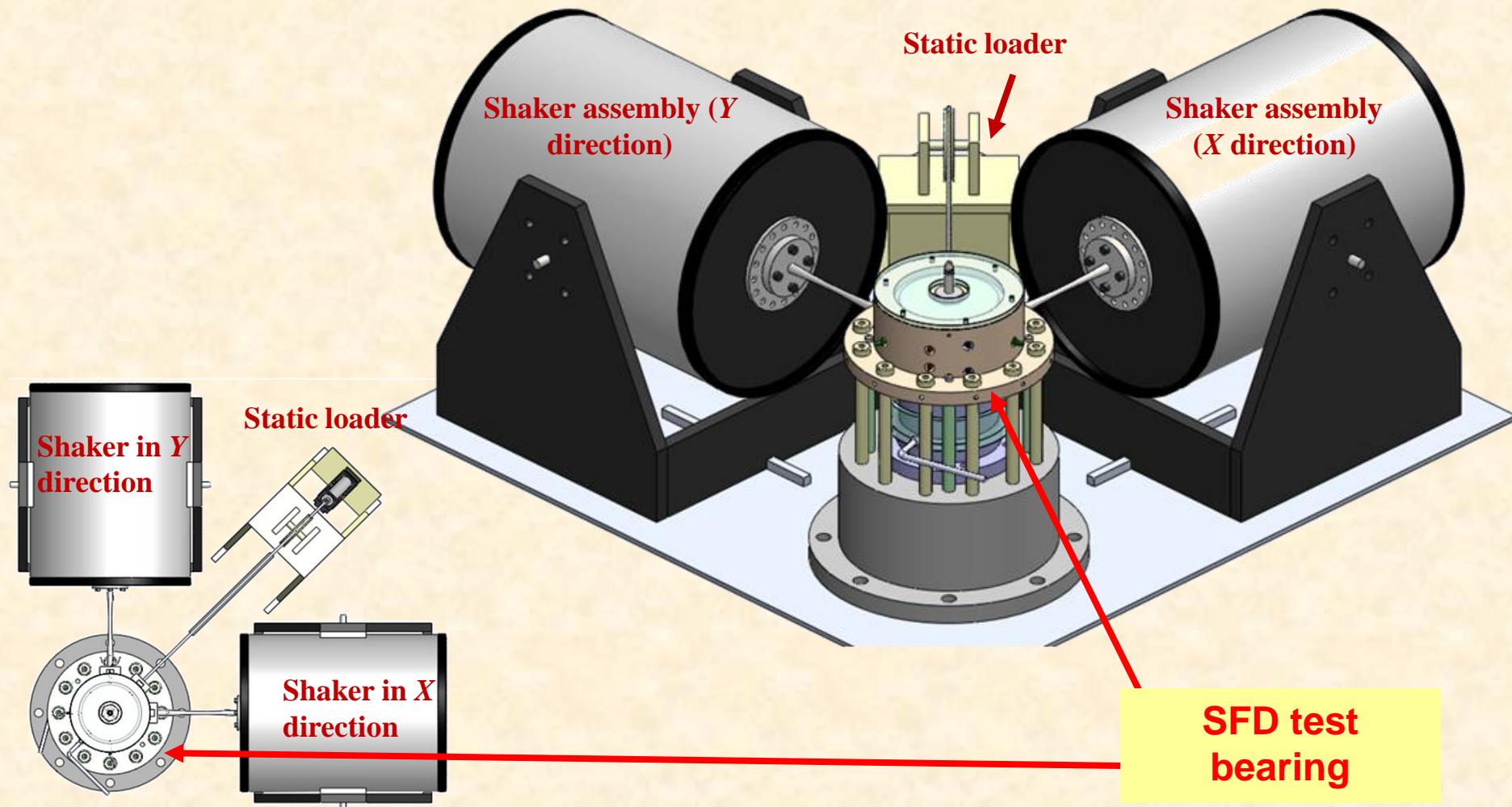
Sanjeev Seshagiri, Paola Mahecha

Research Assistants

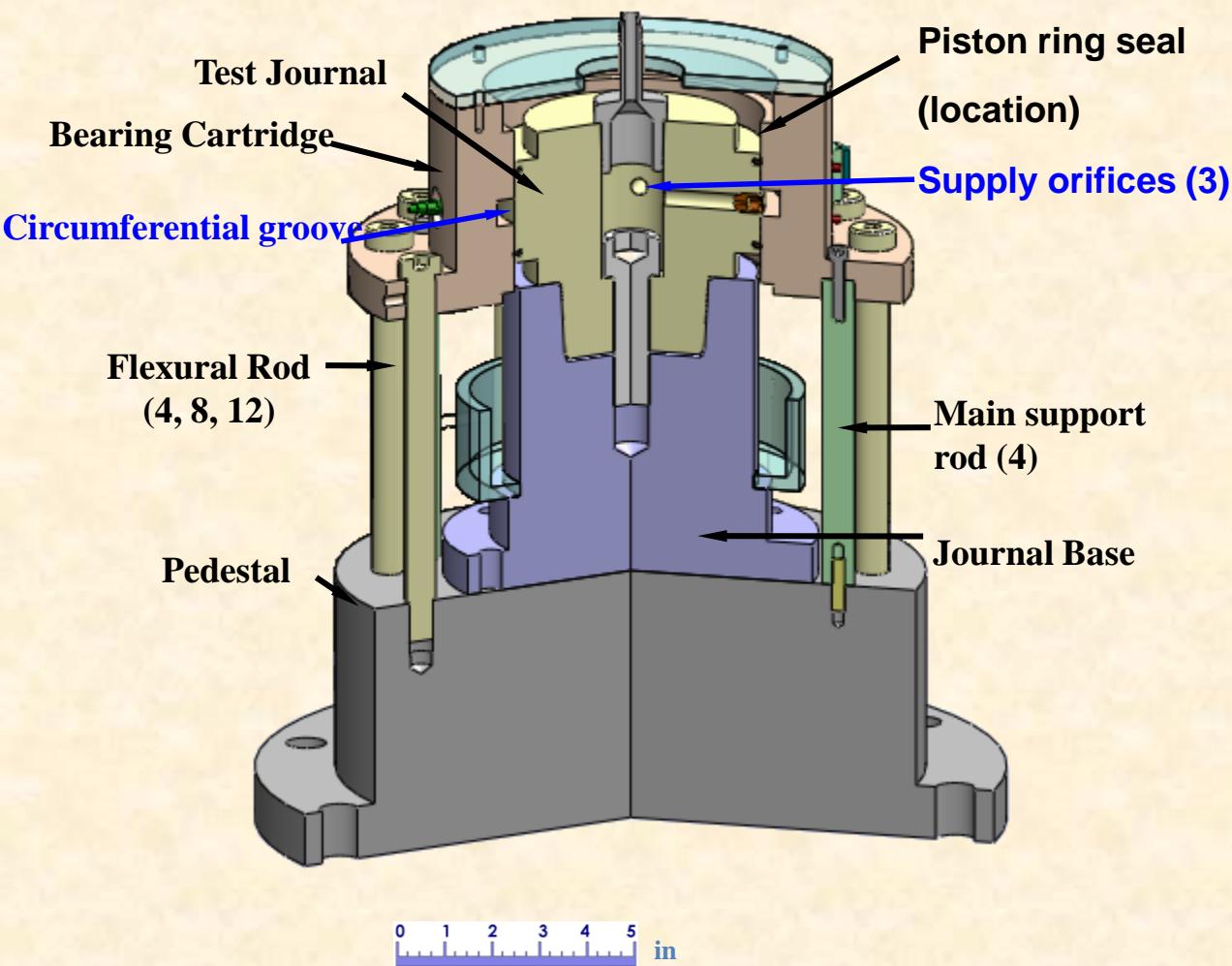
Sponsor: Pratt & Whitney Engines

SFD EXPERIMENTAL TESTING & ANALYTICAL METHODS DEVELOPMENT³²

P&W SFD test rig



P & W SFD Test Rig – Cut Section



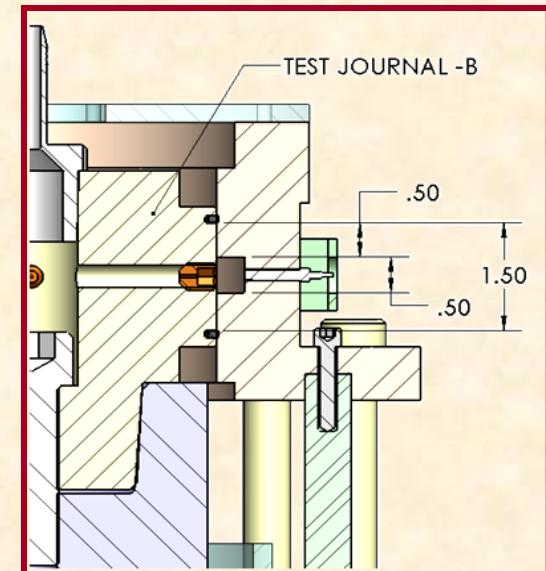
Test rig main features

Journal diameter: 5.0 inch

Film clearance: 5.1 mil

Film length: 2 x 0.5 inch

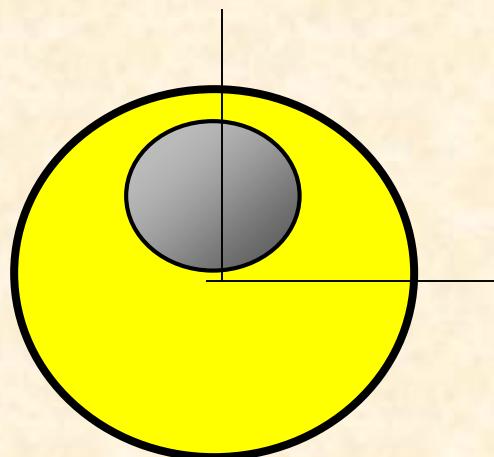
Support stiffness: 22 klf/in



Objective & task

Evaluate dynamic load performance of SFD.

Dynamic load measurements: circular orbits (centered and off centered) and identification of test system and SFD force coefficients

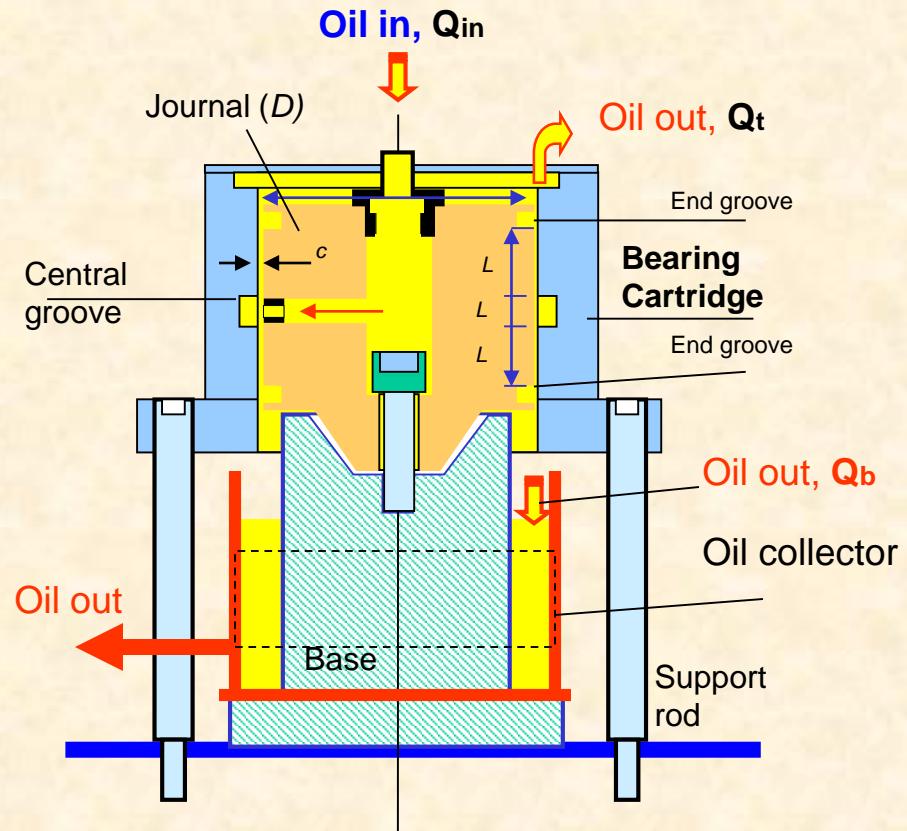


Circular orbit tests

- Frequency range: 5-85 Hz
- Centered and off-centered, $e_s/c = 0.20, 0.40, 0.60$
- Orbit amplitude $r/c = 0.05 - 0.50$

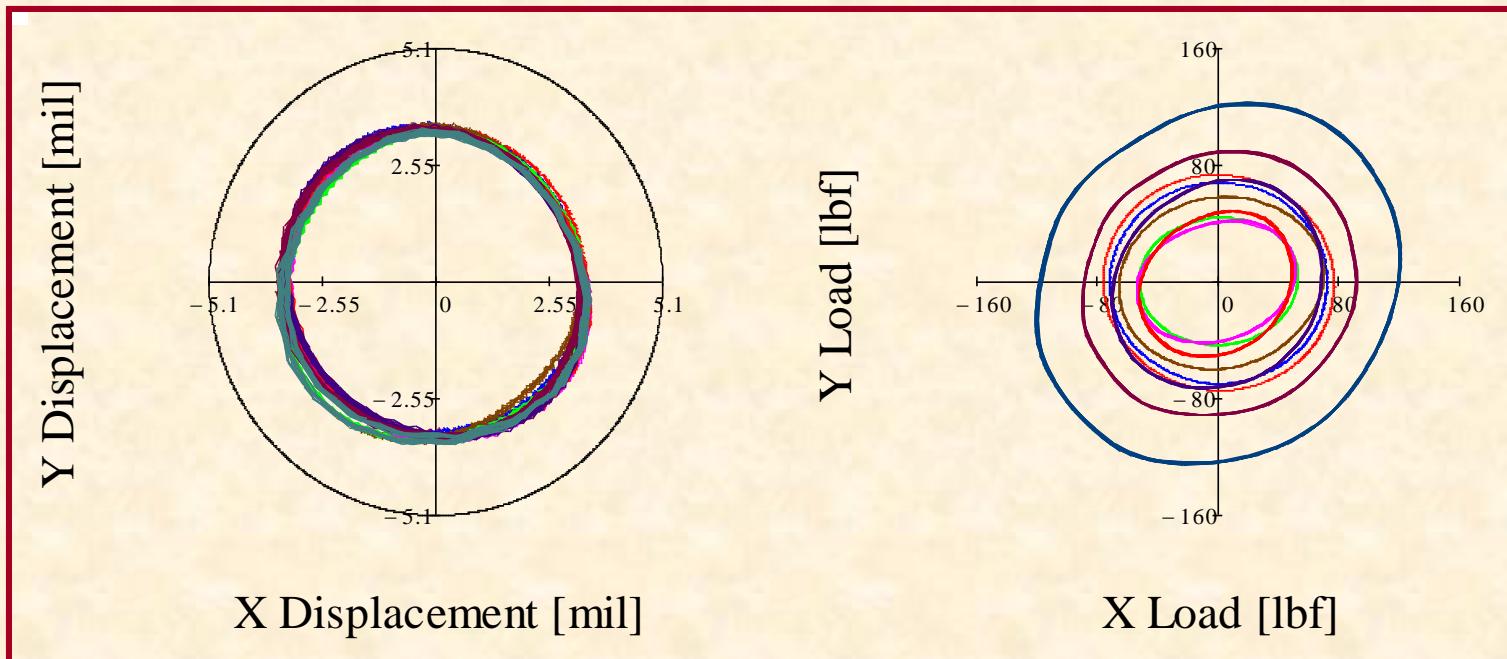
ISO VG 2 Oil

Viscosity at 73.4 °F [cPoise]	2.95
Density [kg/m ³]	784
Inlet pressure [psig]	7.5
Outlet pressure [psig]	0
Radial Clearance [mil]	c
Journal Diameter [inch]	5.0
Central groove length [inch]	L
Land length, L [inch]	L
Total Length [inch]	$3L$



Typical circular orbit tests

- Frequency range: 5-85 Hz
- Centered $e_s=0$
- Orbit amplitude $r/c=0.66$

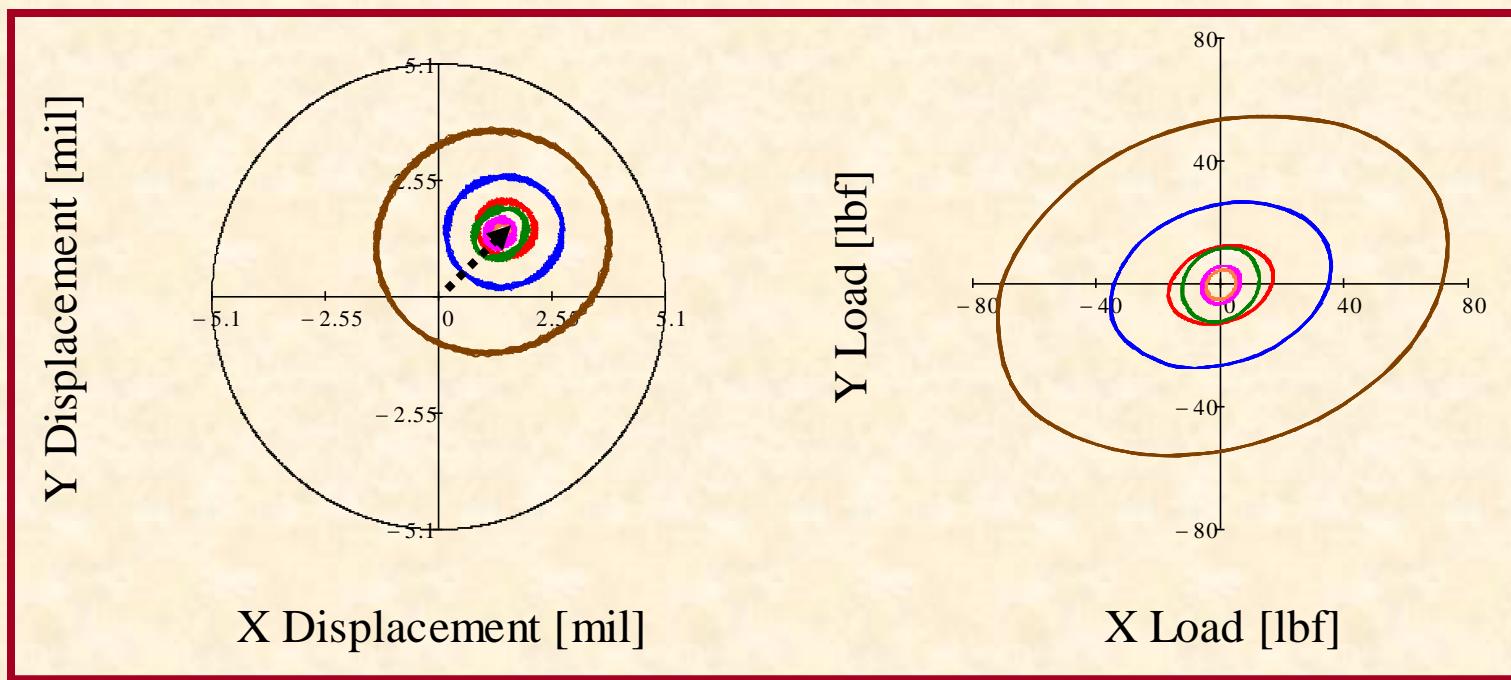


↑ motion (y vs. x)

↑ Forces (f_y vs. f_x)

Typical circular orbit tests

- Frequency: 85 Hz
- Off-centered at $e_s/c = 0.31$
- Orbit amplitude $r=0.05 - 0.5$

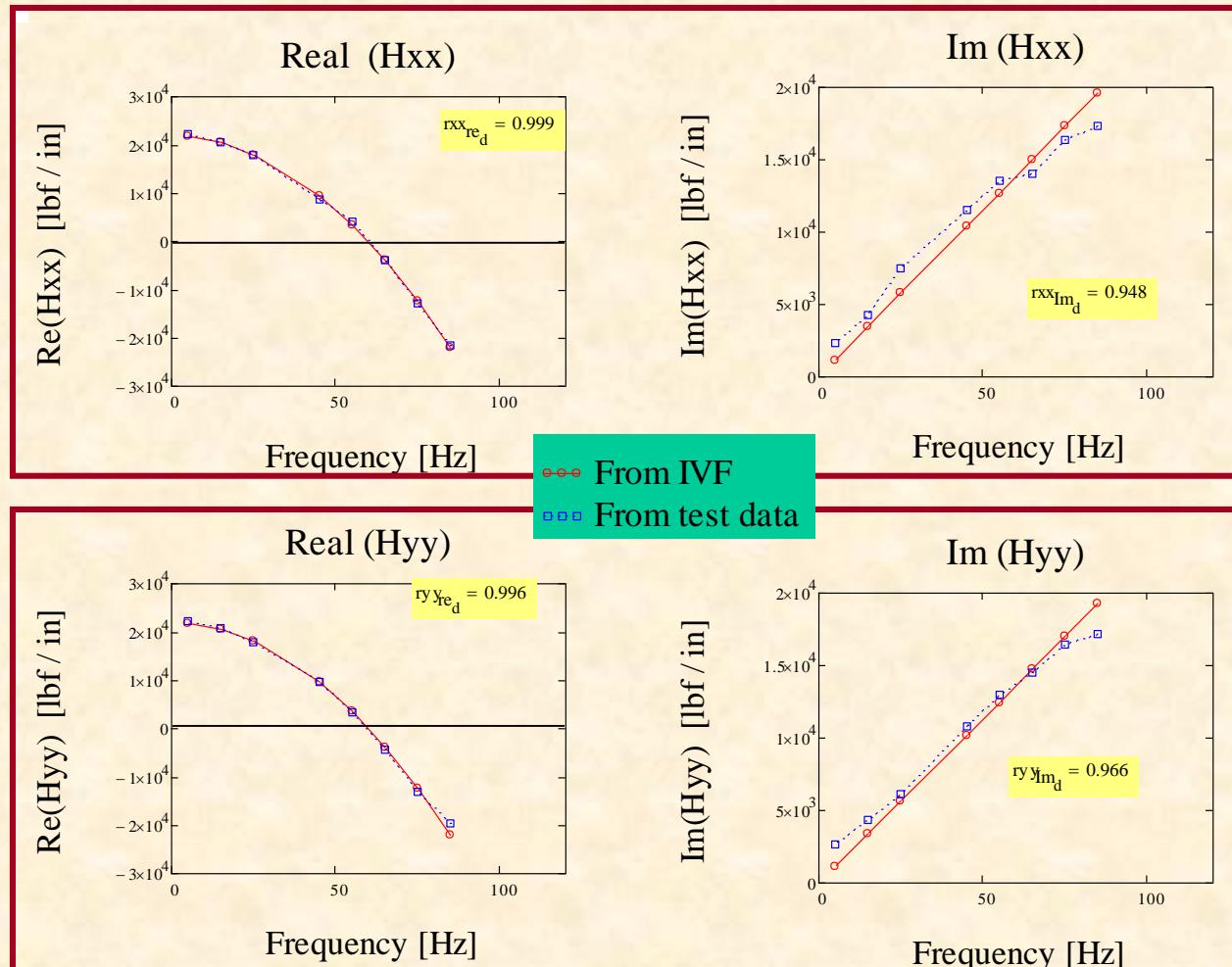


↑ motion (y vs. x)

↑ Forces (f_y vs. f_x)

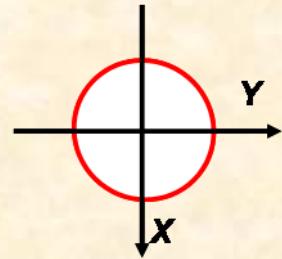
Typ system direct complex stiffnesses

H_{XX}



H_{YY}

$r/c = 0.66$,
centered $e_s = 0$

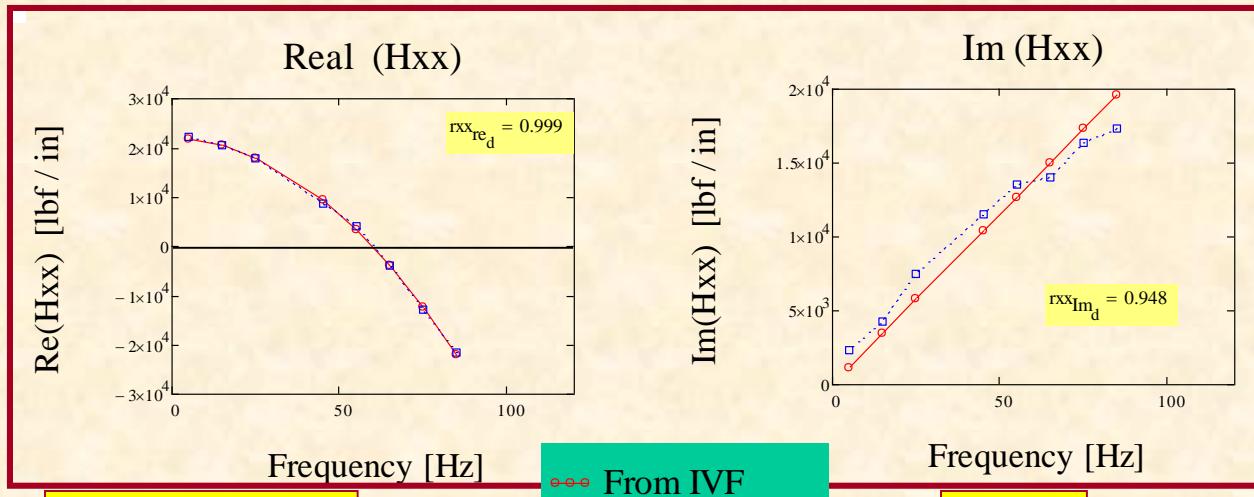


Real part

Imaginary part

Typ. system direct complex stiffness H_{xx}

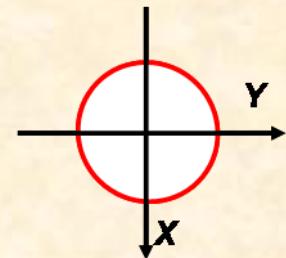
H_{xx}



$K - \omega^2 M$

ωC

$r/c = 0.66$,
centered $e_s = 0$



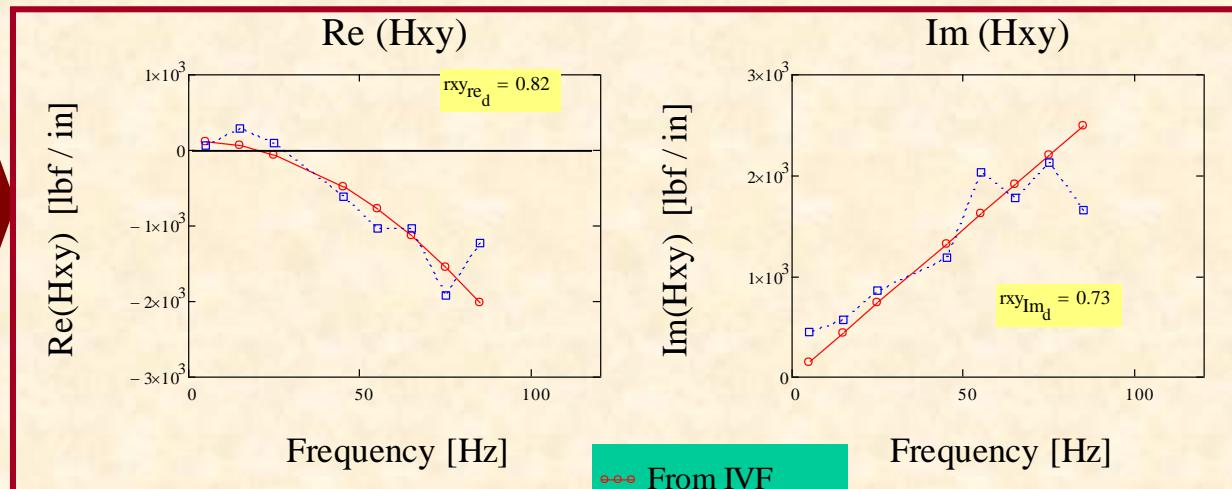
Excellent correlation between
test data and physical model

REAL PART = dynamic stiffness

IMAGINARY PART proportional to viscous damping

Test cross-coupled complex stiffnesses

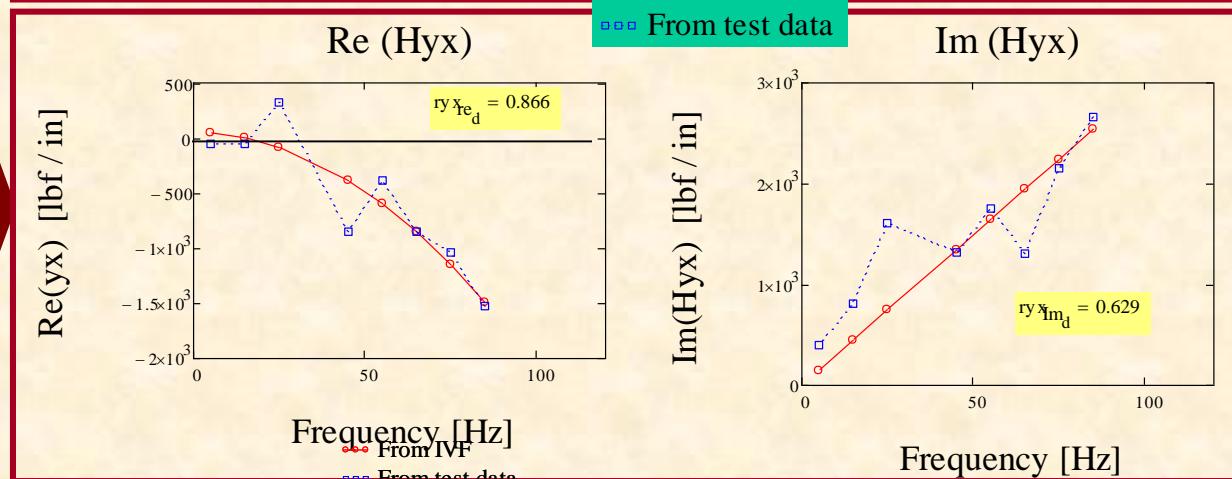
H_{XY}



$r/c = 0.66$,
centered $e_s = 0$

One order of magnitude lesser than direct impedances
= Negligible cross-coupling effects

H_{YX}



Real part

Imaginary part

SFD force coefficients

SFD

Difference between lubricated system and dry system (baseline) coefficients

$$C_{SFD} = C_{lubricated} - C_s$$

$$M_{SFD} = M_{lubricated} - M_s$$

$$K_{SFD} = K_{lubricated} - K_{sh}$$

DRY system parameters

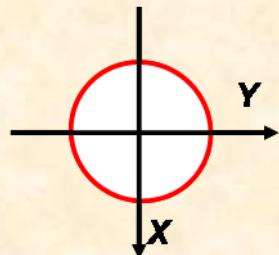
$$K_s = 21 \text{ klf/in}$$

$$M_s = 40 \text{ lb}$$

$$C_s = 7 \text{ lbf-s/in}$$

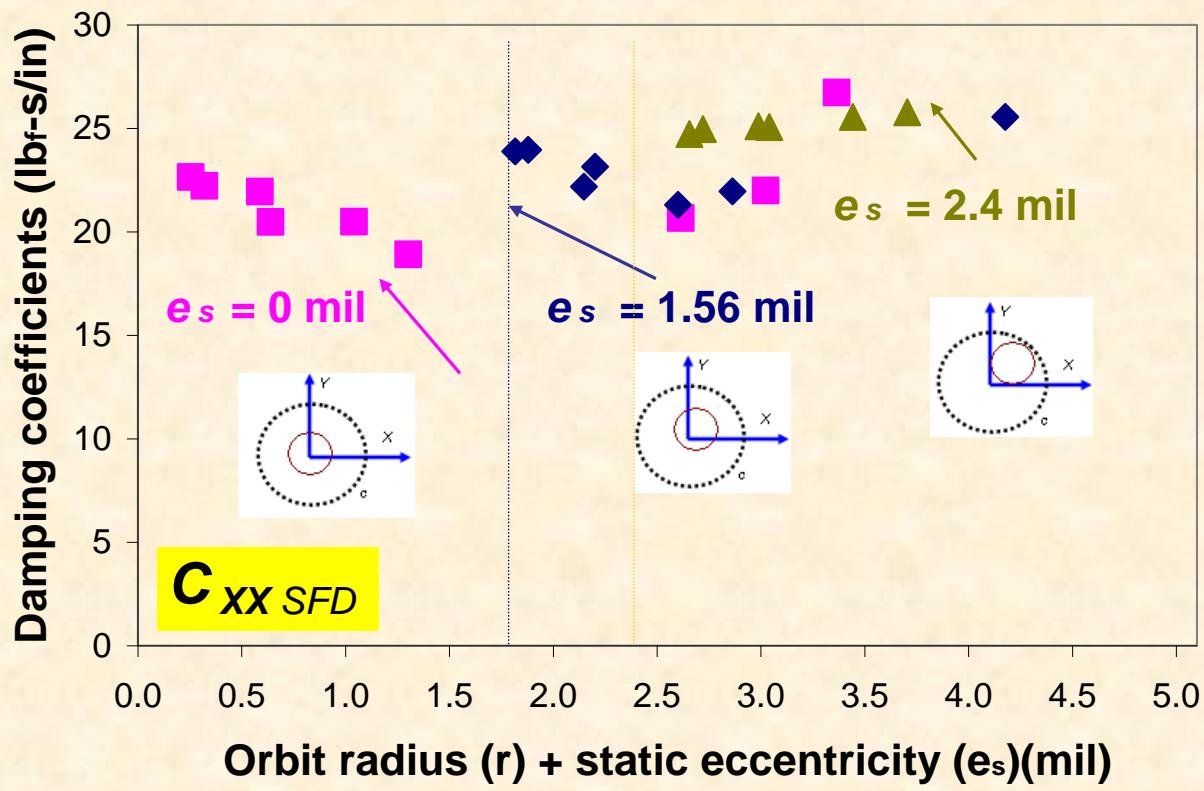
Nat freq = 73-75 Hz

Damping ratio
= 0.04

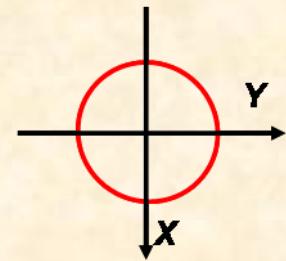


SFD damping coefficients

C_{XX}



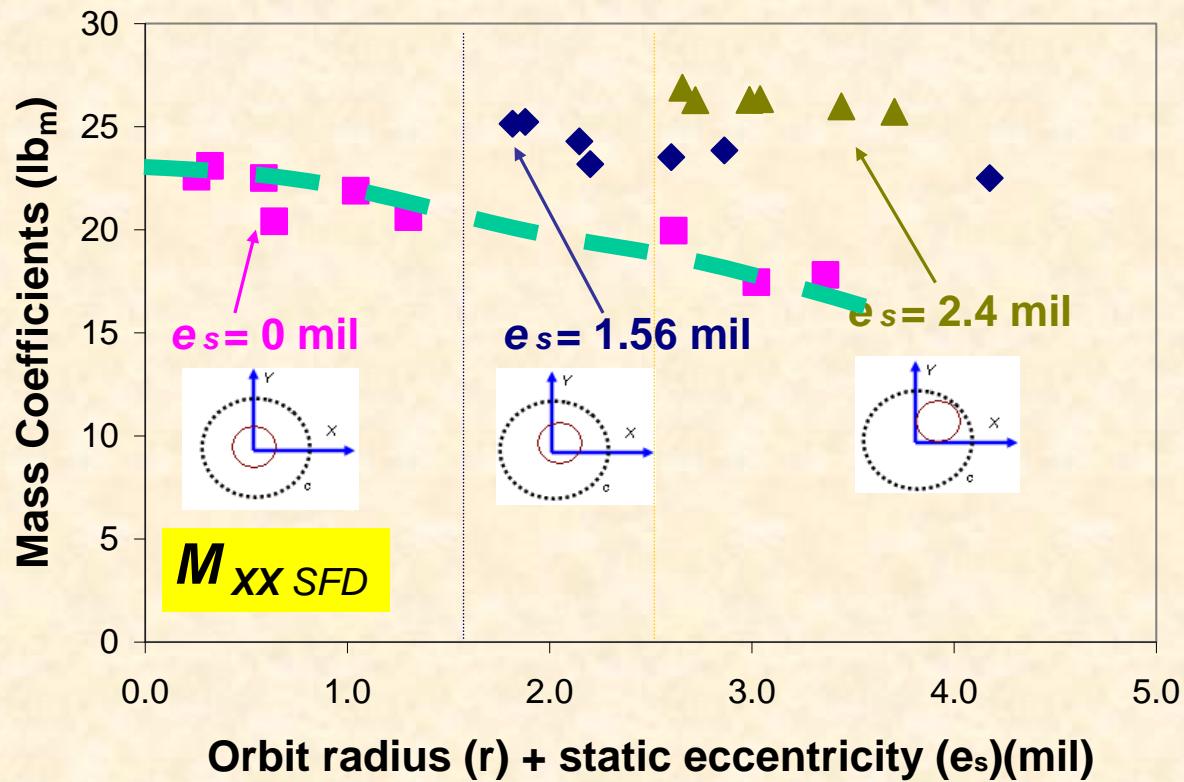
Damping increases mildly as static eccentricity increases



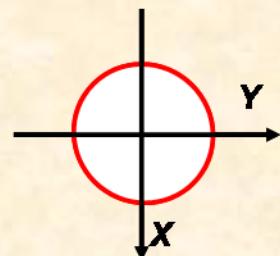
$C_{YY} \sim C_{XX}$ for circular orbits, independent of static eccentricity

SFD mass coefficients

M_{XX}



$M_{XX} \sim M_{YY}$ decreases with orbit radius (r) for centered motions. Typical nonlinearity



Conclusions

- **SFD test rig: completed measurements of dynamic loads inducing small and large amplitude orbits, centered and off-centered.**
- **Identified SFD damping and inertia coefficients behave well. IVFM delivers reliable and accurate parameters.**
- **Comparison to predictions are a must to certify the confidence of numerical models.**

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Questions (?)

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